SOLUTION OF A GENERALIZED AIR POLLUTION MODEL BY ORTHOGONAL COLLOCATION

A Thesis

Presented to

the Faculty of the Department of Chemical Engineering The University of Houston

In Partial Fulfillment of the Requirements for the Degree of Master of Science in Chemical Engineering

> by Miguel T. Fleischer December, 1975

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Finally, I am glad to dedicate this thesis to my wife, Jackie, and my son, David, whose patience and love have made all of this possible.

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ABSTRACT

Turbulent atmospheric diffusion from single or multiple point sources was simulated using the K-theory and solved by a new numerical technique, orthogonal collocation. The physical and chemical behavior of pollutant species in the atmosphere was described by the 3-dimensional, unsteady-state diffusion equation including chemical reactions. Orthogonal collocation was used to reduce the partial differential equation governing the mean concentration of contaminants to first-order ordinary differential equations. This system of equations was then solved in a digital computer.

Mean wind velocities and turbulent diffusivities were represented by empirical equations. Several meteorological parameters were included in these equations so that a variety of atmospheric conditions can be simulated. These parameters and other information required to solve an air pollution problem must be specified as input data by the user.

The present method was evaluated by comparing the results to existing experimental atmospheric concentration profiles. Good agreement was found in all cases. In addition, the sensitivity of the present model to variations in atmospheric conditions was analyzed by means of a parametric study. Proper responses were observed in all cases.

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CHAPTER I INTRODUCTION

There has been increasing interest in bringing the air pollution problem under control. This problem can be represented as a system consisting of three basic components: emission sources, the atmosphere, and receptors. Once the pollutants are emitted from sources, they are transported, dispersed, and transformed according to the laws of physics and chemistry throughout the entire atmosphere. Finally, these air pollutants are detected by receptors, such as human beings, animals, plants, and materials, producing in some cases undesirable effects.

In order to understand the cause-effect relationship of pollutant emission and dispersion on the air quality, a study of the three components previously discussed should be carried out. Understanding of the physical and chemical behavior of pollutant species in the atmosphere plays then an important role in the relation of source emissions to air quality standards. The objective of an air pollution model is to describe mathematically the spatial and temporal history of contaminants released into the atmosphere.

Several air pollution models have been developed so far, but they have had only limited success. An improved dispersion model based on the K-theory and solved using a new numerical technique is presented in this thesis. The atmospheric processes involving air pollutants are described by the 3-dimensional, unsteady-state diffusion equation including chemical reactions. Orthogonal collocation, a weighted residual method, reduces the partial differential equations governing the mean concentration of pollutant species to first-order ordinary differential equations. This system of equations is solved then in a digital computer.

The present work is validated with existing experimental data. In addition, its sensitivity to variations in atmospheric conditions is analyzed by means of a parametric study.

Chapter II

THEORETICAL BACKGROUND

There are two fundamental ways of describing the physical and chemical behavior of pollutant species in the turbulent atmosphere. The first is the so-called Eulerian approach, where the behavior of species is described relative to fixed coordinates. The second is the statistical approach, where concentration changes are described from a statistical point of view by considering the paths of individual elements of fluid, and is thus Lagrangian in nature.

The objective of any air pollution model is to predict pollutant concentrations at given points. These concentrations are caused by emissions from sources, and therefore a source-oriented point of view is the natural one in this case.

Eulerian Approach

Consider s species in a fluid. The concentration of each must, at each instant, satisfy a material balance taken over a volume element. Therefore, the concentration of each species must satisfy the continuity equation usually known as the instantaneous diffusion equation,

$$\frac{\partial C_{i}}{\partial t} + \frac{\partial}{\partial x_{j}} u_{j}C_{i} = \frac{\partial}{\partial x_{j}} \{D_{i} \frac{\partial C_{i}}{\partial x_{j}}\} + R_{i}(C_{1}, \dots, C_{s})$$
(2.1)
$$i = 1, 2, \dots, s$$

where the subscript j represents the three coordinate
directions: x(axial), y(lateral), and z(vertical); and
C_i = the instantaneous concentration of species i
u_j = the jth component of the fluid velocity
D_i = the molecular diffusivity of species i in the carrier
fluid, and

 R_i = the rate of generation of species i.

Since atmospheric flows are turbulent, it is conventional to divide the instantaneous quantities into mean and fluctuating facts,

$$C_{i} = \langle C_{i} \rangle + C_{i}'$$

$$u_{j} = \overline{u}_{j} + u_{j}'$$
(2.2)

It should be noted that the mean fluid velocities are usually determined by temporal averaging and the mean concentrations always represent ensemble averages. This is the reason why a different notation has been used for the mean values of the velocities and the concentrations.

Substitution of equation (2.2) into equation (2.1) gives

$$\frac{\partial}{\partial t}(\langle C_{i} \rangle + C_{i}') + \frac{\partial}{\partial x_{j}} \{ (\bar{u}_{j} + u_{j}') (\langle C_{i} \rangle + C_{i}') \} = \frac{\partial}{\partial x_{j}} \{ D_{i} \frac{\partial}{\partial x_{j}} (\langle C_{i} \rangle + C_{i}') \} + R_{i} (\langle C_{i} \rangle + C_{i}') \} + R_{i} (\langle C_{i} \rangle + C_{i}') \}$$
(2.3)

Taking an average of equation (2.3) over a large ensemble of realizations of the turbulence, one obtains the following equation governing <C;>

$$\frac{\partial \langle C_{i} \rangle}{\partial t} + \frac{\partial}{\partial x_{j}} (\bar{u}_{j} \langle C_{i} \rangle) = \frac{\partial}{\partial x_{j}} \{ D_{i} \frac{\partial \langle C_{i} \rangle}{\partial x_{j}} - \langle u_{j}^{\dagger} C_{i}^{\dagger} \rangle \} + \langle R_{i} (\langle C_{1} \rangle + C_{1}^{\dagger}, \dots, \langle C_{s} \rangle + C_{s}^{\dagger}) \rangle$$

$$(2.4)$$

The most common means of representing the turbulent fluxes $\langle u_j^{\prime} C_i \rangle$ is by the so-called K-theory, in which a turbulent diffusivity is defined by

$$-\langle u_{j}' C_{i}' \rangle = K_{jj} \frac{\partial \langle C_{i} \rangle}{\partial x_{j}}$$
(2.5)

Ignoring the molecular diffusion when compared with the turbulent diffusion, and assuming the atmosphere to be incompressible, the final expression for the diffusion equation becomes

$$\frac{\partial \langle C_{i} \rangle}{\partial t} + \bar{u}_{j} \frac{\partial \langle C_{i} \rangle}{\partial x_{j}} = \frac{\partial}{\partial x_{j}} \{ K_{jj} \frac{\partial \langle C_{i} \rangle}{\partial x_{j}} \} + \langle R_{i} (\langle C_{l} \rangle, \dots, \langle C_{s} \rangle) \rangle$$
(2.6)

It should be pointed out that the effect of concentration fluctuations on the rate of reaction, i.e., terms such as $(C_i^C)^{>}$ that arise from cases where R_i is a nonlinear function of the C_i, were neglected in the development of equation (2.6). The conditions for this approximation to be valid will be discussed in Chapter IV.

Lagrangian Approach

The Lagrangian approach to turbulent diffusion is concerned with the behavior of representative fluid particles. The development of a general equation for the mean concentration $\langle C(\mathbf{x},t) \rangle$ follows that of Seinfeld [22].

$$= \iint_{-\infty-\infty}^{\infty} \bigcup_{0}^{\infty} Q(\mathbf{x},t/\mathbf{x}_{0},t_{0}) < C(\mathbf{x}_{0},t_{0}) > d\mathbf{x}_{0} + \iint_{-\infty-\infty-\infty}^{\infty} \iint_{0}^{\infty} Q(\mathbf{x},t/\mathbf{x}',t') S(\mathbf{x}',t') dt' d\mathbf{x}' \quad (2.7)$$

where $Q(\mathbf{x}, \mathbf{t}/\mathbf{x}^*, \mathbf{t}^*)$ is the transition probability density for the particle, that is, the probability density that if the particle is at a position \mathbf{x}^* at time \mathbf{t}^* will undergo a displacement to \mathbf{x} at t, and $S(\mathbf{x}, \mathbf{t})$ is the spatial-temporal distribution of particle sources.

Therefore, the first term on the right-hand side of equation (2.7) represents those particles present at t_0 , and the second term on the right-hand side accounts for particles added from sources between t' and t.

As can be observed from equation (2.7), the applicability of the Lagrangian approach rests on the ability to evaluate the transition probability Q. This is a difficult task because complete knowledge of the turbulence properties required to determine Q is generally not available. This problem has been usually overcome by assuming that Q obeys a multidimensional Gaussian distribution, but it is known that this assumption is valid only if the turbulence is stationary and homogeneous. By using that assumption then, one is limited to apply the Lagrangian approach only to some atmospheric cases.

Moreover, the development of equation (2.7) was done for cases where the particles were not undergoing chemical reactions. First-order reactions can be included rather easily in the development of a similar equation, but there is no convenient way of incorporating nonlinear chemical reactions in the Lagrangian approach to turbulent diffusion.

The Eulerian method is very useful in air pollution modeling because the Eulerian statistics are easy to measure and as it will be seen later, possible to represent in an analytical form suitable for computer programs. In addition, the mathematical expressions can be directly applied to situations where chemical reactions take place.

Unfortunately, the Eulerian approach can be solved only by approximate solutions, e.g., the K-theory, and this leads to the problem of accurately predict the eddy diffusivities.

The main objective of the present work is to obtain a model which can be applied to any atmospheric dispersion problem with chemical reactions. Because of the previous analysis of the two approaches that can be used to describe atmospheric diffusion, it was decided to use the Eulerian method as the basis of the present model. As will be seen later, this is the most common approach used although it leads to limited success. The reason being the numerical techniques with which the equations have been solved. Another objective of the present work is to investigate the possibility of a better numerical method for solving the general diffusion equation.

Chapter III

REVIEW OF PREVIOUS WORK

Eulerian Models

The basic mathematical statement for description of the temporal and spatial distribution of chemical species by this approach is the mass balance or continuity equation. This expression, equation (2.6) in this study, plus the appropriate initial and boundary conditions complete the general description.

Exact solutions to this set of equations have not been published. However, it is possible to obtain approximate solutions using numerical techniques.

During the past years, several models have been presented in the literature, ranging from very simple ones like the box model to general cases solved by finitedifference techniques.

Box Model

The simplest air pollution model, though not the first, is the box model, which has been discussed by Sklarew [25]. This model consists of an imaginary box which bounds the atmosphere over a city with the top of the box being at a constant mixing height, the height at which there is no further vertical diffusion. The concentration in the box

is calculated from a simple material balance on the air flow and pollutant emissions, thus, it is not sufficient for anything more than gross estimates, by offering a quick result for long-term concentrations.

Grid Model

Sklarew [25] proposed a different approach on urban air pollution modeling, the grid model. In this model, the region of interest is divided into a three dimensional grid of cells where the diffusion equation of the Eulerian approach is solved. The time-dependent solution is obtained numerically from difference equations.

Lagrangian particles representing a definite amount of pollutant are advected and diffused through the Eulerian grid. In order to do this, a pseudo-velocity v* is defined as

$$\mathbf{v}^* = \mathbf{\bar{u}} - \mathbf{D}'\nabla \mathbf{C}/\mathbf{C} \tag{3.1}$$

where D' is a diffusivity tensor and \overline{u} is the wind velocity averaged throughout the cell surrounding each grid point. The defined v* then is just the sum of the wind velocity and the velocity corresponding to the diffusion flux. Therefore, the diffusion equation is reduced to

$$\frac{\partial C_i}{\partial t} = - \nabla (v^* C_i) + R_i + \text{sources}$$
(3.2)

This equation is solved in time-steps in the following sequence: the pollutant concentration in each cell of the grid is given by the particles in the cell. The cell average concentration is updated by advancing the chemical reactions for a time-step and by adding (and/or substracting) pollutant from sources (and/or sinks) within the cell. Finally, the Lagrangian particles are advected and diffused using the pseudo-velocity, and thus transported at the boundary of each cell for a specific period of time.

As it can be observed, this model is a mixture of the Eulerian and Lagrangian approaches. In spite of being a more general model as compared to the previous one, it has the disadvantage that the grid size necessary for a desired accuracy can result in prohibitive computing time.

Two Dimensional Unsteady State Models

Much of the recent work on air pollution modeling centers on solving a simplified diffusion equation, a two dimensional unsteady state case, which includes only the x-component of the wind velocity and the vertical turbulent diffusion coefficient. In almost all these models, the solutions are obtained by using a finite-difference technique.

For these cases, equation (2.6) for a single species becomes

$$\frac{\partial \langle C \rangle}{\partial t} + \bar{u} \frac{\partial \langle C \rangle}{\partial x} = \frac{\partial}{\partial z} \{ K_z \frac{\partial \langle C \rangle}{\partial z} \}$$
(3.3)

Runca and Sardei [21] solved equation (3.3) using a mixed Lagrangian-Eulerian finite difference scheme. In this model, the emission rate of the source is taken as a boundary condition. The wind velocity and diffusion coefficient are considered to be only functions of z.

Equation (3.3) is solved with the method of fractional steps: the concentration field at time $t+\Delta t$ is obtained from that at time t by separating the contributions due to the advection and diffusion terms of equation (3.3).

In the first step, the following advection equation is solved by a Lagrangian technique over the time interval Δt with the concentration field at time t as initial condition:

$$\frac{\partial C}{\partial t} + u(z) \frac{\partial C}{\partial x} = 0$$
 (3.4)

The diffusion equation,

$$\frac{\partial C}{\partial t} - \frac{\partial}{\partial z} \{ K_{z}(z) \frac{\partial C}{\partial z} \} = 0 \qquad (3.5)$$

the second step, is solved with a conventional Eulerian finite-difference scheme over the same time interval. The initial condition is provided by the concentration field obtained from the first step, and the solution of the second step is an approximation of the concentration field at time $t+\Delta t$.

A different model was presented by Egan and Mahoney [6], where equation (3.3) was also solved, but the source emission rate Q, was included in the same governing equation:

$$\frac{\partial C}{\partial t} + \bar{u} \frac{\partial C}{\partial x} = \frac{\partial}{\partial z} \{ K_z \frac{\partial C}{\partial z} \} + Q$$
(3.6)

To simulate transport through an urban area, the region was divided into a number of grid elements, and the partial derivatives in equation (3.6) were approximated by finite differences corresponding to the dimensions of the urban grid elements. Equation (3.6) was solved by an unconventional scheme using moments of concentration distribution.

A slightly more difficult diffusion equation was solved by Eschenroeder and Martinez [7]. A simplified chemical kinetic scheme was included in the governing equation,

$$\frac{\partial C_{i}}{\partial t} + \bar{u} \frac{\partial C_{i}}{\partial x} = \frac{\partial}{\partial z} \{ K_{z} \frac{\partial C_{i}}{\partial z} \} + R_{i}$$
(3.7)
$$i = 1, 2, ..., s$$

where R_i is the production rate for the ith species and s is the number of species.

The numerical solution of equation (3.7) followed a Crank-Nicholson type implicit finite difference scheme. Since fields for \overline{u} and K_z were prescribed by meteorology inputs, the nonlinearity was confined only to some members of the R_i terms. In a later paper, Eschenroeder and Martinez [8] reported that a number of difficulties were encountered in using the Crank-Nicholson method and the approach was abandoned.

General Solutions

Some recent work has been done to solve the general expression of the diffusion equation. One of these models was developed by Roth et al [20]. The governing equation solved was the following:

$$\frac{\partial \langle C_{i} \rangle}{\partial t} + \overline{u} \frac{\partial \langle C_{i} \rangle}{\partial x} + \overline{v} \frac{\partial \langle C_{i} \rangle}{\partial y} + \overline{w} \frac{\partial \langle C_{i} \rangle}{\partial z} = \frac{\partial}{\partial z} \{K_{z} \frac{\partial \langle C_{i} \rangle}{\partial z}\} + R_{i}(\langle C_{1} \rangle, \dots, \langle C_{s} \rangle) + Q_{i}(x, y, z, t)$$
(3.8)
$$i = 1, 2, \dots, s$$

where K_z is the vertical eddy diffusivity, R_i is the rate of formation of species i by chemical reaction, and Q_i is the rate of emission of species i from sources.

Equation (3.8), plus initial and boundary conditions, was applied for the prediction of pollutant concentrations over a fifty mile square area. This region was divided into a grid of 625, 2x2 mile squares, where 198 grid squares were source-free. The grid actually used in the solution of equation (3.8) was a three-dimensional array of ten layers of cells occupying the space between the ground and the inversion base. Therefore, each cell had a two mile square base and a height of (H-h)/10, being H and h the elevations of the inversion base and ground above sea level, respectively.

To represent the surface winds, Roth et al. constructed maps of wind speed and wind direction for hourly time intervals, using data gathered at the network of groundbased monitoring stations. In the absence of wind field aloft data, the surface values were used as the basis to calculate wind velocities to all levels between the ground and the inversion base.

A model for the vertical turbulent diffusivity was developed based on the work done by Eschenroeder and Martinez [7].

Finally, the model was completed by a simplified kinetic mechanism involving 12 species and 14 reaction steps. The mathematical representation included four coupled nonlinear differential equations (five other species were expressed by steady-state relations, and the remaining three species were products that could have also been represented by differential equations, but it was not done) that were solved by a modified Gear's method.

A fractional step finite-difference method was selected to be the numerical technique for the model. In this type of solution, a multidimensional problem is replaced by a succession of simpler lower dimensional problems. Therefore, the four-dimensional partial differential equation (3.8) in

(x,y,z,t) was split into three two-dimensional equations in (x,t), (y,t), and (z,t), with the inclusion of the reaction and elevated source terms in the (z,t) fragment. The solution in (z,t) is implicit, while the solution in (x,t) and (y,t) is explicit. Each of these two-dimensional equations was then integrated in succession over one time step, and the terms in each of the three partial differential equations was approximated by finite-difference expressions.

It should be pointed out that this model cannot handle point or line sources, the emissions therefore averaged over relatively large distances.

Another recent model in which a solution to the general expression of the diffusion equation was obtained, was developed by Shir and Shieh [24]. This model was used to study SO₂ distributions in the St. Louis metropolitan area during 25 consecutive days. The region of interest was divided into a three-dimensional grid system of 30x40x14 = 16,800 grid points. The horizontal grid sizes Δx and Δy were of 1524m and the vertical size 20m, 25m, or (H-200)/4 m, values depending on the heights of the vertical grids (H represents the mixing height). The horizontal grid sizes were chosen according to the grid size of the emission source inventory. The equation solved in this model was the diffusion equation for a single species,

$$\frac{\partial C}{\partial t} + \nabla VC = K_{H} \nabla_{H}^{2} C + \frac{\partial}{\partial z} \{K_{V} \ \frac{\partial C}{\partial z}\} + Q + R \qquad (3.9)$$

where C is the mean concentration of SO_2 , V = (u,v,w) is the mean wind vector, Q is the source strength rate, R is the chemical reaction rate, K_H is the horizontal eddy diffusivity, and K_v , is the vertical eddy diffusivity.

The hourly averaged surface wind field for the total region was obtained by using a weighted interpolation scheme. Data collected at some stations were interpolated to a square grid, which had a size of five area source dimensions. From this wind field, a linear interpolation was used to obtain a wind vector at each grid point.

Since upper layer wind data were not available, the vertical wind profiles at each grid location was assumed to be of power law from

$$\mathbf{v} = \mathbf{v}_{\mathbf{s}} \left(\frac{\mathbf{z}}{\mathbf{z}_{\mathbf{s}}}\right)^{\mathbf{p}} \tag{3.10}$$

where V and V_s are the upper and surface wind at the height z and z_s , respectively. The power constant p was determined by using equation (3.10) and data gathered at heights of 140m and 39m. The vertical component of the wind vector was calculated from the horizontal winds through the continuity equation. Finally, a constant wind direction with height was assumed in the model.

A turbulence transport model developed by Shir [23] was used to calculate the eddy diffusivities. Under neutral conditions, the vertical component of the eddy diffusivity vector is expressed as

$$K_v = u_* l$$
 , $l = k_o z \exp(-\frac{4z}{H})$ (3.11)

where u_* , k_0 and H are the friction velocity, the von Kármán constant, and the height of the boundary layer, respectively.

Under non-neutral conditions, an eddy diffusivity, K_s , at the surface layer (z=10m) is calculated using another model, and then extrapolated to higher altitudes by the assumption that

$$K_{v} = K_{s} \frac{\ell}{\ell_{s}}$$
(3.12)

In this model, the horizontal eddy diffusivity was assumed to be constant.

Finally, the time-dependent source emission rate was averaged over a 2-hour period for each source, and the chemical reaction rate of SO₂ expressed by

$$R = -kC \tag{3.13}$$

where k is the reaction rate constant with a given value of 10^{-4} sec⁻¹.

A second-order, central finite-difference scheme was used to integrate the advection and horizontal terms, and the Crank-Nicholson method was used for the vertical diffusion term.

In this model, 2-hour and 24-hour averaged variations of SO₂ concentrations for the 25-day period were obtained, and compared with experimental measurements for the same term average concentrations, at 10 monitoring stations. In their analysis of the results, Shir and Shieh concluded that the 2-hour data were consistently larger than the 24-hour data.

Statistical Models

As it was previously discussed, one of the problems in the Lagrangian approach is the evaluation of the transition probability density function. The most common way that has been used to overcome this problem has been to assume that the transition probability density function, $Q(x,t/x^*,t^*)$ obeys a Gaussian distribution. This assumption has given rise to several models, which will be discussed next.

Gaussian Plume Models

The best known of the practical models based on statistical theory is the Gaussian Plume model. This model was the first recognized continuous point source air pollution model. Most of the existing urban air pollution models are Gaussian Plume models. The concentration of

gas or aerosols at a point (x,y,z) due to a continuous source with an effective emission height L, is given by

$$= \frac{Q}{2\pi\sigma_{y}\sigma_{z}u} \exp \left[-\frac{1}{2}\left(\frac{y}{\sigma_{y}}\right)^{2}\right] \times \left(\exp \left[-\frac{1}{2}\left(\frac{z-L}{\sigma_{z}}\right)^{2}\right] + \exp \left[-\frac{1}{2}\left(\frac{z+L}{\sigma_{z}}\right)^{2}\right]\right) \quad (3.14)$$

The following assumptions are made:

- 1) The dispersion process is at steady state conditions.
- 2) The plume spread has a Gaussian distribution in both the horizontal and vertical planes, with standard deviations of plume concentration distribution σ_y and σ_z , respectively, which are function of only the atmospheric stability and distance x from the source.
- 3) The mean wind speed affecting the plume, u, is oriented in the x-axis and is a constant value for any height.
- 4) A constant emission rate of pollutants, Q.
- 5) A total reflection at the earth's surface.
- 6) No wind shear; and
- 7) The pollutants are chemically non-interacting.

Typically then, in these models the rate of dispersion is a function of the atmospheric stability class and the travel time or distance from the source. The lateral and vertical distributions are assumed to be Gaussian around the plume centerline. The relation between the horizontal and vertical dispersion coefficients, σ_y and σ_z , and the Pasquill-Gifford stability categories and downwind distance from the source can be found in Turner [30].

It can be observed then that the Gaussian Plume formula is not flexible enough to include all possible variations that the air motion experiences under urban atmospheric conditions.

Gaussian Puff Models

To describe more accurately the general unsteady state atmospheric diffusion case, some Puff models have been developed by Roberts et al [19] and others. In a Puff model, source emissions are broken into a series of instantaneous puffs instead of a continuous plume. The distribution within a puff is assumed Gaussian in the three directions.

In the Puff model, the entire cloud or puff is assumed to be simultaneously transported along a trajectory given by the mean flow. It is also assumed that the constant standard deviations (or diffusivities) are independent of height, up to an inversion height, that there is no wind shear, and that the pollutants are chemically non-interacting.

Monte Carlo Methods

In a Monte Carlo method, averages of a desired quantity, in this case concentration, are obtained by repeating the process many times using the same initial conditions. Each repetition is called a realization and is controlled by the random forces acting during its particular flight.

A recent model using a Monte Carlo method was developed by Bullin [4]. In this model, the turbulent diffusion process was simulated by allowing a large number of particles representing a definite amount of pollutant to diffuse through the flow field according to the following stochastic Markovian equation

$$\frac{dx_{j}}{dt} = u_{j}(x_{j},t) + \left[\frac{2}{P_{j}}\right]^{1/2} N_{j}(t)$$
(3.15)

where u_j is the instantaneous velocity in the jth direction, D is the molecular diffusivity, N_j is the jth independent Gaussian white noise with zero mean, P_j is the power spectral density of the jth white noise, and the subscript j denotes the three coordinate directions in the Cartesian space.

After a specified time, the location of each particle was recorded and the concentration distribution was calculated by counting the number of particles within cells of specified size and dividing by the cell volume. This model was solved on a hybrid computer, where equation (3.15) was programmed for repeated solutions on an analog computer with stochastic variables u_j and N_j being provided as inputs.

As it can be observed, the Gaussian distribution assumption was again incorporated in the model, and no reactions were involved in the simulation of turbulent diffusion in the atmosphere.

Chapter IV

ORTHOGONAL COLLOCATION

In view of all air pollution problems and the need for air quality improvement, air pollution modeling has great practical importance. This is the reason why much consideration has been given to this subject during recent years.

Several air pollution models have been developed so far, but since this subject is very complex, most of those models have been simplified. It is evident that the more simplifications included the less applicable a model is and the poorer the results can be.

It is well known that a major factor that characterizes diffusion processes in the atmosphere is the state of atmospheric turbulence. A general model which includes temporal and spatial variations of meteorological parameters can provide a good description of atmospheric diffusion processes.

As it was discussed in Chapter II, the Eulerian approach will be used for solving the atmospheric diffusion problem in the present work.

A finite-difference scheme has been the most widely used numerical technique for solving the partial differential equations resulting from a model based on the

Eulerian approach. The application of this technique to a general problem gives rise to a very complex model, usually with a large number of grids and in most of the cases then requires much computer time to obtain accurate results. In addition, most of those models cannot handle point sources, and artificial diffusion errors are usually present in the results.

There is a need then to keep working with the diffusion equation in its more general from as a tool for solving air pollution problems, and to investigate the feasibility of using another type of numerical technique which could have better properties than the finite-difference method. This new technique, orthogonal collocation, is discussed next.

Theory

The orthogonal collocation method belongs to the class of weighted residual methods. It was presented by Villadsen and Stewart [31], Finlayson [11], and Villadsen [32], who give details about its theory, some of which will be presented next. They also discuss several applications, mainly one or two-dimensional problems.

The method of weighted residuals is a general method of obtaining solutions to differential equations. The unknown solution is expanded in a set of trial functions, which are specified, but with adjustable constants (or functions), which are chosen to give the best solution to the differential equation.

The trial function is chosen in such a way that will satisfy the boundary conditions for all selections of the adjustable constants. This trial function then is substituted into the differential equation forming a residual. If the trial function were the exact solution, the residual would be zero. In the weighted residual method, the adjustable constants are chosen so that the residual is forced to be zero in an average sense.

Let us begin considering a boundary-value problem in one independent variable, x. A general type of differential equation can be written as

$$L^{V}(y) = 0 \quad \text{in } V$$

$$L^{S}(y) = 0 \quad \text{in } S \qquad (4.1)$$

where x is the position vector and S the surface or boundary. In the collocation method, the dependent variable y is approximated by a series expansion containing N undetermined parameters. These parameters are then calculated by applying equation (4.1) at N pre-selected points.

In the orthogonal collocation method, the series expansion consists of a set of orthogonal polynomials and the collocation points are chosen as the zeroes of the

polynomials, which make the weighted residuals to be zero in an average sense.

Therefore, for the system given by equation (4.1), the solution is approximated by

$$y(x) = y_0(x) + \sum_{i=1}^{N} a_i P_i(x)$$
 (4.2)

where

$$P_{i}(x) = \sum_{j=0}^{i} c_{j} x^{j}$$
(4.3)

is a polynomial such that successive polynomials are orthogonal to all polynomials of order less than i, with respect to some weighting function $w(x) \ge 0$:

$$\int_{a}^{b} w(x) P_{n}(x) P_{i}(x) dx = 0 , n=0,1,...,i-1$$
(4.4)

Depending on the weighting function w(x) and the interval a $\leq x \leq b$, several types of polynomials can be obtained. Choosing w(x) = 1, a = -1, b = 1, and the first polynomial $P_0(x) = 1$, the resulting polynomials, $P_1(x)$, are called Legendre polynomials.

If the weighting function is defined as

$$w(x) = (1-x)^{\alpha} (1+x)^{\beta}$$
 (4.5)

and the same previous interval is used, the Jacobi polynomials, $P_i^{(\alpha,\beta)}(x)$, are obtained [1]. It follows then that

$$P_{i}(x) = P_{i}^{(0,0)}(x)$$
 (4.6)

The polynomial $P_i(x)$, as defined by equation (4.3), has i roots in the interval a $\leq x \leq b$, which serve as collocation points. From now on, let N equal the number of those interior collocation points.

Let us now consider polynomials which have additional convenient properties: the solution of a problem is sought in the domain $0 \le x \le 1$ and is required to be symmetric about x=0. Then, it can be expanded in terms of powers of x^2 . A suitable trial function is

$$y(x) = y(1) + (1-x^2) \sum_{i=0}^{N-1} a_i P_i(x^2)$$
 (4.7)

where the a_i are undetermined constants and the $P_i(x^2)$, polynomials of degree i in x^2 that can be constructed using an orthogonality condition like equation (4.4):

$$\int_{0}^{1} (1-x^{2}) x^{a-1} P_{n}(x^{2}) P_{i}(x^{2}) dx = G_{i} \delta_{in}$$
(4.8)
$$n = 1, 2, \dots, i-1$$

where a = 1,2,3, for planar, cylindrical, or spherical geometry.

The polynomials defined by equation (4.8) are Jacobi polynomials [20], and the constant G_i is given by

$$G_{i} = \frac{\left[\Gamma\left(\frac{a}{2}\right)\right]^{2}\Gamma(i+1)\Gamma(i+2)}{(4i+a+2)\Gamma(i+\frac{a}{2})\Gamma(i+\frac{a}{2}+1)}$$
(4.9)

The coefficients of orthogonal polynomials in a new interval $0 \le x \le 1$ can also be computed from the old interval polynomials, $-1 \le x \le 1$, by the following relation [23]:
If
$$f_{i}(x) = \sum_{j=0}^{i} c_{j}x^{j}$$
, $f_{i}^{*}(x) = f_{i}(2x-1)$
$$= \sum_{j=0}^{i} c_{j}^{*}x^{j}$$
(4.10)

where $f_i^*(x)$ stands for the shifted polynomial. It should be pointed out that if the interval of orthogonality is changed, the weighting function w(x) changes for the Jacobi polynomials, and remains the same (=1) for the Legendre polynomials.

A set of N equations is needed in order to determine the N coefficients a_i , and therefore completely define the solution in the form of the trial function. This can be obtained by substituting equation (4.7) into the differential equation (4.1) and setting the residual formed equal to zero at the N collocation points x_j . These points are the roots to the Nth polynomial, $P_N(x^2) = 0$ at x_j .

The application of the method as described above becomes more confused as N increases (and also if the dimension of the differential equation increases). As a simpler and more attractive alternative to this method, as it will be seen later, the collocation equations can be obtained in terms of the solution at the collocation points, $y(x_j)$. In this method, if the solution at a different point than a collocation point needs to be obtained, it has to be interpolated by using all the known values of $y(x_j)$ at the collocation points.

For this purpose, equation (4.7) can be rewritten as

$$y(x) = \sum_{i=1}^{N+1} d_i x^{2i-2}$$
 (4.11)

where the d_i are undetermined coefficients and the (N+1) collocation point is at x=1. If equation (4.11) is evaluated at the collocation points, the following expression is obtained:

$$y(x_j) = \sum_{i=1}^{N+1} x_j^{2i-2} d_i$$
 (4.12)

The same procedure can be done for the first derivative and the Laplacian,

$$\frac{dy}{dx} \Big|_{x_{j}} = \sum_{i=1}^{N+1} \frac{dx^{2i-2}}{dx} \Big|_{x_{j}} d_{i}$$
(4.13)

$$\nabla^{2} \mathbf{y} \Big|_{\substack{\mathbf{x} \\ \mathbf{j} \\ \mathbf{i} = 1}}^{\mathbf{N}+1} \nabla^{2} (\mathbf{x}^{2\mathbf{i}-2}) \Big|_{\mathbf{x} \\ \mathbf{j}} \Big|_{\mathbf{x} \\ \mathbf{j}} \Big|_{\mathbf{i}}$$
(4.14)

These equations can be rewritten in matrix notation as follows (the square matrices have (N+1)x(N+1) elements):

$$\mathbf{y} = \overline{\mathbf{Q}}\mathbf{d} \tag{4.15}$$

$$\nabla y = \overline{R} d_{2}$$
(4.16)

$$\nabla^2 \mathbf{y} = \overline{\mathbf{T}} \mathbf{d}$$
(4.17)

where

$$Q_{ji} = x_{j}^{2i-2} , \quad R_{ji} = \frac{dx^{2i-2}}{dx} \Big|_{x_{j}},$$
$$T_{ji} = \nabla^{2} (x^{2i-2}) \Big|_{x_{j}}$$
(4.18)

The first derivative and Laplacian can be written in terms of the solution $y(x_j)$ by solving d from equation (4.15) and substituting it into equations (4.16) and (4.17):

$$\nabla \underline{\mathbf{y}} = \overline{\overline{\mathbf{R}}} \ \overline{\overline{\mathbf{Q}}}^{-1} \underline{\mathbf{y}} = \overline{\overline{\mathbf{A}}} \underline{\mathbf{y}}$$
(4.19)

$$\nabla_{\sim}^{2} y = \overline{\overline{T}} \ \overline{\overline{Q}}^{-1} \underline{y} \equiv \overline{\overline{B}} \underline{y}$$

$$(4.20)$$

Finally then, the derivatives can be expressed as

$$\nabla y = \frac{dy}{dx} \begin{vmatrix} N+1 \\ = \Sigma \\ i=1 \end{vmatrix} for j=1, \dots, N+1 \quad (4.21)$$

$$\nabla^{2} y = \left[\frac{d^{2} y}{dx^{2}} + \frac{a}{x} \frac{dy}{dx}\right]_{x} = \sum_{j=1}^{N+1} B_{ji} y_{i}$$
for j=1,...,N+1 (4.22)

where a=0,1, or 2 for plane parallel, cylindrical, or spherical symmetry, respectively, and





Integrals of the solution over the volume V can be calculated with high accuracy via the summation formula

$$\int_{0}^{1} f(x) x^{a-j} dx = \sum_{j=1}^{N+1} W_{j} f(x_{j})$$
(4.26)

where the quadrature weights are given by

$$[W_{j}] = \left[\int_{0}^{1} x^{0+a-1} dx, \int_{0}^{1} x^{2+a-1} dx, \dots, \int_{0}^{1} x^{2N+a-1} dx\right] [Q]^{-1}$$
(4.27)

Examination of equations (4.23), (4.24), and (4.25)reveals that only the collocation points are needed in order to compute matrices A and B. Once the weighting function w(x), the interval of integration a $\leq x \leq b$, and the number of interior points are specified, these collocation points can be easily calculated.

Let us now consider a more general problem, such as the problem being studied in the present work, in which the symmetry property is removed. In this case then, both even and odd powers of x are included in the orthogonal polynomials in the interval 0 to 1.

In first order differential equations, where only one initial condition needs to be satisfied, a suitable trial function can be of the form

$$y(x) = y(0) + x \sum_{i=0}^{N} a_i P_i(x)$$
 (4.28)

In this case, y is a polynomial of degree (N+1) in x, and (N+1) equations are needed in order to compute the undetermined constants a_i . These equations can be obtained by substituting equation (4.28) into the first order differential equation at the N interior collocation points, and at the end of the interval x=1.

For second-order differential equations, a general expression for the trial function can be of the form

$$y(x) = b+cx+x(1-x)\sum_{i=1}^{N} a_i P_{i-1}(x)$$
 (4.29)

This expression contains (N+2) constants that can be determined by N conditions provided by the residuals evaluated at the N interior collocation points, the N roots to $P_N(x)=0$, and two conditions provided by boundary conditions at x=0 and x=1.

If those two boundary conditions were given as y(0) and y(1), equation (4.29) would remain as

$$y(x) = y(0)[1-x]+y(1)x+x(1-x)\sum_{i=1}^{N} a_i P_{i-1}(x)$$
(4.30)

and therefore only N equations, at the N collocation points, are needed to specify the solution, y(x).

In equation (4.28) as well as in equation (4.30), the polynomials are defined by equation (4.4) with a=0and b=1. A general expression for the weighting function w(x) can be obtained from equation (4.8) in the following way:

$$\int_{0}^{1} (1-u^{2})^{\alpha} u^{\alpha-1} P_{i}(u^{2}) P_{j}(u^{2}) du = G_{i} \delta_{ij}$$
(4.31)

Substitute $x=u^2$, dx=2udu:

$$\int_{0}^{1} (1-x)^{\alpha} x^{\frac{a}{2}} \int_{1}^{-1} P_{i}(x) P_{j}(x) dx = 2G_{i} \delta_{ij}$$
(4.32)

or
$$\int_{0}^{1} (1-x)^{\alpha} x^{\beta} P_{i}(x) P_{j}(x) dx = G_{i} * \delta_{ij}$$
(4.33)

Therefore, for a non-symmetric problem, the weighting function is

$$w(x) = (1-x)^{\alpha} x^{\beta}$$
 (4.34)

This weighting function, as it was discussed before, corresponds to shifted Jacobi polynomials, and it follows then that if $\alpha=\beta=0$, i.e., w(x)=1, the polynomials obtained are shifted Legendre polynomials.

In this case, it is also convenient to express the collocation equations in terms of the solution at the collocation points, $y(x_j)$, rather than in terms of the undetermined parameters, a_j . The matrices representing the first and second derivatives, and the approximate expression representing integrals, are given by the same formula as equations (4.19), (4.20), and (4.26), but with different definitions for the elements of the matrices, which now have (N+2)x(N+2) elements:

$$Q_{ji} = x_j^{i-1}$$
 (4.35)

$$R_{ji} = (i-1)x_j^{i-2}$$
 (4.36)

$$T_{ji} = (i-1)(i-2)x_j^{i-3}$$
 (4.37)

In this case, the quadrature weights are

$$[W_{j}] = \left[\int_{0}^{1} x^{0} dx, \int_{0}^{1} x dx, \dots, \int_{0}^{1} x^{2N+1} dx\right] [Q]^{-1} (4.38)$$

Finally, the derivatives can be expressed in terms of the solution, $y(x_j)$, at the collocation points, following the same procedure as for the symmetrical case, but with different values for matrices A and B:

$$\nabla y = \sum_{i=1}^{N+2} A_{ji} y_i \quad \text{for } j=1,\ldots,N+2 \quad (4.39)$$

$$\nabla^2 y = \sum_{i=1}^{N+2} B_{ji} y_i$$
 for j=1,...,N+2 (4.40)

The procedure to solve a one-dimensional differential equation using this alternative orthogonal collocation method (with matrices A and B) is similar for both, the symmetrical and the non-symmetrical cases. Equations (4.21) and (4.22) for the first case, and equations (4.39) and (4.40) for the non-symmetrical case can be used to express the first and second derivatives in the ordinary differential equation. This equation is then reduced to coupled algebraic equations, with the unknowns being the solutions at the collocation points.

The boundary conditions at x=0 and x=1 can be substituted into those equations so that only N unknowns, at the interior collocation points, and N equations remain to be solved.

As was discussed before, once the interval and the problem are specified (i.e., symmetric or non-symmetric), matrices A and B depend only on the number of collocation points. If the same number of points is always used in the solution of a problem, these matrices have to be computed only once. Therefore, it might be enough to just include them as input data rather than having a subroutine to calculate them every time the program is run again.

Several tables and some algorithms for the interval $0 \le x \le 1$ and different weighting functions and number of collocation points are presented in [1], [11], [31], and [32]. These tables give values of the orthogonal points, matrices A and B, and the quadrature weights W_j for the symmetric case in different geometries, and also for the non-symmetric case.

In order to have more flexibility with respect to the number of points and the weighting function to be used, two subroutines (JCBI and DFOPR presented in Appendix A) were included in the computer program in this study. These algorithms were developed by Michelsen [16].

"JCBI" computes the roots of an Nth degree Jacobi polynomial and the derivatives at these points of the polynomial or the polynomial multiplied by x, by (x-1), or by both (including one or both interval end points x=0 and x=1). In this algorithm, the weighting function is given by equation (4.34).

From the quantities calculated in "JCBI", subroutine DFOPR computes the matrices for the first and second derivatives, and also the quadrature weights.

Let us now consider a more complicated problem, the solution to a two-dimensional differential equation, where the solution ϕ is a function of the independent variables x and y.

The trial function for the symmetrical case (about x=0 and y=0) can be obtained as an extension of equation (4.7). A suitable polynomial expression for $\phi(x,y)$ and subject to the symmetry conditions, and also to the boundary conditions $\phi(x,1) = \phi(1,y) = 0$, is

$$\phi(\mathbf{x},\mathbf{y}) = (1-\mathbf{x}^2) (1-\mathbf{y}^2) \sum_{i=0}^{N-1} \sum_{j=0}^{M-1} \sum_{i=0}^{M-1} \sum_{i=$$

Here, $P_i(x^2)$ and $P_j(y^2)$ are defined in the same way as it was done before, with the use of equation (4.8) for planar geometry.

As has been discussed before, this is not the best way of solving differential equations by orthogonal collocation. It is evident that the confusion becomes worse for two-dimensional problems. Therefore, let us discuss the alternative method where matrices A and B are used. Furthermore, let us analyze non-symmetric cases for being more general cases. The trial function can be rewritten then as

$$\phi(x,y) = \sum_{i=1}^{N+2} \sum_{j=1}^{M+2} d_{ij} x^{i-1} y^{j-1}$$
(4.42)

where N and M are the number of interior collocation points in the x and y directions, respectively.

Let us analyze the x-direction first. Knowing that the partial derivatives with respect to x are at constant y, the following substitution can be made into equation (4.42):

$$e_{i} = \sum_{j=1}^{M+2} d_{ij} y^{j-1}$$
(4.43)

Substituting equation (4.43) into equation (4.42) and evaluating it at the collocation points, x_k , one obtains

$$\phi \Big|_{\substack{x_{k} \in \mathbb{I}}} \sum_{i=1}^{N+2} x_{k}^{i-1} e_{i} \qquad (4.44)$$

The first and second derivatives of equation (4.42) evaluated at the collocation points are

$$\frac{\partial \phi}{\partial \mathbf{x}} \Big|_{\mathbf{x}_{k}} \stackrel{\mathbf{N+2}}{= \Sigma} (\mathbf{i}-1) \mathbf{x}_{k}^{\mathbf{i}-2} \mathbf{e}_{\mathbf{i}}$$
(4.45)

$$\frac{\partial^2 \phi}{\partial x^2} \Big|_{x_k} \stackrel{N+2}{\underset{i=1}{\overset{\sum}{\sum}}} (i-1) (i-2) x^{i-3} e_i \qquad (4.46)$$

Using matrix notation, and solving for e for the first and second derivatives, the previous expressions can be rewritten as

$$\phi = \overline{\overline{Q}}e \qquad (4.47)$$

$$\frac{\partial \phi}{\partial x} = \overline{\overline{R}} \overline{\overline{Q}}^{-1} \phi = \overline{\overline{A}} \phi \qquad (4.48)$$

$$\frac{\partial^2 \phi}{\partial x^2} = \overline{T} \overline{Q}^{-1} \phi = \overline{B} \phi$$
(4.49)

where matrices Q, R, and T are defined by equations (4.35) through (4.37).

The same procedure can be used in analyzing the y-direction. One arrives at the same equations (4.47) through (4.49), but changing x_k by y_k , the collocation points in the y-direction, the derivatives now with respect to y, and performing a similar substitution as before,

$$f_{j} = \sum_{i=1}^{N+2} d_{ij} x^{i-1}$$
(4.50)

It follows then that the same A and B matrices can be used for problems in one or two dimensions.

It should be pointed out that in the latter case, after the first and second derivatives and the boundary conditions are substituted into the differential equation, a system of (NxM) algebraic equations remains to be solved.

Partial differential equations can also be reduced to coupled ordinary differential equations, if it is not desired to go further and reduce them to algebraic equations. When both dimensions are collocated, and the differential equation is reduced to algebraic equations, the notation

used (indices) is very important and can give rise to confusion.

Let us illustrate everything that has been said before by setting the collocation equations for an example presented by Villadsen and Stewart [31].

The differential equation and the boundary conditions are:

$$\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} = -1$$
(4.51)

v=0 at x=+1 and at y=+1

This problem is symmetric about x=0 and y=0. Using the matrix B to represent the second derivative, and collocating the same number of points, N, in either direction, the collocation equations are

$$\sum_{i=1}^{N} B_{ji}v_{ik} + \sum_{i=1}^{N} B_{ki}v_{ji} = -1$$
for
$$j=1,...,N$$

$$k=1,...,N$$

where the index j represents the x-direction, and k the y-direction. The boundary points $v_{N+1,k}=v_{j,N+1}=0$ have already been substituted into equation (4.52). Equation (4.51) then has been reduced to a system of (NxN) algebraic equations, i.e., equation (4.52).

A similar analysis as done previously can be made for problems in more than two dimensions. It follows that,

as derived before, the same matrices and procedure can be used to reduce partial differential equations to coupled ordinary differential equations or algebraic equations by using the orthogonal collocation method.

Comparison to Finite Difference Techniques

Examination of the orthogonal collocation method, as presented previously, where a problem is solved for the solution at the collocation points rather than for the arbitrary coefficients in the expansion, leads to the conclusion that the equations which must finally be solved are more compact and easier to formulate, than those associated with conventional finite difference techniques.

Once a problem is converted into partial differential equations with the initial and boundary conditions known, and each term in these equations expressed as a function of the independent variables, the development of the collocation equations presents no difficulty.

The orthogonal collocation method is therefore a simple numerical technique most suitable for machine computation and with proven fast conversion.

Most of the previous models for solving air pollution problems neglect some of the terms in the general diffusion equation, so that the resulting equation is less complicated to solve. The use of the orthogonal collocation method, as discussed before, will not introduce extra complications in the formulation of the final equations and their solution if those terms are not assumed to be negligible. Therefore, it will not be more difficult if a general case (four independent variables, the three directions and time) is solved, even with a non-linear reaction term.

Another advantage of the collocation method is that a much smaller number of points may be used since the solution at each point is influenced directly by the value at all the collocation points, as is the case for the exact solution, instead of depending directly on only neighboring grid points as is the case in finite difference schemes. In this respect then, the collocation method requires less computation time than a finite difference solution of comparable accuracy. This has been proved in several applications, some of them discussed next.

Ferguson and Finlayson [9] applied orthogonal collocation to a linear unsteady-state diffusion problem in a slab. They reported that the collocation solution is more accurate than finite difference solutions which use three to twelve times as many spatial grid points (they only collocated in the spatial coordinate and solved the resulting ordinary differential equations).

Ferguson and Finlayson also solved the diffusion of mass and energy in a spherical catalyst pellet with an

exothermic first-order irreversible reaction. Here, they again collocated only in the x-direction and these equations were integrated using Hamming's method. In this case, Ferguson and Finlayson demonstrated that the number of collocation points can be about ten times less than the number of finite difference grid points for equivalent accuracy. In terms of time, the best collocation solution was about twenty times (or more) as fast as a finite difference solution of about the same accuracy.

Finlayson [10] used orthogonal collocation to solve a two-dimensional (axial and radial directions) packed bed reactor under the assumptions of constant physical properties, plug flow, and a reaction governed by the conversion and temperature. Collocation was applied in the radial axis, and the resulting differential equations integrated using the Runge-Kutta method and the Euler method.

Comparison between collocation solutions and finite difference calculations using a Crank-Nicholson implicit method were reported to be in some cases from two to four times faster for equivalent accuracy. In some other cases, the collocation method was twice as fast and ten times as accurate or four times as fast and five times as accurate.

Finlayson [11] also applied collocation in the axial direction in place of the Runge-Kutta method used before, and solved the resulting set of algebraic equations by an iterative procedure proposed by Villadsen [32]. The solution obtained by this way was about three times faster than the one using the Runge-Kutta method for equivalent accuracy.

Finlayson [11] also solved the problem of tubular reactors with only axial dispersion (concentration and temperature). For this case, when orthogonal collocation was applied, the governing equations were reduced to a set of nonlinear algebraic equations for the concentration and temperature at the collocation points. He used the Newton-Raphson method to solve them, and compared this solution to those given in the literature using finite difference schemes.

Finlayson used six collocation points in order to obtain the same accuracy as reported for both a 100-grid point finite difference solution for the isothermal case and 481 grid points for the non-isothermal case.

Application to Turbulent Diffusion in the Atmosphere

As was discussed before, only finite-difference has been used previously to solve for the concentration distribution in atmospheric diffusion processes. The analysis

made in the preceding section leads to the conclusion that air pollution modeling could be improved by the use of the orthogonal collocation method as the numerical technique for solving the general diffusion equation. This equation, for a single species which will be the case studied in the present work⁽¹⁾, is repeated here for convenience.

$$\frac{\partial C}{\partial t} + u \frac{\partial C}{\partial x} + v \frac{\partial C}{\partial y} + w \frac{\partial C}{\partial z} = \frac{\partial}{\partial x} (K_x \frac{\partial C}{\partial x}) + \frac{\partial}{\partial y} (K_y \frac{\partial C}{\partial y}) + \frac{\partial}{\partial z} (K_z \frac{\partial C}{\partial z}) + R \qquad (4.53)$$

Some of the attractive properties that orthogonal collocation has are its flexibility and easy handling of any variable or component in the diffusion equation, as compared to other numerical methods used in previous works. The model does not increase in difficulty if all the terms involved in equation (4.53) are incorporated in it. That is, a similar amount of work in formulating the collocation equations, but not computer time in solving them, has to be done if three instead of two dimensions are considered in the model. The same reasoning applies for all the components of the wind velocity and turbulent diffusivity, and the chemical reactions. Therefore, it is

Symbols indicating averaged quantities will be omitted henceforth. All velocities and concentrations, however, continue to be time averaged quantities.

not necessary to assume that some terms are negligible in order to be able to obtain results, which in such a case would be applicable only for some particular situations.

With the present approach then, it is possible to simulate many types of atmospheric conditions. But to do so, it is necessary to express all the properties in equation (4.53) in an analytical form. This has been done in a flexible way by introducing several parameters in the expressions representing those properties, so that different input data can be supplied if different atmospheric conditions are to be tested.

Before applying orthogonal collocation to the present model, let us obtain those expressions for the initial and boundary conditions, turbulent diffusivities, velocity profiles, and chemical reaction.

Initial and Boundary Conditions

The initial conditions necessary for an unsteady state model can be expressed as follows,

 $C = C_B$ at t=0 and any (x,y,z) (4.54) where C_B is a constant and in general can be a function of x, y, and z.

These conditions are required to start the calculations performed by the integrating subroutine, and therefore are generated in the MAIN of the computer program. For all the cases studied in the present work, there was no background concentration and thus C_B was equated to zero.

The boundary conditions at the ground, z=0, and at the inversion base, z=H, are obtained by assuming that both completely reflect the diffusing pollutant, i.e.,

$$\frac{\partial C}{\partial z} = 0$$
 at z=0, H for any x and y (4.55)

These two boundary conditions for the z direction are converted into collocation equations and then incorporated in the diffusion equation, as it will be seen later in this chapter. If for a particular case the gradient in equation (4.55) is not equal to zero, e.g.

 $-K_z \frac{\partial C}{\partial z} = Q_s(x,y,t)$ at z=0 for any x and y (4.56) where $Q_s(x,y,t)$ is the surface flux of the pollutant studied, the collocation equations resulted from conditions (4.55) have to be changed.

The bulk transport of the pollutant is due primarily to the effect of the x-component of the wind velocity. Therefore, it can be assumed that at certain distance in the y-direction, the mean concentration is negligible, i.e.

$$C \rightarrow 0$$
 at $y = -y_{max}$, y_{max} for any x and z (4.57)
where $2 \cdot y_{max}$ is the entire region of interest. In general,
 $C = f(x,z,t)$ at $y = -y_{max}$, y_{max}

for any x and z (4.58)

This can also be included in the model by changing the resulting collocation equations from conditions (4.57).

For the purposes of solving equation (4.53), it is necessary to define the location of the sources. These sources can be defined with the aid of a boundary condition in the x-direction.

Let us consider a general case, i.e., a multiple source problem, where there is more than one source involved. If the position of the ith source is defined by the coordinate point (JS(i), KS(i), LS(i)), the boundary condition related to the ith source is given by

 $C = \begin{cases} C_{o}(i,t) & \text{at } x=JS(i), y=KS(i), z=LS(i) \\ 0 & \text{at } x=JS(i) \text{ and any other } y \text{ and } z \end{cases}$ (4.59) where $C_{o}(i,t)$ is the ith source concentration, and in general a function of time. For the purpose of studying different situations, each C_{o} , assumed constant in the present model, and its position is supplied as input information. The way of handling multiple source will be discussed in detail in Chapter V. As a convention, and knowing that for any case there will be at least one source, the first source will always be at x=0, that is JS(1)=0. Therefore, if the problem involves only one source, the boundary condition in the x-direction is given by

$$C = \begin{cases} C_{o}(1) & \text{at } x=0, y=KS(1), z=LS(1) \\ 0 & \text{at } x=0 \text{ and any other y and } z \end{cases}$$
(4.60)

Sometimes the emission rate (mass/time) and not the source concentration is given as information. This problem can be overcome by the use of quadrature weights in orthogonal collocation and will be discussed later in Chapter V.

There is no need for a second boundary condition in the x-direction, due to the assumption of negligible turbulent diffusivity in this direction. This will be discussed in the next section.

Turbulent Diffusivities, K

Eschenroeder and Martinez [7] assumed a trapezoidal shaped vertical profile from the ground up to the inversion base for K_z , the turbulent diffusion coefficient in the vertical direction. At low levels, K_z increases linearly with z to some constant value. At intermediate levels, K_z then is constant. As the mixing depth top is approached, K_z decreases linearly with increasing z. The ground level and inversion layer values are assumed to be equal. The maximum constant value is expected to vary with wind speed approximately like

$$K_z = 50(\bar{u}+5) m^2/min$$
 (4.61)

where \bar{u} is in m/s. This equation starts applying from an elevation of 40 to 100 meters, depending upon meteorological conditions:

Figure 4.1: Variations of Diffusivity with Height



Eschenroeder et al. [8] reviewed their previous work based on more experimental data and concluded that equation (4.61) gives an estimate of vertical diffusivity only in the neutral stability range, and that temperature gradients are apparently far more influential than wind speed in determining vertical diffusivity values.

As an improvement on their previous approach, the dependence on wind speed was dropped, and the diffusivity profiles were reconstructed to represent the broad stability categories (defined by Pasquill and Gifford [30]), as shown in Figure 4.2.

The knee height Δ , where the constant values for K_z apply, varies between 25m to 75m above the ground, depending on the height range of the model and the mesh interval size.

Eschenroeder et al. assumed constant diffusivity profiles all the way to the top because of their little influence for the space and time scales in their calculations at the top of the mesh.

In the present model, the calculation procedure for the turbulent diffusivities is based upon the latter approach used by Eschenroeder et al. [8]. Using their approach, K_z will depend only on the stability class and the vertical direction. Some slight changes were made in that approach and are now discussed.



Figure 4.2: Vertical Diffusivity Profiles.

An additional stability class, between the neutral and unstable, was incorporated in the model so that the six classes presented by Turner [13] could be taken into account.

The knee height Δ , was made dependent on the stability class, as suggested by Pasquill [18], so that the more unstable the conditions, the higher the knee.

The last change made to Eschenroeder's approach was to remove their assumption of constant diffusivity profiles from intermediate levels all the way to the top. As the mixing depth top is approached then, the turbulent diffusivities decrease linearly with increasing z, and the ground level and inversion layer values are assumed to be equal. This is done for the last 100 meters of the z-dimension, with H>300m.

Taking all these changes into account, Figure 4.2 remains as shown in Figure 4.3.

This model was implemented in the computer program by using the following equations ⁽¹⁾:

For $ISTB \leq 4$:

 $K_{z} = COEFK(ISTB) \cdot z/\Delta(ISTB) + 60., \text{ for } z<\Delta(ISTB)$ $K_{z} = COEFK(ISTB) + 60., \text{ for } \Delta(ISTB) \leq z \leq (H-100)$ $K_{z} = COEFK(ISTB) \cdot (H-z)/100 + 60., \text{ for } z>(H-100)$

(1) In the computer program the units for K_z are m^2/min .



where, COEFK(1) = 2940 $\Delta(1) = 125$ COEFK(2) = 540 $\Delta(2) = 125$ COEFK(3) = 258 $\Delta(3) = 100$ (4.62) COEFK(4) = 90 $\Delta(4) = 75$ For ISTB = 5: $K_z = 60 \text{ m}^2/\text{min}$, for any z

for ISTB = 6: $K_z = 30 \text{ m}^2/\text{min}$, for any z

In order to have a flexible model, the stability class, ISTB, is supplied as input information by the user. Because of the fact that once it is defined it remains the same during the simulation, the algorithm that calculates the turbulent diffusivities at each orthogonal point was incorporated in the MAIN of the computer program.

In the region of interest, the turbulence is almost perfectly isotropic, and even below 100 meters, the degree of isotropy seems to be sufficiently high in order to consider, as Sutton [28] has suggested, that K_y varies in an identical way as K_z . In addition, most of the previous models, discussed in Chapter III, considered a constant turbulent diffusivity in the y direction. This suggested then to relate, in the present model, K_y to K_z in the following manner:

$$K_{y} = \propto K_{z}'(ISTB)$$
(4.63)

where \propto is a positive constant which has to be supplied as input information by the user, and K' is the maximum constant value for K_z and thus dependent only on the stability class.

In the present model, the turbulent diffusivity in the x-direction will be neglected as compared to the mean wind velocity. The reason being that its behavior is not well understood and in most of the cases it has no significant effect on the results [24].

Velocity Profiles

Davies [5], Pasquill [17], Smith [26], Tennekes and Lumley [29], and others have developed expressions for the velocity profiles especially in the mean flow direction. Two equation forms have been used often in previous studies. A semi-logarithmic form for rough flows, of the type

$$\bar{u}(z) = \frac{u_{\star}}{k_0} \ln(\frac{z}{z_0})$$
 (4.64)

where u_{\star} is the friction velocity, k_{O} the von Karman's constant, and z_{O} a length characterizing the effect of the surface roughness.

The second is a power law equation which has the following form:

$$\bar{u}(z) = \bar{u}_1 \left(\frac{z}{z_1}\right)^m$$
 (4.65)

where \bar{u}_1 is the mean velocity at a reference height z_1 , and m is a constant.

For this problem, the latter form describes in a better way the mean velocity in the x-direction. Therefore, equation (4.65) will be used in the diffusion equation.

This form of the velocity profile is used for the lower part of the z dimension, that is from the ground level to the knee height Δ , previously defined, and from this elevation up to the mixing depth top a constant value for the velocity is used, UST. This means then that u will depend on the stability class, z, and the parameters u₁, z₁, m; and UST.

Usually UST, the geostrophic wind speed, is the easiest value to obtain whether by experimental data or previous records of the zone in question. This is the reason why UST is required as input information in the model.

Having in mind further adjustments for u, the constant m has also to be provided as input information by the user. This exponent m is a number which varies from 0.1 to 0.4, depending on the roughness of the ground surface as well as on atmospheric stratification. The rougher the terrain, i.e., the larger the surface obstructions, the thicker will be the affected layer of air, and the more gradual will be the increase of velocity with height. Thus, as the roughness increases, the exponent m increases, as it is shown in Table 4.1 which was presented by Seinfeld [22]:

TABLE 4.1 : ESTIMATES FOR m in eq.(4.65)

Type of terrain

	open country	suburbs	urban	
m	.16	.28	.40	

The stability limits of m, also presented by Seinfeld [22], are:

٢	.83	very stable	
m = {	1/7	neutral	
ί	.02	very unstable	

For purposes of continuity calculations (see Chapter V) the ground level value for the mean wind velocity in the x-direction, UGR, is also taken into consideration in the model and is supplied as input information. If its value is not known at the moment of simulating a case, as it happened in this work, it can be assumed to be equal to the friction velocity, u_* .

The algorithm for obtaining u then, as used in the MAIN program, is described by the following equations: For ISTB ≤ 4 :

	u =	UGR	for $z = 0$	
	u =	UST $\left(\frac{z}{\Delta(ISTB)}\right)^{AM}$	for 0 <z<∆(istb)< th=""><th></th></z<∆(istb)<>	
	u =	UST	for $z \geq \Delta$ (ISTB)	
For	ISTB>4	:		(4.66)
	u =	UGR	for $z = 0$	
	u =	UST	for $z > 0$	

In almost all the previous works in air pollution modeling, the y-component of the mean wind velocity as well as the z-component are assumed to be negligible.

As it was discussed before, the present model does not increase in difficulty by including more than one component for the mean wind velocity. In spite of this, it has been difficult to obtain equation forms for the other two components of the wind velocity. Therefore, it will be assumed that W, the vertical component is negligible, and that V, the y-component is some fraction of u.

To make a general case for this study, let us consider that v also varies with respect to time in a linear

dependence until it reaches its maximum value. Because of this dependency on time, the algorithm used to its calculation, as described by the following equations, was incorporated in a subroutine called FCT:

$$v = Pu \frac{t}{TCH}$$
 for $0 \le t \le TCH$ (4.67)
 $v = Pu$ for $t \ge TCH$

where P is a constant that will determine the direction of the mean wind velocity, and TCH another parameter that will specify the time required for v to change from its value at t=0 to its maximum value, P.u. Both, P and TCH have to be supplied as input information in the computer program by the user.

Chemical Reactions

In the development of equation (2.6), the general diffusion equation, an approximation concerning the chemical reaction term was made: the term $<R_i(<C_1>+C_1', ..., <C_s>+C_s')>$ was replaced by $<R_i(<C_1>, ..., <C_s>)>$, i.e., the effect of concentration fluctuations on the rate of reaction was neglected. This is true only for first-order reactions, however, conditions under which this assumption is valid can be obtained for higher order reactions.

Lamb [15] analyzed this problem for a two-dimensional turbulent fluid transporting a single chemical species

involving a nonlinear chemical reaction (second-order reaction). In this case, the chemical reaction term is expressed by

$$R = -k[\langle C \rangle^{2} + \langle C \rangle^{2} \rangle]$$
(4.68)

He concluded that the replacement of equation (4.68)by $-k < C >^2$ is valid when the reaction process is slow compared with turbulent transport and the characteristic ' length and time scales for changes in the mean concentration field are large compared with the corresponding scales for turbulent transport.

Although these conditions were derived for a relatively simple two-dimensional case, the essential aspects of those restrictions were found to apply to the general three-dimensional equation as well. In addition, they give a good indication of the conditions of validity for general R_i.

Let us turn our attention now to the mechanisms of removal of sulfur dioxide in the atmosphere, since this will be the chemical species used to validate the present model.

Although a great deal of importance has been given to sulfur oxides, in particular sulfur dioxide, the chemistry of sulfur dioxide in the atmosphere is still far from being fully understood.

The removal of SO₂ in the atmosphere is quite complex and can take place through several mechanisms. It has been suggested that atmospheric SO₂ can undergo oxidation to sulfates by mainly two mechanisms: catalytic (heterogeneous) oxidation and photochemical (homogeneous) oxidation. However, sulfur dioxide is removed from the atmosphere not only by oxidation, but also by sedimentation, rainout and washout.

Photochemical oxidation of SO_2 is apparently a gas-phase process consisting of several chemical reactions. In the presence of air, SO_2 is oxidized to SO_3 when exposed to solar radiation, and if water is present, the SO_3 is rapidly converted to sulfuric acid. The conversion of SO_2 to SO_3 involves excited SO_2 molecules, oxygen, and oxides of sulfur other than SO_2 . Although photochemical oxidation of SO_2 can take place in clean air, the more important process of SO_2 photooxidation occurs in atmospheres containing hydrocarbons and oxides of nitrogen. In this case, the rate of conversion of SO_2 to SO_3 increases markedly over that observed in pure air.

Catalytic oxidation is the principal process for SO₂ conversion under conditions of high humidity and high particulate concentration. It occurs in aqueous solution, involves both water and dissolved O₂, and requires the presence of a catalyst. The overall reaction can be expressed as

$$2SO_2 + 2H_2O + O_2 \xrightarrow{\text{cat.}} 2H_2SO_4$$
 (4.69)

Catalysts for this reaction include several metal salts, such as sulfates and chlorides of manganese and iron. However, most of the recent work has been dedicated to manganous salts only.

The same reaction can also occur without the presence of a catalyst, as discussed by Worley [34], and shown in the following equations,

$$SO_{2} + H_{2}O \longrightarrow HSO_{3} + H^{+}(seconds)$$

$$2HSO_{3}^{-} + O_{2} \xrightarrow{aq} 2SO_{4}^{-} + 2H^{+}(hours)$$

$$(4.70)$$

In the sedimentation or deposition process, SO_2 behaves as if it were a particle with a settling velocity. Rainout and washout also serve as additional removing mechanisms. Rainout is the incorporation of gaseous SO_2 in cloud droplets, while washout is the removal of SO_2 by rain falling through air masses below the cloud level. Once SO_2 is absorbed, it is converted via oxidation (4.70)the heterogenous aspect causes apparent SO_2 level to fall.

As it can be observed, a model that would take into account all the possible mechanisms for SO₂ removal in the atmosphere and all the variables involved would be quite complex.
Because of the fact that the processes previously discussed are not very well understood yet, and the incorporation and thus the study of such a complex model is not the main objective in the present work, only photochemical oxidation will be considered here. Furthermore, the reactions taking place in a system of SO_2 , hydrocarbons, NO_x , and air are probably the least well understood of all those in atmospheric chemistry, and thus will not be considered either.

Since data for photochemical oxidation in clean air are extremely scattered, a simplified first order reaction model will be used to represent SO₂ removal from the atmosphere,

$$R = -kC \tag{4.71}$$

where k is the reaction rate constant. The value of 5.8×10^{-5} per minute or 0.35% loss of SO₂ per hour, used by Hallidy and Anderson [12] in their work will be the rate constant in the present model, and must be supplied as input information in the computer program.

Development of the Collocation Equations

The diffusion equation to be solved, and the corresponding initial and boundary conditions, under the assumptions previously discussed, are:

$$\frac{\partial C}{\partial t} + u \frac{\partial C}{\partial x} + v \frac{\partial C}{\partial y} = \frac{\partial}{\partial y} (K_y \frac{\partial C}{\partial y}) + \frac{\partial}{\partial z} (K_z \frac{\partial C}{\partial z}) + R \qquad (4.72)$$

$$C = 0$$
 at t=0, and any (x,y,z) (4.73)

$$\frac{\partial C}{\partial z} = 0$$
 at z=0, H for any x and y (4.74)

$$C^{(*)} = \begin{cases} C_{0}(1) & \text{at } x=0, y=KS(1), z=LS(1) \\ 0 & \text{at } x=0, \text{ and any other } y \\ and z \end{cases}$$
(4.75)

$$C = 0 \qquad \text{at } y = -y_{\text{max}}, y_{\text{max}} \qquad (4.76)$$
for any x and z

In equation (4.72), the velocity profiles are assumed to be,

$$\mathbf{u} = \psi(\mathbf{z}, \text{ISTB}) \tag{4.77}$$

$$\mathbf{v} = \eta(\mathbf{z}, \mathbf{t}, \text{ISTB}) \tag{4.78}$$

the turbulent diffusivities,

$$K_{y} = \xi (ISTB)$$
(4.79)

$$K_{z} = \zeta(z, ISTB)$$
(4.80)

and the rate of reaction,

$$R = -kC \tag{4.81}$$

For this problem, let us transform the spatial coordinates to yield limits of 0 to 1 by the following procedure:

^(*) The multiple sources case will be discussed in Chapter V.

$$\mathbf{x}^{\star} = \frac{\mathbf{x}}{\mathbf{x}_{\max}}$$
(4.82)

$$y^* = \frac{\frac{y}{y_{max}} + 1}{2}$$
(4.83)

$$z^* = \frac{z}{H} \tag{4.84}$$

For simplicity, from now on the asterisk will be dropped out from the independent variables, x^* , y^* , and z^* .

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Substituting equations (4.77) through (4.84) into equation (4.72) one obtains

$$\frac{\partial C}{\partial t} + R_1 \frac{\partial C}{\partial x} + R_2 \frac{\partial C}{\partial y} - R_3 \frac{\partial C}{\partial z} = R_5 \frac{\partial^2 C}{\partial y^2} + R_6 \frac{\partial^2 C}{\partial z^2} - kC \qquad (4.85)$$

where

$$R_1(z) = \psi(z, ISTB) \frac{1}{x_{max}}$$
 (4.86)

$$R_2(z,t) = \eta(z,t,ISTB) \frac{1}{2y_{max}}$$
 (4.87)

$$R_{3}(z) = \frac{d\{\zeta(z, ISTB)\}}{dz} \frac{1}{H^{2}}$$
(4.88)

$$R_5 = \xi (ISTB) \cdot \frac{1}{4y_{max}^2}$$
 (4.89)

$$R_{6}(z) = \zeta(z, ISTB) \frac{1}{H^{2}}$$
 (4.90)

There are many ways in which orthogonal collocation can be applied to equation (4.85), but it is chosen to collocate only in the three directions, x, y, and z, so that a system of ordinary differential equations with respect to time is left to be solved. The reason of doing this is because the solution of the diffusion equation is primarily wanted at any instant of time.

If the emission source is put in the middle of the interval in the y-direction, and only the x-component of the wind velocity is taken into account, then by examining the boundary conditions it can be concluded that this problem would be symmetric with respect to y. Since this is only a particular case, it is better to apply the equations for a non-symmetric case to every direction.

In doing this, let N_x , N_y , and N_z be the number of collocation points in the x, y, and z directions, respectively, and C_{jkl} the mean concentration at the point (x_j, y_k, z_l) .

Equation (4.85) remains then as,

$$\frac{dC_{jkl}}{dt} + R_{1}(l) \sum_{i=1}^{N} A_{ji}^{(1)}C_{ikl} + R_{2}(l,t) \sum_{i=1}^{N} A_{ki}^{(2)}C_{jil} - \frac{N_{z}^{+2}}{i=1} - R_{3}(l) \sum_{i=1}^{N} A_{li}^{(3)}C_{jki} = R_{5} \sum_{i=1}^{N} B_{ki}^{(2)}C_{jil} + R_{6}(l) \sum_{i=1}^{N} B_{li}^{(3)}C_{jki} - kC_{jkl} - kC_{jkl}$$

As discussed before, matrices A and B depend on the number of collocation points. For a general case, that number of points can be different for each direction and therefore, A and B would be different. Furthermore, N_x , N_y , and N_z could be changed between runs. These reasons lead to the conclusion that it is better to include subroutines JCBI and DFOPR in the computer program rather than putting matrices A and B as input data each time the collocation points are changed.

In equation (4.91), the indices of matrices A and B represent the direction and thus the way they are computed, i.e., (1), (2), and (3) stand for the x, y, and z directions, respectively.

The application of orthogonal collocation to the boundary conditions, equations (4.74) through (4.76), gives the following expressions:

$$N_{z}^{+2} = 0 \quad \text{at } z=0 \quad (4.92)$$

$$N_{z}^{+2} = 0 \quad \text{at } z=0 \quad (4.92)$$

$$N_{z}^{+2} = 0 \quad \text{at } z=1 \quad (4.93)$$

$$i=1 \quad N_{z}^{+2}, iC_{jki} = 0 \quad \text{at } z=1 \quad (4.93)$$

$$C_{lkl} = \begin{cases} C_{0}(1) & \text{at point source} \\ 0 & \text{elsewhere} \end{cases}$$

$$C_{jll} = 0 & \text{at } y=0 \qquad (4.95)$$

$$C_{jN_y+2l} = 0$$
 at y=1 (4.96)

Equations (4.92) and (4.93) can also be written as

$$A_{11}^{(3)}C_{jkl} + \sum_{i=2}^{N_{z}+1} A_{1i}^{(3)}C_{jki} + A_{1N_{z}+2}^{(3)}C_{jkN_{z}+2} = 0 \quad (4.97)$$

$$A_{N_{z}+2}^{(3)}C_{jkl} + \sum_{i=2}^{N_{z}+1} A_{N_{z}+2i}^{(3)}C_{jki} + \sum_{i=2}^{N_{z}+2i} A_{N_{z}+2i}^{(3)}C_{jki} + A_{N_{z}+2N_{z}+2}^{(3)}C_{jki} + A_{N_{z}+2N_{z}+2}^{(3)}C_{jki} + A_{N_{z}+2N_{z}+2}^{(3)}C_{jkN_{z}+2} = 0 \quad (4.98)$$

Solving for C_{jkl} and C_{jkN_z+2} one obtains:

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$$C_{jkN_{z}+2} = \frac{\sum_{i=2}^{N_{z}+1} (A_{11}^{(3)} A_{N_{z}+2i}^{(3)} - A_{N_{z}+21}^{(3)} A_{1i}^{(3)}) C_{jki}}{A_{N_{z}+21}^{(3)} A_{1N_{z}+2}^{(3)} - A_{11}^{(3)} A_{N_{z}+2N_{z}+2}^{(3)}}$$
(4.99)

$$c_{jkl} = -\frac{{}^{(3)}_{A_{lN_{z}+2}C_{jkN_{z}+2} + \sum_{i=2}^{x} A_{li}^{(3)}c_{jki}}{A_{l1}^{(3)}}$$
(4.100)

These expressions can be simplified by defining,

$$DEN = A_{N_{z}+21}^{(3)}A_{1N_{z}+2}^{(3)} - A_{11}^{(3)}A_{N_{z}+2N_{z}+2}^{(3)}$$
(4.101)

APA1(i) =
$$\frac{A_{11}^{(3)}A_{N_{z}+2i}^{(3)} - A_{N_{z}+21}^{(3)}A_{1i}^{(3)}}{DEN}$$
(4.102)

so that equation (4.99) remains as

$$C_{jkN_z+2} = \sum_{i=2}^{N_z+1} APAI(i)C_{jki}$$
(4.103)

In cases where there is no inversion layer at the maximum elevation, H, as it has been assumed so far, a new boundary condition at this point can be assumed as

C = 0 at z=H for any x and y (4.104) This condition can be easily included in equation (4.103):

$$C_{jkN_z+2} = \sum_{i=2}^{N_z+1} APA1(i) C_{jki} INVRS$$
(4.105)

where INVRS=1 means that there is inversion at z=H, and INVRS=0 means that there is no inversion and the concentration at z=H is equal to zero.

By substituting equations (4.94) through (4.96), equation (4.91) remains as follows:

$$\frac{dc_{jk\ell}}{dt} + S(j) + R_{1}(\ell) \sum_{i=2}^{N_{x}+2} A_{ji}^{(1)} c_{ik\ell} + R_{2}(\ell,t) \sum_{i=2}^{N_{y}+1} A_{ki}^{(2)} c_{ji\ell} - R_{3}(\ell) A_{\ell1}^{(3)} c_{jk1} - R_{3}(\ell) \sum_{i=2}^{N_{z}+1} A_{\elli}^{(3)} c_{jki} - R_{3}(\ell) A_{\ell N_{z}+2}^{(3)} c_{jk N_{z}+2} = R_{5} \sum_{i=2}^{\Sigma} B_{ki}^{(2)} c_{ji\ell} + R_{6}(\ell) B_{\ell1}^{(3)} c_{jk1} + R_{6}(\ell) \sum_{i=2}^{N_{z}+1} B_{\elli}^{(3)} c_{jki} + R_{6}(\ell) B_{\ell N_{z}+2}^{(3)} c_{jkN_{z}+2} - kc_{jk\ell}$$

$$(4.106)$$

And rearranging,

.

$$\frac{dC_{jk\ell}}{dt} = -S(j) - R_{1}(\ell) \sum_{i=2}^{N_{x}+2} A_{ji}^{(1)}C_{ik\ell} + + [R_{3}(\ell)A_{\ell1}^{(3)} + R_{6}(\ell)B_{\ell1}^{(3)}]C_{jk1} + + [R_{3}(\ell)A_{\ell N_{z}+2}^{(3)} + R_{6}(\ell)B_{\ell N_{z}+2}^{(3)}]C_{jkN_{z}+2} + \frac{N_{z}+1}{i=2} [R_{3}(\ell)A_{\ell i}^{(3)} + R_{6}(\ell)B_{\ell i}^{(3)}]C_{jki} + i=2 [R_{5}B_{ki}^{(2)} - R_{2}(\ell,t)A_{ki}^{(2)}]C_{ji\ell} - kC_{jk\ell}$$
(4.107)

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or,

$$\frac{dC_{jkl}}{dt} = -S(j) - R_{1}(l) \sum_{i=2}^{N_{x}+2} A_{ji}^{(1)}C_{ikl} + R361(l)C_{jk1} + R362(l)C_{jkN_{z}+2} + \sum_{i=2}^{N_{z}+1} R361(l,i)C_{jki} + \frac{N_{y}+1}{i=2} R521(l,k,i,t)C_{jil} - kC_{jkl}$$
for $j=2, \ldots, N_{x}+2$

$$k=2, \ldots, N_{y}+1$$

$$l=2, \ldots, N_{y}+1$$

where

$$S(j) = \begin{cases} R_{1}(LS(1))A_{j1}^{(1)}C_{0}(1) \text{ at } k=KS(1) \\ \ell=LS(1) \\ 0 \text{ elsewhere} \end{cases}$$
(4.109)

and C_{jkl} and C_{jkN_z+2} are given by equations (4.100) and (4.105), respectively.

Equation (4.108) gives then a set of $(N_x+1)(N_y)(Nz)$ first-order ordinary differential equations to solve for the concentration as a function of time at the orthogonal collocation points in the three directions, x, y, and z.

Chapter V

FORMULATION OF CALCULATIONAL SCHEME

Calculation Procedure

It has previously been shown that the solution to the turbulent diffusion process in the atmosphere can be obtained by the use of the K-theory, which can be accomplished by solving a partial differential equation. Orthogonal collocation simplifies this, reducing that equation to a system of first-order ordinary differential equations with respect to time.

The basic calculational procedure then is to solve that system of equations (4.108) on a digital computer. There are several methods that can be used for this purpose, but in this work it was decided to use RKGS, a subroutine furnished by IBM [14] which in essence is a fourth-order Runge-Kutta method.

In addition to RKGS, two other subroutines must be supplied: A function subroutine, called FCT in the computer program (see Appendix A), where the system of firstorder ordinary differential equations to be solved are provided; and an output subroutine, called OUTP in the computer program, which will print the results computed by RKGS. In RKGS, the integration is performed with respect to time, and therefore the solution of the problem can be obtained at any moment. However, it is usually desired to print it only at every defined interval of time. Having this in mind, an algorithm that uses a variable called PRDEL was incorporated in OUTP so that it will allow the computer to print the results only every PRDEL units of time.

Due to the nature of orthogonal collocation, small oscillations can be introduced in the solution of the problem. Although these are part of the true solution of the collocation equations, there might be small negative concentrations as output for points that follow zero levels of concentration. Because of this, and by suggestions of Stewart [27] and Villadsen [33], a subroutine called ZERO is always called before printing the results. This subroutine will convert all the concentration values that follow the zero level to zero. This testing is done in all three directions.

Because of the fact that actual data are available as average concentrations, the computer program contains a subroutine called AVG, which will convert the solution to a time average solution, so that a comparison to experimental data can be performed. To do this, AVG subroutine computes the time average concentrations at

every collocation point for three consecutive points in time; i.e., θ , t, and Ω , where t is any point in time, θ (=t-PRDEL) is the time between the pollutant release(t=0) and the initiation of the averaging time, and Ω (=t+PRDEL) is the end of the averaging time. In order to reduce the program storage requirement, AVG subroutine is performed by direct-access input/output statements.

Multiple sources

In Chapter IV, orthogonal collocation was applied to problems involving only one point source. The same resulting collocation equations can be used to multiple source cases, but the calculational procedure is different.

As it was previously discussed, the concentration at the source is considered here as the boundary condition in the x-direction. If the problem involves only one source, the input information required to solve it consists of $C_0(1)$, the concentration at the source, KS(1) and LS(1), the location of the point source in the y and z directions, respectively (JS(1) in this case is equal to zero, i.e., x=0), and XMAX, YMAX, and H, the dimensions in the three directions. The calculation of the solution starts at TINIT, the initial time and ends at ENDS, the end of the simulation time. Intermediate results are computed at every interval of integration time, PRMT(3),

but are printed only every PRDEL units of time. TINIT, ENDS, PRMT(3), and PRDEL are also required as input information.

The calculation procedure for cases involving more than one source is different than the one previously presented. The reason being the interaction that occurs on the concentration distribution due to the multiple sources.

In order to describe the calculation procedure for multiple sources, let us analyze a situation that involves three point sources. In this case, the source concentrations, $C_0(i)$ are still considered as boundary conditions in the x-direction, but while the first source is located at x=0, the following ones can be located in general at x>0. The entire domain in the x-direction is divided into three portions, so that the first source is located at x=0, the second at XMAX(1), and the third at XMAX(2). In order to define completely each source, its position in the y and z directions must be supplied by the values of KS(i) and LS(i), respectively, where i indicates the number of the source. This case then can be represented as it is shown in Figure 5.1.

The input information given by TINIT, ENDS, PRMT(3), PRDEL, NX, NY, NZ, YMAX, and H is the same for each source. The number of sources, NSRCS, must also be supplied as input information, and in the present model



NSRCS < 3. If more than three sources are to be considered, changes should be made in the dimension and common statements, initial conditions, and RKGS call statements.

The results for the first portion of Figure 5.1 are the same as if they were computed for a single source problem, but for the next portions this is different. The concentrations in the second portion not only depend on $C_0(2)$, but also on the mass flux that comes from the first portion. The same reasoning applies for subsequent portions.

The calculation procedure then is as follows: at every interval of integration time, the concentration values at any orthogonal point in the y-z plane at x=XMAX(1) are recorded and used, with $C_{c}(2)$, as boundary conditions for the second portion. The same procedure is also utilized for the following portions. This method can be implemented in the computer program by two ways: (1)Integrating the three portions at the same time for every interval of time, printing every PRDEL units of time, and continuing to do this until ENDS is reached. This can be done by integrating first the last portion for one interval of time. The values recorded for the preceding portion at the previous interval of time are used as boundary conditions for the interval of time and the portion in question. After this calculation is over,

the same procedure can be applied for the next portion going backwards, and continuing to do this until the first portion is reached. In this way, the same storage can be used to record the values at x=XMAX(1) and XMAX(2) for every interval of time. Furthermore, once the values are printed, the concentration distribution at time= 0 are not used any more and that storage can be utilized for the next PRDEL units of time.

(2) Integrating the first portion between TINIT and ENDS and recording all the values, then integrating the next portion i all the way, using the values at X=XMAX(i-1) for every interval of integration time recorded previously, and continuing to do this until the last portion is reached. Then, at every PRDEL units of time the solution for that portion and the ones in the disk can be printed. The simulation stops when ENDS is reached for that last portion. It is evident that this method will require more storage capacity than the previous one.

An analysis made on both methods indicated less computational time for the latter way. Therefore, the second method was put in the system.

It should be pointed out that the position of any source can be located only at orthogonal collocation points. That position, given by KS and LS, must be supplied with point numbers. For example, if $N_z=7$ and

the location of a source is at one half of the z-diamension, LS should be 5 (the total number of collocation points is 9, including z=0 and z=1).

Continuity

In the present model, the concentration at the source, $C_0(i)$, is used to represent the source, and therefore is a required input information. Unfortunately, the emission rate is, in many cases, the only information that is available. One of the objectives of the present study is to obtain a general model that would also compute the concentration distribution when only the emission rate of the source is given.

There is a need then to develop a calculational procedure which would compute a concentration equivalent to the emission rate. The concept of continuity can be used for this purpose. Moreover, the flux at any x is a valuable piece of information that can be obtained from the results of an air pollution model. Therefore, a general continuity calculation will be obtained and then used for the particular problem of computing $C_0(i)$ from the emission rate Q(i).

The flux across any plane normal to the x axis can be expressed by the following equation:

$$\iint_{0-YMAX}^{H YMAX} C \cdot u \, dydz = Q_{x_{j}}$$
(5.1)

where C is the mean concentration at any point in the y-z plane, u its corresponding x-component of the mean wind velocity, and Q_{x_j} the mass rate at x=x_j.

If there was no mechanism of removal in a model involving one source, the pollutant would be not created or lost within the region of interest, and at steady state Q_{x_j} would be the same for any value of x, and equal to Q(1). For a multiple source case, Q_{x_j} would be equal to the sum of the constant emission rates for a particular source and its precedings. In such a case, the continuity condition should be included in the model, but since removal is taken into account in the present study, the values of Q_{x_j} will vary along the x direction.

Transforming the spatial coordinates into an interval between 0 and 1, equation (5.1) remains as follows:

$$\int_{0}^{1} \int_{0}^{1} C \cdot u (2 \cdot YMAX \cdot H) dy dz = Q_{x_{j}}$$
(5.2)

Using the quadrature weights, the double integral can be transformed into a double summation leading to the following collocation equation

$$N_{y}^{+2} N_{z}^{+2}$$

$$\sum_{k=1}^{y} \sum_{k=1}^{z} W^{(2)}(k) W^{(3)}(k) C_{jkl}^{u}(k) (2 \cdot YMAX \cdot H) = Q_{xj} (5.3)$$
for j=2, ..., N_y+2

But from the boundary conditions in the y direction,

$$C_{jll} = C_{jN_{y}+2l} = 0$$
 $j=2, ..., N_{x}+2$ (5.4)
 $l=1, ..., N_{z}+2$

Finally then, equation (5.3) remains as

$$Q_{x_{j}} = \sum_{k=2}^{N_{y}+1} \sum_{k=2}^{N_{z}+2} W^{(2)}(k) W^{(3)}(k) C_{jkl}^{u}(k) (2 \cdot YMAX \cdot H)$$
(5.5)
for j=2, ..., N_x+2

Equation (5.5) was included in the model (in subroutine OUTP) so that the mass flux can be known at any time and at any position in the x direction. In a case of no removal, these values, if they were equal as it will be shown in the next chapter, serve as a proof of the validity of the model and the numerical technique used.

In the case of using equation (5.5) to calculate the equivalent $C_0(i)$, Q_{xj} is equated to the constant rate of emission, Q(i), and all the concentrations but the one at the point source are made equal to zero. Equation (5.5) becomes then

$$Q(i) = W^{(2)}(KS(i))W^{(3)}(LS(i))C_{O}(i)u(LS(i))(2 \cdot YMAX \cdot H) (5.6)$$

.

for i=1, ..., NSRCS

Therefore, whenever Q(i) is the only information available, the equivalent concentration at the source can be calculated from the following equation:

$$C_{0}(i) = \frac{Q_{(i)}}{W^{(2)}(KS(i)) W^{(3)}(LS(i)) u(LS(i)) 2 \cdot YMAX \cdot H}$$
(5.7)

Chapter VI

PRESENTATION AND ANALYSIS OF RESULTS

The objective of the current study was to demonstrate the suitability of the present method to simulate diffusion in the atmosphere. To do this, experimental cases are simulated with the present model, and the results compared to the corresponding experimental data. Unfortunately, at the present moment there are almost no experimental data available in the literature. This is the reason why only the Project Prairie Grass diffusion data at O'Neill, Nebraska, were considered. Four cases from this Project were simulated and the results compared to the experimental data.

In addition, a parametric study was performed, and the sensitivity of the present model to variations in the atmospheric conditions analyzed. This was done by simulating several hypothetical cases.

Comparison to Experimental Values

The atmospheric diffusion data from Runs #20, #24, #45, and #54 of the Project Prairie Grass data [2, 3, 13] were used as a test of the present method. Project Prairie Grass was a field program designed to provide experimental data on the diffusion of a tracer gas in the atmosphere. The sulfur dioxide tracer gas was released for ten minutes over a flat prairie at O'Neill, Nebraska. The emission was done at about 50 centimeters above the ground and the gas was sampled along semicircular arcs from 50 meters to 800 meters from the source. Samplers were placed at 1.5m above the ground. In addition, concentration profiles along the vertical were measured from samplers located at nine levels on each of six towers positioned along the 100m arc.

In the experiments, the entire sampling network was put in operation just before the start of the gas release and the operation continued until the tracer was transported beyond the 800m arc. Although the actual measurements were of total exposure for each gas release, the investigators reported average concentrations for a ten minute sampling time. It was estimated that the concentration measurements were accurate to within about 10%.

Several meteorological measurements were taken during the tracer release. Among others, the mean wind velocity at two meters above ground. These values were used in the present model to simulate the mean wind velocity profile.

The source strengths Q, and the mean wind velocities at 2 meters u₁, for the four cases simulated in the present study were as follows:

Table 6.1 : Source Strengths and Mean Velocities (at 2m) for Experimental Cases Simulated

Run No.	Q(g/s)	u ₁ (m/s)		
20	101.2	9.38		
24	41.2	5.86		
45	100.8	6.02		
54	43.4	3.94		

In the present model, concentration profiles were simulated using the power law equation for the mean wind velocity as

$$\bar{u} = u_1(\frac{z}{z_1}) \cdot 14$$
 (6.1)

where u_1 corresponds to the value given in Table 6.1 for an elevation of $z_1 = 2m$, and z = elevation in meters.

It can be observed that the source emission rate is the only available information related to the source. Therefore, the equivalent concentration at the source, $C_0(1)$, was calculated using equation (5.7) with the source strength Q(1) given by Table 6.1. Since no data for the turbulent diffusivities were available, the four cases were simulated assuming α =1, i.e., the turbulent diffusivities in the y and z directions were the same; and a stability class 4, i.e., neutral stability. In addition, INVRS was made equal to zero, i.e., no inversion at the maximum elevation; and a firstorder reaction with k = 5.8x10⁻⁵ min⁻¹ was assumed for all cases.

In addition to the experimental values, the concentration profiles calculated by the present method were also compared to the ones (#45 and #54) obtained by the statistical method developed by Bullin [4].

Simulated vertical profiles at the centerline for Runs #20, #24, and #45, and at 20m from the centerline for Run #54 (no experimental data for the centerline were available) are compared with the limited experimental data available in Figures 6.1 through 6.4. There is good agreement between experimental and simulated vertical profiles except for Run #45, where the concentration values near ground level were much lower than the experimental ones. The reason for this discrepancy is a higher simulated velocity profile, u_s , as compared to the actual one u_a . This difference, especially near ground level, can be observed in Table 6.2.

labre	0.2	÷	Comparison	Detween	ACC		ina
			Theoretical	L Values	for	the	Mean
			Velocity in	n Run #4!	5		

DO

z (m)	u _a (m/s)	u _s (m/s)
0.25	3.78	4.50
0.5	4,60	4.96
1.0	5,31	5,46
2.0	6.02	6.02
4.0	6.65	6.63
8.0	7.35	7.31
16.0	7.88	8.05

Another reason for the discrepance in Run #45 can be due to a possible smaller simulated turbulent diffusivity in the z direction. This was checked by simulating the same problem, but with stability class 3, i.e., semi-unstable. The results are also plotted in Figure 6.3, and it can be observed that the concentration values increased and thus the difference between the experimental and theoretical values decreased.

and

Horizontal concentration profiles for the four cases at 1.5m above ground and at downwind distances of 200m and 400m are shown in Figures 6.5 through 6.8. In general, the agreement between experimental and simulated values is good.

The mass flux (g/s) across y-z planes at any x were also calculated using the results obtained for the concentration. These values are given in Figures 6.1 through (6.8) for all the four cases analyzed. The comparison of these values with the experimental data shows good agreement with an average difference of about 1.39%, and a maximum of 2.67%.











Figure 6.5: Comparison of Horizontal Concentration Profile.



y (m)

Figure 6.6: Comparison of Horizontal Concentration Profile.











Hypothetical Cases

Concentration profiles were simulated for all the hypothetical cases shown in Table 6.3. Vertical profiles at the centerline (y=0) and at several downwind locations for cases 4 through 6 are shown as a function of time in Figures 6.9 through 6.12. As it would be expected, there is a peak in the concentration at the elevation of the effective emission height. The value of this concentration peak increases as the atmospheric stability increases.

The concentration distribution for cases 1 through 7, at the steady state and for different x and y locations are shown in Tables 6.4 and 6.5. These correspond to an elevation of the effective emission height and the ground level, respectively.

Tables 6.6 and 6.7 show case 8, where a velocity in the y direction is included. This case is explained as a function of time so that the development of the plume can be clearly visualized.

This parametric study, i.e., cases 1 through 8, will be analyzed in more detail in the next section.

A multiple source case is shown in Table 6.9, and each source acting individually is shown in Table 6.8. The effect of their interaction can be obtained by comparing both Tables.

CASE	NO. OF SOURCES	SOURCE ⁽¹⁾ LOCATION (m)	SOURCE STRENGTH (kg/s)	STABILITY CLASS ⁽²⁾	ALPHA	UST (m/s)	RATE OF ⁽³⁾ REACTION (min ⁻¹)	АМ	P(%)
1	1	(0,0,150)	5	D	1	5	0	.25	0
2	1	(0,0,150)	.5	D	2	5	0	.15	0
3	1	(0,0,150)	5	D	2	5	0	.25	0
4	1	(0,0,150)	5	D	2	5	5.8x10 ⁻⁵	.25	0
5	1	(0,0,150)	5	В	2	5	5.8x10 ⁻⁵	.25	0
6	l	(0,0,150)	5	E	2	5	5.8x10 ⁻⁵	.25	0
7	l	(0,0,252.5)	5	D	2	5	5.8x10 ⁻⁵	.25	0
8	1	(0,0,150)	5	D	2	5	5.8x10 ⁻⁵	.25	5
9	l	(0,-284.1,	3	D	2	5	5.8x10 ⁻⁵	.25	0
		252.5)							
10	2	(0,0,150)	5	D	2	5	5.8x10 ⁻⁵	.25	0
		(10000,-284.1,	, 3	D	2	5	5.8x10 ⁻⁵	.25	0
		252.5)							

Table 6.3 : Hypothetical Cases Simulated

(1) The value in the z direction refers to the effective emission height.

(2) Using Pasquill-Gifford definition for stability categories.

(3) Rate of reaction for a first-order reaction.
It should be noted that the concentration distribution due to each source acting individually is at the steady state. This occurs when the time between the start of release and the initiation of the averaging time, Θ , is equal to 80 minutes and the end of the averaging time, Ω , is equal to 90 minutes. However, when both sources are put together, the steady state is reached after a longer time.

For all of the cases simulated without reaction, the differences between the mass flux at any x location, at steady state, and the constant source emission rate were not more than 4% of the emission rate. These results show the validity of the model in computing the concentration distribution, and also that the present method can handle continuity calculations.









Table 6.4Concentration Distribution for Hypothetical Cases1Through 7(Θ = 80 min, Ω = 90 min)

X DIRECTION (METERS)

0.0 337.7 1694.C 3806.9 6193.1 8306.0 9662.3 1000C.C

CONCENTRATION (MG/CU.M)

2 DIRECTION = 15C.C M

.

Y=-664.4 + Y=-519.1 + Y=-284.1 + Y= 0.0 + Y= 264.1 + Y= 519.1 + Y= 664.4 +	C.0 C.C O.0 35.45 C.C C.C C.0	C.O C.C 0.16 33.11 C.16 0.0 C.O	0.0 C.C 0.61 26.06 C.61 0.0 0.0	0.0 C.0 1.C0 19.48 1.C0 C.0 0.0	0.0 0.0 1.24 15.15 1.24 0.0 C.C	6.0 0.0 1.37 12.72 1.37 0.0 C.C	C.C 0.C 1.44 11.57 1.44 0.0 0.C	C.C 0.C 1.45 11.32 1.45 0.0 C.C	Case	#1
¥=-664.4 H ¥=-519.1 H ¥=-284.1 M ¥= 0.0 H ¥= 284.1 M ¥= 519.1 M ¥= 664.4 H	C.C 0.0 5.45 C.O 0.0 C.O	C.C 0.0 0.31 32.E8 0.31 C.C C.O	C.C 0.0 1.16 25.22 1.16 0.0 0.0	C.0 C.0 1.81 18.18 1.81 C.0 O.0	0.0 0.C 2.12 13.63 2.12 C.C 0.0	0.0 2.24 11.11 2.24 C.C 0.0	0.C 2.28 9.52 2.28 C.C 0.C	0.0 2.28 9.67 2.28 C.C 0.0	Case	#2
Y=-664.4 M Y=-519.1 M Y=-284.1 K Y= C.O M Y= 284.1 K Y= 519.1 M Y= 664.4 M	C.C O.U C.C 35.45 O.O C.O C.O	C.C 0.0 C.31 32.E9 0.31 C.0 C.C	C.C G.D 1.16 25.22 1.16 C.O C.C	C.O C.O 1.81 18.19 1.81 C.O C.O	0.0 C.C 2.12 13.65 2.12 0.0 0.0	0.0 C.C 2.25 11.14 2.25 C.O 0.0	0.0 C.C 2.29 9.96 2.25 C.C 0.C	C.C 2.3C 9.71 2.3C C.C C.C	Case	#3
$Y = -664.4 \mu$ $Y = -513.1 \mu$ $Y = -284.1 \mu$ $Y = 0.0 \mu$ $Y = 284.1 \mu$ $Y = 519.1 \mu$ $Y = 664.4 \mu$	C.O C.O 35.45 O.C C.C O.O	C.0 C.31 32.28 C.31 C.C 0.0	C.O 0.0 1.16 25.22 1.16 C.O 0.0	0.0 0.0 1.81 18.19 1.81 C.C 0.0	C.C 0.0 2.12 13.63 2.12 0.C C.O	C.C 0.0 2.24 11.12 2.24 0.0 0.0	C.C C.C 2.29 5.54 2.29 0.C 0.C	C.C C.C 2.29 9.65 2.25 0.0 C.C	Case	#4
$\begin{array}{rrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrr$	C.C G.C 35.45 C.C G.O C.O	C.C C.C 0.95 26.93 C.95 0.0 C.O	C.C 2.13 12.15 2.13 C.O O.C	0.0 C.0 2.11 6.44 2.11 C.0 0.0	0.10 0.2e 1.5ć 3.98 1.96 C.28 0.10	0.41 0.41 1.81 3.19 1.81 C.41 0.41	1.24 0.33 1.78 2.72 1.78 C.23 1.24	1.8C 0.27 1.7E 2.64 1.78 C.27 1.EC	Case	#5
Y=-664.4 W Y=-519.1 P Y=-2E4.1 M Y= 0.0 M Y= 284.1 M Y= 284.1 M Y= 519.1 M Y= 664.4 W	0.0 C.0 25.45 0.0 C.0 C.C	0.0 C.C C.13 34.38 C.13 C.0 C.C	0.0 0.58 3C.61 0.58 0.0 C.C	0.0 C.0 1.08 26.02 1.08 0.0 C.0	C.C C.C 1.47 22.13 1.47 0.C 0.0	C.C C.C 1.70 19.48 1.7C 0.0 0.0	0.C C.C 1.82 18.C6 1.£2 C.C 0.C	0.0 C.C 1.85 17.74 1.85 C.C 0.0	Case	#6
$Y_{2} - EE4.4 P$ $Y_{2} - 519.1 P$ $Y_{2} - 284.1 P$ $Y_{2} C.0 P$ $Y_{2} E4.1 P$ $Y_{2} E4.1 P$ $Y_{3} E4.1 P$ $Y_{4} E64.4 P$	C.C O.O C.C C.C O.O C.C	C.C C.C 0.01 1.C8 C.C1 0.0 C.C	0.0 C.C 0.18 3.91 C.18 C.0 C.0	0.0 C.0 C.57 5.70 C.57 C.0 C.0	0.0 0.0 6.95 6.14 0.95 C.C C.C	C.0 0.0 1.21 6.C0 1.21 0.0 C.C	C.C 0.C 1.23 5.El 1.23 0.C C.C	C.C 0.C 1.36 5.76 1.36 0.0 C.C	Case	#7
Z DIRE	CTICN =	252.5 ₽								
Y=-864.4 H Y=-519.1 H Y=-284.1 M Y= 0.0 H Y= 284.1 H Y= 519.1 H Y= 519.1 H	C.O C.C 32.J9 O.C C.C O.O	C.O C.29 30.35 C.29 C.C 0.0	0.0 0.0 1.11 24.CC 1.11 C.C C.O	C.0 0.0 1.77 17.79 1.77 C.0 C.0	C.C 0.0 2.11 13.56 2.11 C.C C.C	C.C 0.0 2.24 11.13 2.24 0.0 G.C	C.C 2.28 5.56 2.28 0.C 0.C	C.C 2.29 9.70 7.25 C.C C.C	Case	#7

Table 6.5 Concentration Distribution for Hypothetical Cases 1 Through 7 ($\theta = 80 \text{ min}, \Omega = 90 \text{ min}$)

* DIRECTION (HETERS)

C.O 337.7 1694.C 3806.9 6193.1 8306.0 9662.3 100CO.0

CONCENTRATION (MG/CL.*)

.

Z DIRECTION = 0.0 P

۰.

	0.0 C.C O.O C.C C.C C.C	C.O C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	C.0 0.0 C.09 1.59 0.09 C.0 Q.0	C.C Q.Q C.3e 4.C3 0.38 0.0 C.C	C.C 0.68 5.48 0.68 0.68 0.6 C.C	C.C O.C O.27 6.C8 G.£7 O.C C.C	C.C 0.91 6.2C C.51 0.0 C.C	Case	% 1
$Y = - \xi \xi 4.4 + Y$ Y = - 519.1 + Y Y = - 284.1 + Y Y = C.0 + Y Y = 284.1 + Y Y = 519.1 + Y $Y = \xi 64.4 + Y$	C.C O.O O.O C.O C.O	C.Q C.C C.C C.C C.C C.C C.O	0.0 c.c 0.0 c.c c.c 0.0	0.0 C.0 0.12 1.18 C.12 C.0 G.0	0.0 0.0 C.52 3.13 0.52 C.C C.C	0.0 0.0 C.52 4.25 0.92 0.0 0.0	C.C 0.0 1.16 4.70 1.16 0.C 0.C	C.C 0.0 1.22 4.75 1.22 0.0 C.C	Case	#2
Y=-664.4 M Y=-519.1 M Y=-284.1 M Y= C.C M Y= 284.1 M Y= 519.1 M Y= 664.4 M	C.0 C.0 C.0 C.0 C.C C.C C.C	C.C C.O C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.15 1.48 C.15 C.0 O.0	0.C C.C C.63 3.59 C.63 C.C O.O	0.0 G.C 1.CE 4.72 1.CB C.C 0.0	0.0 C.C 1.34 5.15 1.34 C.C C.C	0.C 0.0 1.4C 5.23 1.40 C.C C.C	Case	#3
	0.0 C.0 C.C 0.0 C.C C.0	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.0 0.15 1.48 0.15 0.0 C.0	C.C C.C J.58 C.63 O.O O.O	0.0 C.C 1.08 4.72 I.Ce 0.0 0.0	0.C C.C 1.34 5.14 1.34 0.C 0.0	0.0 C.C 1.4C 5.22 1.4C C.C 0.0	Case	#4
V = -664.4 M V = -519.1 M V = -284.1 M V = 0.0 M V = 284.1 M V = 519.1 M V = 519.1 M	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 C.0 C.0 C.0 C.0 C.C	0.0 0.27 1.10 0.27 C.C 0.0	C.0 C.0 1.18 3.03 1.18 C.0 C.0	C.C8 0.38 1.64 2.53 1.64 C.38 C.C8	C.C4 0.54 1.75 2.8C 1.75 0.54 C.C4	0.05 0.54 1.76 2.55 1.76 0.54 C.C5	C.CS 0.52 1.77 2.52 1.77 0.52 C.CS	Case	∜ 5
Y=-664.4 P Y=-519.1 P Y=-284.1 P Y= G.G P Y= 284.1 M Y= 284.1 M Y= 519.1 P Y= 664.4 P	C.C O.O C.C O.O C.C C.O	C.C 0.0 C.C 0.0 0.0 C.O	C.C 0.0 C.C 0.0 0.0 0.0	C.O C.O C.O C.O C.O C.O	0.0 0.0 0.02 C.35 0.02 C.C C.C	0.0 0.13 1.44 0.13 C.C 0.0	0.C 0.C 2.14 0.22 C.C 0.C	C.C C.Z 2.31 0.24 C.C 0.0	Case	#6
Y=-664.4 F Y=-519.1 F Y=-284.1 M Y= 0.0 F Y= 284.1 M Y= 519.1 F Y= 664.4 F	C.O C.C C.C O.O C.C Q.O	C.C C.C C.C C.C C.C C.C	0.0 0.0 C.C 0.0 0.0 C.C 0.0	C.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 C.02 C.C5 0.02 0.0 C.0	C.C 0.0 0.08 C.34 0.08 0.0 0.0	C.C C.C 0.14 C.56 C.14 O.C 0.C	C.C C.C C.62 C.16 C.62 C.16 C.C	Case	#7

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Table 6.6 Concentration Distribution for Hypothetical Case #8

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DIRECTION (METERS)

0.0 337.7 1694.0 3806.9 6193.1 8306.0 9662.3 1CCCC.C

CONCENTRATION (MG/CL.M)

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Y=-664.4 F Y=-519.1 F Y=-284.1 M Y= 0.0 F Y= 284.1 F Y= 519.1 M Y= 664.4 F	0.0 C.C 35.45 C.C C.C C.C	0.0 C.C 22.36 C.E8 C.C 0.0	0.0 C.C 11.81 1.93 C.C 0.0	C.O C.O 1.08 C.O C.O C.O	C.C C.O C.C C.C C.O C.O	C.C 0.0 0.0 C.C 0.0 0.0 0.0	C.C 0.C 0.C C.C C.C 0.C 0.C	C.C C.C C.C C.C C.C C.C C.C C.C	({	5 Э	8 8	0 10	min min
Y = -664.4 P Y = -519.1 M Y = -264.1 M Y = 0.0 P Y = 264.1 P Y = 519.1 M Y = 664.4 P	C.O C.C 35.45 C.C C.C C.C	0.0 0.0 C.C 33.13 1.53 C.C 0.C	0.0 0.0 C.C 19.83 4.78 C.C 0.0	0.0 0.0 5.07 2.84 C.0 C.0	C.C C.C C.C C.C C.C C.C	C.C C.U C.C Q.Q O.O C.C	C.C 0.C C.C C.C C.C C.C	C.C 0.0 C.C C.C 0.0 C.C	() {	3	11 11	5 15	min min
$Y = - \{ \{ \{ \{ \} \} \} \}$ $Y = - \{ \{ \{ \} \} \}$ $Y = - 284 \cdot 1 \} \}$ $Y = 0 0 \} \}$ $Y = 0 0 \} \}$ $Y = 2\{ \{ \{ \} \} \}$ $Y = 519 \cdot 1] M$ $Y = 664 \cdot 4 \}$	C.C 0.0 35.45 C.C 0.0 C.O	C.O C.C C.O 32.60 1.95 C.O C.C	0.0 C.C 0.0 23.57 7.64 0.0 0.0	0.0 C.0 9.19 7.40 C.0 C.0	0.0 C.0 1.23 1.72 C.12 C.C	0.0 0.0 0.0 0.0 0.0 C.0	0.C 0.C C.C 0.C 0.C C.C	C.C O.C C.C O.C O.C G.C	(5	=	10 20	min min
$Y = -664.4 \mu$ $Y = -519.1 \mu$ $Y = -284.1 \mu$ $Y = G.0 \mu$ $Y = 284.1 \mu$ $Y = 519.1 \mu$ $Y = 664.4 \mu$	C.C 0.0 C.0 35.45 0.0 0.0 C.C	C.C 0.0 32.77 2.33 C.C C.C	C.C 0.0 22.19 9.6C 0.0 0.0	C.O C.O 1C.86 14.26 C.O4 O.O	C.C C.C 0.0 3.87 9.88 3.29 0.0	0.0 0.0 0.79 3.30 2.05 0.21	0.C 0.C 0.16 0.82 C.59 0.14	0.C C.C C.C 0.08 0.45 C.33 0.C5	({ {	5	11 11	20 30	min min
$Y = - \epsilon \epsilon 4 . 4 F$ Y = - 519 . 1 F Y = - 284 . 1 F Y = C . 0 F Y = 284 . 1 F Y = 519 . 1 F Y = 664 . 4 F	C.0 C.C 0.0 35.45 C.C 0.0 C.0	C.O C.C 32.71 2.29 0.0 0.0	0.0 C.0 22.29 9.71 0.0 0.0	0.0 C.0 9.92 14.60 C.12 C.0	0.0 C.C 2.65 12.15 6.76 C.C	0.0 0.0 0.0 0.0 0.0 C.C C.C	C.C C.C C.C 0.0 C.C C.C	C.C C.C C.C C.C C.C C.C C.C	({ {	9		40 50	min min
Y=-664.4 H Y=-519.1 H Y=-284.1 M Y= C.0 F Y= 264.1 H Y= 519.1 H Y= 664.4 H	C.O C.C 0.0 35.45 ⁻ C.C C.C C.O	0.0 C.C 0.0 32.71 2.28 C.O C.O	0.0 C.0 0.0 22.26 9.73 C.0 C.0	0.0 C.0 9.92 14.57 0.10 C.0	0.0 0.0 2.64 12.21 6.79 C.C	0.0 0.0 C.0 C.C 0.0 C.C	C.C 0.C 0.C 0.C 0.C C.C	0.C 0.C 0.0 C.C 0.0 0.0 C.C	({ {	5	=	80 90	min min

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Table 6.7 Concentration Distribution for Hypothetical Case #8

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X DIRECTION (METERS)

C.O 237.7 1694.C 3806.9 6193.1 8306.0 4662.3 10000.0

CONCENTRATION (PG/CU-P)

2 CIRECTI	CN =	G.0 M										
North A N	0.0	0.0	0.0	0.0	C .C	0.0	0.0	0.0				
V==510.1 P	6.0	0.0	0.0	C.0	č.c	c.c	· c.c	¢.¢	~	_	^	
Nam 286.1 N	6.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	Θ	=	U	mın
V= 0.0 M	C . 0	Č.C	c.0	C.06	C.0	0.0	0.0	0.0	Ω	=	10	min
V= 284.1 M	0.0	0.0	0.0	0.01	c.c	C.C	c.c	c.c			T 0	1.(111
Y# 519.1 P	c .a	0.0	0.0	0.0	0.0	0.0	0.0	c.c				
Y= 664.4 4	c. 0	c.c	C.0	c.o	0.0	0.0	0.0	0.0				
•												
¥=-664.4 M	0.0	0.0	0.0	C.O	C.C	0.0	0.0	0.0				
¥=-519.1 M	C.0	0.0	0.0	C.O	c.c	c.c	0.C	c.c	~		-	
Y=-284.1 M	C.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	Θ	=	5	mın
Y= 0.0 H	C.0	0.0	0.0	0.27	0.0	0.0	0.0	0.0	0	_	15	min
¥= 284.1 M	c.o	0.0	0.0	C.11	c.c	c.c	C.C	C.C	44		1.0	111-1-11
Y= 519.1 M	c.c	C.0	0.0	0.0	0.0	0.0	C.C	0.0				
¥= 664.4 M	C.0	c.c	C.C	C.O	C. 0	0.0	0.0	0.0				
¥=-664.4 ¥	C.0	0.0	0.0	C.0	c.c	c.c	c.c	c.c				•
¥==519.1 ¥	C.C	6.6	0.0	0.0	0.0	0.0	C.C	C.C				
Y=-284.1 M	C.C	c.c	C.C	C.0	0.0	0.0	0.0	0.0	Θ	=	10	min
Y= 0.0 P	0.0	0.0	c.o	C.63	C.13	C.C	C•C	c.c	õ		20	
Y= 284.1 M	0.0	C.C	0.0	0.44	0.16	0.0	c.c	C.C	22	=	20	mın
¥= 519.1 M	C.C	c.c	c.0	C.0	0.0	0.0	0.0	0.0				
Y= 664.4 M	0.0	0.0	0.0	C.0	C.C	C.0	0.0	c.c				
No. 141 1 1		c o		• •	• •							
1=-CC4.4 P	<u> </u>	6.0	0.0	0.0	0.0	0.0		0.0	_			
V284 1 W	0.0	0.0	0.0	6.0	C .C	C.C	C.C	6.6	- (-)	=	20	min
N= C.0 V	0.0	0.0	0.0	0.96	0.84	0.14	C.C	0.0	0		20	min
V# 284.1 M	0.0	6.64	Č .Č	1.21	1.84	0.47	0.0	0.0	36	-	50	штп
V# 519.1 M	0.0	0.0	0.0	0.06	C.49	0.26	0.01	0.0				
N# 664.6 W	0.0	C.0	0.0	6.0	6.6	0.01	0.01	0.01				
•												
¥=-664.4 P	c.o	0.0	0.0	C.O	c.c	C.C	c.c	c.c				
Na-519.1 M	C.C	C.O	0.0	0.0	0.0	0.0	C.C	c.c	6		10	min
Y=-284.1 M	c.c	c.c	c.o	C.0	0.0	0.0	0.0	0.0	0		40	111711
Y= 0.0 P	0.0	0.0	0.0	0.80	C.74	C.C	C •C	C.C	Ω	=	50	min
¥= 284.1 M	C.C	C.C	0.0	1.17	3.26	0.0	c.c	C.C				
V* 519.1 M	C.C	C.C	C.0	C.02	1.77	0.0	0.0	. 0.0		•		
Y= 664.4 P	0.0	0.0	0.0	C.0	C.C	0.0	0.0	C.C				
¥=-664.4 M	0.0	0.0	c.0	c.o	c.c	C.C	C.C	C.C				
¥=-519.1 H	0.0	C.C	C.0	0.0	0.0	0.0	C.C	0.0	\circ		00	
Y=-284.1 M	C.C	c.c	C.C	C.O	c.o	0.0	0.0	0.0	Θ	=	βÛ	min
¥= 0.0 P	0.0	0.0	0.0	0.20	C.73	0.0	0.0	c.c	Ω	=	90	min
¥= 284.1 M	C.O	c.a	0.0	1.18	3.26	C.O	0.0	c.c				
¥# 519.1 M	c. c	c.c	c.c	C.03	1.76	C.O	0.0	0.0				
Y= 664.4 "	0.0	0.0	0.0	0.0	C.O	0.0	0.0	0.0				

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1	.1	0

Table 6.8 Concentration Distribution

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B DIRECTICK (PETERS) 8.0 337.7 1694.6 3006.9 6143.1 83C4.C 5662.3 1CCCC.C 0.0 270.1 1355.2 3C45.5 4554.5 4644.8 7725.9 80C0.C

				CCACENT 8844444	R#TICN (P	G/CU.#1		•				ECNCEN 44444	TRATICHI	\$G7CL.\$1		
2 DIRECT	1CK +	553.0	•	•												
Y=-644.4 P Y=-519.1 P Y=-264.1 P Y= C.0 P Y= 284.1 H Y= 519.1 H Y= 519.1 H	C.C C.C C.C C.C C.C C.C	C.C 0.0 C.C C.C 0.0 C.C	e.e c.e c.e c.c c.c c.c	0.0 0.0 C.0 C.0 C.0	6.0 6.0 0.0 6.0 6.0	6.0 0.0 0.0 0.0 6.0 6.0	0.0 C.C G.C O.D C.C	C.C C.C C.C 0.0 C.C	C.0 C.0 C.0 C.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C G.C C.C C.C G.C G.C	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	6.C 0.C C.C 0.C 6.C	C.C C.C C.C C.C C.C C.C
2 EINECT	tan +	497.1	•													
Y=={{4,4 x Y==514,1 x Y==24,1 x Y==0.0 x Y= 204,1 x Y= 204,1 x Y= 519,1 x Y= 44,4 x	C.C C.O C.O C.O C.O	C.C C.C C.C C.C C.C C.C	C.C 0.0 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C Q.Q C.C C.C Q.Q C.C	0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	C.0 0.0 C.C C.0 C.0 G.0	C.C 0.0 C.0 C.0 C.0 C.0	C.C C.O C.O C.O C.O	C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C	0.0 C.C C.C C.C C.Q	0.C 0.C 0.C C.C C.C	c.c c.c c.c c.c c.c
E CIRECT	1CN +	439.7	•													
Y=-664.6 P Y=-514.1 P Y=-264.1 M Y=-0.0 P Y=-284.3 P Y=-519.1 P Y=-664.4 P	0.0 C.C C.C 0.0 C.C C.C	9.0 2.0 2.0 2.0 2.0 2.0 2.0	6.0 5.0 6.0 6.0 6.0 6.0	0.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0	0.C C.C 0.C C.C 0.C	0.0 5.0 6.0 6.0 6.0 6.0	0.0 2.0 2.0 2.0 2.0 2.0 2.0	0.0 0.0 c.0 c.0 c.0 c.0 c.0	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 C.C 0.0 C.C 0.0	0.0 c.c 0.0 c.c c.c c.0	0.C 0.C 0.C 0.C C.C 0.C	0.0 c.c c.c c.c c.c
2 DIRECT	1CN ÷	355.0 (,													
Y=-664.4 P Y=-619.1 P Y=-644.1 P Y= 0.0 P Y= 284.1 P Y= 519.1 P Y= 666.4 P	c.c c.c c.c c.c c.c	0.0 C.0 C.C C.C C.C 0.0	0.0 C.0 G.0 G.0 C.C C.C	C.C C.C C.C C.C C.C C.C	C.C 0.0 C.C C.C C.0	C.C 0.0 C.C 0.0 0.0 0.0	C.C 0.0 0.C 0.C 0.C	c.c c.c c.c c.c c.c c.c	8.0 C.C C.C 0.0 C.C C.C C.C	C.Cl C.43 G.GO C.C C.C O.4	0.0 0.09 1.20 0.04 0.0 0.0	C.0 C.06 C.51 C.03 O.0 C.0 C.0	C.C C.C C.C C.C C.C C.C	C.C G.G C.C C.C G.O O.O	C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C
L DIRECT	ICN =	252.5 1	,					;								
Y=={{4,4 p Y==513.1 p Y==284.1 p Y= C.C p Y= 284.1 p Y= 314.1 x Y= 514.1 x	C.C C.J C.J C.J C.J C.J C.J C C.J C C.J C C C C	C.0 C.C C.83 C.C1 0.0 C.0	0.0 C.C 1.74 C.CE 0.0 0.0	0.0 C.03 C.03 C.03 C.0 C.0	0.0 C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.C 0.C 0.C 0.C 0.C 0.C	C.C C.C C.C C.C C.C C.C	C.0 C.C 21.27 C.0 G.C G.C G.O	C.C C.19 13.58 0.16 C.C 8.0 0.0	0.0 C.54 9.36 C.31 C.C 0.0 0.0	C.0 C.23 2.13 C.14 C.0 C.0 C.0 C.0	0.0 C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C
2 CIRECT	1Ch -	110.0 /	r -													
Y=-264.4 F Y=-314.1 * Y=-284.1 P Y= 6.6 F Y= 284.1 F Y= 319.1 F Y= 664.4 F	C.C 0.0 5.45 C.O C.O C.O	C.C 0.0 C.24 22.17 0.24 C.C G.D	C.O C.O G.S2 I2.04 C.52 C.O O.O	C.0 C.1C 1.15 C.10 C.0 Q.0	0.0 C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.C 0.C 0.C 0.C 0.C 0.C	0.0 0.0 0.0 0.0 0.0 0.0	C.C G.O C.O C.O C.O C.O	C.C G.G1 C.43 C.C0 O.Q C.C C.C	C.C 0.02 1.19 0.04 0.0 0.0 0.0	C.0 C.G& C.51 G.C3 C.C C.O O.O	C.C C.C C.C C.C C.C C.C C.C	0.0 C.C C.C 0.0 C.C C.C 0.0	0.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0
E CIRCCTI	ICN +	45.3 P	I													
9664.4 » 9519.1 » 9284.1 » 9- 284.1 » 9- 519.1 » 9- 519.1 » 9- 664.4 M	0.0 C.C C.C 0.0 C.C C.D	C.C C.C C.C C.C C.C C.C C.C	0.0 0.12 2.50 0.12 0.12 0.12 0.0	8.6 . C.0 0.04 C.41 C.04 0.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	8.C C.C G.G C.C G.C G.D	0.0 C.C C.C C.C C.C 0.C	0.0 C.0 C.0 0.0 C.0	0.0 0.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 C.C C.C G.O	0.0 0.0 0.0 0.0 0.0 0.0	0.C C.C 0.C C.C C.C	0.0 C.C C.C C.C C.C C.C
E DIRECTI	ICK 4	12.9 P	,													
Y=-684.4 P Y=-513.1 P Y=-64.1 P Y= 0.0 P Y= 284.3 P Y= 519.1 P Y= 644.4 P	C.C C.C O.O C.C C.C O.O	0.0 C.C 0.0 C.C C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.CO C.CO C.O C.O C.O	C.C C.C C.C C.C C.C C.C	C.C G.O C.C O.O O.O C.S	C.C G.C G.C G.C G.C G.C	C.C 0.0 C.C C.C 0.0 C.C	C.0 C.0 0.0 0.0 C.0 0.0	0.0 C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	C.O C.O C.O C.O C.O C.O	C.C 0.0 C.C 0.0 C.C C.C	C.C G.G C.C C.O C.O C.O	C.C G.C C.C C.C C.C D.C	C.C 0.0 C.C 0.C C.C
& DIRCCT	10h - '	c.o #	I													
V264.6 P V519.1 P V28.1 P V- C.0 P V- 76.1 P V- 319.1 V V- 519.1 V	6.0 0.0 6.0 6.0 0.0 0.0	C.C G.C C.C C.C G.G G.G	C.C 0.Q 0.Q C.C 0.Q C.C 0.D	0.0 C.C C.C C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0	C.C O.C O.C O.C C.C C.C	6.C 0.0 C.C 0.0 0.0 C.C	C.C C.C C.C C.C C.C C.C C.C C.C C.C	0.0 (.) 0.0 (.) 0.0 0.0 0.0	0.0 C.0 G.0 C.C C.C Q.0	0.0 C.0 C.0 C.0 C.0 C.0	0.0 c.c c.c c.c c.c c.c c.c	0.0 C.C 0.0 0.0 0.0 C.C	C.C C.C C.C C.C C.C C.C	C.C C.C D.C D.C C.C
CRIGA/SECI-	.119.76	3491.84 2	400.36	295.58	0.0	6.0		e.c	2117.46	2044.21 1	757.14	498.90	9.0	a.o	e.c	c. c

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Table 6.8 Continuation $(\Theta = 10 \text{ min})$ $(\Omega = 20 \text{ min})$

111 # GIRECTICK (PETERS)

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DERECTICA IPETERSI

	e.c	337.7	1694.0	3804.9	4193.1	4364.6	\$462.3	10000.0	0.0	270.1	1355.2	3045.5	4594.9	£644.8	1125.5	*****
				CONCENTA 44444444	AT 204 6PG	/21.21						CCACENT 4++++++	RATICN (>	G/CL.#1		
2 CIRECTI	Ch •	505.0 M					•									
7=-664.4 # 7=-514.1 # 7=-286.1 # 7= C.C # 7= C.C # 7= S16.1 # 7= 664.4 #	C.C 0.0 C.O C.O C.O C.O C.O	° 0.0 0.0 0.0 0.0 0.0	C.C C.C 0.C 0.0 0.0 0.0 0.0 0.0	0.0 c.0 c.0 c.0 c.0 c.0	6.0 C.C C.C C.C C.C C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0	6.6 6.0 0.0 0.0 6.0 6.0	0.0 0.0 0.0 0.0 0.0 C.C 0.0	6.0 6.0 6.0 6.0 6.0 6.0	C.C C.C C.C C.C C.C C.C 0.0	Q.0 C.C C.0 C.0 C.0 C.0 C.0	0.0 C.0 C.0 C.0 C.0 C.0	0.0 0.C0 0.C2 0.C 0.0 C.C 0.0	0.0 0.00 0.00 0.0 0.0 0.0	C.C 0.C 0.C 0.C 0.C C.C	C.C C.C C.C C.C C.C
2 CIRECTI	Ch +	492.1 P														
Y==666.4 4 Y==519.1 P Y==366.1 P Y==0.0 P Y==264.1 P Y==264.1 P Y==264.4 P	0.0 C.0 C.C C.C C.C C.C	8.0 C.C C.C C.C C.C C.C	0.0 0.0 c.c 0.0 0.0	6.0 6.0 6.0 6.0 6.0 6.0 6.0	6.0 6.0 6.0 6.0 6.0 6.0	0.0 C.C 0.0 C.C C.C 0.0	0.0 c.c c.c c.c c.c	0.0 0.0 0.0 C.C C.C 0.0	C.C 0.0 0.0 C.C 0.0 0.0 C.0	C.C 0.0 C.C 0.0 C.C C.C	C.C C.O C.C Q.O Q.O	C.0 G.0 C.0 C.0 C.0 C.0	C.0 C.03 C.C0 C.C C.C C.C	0.0 0.00 0.00 0.0 0.0 0.0	0.c c.c c.c c.c c.c c.c	
2 DIRECTI	CN =""	439.7 P														
Y=-664.4 # Y=-519.1 # Y=-284.1 M Y=-0.0 # Y=284.1 # Y=519.1 # Y=664.4 #	C.C C.C 0.0 0.C C.C 0.0	C.0 C.C C.C C.C C.C C.C	0.0 0.0 C.C C.0 C.0 C.0	C.C 0.0 0.0 0.0 C.C C.C	C.C 0.0 0.00 C.C2 0.00 C.C	C.C 0.0 0.0 C.C 0.0 0.0	C.C G.C G.C G.C G.C G.C	C.C 0.0 C.C C.C C.C C.C	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 C.0 0.0 C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.0 0.0 C.0 C.0 C.0	C.C1 0.G3 C.C0 C.C C.C C.C	C.Cl 0.C3 0.C0 C.C G.O Q.O	0.C C.C C.C C.C C.C C.C	n.0 C.C C.C C.C C.C O.0
2 DIRECTI	CN -	3:5.C P														
Y=-((4.4 # Y=-517.1 # Y=-284.5 # Y= C.C # Y= 244.1 # Y= 519.1 # Y= 664.4 #	C.C C.C C.C C.C C.C C.C	C.C G.C C.C C.C G.Q C.C	0.0 C.C C.C C.C C.O C.O	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.02 0.02 0.00 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	C.C G.C G.C G.C G.C C.C	6.C 0.0 C.C 0.0 0.0 0. C	0.0 C.0 0.0 C.0 C.C 0.0	0.0 C.21 C.25 C.01 C.C C.C 0.0	0.0 0.14 2.26 0.08 0.0 C.C 0.0	C.0 0.36 2.87 C.22 0.0 C.0 C.0	C.C 0.28 1.55 C.18 G.C G.C C.C	C.C 0.07 0.30 C.C4 6.0 0.0 C.C	C.C 0.C C.C C.C C.C C.C	C.C G.G C.C C.C G.G C.C
2 CIRECTI	GN -	252.5 ×	l													
Y664.4 F Y519.1 # Y284.1 F Y- 284.1 F Y- 284.1 F Y- 284.1 F Y- 519.1 F Y- 664.4 F	C.C G.O C.O C.O C.O	C.C 0.0 C.CL 1.03 0.01 C.C C.C	C.0 G.0 G.19 4.00 G.19 C.0 C.0	C.0 C.37 3.80 0.37 C.0 G.0	C.C C.13 C.91 0.13 C.C Q.Q	0.0 6.0 6.0 6.0 6.0 6.0 6.0	0.C c.c c.c c.c c.c c.c c.c	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.C 21.27 C.C C.C 0.0 0.0	C.C C.25 20.C1 C.13 C.C 0.C	0.0 1.C1 16.73 0.50 C.C C.0 0.0	0.0 1.32 10.75 0.80 C.0 C.C	0.0 C.75 4.CE 0.47 C.0 C.C 0.0	0.0 0.15 C.65 C.10 0.0 C.C 0.0	C.C C.C C.C G.C C.C C.C	C.C 0.0 C.C C.C C.C C.C
2 CIRECTI	CN =	150.0 #	I.													
Y==444.4 m Y==519.1 p Y==224.1 p Y== G.G p Y== G.G p Y== G.G f Y== 514.1 p Y== 514.4 p Y== 624.4 p	0.0 C.C 25.45 0.0 C.C C.C	0.0 C.G C.30 32.72 0.30 C.0 C.C	C.0 C.0 1.19 25.44 1.19 G.0 C.C	8.0 C.C 1.20 12.27 1.20 0.0 C.D	C.C 0.31 2.14 C.21 0.0 0.0	0.0 C.C 0.0 C.C 0.0 0.0	0.C C.C 0.C C.C C.C 0.C	0.0 C.C C.G C.C C.C C.C G.O	C.C G.D C.C C.C C.C C.C	C.C 0.01 C.55 C.C1 0.0 0.0 C.C	C.C 0.13 2.25 C.C2 C.O 0.0 0.0	C.C C.Je 2.87 C.22 C.O C.O G.O	C.0 C.2 1.55 C.18 C.C C.C C.C	C.0 C.07 C.3C 0.04 C.0 C.C 0.0	0.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C C.C
2 DIRECTIO	CA •	e5.3 P							1							
Y==664.4 P Y==215.1 P Y==204.1 P Y== 0.0 P Y== 264.1 P Y== 519.1 R Y== 664.4 R	C.O C.O C.C C.C C.C D.O	0.0 C.C1 1.73 C.G1 C.C 0.0	0.0 0.2 5.99 0.29 0.0	6.0 0.0 C.49 4.96 0.49 0.0 C.0	C.C G.O G.15 1.C4 0.15 0.0 G.O	C.C 0.0 C.C 0.0 0.0 0.0	C.C 0.0 C.C 0.C 0.C 0.C	C.C 0.0 C.C C.C 0.0 C.C	6.0 C.0 C.0 C.0 C.0 C.0	0.0 C.C 0.0 C.C C.C	C.0 C.0 0.0 0.0 0.0 C.0	C.0 G.0 C.0 C.0 C.0 C.0	C.C C.C1 C.C4 C.C0 C.C O.O C.C	C.0 C.C1 C.03 G.CC C.C G.0 G.0	0.C C.C C.C C.C C.C C.C Q.C	0.0 C.C C.C C.C C.C C.C C.C
E DIRECTI	Ch -	12.9 P							•							
Y=-(24.4 P Y=-519.1 P Y=-26.1 P Y= 2.6 P Y= 324.1 P Y= 519.1 P Y= 444.4 P	C.C 0.0 C.C C.C C.C C.C C.C	C.C 0.0 C.C C.C 0.0 0.0	0.0 C.C C.O C.O C.O O.O	0.0 C.0 C.09 C.09 C.0 C.0	0.0 0.2 0.24 0.24 0.24 0.24 0.24 0.24 0.	C.C C.C C.C C.C	0.0 C.C C.C 0.C C.C C.C	C.C C.C C.C 0.0 C.C C.C	8.0 C.C 0.0 C.C C.C C.C	0.0 C.C C.C C.C C.C C.C	6.0 9.0 6.0 6.0 6.0	C.C 0.0 C.C 0.0 C.C 0.0 C.C	C.C G.C C.C G.C C.C C.C	C.C C.C C.C Q.C C.C C.C	C.C O.C C.C C.C C.C C.C	C.C C.C C.C C.C D.D C.C
S CIRECTI	CR -	0.0 M	i													
Y644.4 P Y514.1 P Y264.1 P Y- 0.0 P Y- 284.1 P Y- 284.1 P Y- 319.1 P Y- 644.4 M	C.C C.C C.C C.O C.O	C.C 0.0 0.0 C.O 0.0 C.O C.O	C.0 0.0 0.0 0.0 0.0 0.0	6.0 0.0 C.06 C.81 0.08 G.0 9.0	0.0 C.C3 C.C3 C.C3 C.C3 C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	8.0 0.0 0.C 0.0 0.0 C.C 8.0	E.0 C.C G.C C.O C.O C.O G.G	C.C C.C C.C C.C C.C C.C C.C	6.0 6.0 9.0 6.0 6.0 6.0 6.0	6.0 C.C G.C C.G C.G C.G	0.0 C.C 0.0 C.C C.C G.O	C.C C.C Q.C C.C C.C C.C	C.C C.C C.C C.C G.C C.C C.C	C.C C.C C.C C.C C.C C.C

8.6 2999.86 3014.46 3183.22 2645.00 1240.97 231.19 C.C c.c

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Table 6.8 Continuation ($\Theta = 20 \text{ min}$) B Direction (Returs) ($\Omega = 30 \text{ min}$) I DIRECTION (PETERS) . 6.8 \$37.7 1494.0 3804.9 4193.1 8304.0 9662.3 10000.0 8.0 \$70.1 1355.2 3045.5 4954.5 4644.8 7725.4 ECCC.C

				CONCENT	RATION ().	G/CL.#1						CONCEN 000000	TRATICN (#G/CL.#}		
8 CIRECT	ICh +	505.0	•			•										
V484.4 X V514.1 P V264.1 P V- 264.1 P V- 264.1 P V- 264.1 P V- 219.1 P V- 644.4 P	0.0 C.0 C.C 0.0 C.C C.C	0.0 5.0 5.0 5.0 6.0 6.0	C.C 0.0 C.C C.C C.C C.C	6.0 6.0 6.0 6.0 6.0 6.0	C.C C.C C.C C.C C.C C.C C.D	0.0 C.C 0.0 C.C 0.0 C.C	0.0 C.C 0.C0 0.C0 C.CC C.C 0.0	0.0 C.C 0.CC 0.00 0.CC C.C 0.G	8.8 6.0 6.0 6.0 6.0 6.0	0.0 0.0 c.c c.c c.c c.c	8.8 5.0 5.0 6.0 6.0 6.0	C.C 0.0 C.C C.0 C.0 C.0	C.C1 0.05 C.C1 C.C 0.0 0.0	C.0 C.01 C.0 C.0 C.0 C.0 C.0	0.C C.C1 0.C4 0.C1 C.C C.C	0.0 6.0 6.0 6.0 6.0
Z GIRECT	ICN -	492.1	,													
V264.4 P V214.1 P V244.1 P V244.1 P V244.1 P V244.1 P V244.1 P V464.4 P	C.Q C.C Q.U Q.C C.Q O.O	0.0 C.0 C.C C.C C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 C.C 0.0 C.C C.O	C.C 0.0 C.C 0.0 0.0 0.0	t.t c.c c.cc c.cc c.cc o.c	C.C G.C G.C C.CC G.C C.C	C.C C.C C.C C.C C.C C.C S.O	6.0 C.C C.C C.O C.C C.C	8.0 C.C 0.0 G.O C.O	C.C G.O C.O C.O C.O C.O	C.C 0.01 C.07 C.CJ 0.0 C.0 0.0	C.C 0.01 0.C6 C.C1 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C C.C C.C	C.C C.C1 C.C1 C.C C.C C.C
8 DIRECT	1CA -	429.7														
Y((4.4 P Y513.1 R Y284.1 P Y- C.0 P Y- 284.1 P Y- 519.1 P Y- 664.4 P	C.0 C.C C.C C.C C.C C.C	¢.0 c.c c.o c.c c.c c.c	0.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.C1 C.C1 C.C1 C.C1 C.C1 C.C	C.C C.C C.C C.C C.C	0.C 0.C 0.C1 0.C0 0.C C.C	C.CC C.CC C.C1 C.C1 C.C1 C.C1 C.C1 C.C1	6.8 6.0 9.0 0.0 6.0 6.0	Ċ.C C.C C.O C.C C.C C.O	0.0 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C C.C	0.0 C.C2 C.12 0.01 C.0 G.C C.C	0.0 0.0 0.32 0.0 0.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C	C.C C.C7 C.26 C.C C.C C.C
Z CIRECT	- A31	355.0 1	ł													
Y=-&&&.4 P Y=->19.1 P Y=->24.1 P Y= G.C H Y= JA4.1 P Y= 513.1 P Y= &&A.4 R	C.0 C.0 C.0 C.0 C.0 C.0	C.C G.G C.C G.G C.G C.G	C.0 0.0 0.0 C.0 C.0 0.0 0.0	C.C C.C C.C C.C C.C C.C C.C	0.0 C.C C.C4 0.25 C.C4 C.C 0.0	0.0 0.0 C.C5 0.25 0.05 C.C 8.0	0.0 0.C 0.C2 0.10 0.C2 C.C 0.0	0.0 0.0 0.01 0.01 0.01 0.0	6.0 0.0 0.0 C.0 C.0 C.0 C.0	C.C 0.01 C.58 C.C0 0.0 C.0 C.0	C.C 0.13 2.2C C.O C.O C.O C.O	6.0 6.42 3.42 6.0 6.0 6.0 6.0	6.0 C.72 J.75 C.46 C.C C.C C.C	0.0 0.66 2.77 0.53 0.5 0.6 0.6	0.C 0.47 1.75 0.31 0.C C.C C.C	C.C 0.41 1.1C C.27 0.0 C.C C.C
2 CIRLOT	164 -	252.5 /	,													
V-664.6 # V-519.1 # V-224.1 # V= 0.0 # V= 0.0 # V= 284.1 # V= 519.1 #	0.0 C.C C.C C.C C.C C.C	0.0 C.C 1.11 C.O2 C.C C.C	0.0 0.17 3.04 0.17 0.0 C.0	C.C C.C O.59 5.85 C.59 G.O C.C	C.C C.C Q.76 4.96 C.76 C.76 C.0 C.0 C.0	0-0 C.C C.39 C.39 C.39 C.0 O.0	0.C C.C 0.12 C.12 C.12 G.C 0.C	0.0 C.C 0.27 0.32 C.C7 C.C 0.0	0.0 C.0 21.27 C.C 0.0 C.0 C.0	8.0 C.26 2C.16 C.14 G.0 C.C C.C	G.0 C.98 16.45 C.5e C.0 C.0 C.C	C.0 1.57 12.67 0.95 C.C 0.0 C.C	C.C 1.84 9.74 1.17 C.C 0.6 C.Q	6.0 1.÷1 5.88 C.92 C.C 0.0 0.0	C.C C.SZ 3.44 O.CL C.C C.C O.C	0.8 2.5C 0.53 C.C C.C 0.0
2 DIRECT	ICN -	156.0 0	,													
9==2(4.4 # 4==514.1 # 7==284.1 # 9= 0.0 # 9= 224.1 # 9= 513.2 # 9= 664.4 #	C.0 C.C C.C 35.45 C.C C.C Q.O	0.0 0.0 0.0	0.0 0.0 1.14 25.0e 1.14 C.C C.G	C.0 0.0 1.86 18.50 1.86 C.0 C.0	C.C 0.0 1.71 11.C5 1.71 0.0 0.C	C.C 0.0 0.73 3.74 C.73 0.0 C.C	C.C C.C 0.21 C.S4 0.23 0.C C.C	C.C 0.12 C.52 C.12 0.C C.C	6.0 6.0 6.0 6.0 6.0 6.0	C.C C.C1 C.:P O.GO C.C C.C O.Q	C.0 0.13 2.19 C.08 0.0 C.C C.C	C.0 0.42 3.41 C.26 0.0 C.0 C.0	C.C 0.72 3.79 C.46 0.0 C.C C.C	C.C Q.67 2.77 C.43 C.C C.C C.C	C.C C7 L.76 C.21 C.C C.C	C.C 0.41 1.51 C.26 C.C C.C C.C
Z DIACCTI	ica =	45.3 P	I													
Y==264.4 # Y==313.1 # Y==264.1 # Y== 264.1 # Y== 264.1 # Y== 519.1 # Y== 464.4 #	C.C 0.0 C.C C.C C.C C.O	C.Q C.C Q.Q3 1.86 C.C3 0.Q 0.Q	0.0 C.C 5.75 C.25 0.0 8.0	0.0 C.0 0.80 7.81 C.80 C.0 C.0	0.0 0.0 5.96 0.43 0.4 0.6	0.0 0.0 C.41 2.00 0.41 0.0 0.0	C.C 0.0 C.IC C.4 0.10 0.0 0.0	C.C 0.0 0.22 0.C! 0.C! 0.C	C.0 C.0 0.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C O.C C.C C.C C.C O.D O.D	C.C C.C C.C C.C C.C C.C	G.C C.C C.C C.C C.C	0.0 c.c3 c.16 0.02 0.0 c.c c.c	0.09 0.09 0.04 0.06 0.0 0.0 0.0	C.C 0.CB C.C 0.C 0.C 0.C 0.C C.C	C.C 0.0# C.27 C.C 0.0 C.C
8 EIRECT	[CN =	12.5	i													
Y==664.4 H Y==519.1 P Y==264.1 P Y= 264.1 P Y= 264.1 P Y= 264.1 P Y= 519.1 P Y= 644.4 P	C.C C.C C.C C.C C.C C.C	C.C 0.0 C.C 0.00 C.C C.0 C.0	C.C C.C C.C C.C C.C C.C C.C	C.0 C.20 1.75 G.20 C.0 C.0	C.O C.O 0.37 7.23 C.37 C.C 0.0	0.0 0.13 0.41 0.41 0.41 0.41 0.41 0.41 0.41	0.C 0.C 0.C1 0.C1 0.CJ C.C 0.O	0.0 C.C C.C 0.0 C.C C.C 0.0	C.0 C.0 C.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 0.C C.C C.O C.Q	C.C 0.0 C.C C.C C.O C.O	C.0 C.0 C.0 C.C C.C Q.D	C.C1 0.02 0.02 C.C C.C C.C C.C	0.0 C.C1 C.C3 0.01 0.0 C.C 0.0	0.C 0.Tl C.C4 0.Cl C.C C.C 0.C	C.C C.C1 C.C4 C.C1 D.D C.C G.C
e émecri	ICN +	6.0 /	,			•		•	:							
Y=-664.4 P Y=-519.1 P Y=-284.1 P Y	0.0 C.C C.C C.Q C.Q	6.0 F.0 C.Cl G.C C.C C.C	6.0 8.0 0.0 0.0 0.0 6.0 6.0	6.0 C.C 0.10 1.41 0.18 6.0 6.0	C.C 0.35 2.C5 C.35 0.0 Q.C	0.0 C.C O.11 0.32 C.11 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	8.0 6.0 6.0 6.0 6.0 6.0	6.0 6.0 6.0 6.0 6.0 6.0 6.0	C.Q Q.Q C.Q C.Q C.Q C.Q C.Q C.Q	C.0 C.0 C.C C.C C.C C.C C.O	C.C G.CC G.CC C.C G.G C.C	C.CC 0.01 C.CC C.C C.C 0.0 0.0	0.C C.Cl C.CJ C.CO C.C C.C 0.C	0.0 C.Cl G.Cl G.Cl C.C G.C 0.0

CESCENSECI- 4999.76 5100.34 5087.10 5149.54 3894.23 1490.48 390.11 216.25 2999.86 3043.10 3119.47 3129.05 3022.61 2169.12 1382.65 1189.79

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$(\Theta = 40 \text{ min})$ $(\Omega = 50 \text{ min})$

. B DIRECTION (FETERS) 6.6 337.7 1694.0 3806.9 6193.1 8306.0 9CEZ.3 1CCCC.C C.O 270.1 1355.2 3045.5 4954.5 6E44.8 7725.9 ECCC.C

				CONCENT	RAT 10H (M)	G/CL.#1							CCACEN 644444	TRATICN (P	G/CL.P3		
2 619601	ICN -	\$05.0	•			•											
Y{{4.4 K Y519.1 P Y34.1 P Y- 6.6 M Y- 284.1 P Y- 519.1 P Y- 519.4 M	0.0 C.C C.C 0.0 C.C C.C	0.0 C.0 C.C C.C C.C	6.0 0.0 C.C 0.0 C.C C.C	6.0 0.0 0.0 0.0 0.0 6.0	C.C C.C C.C C.C C.C C.C C.C C.C	8.0 C.C C.C C.C C.C O.O	0.0 0.00 0.01 0.01 0.00 0.0	0.0 C.C O.CC O.O1 C.CC C.C G.O		0.0 C.0 C.C C.0 C.0 C.0	8.0 C.C C.C C.C C.C C.C	0.0 0.0 C.0 C.0 C.0 C.0	C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C1 0.05 C.C1 C.C 0.0 C.C	0-0 C.C1 C.O1 C.C 0.C 0.C	0.C9 0.C9 0.C2 C.C C.C C.C	0.0 C.C C.U C.C C.C C.C
2 DIRECT	tch +	492.2	,											•			
Y==664.4 P Y==513.1 P Y==284.1 P Y= 0.0 P Y= 284.1 P Y= 515.1 P Y= 664.4 P	C.0 C.0 C.C C.C C.C C.C	6.0 C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 C.0 C.0	C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C	C.C 0.0 C.CC 0.0 0.0 0.0	C.C G.C G.C C.C1 G.C G.C G.C	C.C 0.00 C.C2 0.CC 0.C C.C		C.C C.C C.C C.C C.C C.C C.C	C.C C.C G.O C.C B.O	0.0 0.0 0.0 0.0 0.0 0.0	C.O C.O C.O C.O C.O C.O	C.C 0.01 C.C7 C.C1 0.0 C.C C.C	C.C G.C7 C.C2 C.C1 G.O O.O C.C	C.C G.C3 G.12 C.C2 C.C G.C C.C	C.C C.C C.C C.C C.C C.C
& DIRECT	ICN =	425.7	,														
Y=-{{4,4 p Y=-519.1 p Y=-284.1 p Y= C.0 p Y= 2{{4.1 p Y= 519.1 p Y= 664.4 p	C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0	0.0 C.C C.0 C.0 C.0	0.0 0.0 0.08 0.01 C.C C.C	0.0 0.0 0.01 0.01 0.01 0.0	C.C 0.C C.C3 0.11 0.C3 0.C C.C	C.C 0.0 C.CJ C.1] 0.0] 0.0 C.C		C.0 C.C C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C 0.0	0.0 C.C G.O C.C C.C C.O	0.0 C.0 G.0 C.0 C.0	0.0 C.C2 C.13 C.01 C.0 C.C C.C	0.C 0.11 0.45 0.07 0.0 0.0 0.0	C.C 0.17 C.22 C.11 C.C C.C C.C	C.C C.12 C.C C.C C.C
2 DIRECT	ICK =	355.0 (•								•						
Y=-264.4 P Y=-510.1 P Y=-264.1 P Y= C.C P Y= 284.1 4 Y= 519.1 P Y= 864.4 P	C.C 0.0 C.C C.O C.C	C.C C.C C.C C.C C.C	C.C 0.0 0.0 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0	0.0 C.C 0.31 G.C5 C.C 0.0	0.0 6.0 6.17 0.85 0.17 6.C 0.0	0.0 0.2 1.16 0.27 C.C 0.C	0.0 0.25 1.21 0.29 C.C 0.0		C.C C.C C.C C.C C.C	C.C 0.01 C.S7 C.CC 0.0 0.0 C.C	C.C G.13 Z.2C C.C? G.O G.O	C.0 0.42 3.40 C.26 C.0 C.C 0.0	C.0 C.73 3.84 C.46 C.C C.C C.C	0.0 C.93 3.85 0.61 C.C C.C C.C	0.C 1.C7 1.77 0.28 0.C C.L 0.C	C.C 1.C5 1.75 C.7C 0.G C.C
2 CIRFCT	ICA = .	252.5	,						,								
Y=-664 P Y=-515.1 P Y=-224.1 P Y= 0.0 R Y= 234.1 P Y= 519.1 P Y= 664.6 P	6.0 C.C C.C C.C C.C	C.0 C.01 1.CP C.C1 C.C C.C	0.0 0.18 3.9C C.16 0.0 C.C	C.O C.O G.S7 5.72 C.S7 G.O C.O	C.C C.C 0.94 6.10 C.54 C.0 0.0	0.0 C.C 1.21 6.03 1.21 0.0 0.0	0.0 C.C 1.36 5.92 1.36 C.C 0.0	0.0 C.C 1.35 5.86 1.35 C.C 0.0		0.0 C.0 21.27 C.0 C.0 C.0 C.0	0.0 0.26 2C.14 0.14 0.0 C.C C.C	C.C 0.99 16.48 0.57 0.0 0.0 C.C	0.0 1.57 12.63 C.95 C.0 0.0 C.0	C.C 1.87 9.86 1.10 C.C C.C C.C	C.C 1.57 8.16 1.28 C.C C.C C.C Q.Q	0.C 1.59 7.12 1.12 C.C C.C C.C	0.0 1.55 7.14 1.17 C.C 0.C 0.C
2 DIFECT	ICH -	156.0 /	•														
Y==664.4 F Y==284.1 F Y==284.1 F Y==0.0 F Y== 284.1 F Y== 0.0 F Y== 313.1 F Y== 634.4 F	C.C C.C 35.45 C.C C.C 0.0	C.C C.31 32.89 C.31 C.C 0.0	C.C G.O 1.16 25.2C 1.16 C.C 0.0	C.C 0.0 1.22 18.21 1.52 C.0 C.0	C.C 0.0 2.10 13.17 2.10 6.0 C.C	C.C 0.0 2.26 11.18 2.26 0.0 C.C	C.C C.C 2.34 IC.12 2.34 0.C C.C	C.C 0.0 2.34 9.03 2.34 0.C C.C		C.O C.C O.O C.C C.O C.O	C.01 C.41 C.57 C.C0 C.C C.C	0.0 0.13 2.20 0.08 0.0 0.0 0.0	C.0 0.42 3.39 G.25 0.0 C.0 C.0	C.C C.7J J.24 C.46 O.C C.C	C.C 0.41 3.86 C.El C.C 0.0 C.C	C.C 1.C3 3.74 C.EE C.C Q.C C.C	C.C 1.C 3.77 C.7C C.C 0.0 C.C
& DIRECTI	ION -	e5.3 +	,														
Y== {{4.4 } Y== 519.1 } Y== 284.1 } Y= C.C = Y= 274.1 } Y= 519.1 } Y= 464.4 }	C.C D.C C.C C.C C.C C.C	C.C C.C 0.C2 1.21 C.C2 C.O C.O	0.0 C.C C.27 5.45 C.27 0.0 0.0	9.0 6.0 7.61 6.77 6.0 6.0	0.0 C.0 1.71 7.67 1.21 C.0 C.C	0.0 1.57 7.53 1.57 6.0 9.0	0.C 0.0 1.72 7.31 1.72 0.C 0.C	C.C 0.0 1.72 7.C3 1.72 0.0 C.C		C.C C.C C.C C.C G.C G.O G.O	C.C C.C C.C C.C C.C C.C C.C C.C	Q.0 C.C C.Q C.C C.C C.O Q.O	 c c c c c	8.8 C.C4 C.17 0.02 C.C C.C C.C	0.0 0.13 C.54 C.09 0.0 C.C C.C	C.72 C.72 C.73 C.C C.C C.C	C.C 7.22 C.77 C.14 C.C C.C C.C
8 CIPECTI	ICK =	12.9 #	I														
YEE4.4 R Y519.1 P Y284.1 P Y- 284.1 P Y- 284.1 P Y- 519.1 P Y- 664.4 M	C.Q 0.0 C.Q 0.0 0.0 C.Q	C.C 0.0 C.C 0.0 0.0 0.0 C.C	C.0 0.0 0.0 0.0 0.0 0.0	0.0 C.0 C.17 1.60 C.17 C.C 0.0	0.0 0.65 3.74 6.65 6.0	0.0 5.0 1.12 4.90 1.12 5.0 0.0	0.0 0.0 1.;2 4.78 1.22 C.C 0.0	0.C C.C 1.2C 4.15 1.2C C.C 0.D		C.C C.C C.C C.C C.C C.C C.C	C.C C.D C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C C.C	C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C C.C	0.0 0.11 0.02 0.C 0.C	0.C C.15 C.C4 0.C C.C C.C	C.C C.21 C.C4 U.O C.C G.C
E CINECTI	ICA +	G.O P	I														
Y664.4 = Y319.1 P Y264.1 P Y- 6.0 M Y- 204.1 P Y- 319.3 M Y- 664.6 P	8.0 C.0 C.0 C.0 C.0 C.0	0.0 C.0 0.0 0.0 C.0 C.0	6.0 6.0 6.0 6.0 6.0 6.0 6.0	G.0 0.15 1.46 G.15 0.0 G.0	0.0 C.C Q.63 J.59 C.C Q.C Q.C	0.0 C.C 1.10 4.70 I.1C 0.0 0.0	0.0 C.C 1.20 4.44 1.3C 0.0 0.0	0.0 0.C 1.17 4.44 1.17 0.0 0.0		0.0 C.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 9.0 9.0 5.0 6.0 6.0 6.6	0.0 0.0 0.0 0.0 0.0 0.0 0.0	C.0 0.0 C.0 C.0 C.0 C.0	C.C C.C2 C.C1 C.C 0.0 C.C	C.C C.C? C.C? C.C C.C Q.C O.C	0.C C.C 0.16 0.C3 C.C C.C 0.C	0.0 (.((() 0.04 (.(0.0 0.0

CRECR/SECJ- 4999.26 5074.32 5136.52 5039.74 4868.68 4858.25 4826.14 4773.88 2999.86 3039.09 3126.83 3116.64 3064.48 3024.42 2889.57 2994.58

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DIRECTION (HETERS)

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6.8 Continuation (Θ = 80 min) (Ω = 90 min) x cirectick ipeters) •... 337.7 1654.6 3006.8 6193.1 6306.0 4662.3 10000.0 c.0 270.1 1235.2 3055.5 4954.5 6544.8 7725.9 εςςςς.c

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				CCACEAT	R#T1CK1P	C/CU.F)							CONCENT #######	RATICS (P	6/CL.#} *******		
E DIRECT	10N +	565.0	•			•											
Y==684.4 P Y==517.1 P Y==264.1 P Y= 264.1 R Y= 514.1 R Y= 644.4 P	C.C 0.0 0.0 C.C 0.0 0.0	C.C 0.0 0.C 0.C 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	6.0 0.0 0.0 0.0 0.0 0.0 0.0	0.C 0.C 0.C1 0.C0 0.C 0.C	C.C 0.C C.C1 0.C 0.C 0.C		0.0 C.0 C.C C.0 C.0 C.0	0.0 C.0 C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 C.C	8.8 6.0 6.0 6.0 6.0 6.0	C.C C.CI C.CI C.C Q.Q C.Q	0.0 C.Cl C.Cl C.Cl C.C C.0 0.0	8.C 0.10 0.22 C.C 0.C	0.0 (.() (.) (.) (.) (.) (.)
2 LIRECT	104 -	492.1	R														
Y=-664.4 F Y=-519.1 4 Y=-284.1 F Y= 6.0 F Y= 62.1 F Y= 519.1 F Y= 664.4 F	C.C C.C C.C C.O C.O	C.C G.Q C.C C.D C.Q C.Q	C.C C.O C.O C.O C.O C.O C.O	C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 C.C 0.0	0.C 0.D 0.C1 0.C0 C.C C.C 0.Q	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0		C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C C.C	0.0 C.C C.O C.O C.O	C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.01 0.C7 C.C1 0.0 0.C C.C	C.C Q.CZ C.OF C.C1 Q.C D.O C.C	C.C G.C G.12 C.C C.C C.C G.C C.C	C.C G.C4 G.13 C.C3 C.C C.C C.C
2 CIRECT	ICN .	439.7	P										•				
Y==264.4 # Y==519.1 # Y==224.1 # Y==204.1 # Y==204.1 # Y==215.1 # Y==215.1 #	0.0 C.0 C.C G.C C.C C.C	8.8 C.C C.C C.C C.C C.C	0.0 C.0 C.0 C.0 C.0 C.0	0.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.01 C.C C.C 0.0 0.0 0.0	0.0 C.C 0.02 C.C2 0.0 0.0	6.0 C.C 0.C3 0.11 C.C3 C.C 0.0	0.0 C.C 0.12 0.12 0.C 0.0		C.C C.C C.C C.C C.C C.C C.C	6.0 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C C.C C.C C.C	0.0 C.0 C.0 C.0 C.0	0.0 C.CZ C.12 0.01 C.C C.C	0.C 0.11 0.45 0.07 0.0 C.0 0.0	C.C 0.17 C.63 C.11 0.C C.C	C.C 0.15 C.67 C.17 C.17 C.C B.C C.C
8 DIRECT	ICA -	355.0	•														
Y=-664.4 F Y=-519.1 F Y=-284.1 F Y= 0.0 F Y= 284.1 F Y= 519.1 F Y= 664.4 F	C.0 C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C C.C	0.0 C.C C.C C.C C.C C.C C.C	C.0 0.0 0.0 0.0 C.0 C.0	C.C 0.0 C.GS C.JZ 0.05 C.D C.O	C.C 0.0 0.17 C.26 0.17 0.0 0.0	C.C 0.26 1.12 0.ž6 0.0 0.C	C.C G.C Q.22 1.12 C.72 Q.Q C.C		C.C D.C C.C C.C G.C G.C	C.C 0.01 C.57 C.CC 0.0 C.C C.C	C.C 0.13 2.2C C.C C.C 0.D 0.0	C.0 C.42 J.41 C.26 C.0 C.0 C.0	0.C C.73 J.84 C.46 C.C C.C C.C	C.0 C.93 J.86 C.0 C.C C.C	0.C 1.(3 3.}9 0.68 C.C C.C C.C 0.C	C.C 1.C5 3.77 C.7C G.G C.C
2 GIRECTI	IC# •	252.5	,														
Y(24.4 # Y(19.1 # Y28.1 # Y- C.C # Y- 284.1 # Y- 284.1 # Y- 319.1 # Y- 444.4 #	C.C C.C C.C C.C C.C C.C	C.Q C.C 0.01 1.C8 C.C1 0.0 C.O	0.0 C.C 0.18 3.71 C.18 G.O 0.0	0.0 C.0 C.57 5.70 C.57 C.0 C.0	0.0 C.95 4.14 C.95 C.C C.C	0.0 0.0 1.21 6.00 1.21 0.0 C.C	C.C 0.C 1.23 5.61 1.23 0.C C.C	C.C 0.G 1.36 5.76 1.36 0.0 0.C		0.0 C.0 21.27 C.0 C.0 C.0 C.0	0.0 C.26 2C.14 0.14 C.0 C.0 C.C	C.0 C.99 16.48 0.57 0.0 0.0 C.C	C.0 1.57 12.63 C.95 C.0 0.0 C.0	C.C 1.fe 9.A5 1.18 C.C 0.0 C.C	0.0 1.57 4.18 1.29 C.C C.C C.C	0.C 2.CC 7.25 1.13 C.C C.C 0.C	0.0 2.00 7.17 1.34 0.0 0.0
2 CIRECTI	ICN =	150.0 #	•														
Y=-&{{} Y=-517.1 x Y=-317.1 x Y=-324.1 p Y= 324.1 v Y= 324.1 v Y= 519.1 p Y= \$44.4 p	C.C 0.0 C.0 35.45 C.0 C.0 C.0	C.C 0.0 C.31 32.28 0.31 C.O C.C	C.C C.O 1.16 25.22 1.16 C.O 0.0	C.0 C.0 1.81 10.18 1.81 C.0 0.0	C.0. C.C 2.12 13.63 2.12 C.C 0.0	0.0 2.24 11.12 2.24 C.C 0.0	0.C 2.29 9.94 2.29 C.C 0.Q	0.0 2.2 9.69 2.29 C.C 0.0		6.0 C.C C.C C.C C.C C.C C.C	C.C C.C1 C.57 C.CO C.C C.C D.Q	0.0 0.13 2.19 0.08 0.0 C.C 0.0	C.0 0.42 3.39 C.26 0.0 C.0 C.0 C.0	C.C 0.72 3.84 C.46 C.0 G.0 C.C	C.C 0.43 3.87 C.21 C.0 0.0 C.C	C.C 1.C4 7.81 C.E5 C.C 0.C C.C	C.C 1.Ce 3.70 C.71 C.C 0.0 C.C
E CIPECTI	ICA +	61.3 1	,														
Y== 664.4 Y Y== 519.1 P Y== 2t4.1 P Y== 0.C N Y== 0.C N Y== 519.1 P Y== 644.4 P	0.0 C.C C.C 0.0 C.C	0.0 C.C 1.2 C.C 2.0 C.C	0.0 0.27 5. <i>P</i> 6 C.27 0.0 C.C	C.C C.O 0.76 7.59 C.76 0.0 C.C	C.0 C.C 1.22 7.72 1.12 0.0 0.0	C.O C.C 1.55 7.44 1.55 0.0 0.0	0.C C.C 1.72 7.23 1.72 0.C 0.0	0.0 C.C 1.7e 7.17 1.7e 0.C 0.0		C.C 0.0 0.0 C.C 0.0 C.O	C.C G.C C.C C.C C.C G.O G.C	0.0 c.c c.a c.c c.c c.c c.c	0.0 C.C 0.0 C.C C.0 C.0	0.0 C.C3 C.17 0.02 C.0 C.C 0.0	0.C 0.13 C.54 0.09 0.0 C.C C.0	C.C C.20 C.73 C.13 C.C C.C C.C	C.C 0.22 C.77 C.15 C.C C.C C.C
	Ch +	12.9 9	,														
Y664.4 # Y519.1 # Y526.1 # Y- 0.0 # Y- 28.1 # Y- 319.1 # Y- 644.4 #	C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.0 C.17 1.42 0.17 C.0 G.0	C.C Q.63 3.74 Q.65 Q.65 Q.Q C.Q	C.C 0.0 1.10 4.24 1.10 0.0 Q.0	C.C 0.C 1.35 5.24 1.25 0.C 0.C	C.C G.C 1.41 5.31 1.41 0.0 0.0		4.0 6.0 6.2 6.2 6.2 6.0	C.C C.C C.C Q.Q C.C C.C C.C	C.C C.C C.C 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0 C.0 C.0	0.C C.C1 0.05 C.C1 C.C C.C 0.D	0.0 C.C3 C.13 C.C C.C C.C C.C	0.C C.19 0.C C.C C.C 0.C	C.C C.C C.C C.C C.C C.C C.C
2 DIRECTI	CN +	C.0 F	1						•								
9+-664.4 P 9519.1 P 9284.1 P 9- C.O P 9- 284.1 M 9- 519.1 M 9- 644.4 P	C.C D.O C.C C.C D.O C.C	C.C 0.0 C.C C.C 0.0 C.C	8.0 6.6 0.0 6.0 6.0	0.0 C.0 C.15 1.46 C.15 C.0 0.0	0.0 C.0 G.43 3.56 0.43 0.0 C.C	0.0 6.0 1.07 4.72 1.04 0.0 C.C	0.0 1.34 5.14 1.34 0.0 0.0	C.C 0.0 1.4C 5.j; 1.4C 0.0 0.C		0.0 C.D C.O C.O C.O C.O C.D	0.0 C.C C.O C.O C.O C.O	0.0 0.0 0.0 0.0 0.0 0.0	C.0 C.0 C.0 C.0 C.0 C.0	C.C1 C.C2 C.C1 C.C1 C.C C.C C.C	0.0 0.08 0.08 0.02 0.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C C.C C.C	0.0 C.C? C.IS 0.04 C.C C.C 0.C

Table 6.9 Concentration Distribution for Case #10 ($\Theta = 0 \text{ min}, \Omega = 10 \text{ min}$)

DIRECTICK (PETERS)

0.0 337.7 1694.6 3806.9 6193.1 #308.C 4662.3 100000 10270.1 11353.2 13045.5 14454.5 16444.8 17729.9 18000.d

CONCENTRATION (#G/CL.#)

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. Z GIRE	T10K +	505.0) M												
Y=-264.4 M Y=-519.1 M Y=-264.5 M Y= 0.0 M Y= 264.1 P Y= 515.1 M Y= 664.4 M	C.0 0.0 0.0 C.0 0.0 C.0	C.0 0.0 0.0 0.0 0.0 0.0	C.C 0.C 0.C 0.C 0.C 0.C	0.0 0.0 0.0 0.0 0.0 0.0	6.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0	6.C c.c c.s 0.0 c.c	C.C 0.0 C.C 0.0 C.C 0.0 C.C	C.C 0.C 0.C 0.C 0.C C.C	C.C a.o c.o c.c c.o c.o c.c	C.C 0.0 C.C 0.0 C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C 0.C
2 CIREC	TION +	492.1	•												
V684.4 P V519.1 P V284.1 P V- 0.0 P V- 284.1 P V- 519.1 H V- 664.4 P	0.0 0.0 c.0 c.0 c.0 c.0	0.0 0.0 0.0 0.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	6.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 5.0 0.0 5.0 6.0 0.0	0.0 C.C 0.0 C.O C.O	0.0 0.0 0.0 0.0 0.0 0.0	0.0 2.2 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C C.C C.0 C.0	0.0 C.C 0.0 0.C C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.C C.C 0.0 C.C C.C	C.C 0.0 C.C 0.0 C.C C.C	c.c 0.c c.c c.c c.c	C.C 0.C 0.C 0.C 0.C 0.C
2 DIREC	TION -	439.7	٠												
Y664.4 P Y519.1 R Y284.1 M Y- 0.0 P Y- 284.1 P Y- 519.1 M Y- 664.4 P	0.0 C.C C.C C.C C.C C.C	0.0 C.C C.C C.C C.C C.C	8.0 0.0 0.0 0.0 0.0 0.0 0.0	0-0 0-0 0-0 0-0 0-0 0-0 0-0 0-0	C.C 0.0 C.C 0.0 0.0 C.C	C.C G.O C.C G.O C.C C.O	C.C 0.0 C.C 0.C 0.C C.C	C.C 0.0 C.C C.C C.C C.C	C.C C.C C.C C.C Q.C	C.C 0.0 0.C C.C 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0	0.0 c.c c.c c.c c.c	0.0 C.0 0.0 C.C C.C C.C	0.0 0.0 0.c 0.c 0.c C.C
. Z DIRCC	110N -	355.0	F												
Y==264.4 K Y==519.1 H Y==284.1 K Y==284.1 K Y==284.1 K Y==519.1 K Y==664.4 K	C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C Q.C	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 6.0 6.0 0.0 6.0 6.0	0.0 0.0 C.C 0.0 C.C 0.0	0.0 0.0 C.C 0.0 C.C C.C	C.C 0.01 C.43 C.C 0.0 C.C C.C	C.C 0.02 0.2C C.C 0.0 0.0 C.C	C.C 0.06 0.51 C.C3 0.0 0.0 C.C	C.C C.C C.C C.C C.O C.O	C.C C.C C.C C.C C.C C.C	0.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0 0.0
S CIREC	TICN -	252.5	я				·								
Y=-664.4 M Y=-519.1 H Y=-284.1 P Y- 0.0 P Y- 284.1 M Y- 519.1 P Y- 664.4 M	C.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 0.01 C.E3 0.01 0.0 C.C	C.0 0.0 0.08 1.74 0.08 0.0 C.0	C.C C.C C.C; C.33 C.33 C.C3 C.C C.C	0.0 C.C C.C C.C C.C C.C C.C	0.0 C.C 0.0 0.0 C.C 0.0	0.C 0.C 0.C C.C 0.0	0.0 0.C 21-21 0.0 C.C C.C 0.0	0.0 C-19 13.50 C.10 C.0 C.C C.C	0.0 9.34 9.36 0.31 0.C C-C C.C	0.0 0.23 2.13 0.14 0.0 0.0 C.C	C.C 0.0 0.C C.C C.O C.C	C.C C.C C.C C.C G.C C.C	c.c c.c c.c c.c c.c c.c	0.C 0.C 0.C 0.C C.C 0.C
2 DIREC	TICN .	150.0	•												
V=-664.4 P V=-319.1 P V=-284.1 P V= 284.1 P V= 284.1 P V= 519.1 P V= 664.4 T	0.0 0.0 35.45 0.0 0.0	0.0 C.24 22.39 0.24 C.C 0.D	0.0 0-0 C.52 12.04 0.52 C.0 0.0	6.0 0.0 C.10 1.15 C.10 C.0 8.0	C.C 0.0 0.C 0.C 0.0 0.0	C.C 0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.C 0.0 0.C 0.C 0.C 0.C	0.0 0.0 0.0 0.0 0.0 0.0	C.C C.C3 C.43 C.C0 C.C C.0 G.D	0.0 C.CE 1.15 C.C C.C 0.0	0.0 0.51 0.03 C.0 C.C 0.0	0.0 5.0 6.0 6.0 6.0 6.0	C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C	C.C 0.0 0.C C.C C.C C.C
2 DIREC	110N -		۲												
Y==&&4.4 # Y==519.1 # Y==224.1 # Y==0.0 # Y==224.2 # Y==519.1 # Y==&44.4 #	0.0 C.0 C.0 C.0 C.0 C.0	0.0 0.02 1.39 C.02 C.C 0.0	0.0 C.0 0.17 2.58 0.12 0.0 0.0	0.0 0.04 0.41 0.04 0.0 6.0	6.0 0.0 c.c 0.0 6.0 6.0	0.0 0.0 C.C 0.0 C.C	0.C 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0 C.6	C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C D.O 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C C.C	0.0 C.C C.C C.C C.C Q.Q	0-0 5-5 5-5 5-5 5-5 5-5 5-5	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
2 CIREC	110N -	12.4	P												
Y664.4 P Y519.1 4 Y284.1 P Y- 0.0 - Y- 284.1 - Y- 519.1 - Y- 519.1 - Y- 664.4 P	C.0 C.0 C.0 C.0 C.0 C.0	C.0 C.0 C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0	0.0 0.00 0.07 0.07 0.0 0.0 0.0	6.0 0.0 0.0 0.0 0.0	0.0 6.0 0.0 0.0 6.0 6.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.0 C.C 0.0 C.C C.C	C.C 0.0 0.C C.C 0.0 0.0 C.C	C.C 0.0 C.C C.C 0.0 C.C	C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.C 0.C 0.C 0.C 0.C
2 CIPEC	FIGN +	0.0	*				~								
V664.4 P V519.1 P V284.1 P V- 284.1 P V- 284.1 P V- 284.1 P V- 284.4 P	C.C C.C C.C C.C C.C C.C C.O	C.C 0.0 C.C 0.0 C.C	C.C C.C C.C C.C C.C C.C C.Q	6.0 9.0 0.05 C.04 C.C0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C C.C O.O O.O	0.0 C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0	8.0 6.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C C.C	6.6 6.0 6.0 6.0 6.0 6.0 6.0	6.0 0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.0 0.0 0.0 C.C 0.C
G #EG#/SEC)+	4999.76	3491.04	2400.56	255.58	c. 0	0.0	0.0	2797.86	2044.21	1757.14	<98.90	c.¢	e.c	e.c	0.0

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$(\Theta = 10 \text{ min})$ $(\Omega = 20 \text{ min})$

8.0 337.7 1694.0 3806.9 6193.1 8306.0 9662.3 10000.0 10270.1 11355.2 13045.5 14954.5 14644.8 17724.9 1FCCC.C

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CONCENTRATICN (#C/CU.#)

# DIRECT	10N =	sc.s.0	Þ												
9=-644.4 P 9=-519.1 M 9=-284.1 R 9= C=0 P 9= 384.1 M 9= 519.1 M 9= 664.4 P	0.0 C.C C.0 C.0 C.0 C.0	6.0 C.C C.0 C.0 C.0 0.0	6.0 C.0 G.0 G.0 G.0	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 c.c 0.0 0.0 ¢.c	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C O.O G.O	C.C C.C C.C C.C C.C C.C C.C	6.C 6.C 6.C 6.C 6.0	0.0 C.CQ Q.02 C.C C.C C.C G.Q	0.0 C.(C C.(C C.(C C.(C C.(C C.(C	0.0 C.C 0.0 C.C C.C	0.0 0.c 0.c 0.c 0.c 0.c
8 GIRECT	10# -	492.1	M												
Y2(4.4 P Y2(4.1 P Y264.1 P Y- 264.1 P Y- 284.1 M Y- 284.1 M Y- 519.1 P Y- 844.4 P	C.0 C.0 C.0 C.0 C.0 C.0	C.0 0.0 C.0 C.0 C.0 C.0	0.0 0.0 0.0 0.0 0.0 0.0	8.0 5.0 5.0 6.0 6.0 0.0	0.0 0.0 c.c c.c c.c	0.0 0.0 0.0 0.0 0.0 0.0	0.0 c.c c.c c.c c.c c.c	0.0 c.c c.c c.c c.c c.c c.c	9.0 0.0 0.0 0.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C C.C C.C C.C	6.C 0.0 0.C 0.0 0.0 6.C	C.C G.C0 G.03 C.C C.C G.C C.C	C.CC 0.CC C.CC C.C 0.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.c c.c c.c c.c c.c c.c
2 DIRECT	ECN -	439.7	р.												
Y=-664.4 P Y=-519.1 P Y=-224.1 P Y= 0.0 X Y= 284.1 P Y= 519.1 P Y= 664.4 P	C.0 C.0 C.0 0.0 C.C C.C	C.C C.C C.C C.C C.C	C.C 0.0 C.C 0.0 C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.00 0.02 0.0 0.0 0.0	6.0 6.0 6.0 6.0 6.0 6.0	0.0 C.C 0.0 C.C C.C 0.0	0.0 C.C C.C C.C C.C C.C	0.0 C.C C.C C.C C.C Q.C	0.0 C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.Cl C.C3 C.C0 0.0 C.C C.C	C.C G.C1 C.C3 C.CC G.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C C.C	8.C 0.C 0.C C.C 8.C
2 DIRLCT	ICN +	355.0	۲												
Y==664.4 P Y==519.1 P Y==284.1 P Y==284.1 P Y==284.1 P Y==513.1 P Y==664.4 P	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 C.C 0.0 C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 0.00 C.C2 0.00 C.C	C.C 0.0 0.0 0.0 0.0 0.0	C.C C.C 0.0 0.C 0.C 0.C 0.0	C.C C.C C.C C.C C.C 0.0	0.0 C.C1 S.55 C.C1 C.C C.C C.C	0.0 C.14 2.76 G.C8 C.C C.C 0.C	8.0 0.36 2.27 0.22 C.C C.C 0.C	0.0 0.28 1.15 0.18 0.0 C.C 0.0	C.C C.C7 C.2C C.C4 O.C C.C C.C	C.C C.C C.C C.C C.C C.C	C.C 0.C 0.C 0.C C.C C.C
2 DIR.CT	104 -	252.5	•												
Y=-4(4.4 P Y=-113.1 P Y=-28*.1 P Y=-26*.1 P Y=-264.1 P Y=-517.1 P Y=-564.4 P	C.O C.O C.O C.O C.O	6.0 C.C 0.01 1.03 C.01 C.0 C.0	0.0 C.0 G.19 4.00 0.19 0.0 0.0	0.0 C.0 G.37 3.80 0.37 0.0 0.0	0.0 C.0 C.13 C.51 C.13 C.0 C.C	0.0 6.0 6.0 6.0 6.0 6.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.0 21.27 C.C 0.0 0.0 C.C	C.C C.25 20.01 C.13 C.G 0.0 C.C	C-C 1.C1 16.73 C.5E C.C 0.0 0.0	0.0 1.22 10.75 0.60 C.C 0.0 0.0	0.0 C.75 4.08 0.47 C.C C.C 0.C	0.C C.15 C.45 J.10 C.C C.C C.C	0.0 2.C C.C C.C C.C C.C	0.C C.C C.C C.C C.C C.C
2 CIRECTI	icn -	150.0 #	•												
YEE4.4 # Y519.1 # Y284.1 F Y- C.C # Y- 284.1 # Y- 514.1 # Y= 614.4 #	C.0 0.0 35.45 C.0 C.0 C.0	C.C 0.0 C.JO 32.72 C.JC C.0 C.0	0.0 0.0 1.19 25.44 1.19 0.0 0.0	6.0 1.20 12.27 1.20 C.0 0.0	0.0 C.J 2.14 C.J1 C.C 0.0	0.0 c.c c.c c.c c.c c.c	0.0 0.0 0.0 0.0 0.0 C.C	0.0 0.0 0.0 0.0 0.0 C.C C.Q	C.C 0.01 C.55 C.C1 0.0 C.C C.C	C.C 0.13 2.25 C.CE 0.0 C.C C.C C.C	C.C 0.34 2.27 C.22 0.C 0.C C.C	C.C; C.28 1.55 C.18 C.C 0.0 C.C	C.C G.JO C.C4 C.C G.C C.C	8.C c.c c.c c.c c.c c.c	0.0 0.0 0.0 0.0 0.0
2 CIRLOTI	GN .	65.3 M	l												
V464.4 N V519.1 P V264.1 P V- 284.1 P V- 519.1 P V- 519.1 P	C.0 C.0 C.0 C.0 C.0 C.0	C.C G.Q L.73 G.Q C.C C.C	0.0 0.29 5.99 0.25 0.0 C.0	6.0 6.0 6.49 4.96 6.49 6.4 6.0	C.C 0.15 1.04 0.15 C.O 0.0	8.0 C.C C.C C.C C.C C.C	0.0 5.C 0.0 0.C 0.C 0.0	0.0 C.C C.C C.C C.C C.C	0.0 C.C C.C C.C C.C C.C C.C	0.0 0.c 0.0 0.c C.c C.c	0.0 C.C 0.0 0.0 C.C C.O	8.8 0.01 C.04 C.00 C.0 0.0 C.0	C.C1 C.C3 C.CC C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C	6.6 0.0 0.C C.C 0.C 0.C
2 DIRECTI	Ch +	12.9 #									•	•			
Y== 664.4 P Y== 519.1 P Y== 784.1 P Y== 784.1 P Y== 784.1 P Y== 513.1 P Y== 664.4 P	0.0 C.0 0.0 0.0 0.0 0.0	0.0 C.C C.C 0.0 C.C C.C 0.0	0.0 C.C C.C C.C C.C C.C C.C	0.0 0.0 0.09 0.09 0.09 0.09 0.0	C.C 0.04 0.24 0.04 0.04 0.0 C.C	C.C C.O C.O C.O C.O C.O C.O	C.C G.C G.G G.C G.C G.C G.C	C.C C.Q C.C Q.C Q.Q Q.Q	C.C 0.0 0.0 C.C C.D 0.0	0.0 C.C C.C C.C C.C C.C 0.0	8.C C.C 0.0 C.C C.C 8.0	6.0 0.0 C.0 C.0 C.0 C.0 C.0 C.0	C.C C.C C.C C.C C.C C.C C.C	C.C 0.0 0.c 0.0 C.C C.C	C.C 0.C 0.C 0.C 0.0 0.C C.C
& GIRECTI	CR +	C.O P					:								
94-444.4 P 94-334.1 P 94-234.1 P 94 6.0 P 94 324.1 P 94 324.1 P 94 324.1 P	6.0 6.0 6.0 6.0 6.0 6.0	8.0 C.C C.C C.C C.C C.C C.D	0.0 c.q 0.e 0.0 0.0 0.0	6.0 0.0 0.0 0.0 0.0 0 0.0 0 0.0 0 0.0	0.0 0.0 0.03 C.2C 0.03 C.0 C.C	8.8 0.0 0.0 C.C 0.0 C.C	0.C 0.0 0.C 0.C 0.C C.C	0.C 0.0 C.C 0.0 0.0 F.C	C.C 0.0 C.C 0.0 C.C 0.0 C.C	C.C D.O C.C C.C C.C C.C C.C	8.C 0.0 0.C C.C 6.C 6.C	C.C C.C C.C C.C C.C Q.O	0.0 c.c 0.c 0.c c.c c.c 0.0	8.8 c.c c.c c.c c.c c.c c.c	0.0 0.0 0.0 0.0 0.0 0.0

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$(\Theta = 20 \text{ min})$ $(\Omega = 30 \text{ min})$

\$37.7 1694.6 3806.9 6193.1 8306.0 9662.3 10000.0 10270.1 11355.2 13045.5 14454.5 14644.8 17725.4 12CCC;C c..

CONCENTRATICN (PG/CU.P)

E DIRECT	ICN +	\$55.0	*												
Y664.4 P Y517.1 P Y284.1 H Y- 0.0 P Y- 284.1 P Y- 519.1 P Y- 644.4 P	C.0 C.0 C.0 C.0 C.0 C.0	C.O C.O C.O C.O C.O C.O	6.0 6.0 6.0 6.0 6.0 6.0	6.8 0.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 C.C 0.0 0.0 C.C	C.C 0.0 0.0 C.C 0.0 C.D C.D	C.C 0.00 C.C0 0.0 0.0 C.C	0.00 C.CC C.CC C.CC C.CC C.CC	C.C C.C C.C C.C C.C C.C C.C	C.C C.C 0.0 0.0 C.C 0.0 0.0	0.C C.C 0.C C.C 0.0 6.0	C.C C.C1 C.C5 0.01 C.C C.C C.C	6.0 C.C1 C.C1 C.C1 C.C C.C C.C	0.0 C.C1 C.C4 C.C1 C.C C.C	0.C 0.C1 0.C 0.C C.C
& DIRECT	ICN -	492.1	F												
$ \frac{1}{2} - \frac{1}{2} \frac{1}{2}$	C.0 C.0 C.0 C.0 C.0 C.0	C.C 0.0 C.C C.C 0.0 C.C	0.0 0.0 0.0 C.0 0.0 0.0	0.0 C.0 0.0 C.0 0.0	6.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	0.C 0.0 0.C0 0.C0 0.C 0.0 0.C	0.C 0.0 0.CC 0.0C C.C 0.C	C.C C.C C.C C.C C.C C.C	C.C 0.0 C.C C.0 0.0 C.C	C.C 0.0 0.0 C.C C.O C.O	C.C1 G.C7 C.C1 C.C C.C C.C C.O	C.C1 C.C1 C.C1 C.C C.C C.C	C.C1 C.C1 C.C1 C.C1 C.C C.C C.C	8.C C.C1 C.C5 0.C1 C.C C.C
E EIRECT	ICh =	439.7	#												
Y644.4 P Y519.1 P Y264.1 P Y- 264.1 P Y- 264.1 U Y- 319.1 P Y- 664.4 P	C.C C.C C.O O.O C.O C.O	C.C Q.Q C.C Q.Q C.Q C.Q	C.0 0.0 C.0 0.0 0.0 0.0	6.0 6.0 6.0 6.0 6.0 6.0	0.0 0.01 0.07 0.07 0.01 0.07 0.01 0.0	0.0 0.00 0.02 0.00 C.C 0.0	0.C C.C 0.C1 0.C0 0.C0 6.C 0.0	0.0 C.C 0.CC 0.01 0.00 0.00 0.C 0.0	0.0 C.C C.C C.C 0.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	8.8 C.C C.C 6.0 C.C C.C	0.C 0.12 0.12 C.C1 0.C C.C	C.C C.C2 C.C5 C.C C.C C.C	C.C G.C7 G.28 C.C5 C.C G.C C.C	0.C C.C7 0.26 C.C5 C.C C.C G.C
8 DIRECT	ICA +	355.0	•												
Y=-664.6 P Y=-519.1 P Y=-224.1 P Y= 0.0 M Y= 264.1 P Y= 519.1 P Y= 624.4 M	0.0 C.C C.C C.O C.O C.C	0.0 C.C 0.0 C.C C.O 0.0	0.0 C.C C.C 0.0 0.0	0.0 C.C C.C C.O C.O C.O	C.0 0.04 C.25 C.C4 0.0 C.0	0.0 0.05 0.26 C.C5 0.0 0.0	0.C C.C 0.C2 0.10 C.C2 U.O C.C2	C.C 0.01 0.06 C.C1 0.0 0.0	C.C1 C.C1 C.SE O.O2 C.CC C.CC C.CC	0.0 C.13 2.15 C.C7 C.C C.C 0.0	C.C C.42 3,42 C.26 C.C C.C Q.Q	0.0 C.72 J.79 C.45 C.C C.C	C.C C.66 2.17 C.43 G.C C.C C.C	C.C 0.47 1.75 C.31 0.C C.C C.C	C.C C.41 L.TC C.JE C.C C.C
E DIRECT	10k •	252.5	•												
Y=-664.4 P Y=-284.1 P Y=-284.1 P Y= 224.1 P Y= 224.1 P Y= 517.1 P Y= 664.6 P	C.0 C.0 C.0 C.0 C.0 C.0	6.0 C.C 1.11 C.GZ C.O 0.0	0.0 C.0 C.17 3.84 0.17 C.0 0.0	0.0 5.0 5.85 0.59 6.0 6.0	0.0 C.0 0.76 4.96 C.76 C.C	0.0 0.39 2.00 0.39 0.0 0.0 0.0	0.C 0.0 0.12 C.55 0.12 0.0 C.C	C.C 0.0 21.34 C.31 0.C7 D.O C.C	C.C G.76 20.18 C.23 G.C1 G.O1 G.C	C.C C.SE 16.45 C.T4 C.C 0.0 0.C	C.C 1.57 12.48 C.97 C.C 0.0 C.C	Q.C 1.84 9.73 1.16 C.C C.C Q.Q	0.6 1.41 5.89 C.53 C.C C.C G.O	0.C C.SZ 1.43 0.61 C.C C.C C.C	0.0 C.EG Z.SC C.13 C.C C.C G.C
Z DIRICTI	ICN =	150.0 P	•												
$y_{2-} \in \{4, 4, 4, 4\}$ $y_{4-} \ge \{3, 4, 4\}$ $y_{4-} \ge \{4, 4\}$	C.0 0.0 35.45 C.0 C.0 C.0	C.0 0.0 0.32 32.97 C.32 C.0 C.0	0.0 0.0 1.14 25.06 1.14 0.0 0.0	C.0 C.0 1.86 18.50 1.86 C.0 0.0	0.0 0.0 1.71 11.09 1.71 C.C 0.0	0.0 0.0 0.73 3.74 0.73 C.C 0.0	0.0 0.0 0.94 0.94 0.21 C.C 0.C	0.0 0.12 0.52 0.12 C.C 0.C	C.C Q.01 C.£1 C.15 Q.03 C.CC C.C	C.C 0.13 2.18 C.C5 0.0 C.C C.C	C.C 0.42 3.41 C.28 0.0 0.C C.C	C.C C.72 3.79 C.44 C.C 0.0 C.C	C.C C.47 2.77 C.44 C.C C.C C.C	0.C C.47 1.76 C.31 C.C 0.0 C.C	C.C C.41 1.11 C.20 C.C G.C G.C
Z EIRECTI	ICN	45.3 M	ı												
Y==664.4 * Y==514.1 # Y==264.1 # Y==0.0 * Y==284.1 # Y==519.1 # Y==664.6 #	C.C C.C C.C C.C C.C	C.C 0.C C.03 1.P6 0.03 C.C C.C	C.C 0.25 5.75 0.25 0.0 C.C	0.0 0.0 0.80 7.81 0.20 0.0 6.0	6.6 C.C 0.93 5.96 C.43 0.0 0.0 0.0	0.0 C.C C.41 2.06 C.41 0.0 0.0	0.0 C.C 0.10 0.45 C.1C 0.C	0.0 C.C 0.20 C.C5 C.C 0.0	0.0 C.C C.C 0.06 C.C1 C.C C.C	6.0 0.0 0.0 0.0 0.C C.C	0.0 6.0 6.C 0.C 0.C C.C C.0	6.C 0.C3 C.16 C.C2 C.D 0.C C.C	C.C C.36 C.C C.C C.C C.C C.C C.C	C.C 0.08 0.30 C.C5 C.C C.C C.C	0.C 0.27 0.27 0.C5 0.C 0.0 0.0
2 DIRECTI	CN .	12.9 #	ı												
V664.4 P V519.1 P V264.1 P V- 264.1 P V- 519.1 P V- 519.1 P	0.0 C.C C.C C.C C.C C.O C.O	0.0 5.0 0.0 5.00 5.00 5.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 1.75 C.20 C.0 0.0	C.C 0.0 0.37 2.23 0.37 C.O C.O	C.C 0.13 0.61 0.13 0.0 0.0	C.C C.C 0.01 0.C1 0.C 0.C 0.C	C.C 0.0 C.C 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C C.C	0.C C.C C.C C.C C.C C.C	C.C D.O C.Q1 C.Q C.C C.C	6.0 C.C1 C.C2 C.C C.C C.C G.C	C.C1 C.C2 C.C1 C.C C.C C.C	C.C C.C C.C C.C C.C C.C	C.C 0.01 0.C4 C.C1 0.G C.C C.C
2 DIALCIL	0N -	c.c =													
Y444.4 P Y515.1 P Y284.1 H Y- 6.0 P Y- 284.1 P Y- 519.1 P Y- 444.4 P	6.0 C.C C.C C.C G.C B.C	0.0 (.(0.01 (.) (.) 0.0 0.0 0.0	6.0 C.C C.C C.C C.C C.C C.C	6.0 C.0 C.10 1.41 C.18 C.0 G.0	0.0 0.35 2.05 0.35 C.0 6.0	0.0 0.11 C.32 0.11 0.0 C.C	0.0 0.0 0.0 0.0 0.0 0.0 0.0	8.0 0.0 0.0 0.0 0.0 0.0 6.0	C.C C.C C.C C.C C.C C.C	6.C C.C C.C C.C C.C G.C G.C	C.C C.C C.C C.C C.C C.C C.C C.C	0.0 C.CO C.CO C.C C.C C.C C.C	6.0 C.CC C.C1 C.C0 C.C C.C C.C	0.0 C.C1 C.C3 C.G1 0.C C.C 0.9	0.0 0.01 0.00 0.0 0.0 0.0 0.0

@FECA/SEC1+ 4999.76 5100.34 5087.10 5149.54 3694.23 1490.48 390.11 3214.08 3102.40 3108.67 3135.80 3017.97 2172.17 1380.22 1192.65 .

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$(\Theta = 40 \text{ min})$ $(\Omega = 50 \text{ min})$ # EIRECTION (ARTERS)

C.C 337.7 1694.C 3806.9 6193.1 8306.0 9662.3 100C0.0 10270.1 11355.2 13049.5 14554.5 16644.8 17725.9 12CCC.C

	、			CCACENT	RATICN (P	C/CU.#3									
E DIRECT	Itck +	505.0	,		•										
Y{{{,4} } Y{{{,1} } Y{{	0.0 C.C C.C C.C C.C C.C	C.0 C.C C.C C.C C.C C.C	0-0 0-3 0-3 0-0 0-0 0-0 0-0	C.O C.O C.O G.O C.O C.O	C.C C.O C.C O.O C.C C.C	C.C 0.0 C.C 0.0 C.C C.C	C.C G.C G.CO C.Ci G.CO G.C C.C	C-C C.C C.Cl C.Cl C.C C.C C.C	C.C 0.0 C.Ci C.Cc 0.0 C.C	• C.C C.C G.G C.C1 O.O O.O	0.C 0.C 0.C 0.C] 0.C] 0.C 0.0	C.C C.C1 C.C5 C.C2 C.CC C.C0 C.C0	0.0 C.C1 C.C2 C.C C.C C.C	0.0 C.C3 C.C5 C.C C.C C.C	6.C 0.C3 0.C2 0.C2 0.C C.C C.C
2 DIACCI	1CN -	452.1	P	•											
Y==264.4 + Y==319.1 # Y==284.1 P Y= C=0 P Y= J84.1 # Y= J19.1 # Y= 519.1 # Y= 644.4 P	C.C C.C C.Q C.Q C.Q C.Q	C.C C.C C.C C.C C.C C.C C.O C.O	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 6.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0 0.0	6.0 6.0 6.0 0.0 0.0 0.0 0.0	C.C 0.0 C.Cl 0.C0 0.0 C.C	C.C C.C C.C C.C C.C C.C C.C C.C C.C C.C	C.C 0.0 C.C C.C C.C C.C C.C	C.C 0.0 C.C C.C C.C C.C C.C	C.C 0.0 0.0 0.01 0.0 0.0 0.C	C.C C.C1 C.C7 C.C2 C.C0 C.C0 C.C0	C.C C.C C.C C.C C.C C.C C.C	C.C C.C3 C.12 C.C2 C.C C.C C.C	0.C C.C4 C.13 0.C2 C.C C.C C.C
Z CIRECT	ICN =	439.7	R												
T=-EE4.4 P Y=-513.1 P Y=-284.1 P Y= C.C M T= 284.1 X Y= 519.1 P Y= EE4.4 M	C.0 0.0 C.0 0.0 C.0 C.0	C.C G.Q C.C C.C C.Q C.Q	C.0 2.0 0.0 C.0 0.0 0.0	0-0 0-0 0-0 0-0 0-0 0-0	0.0 8.C 0.01 0.09 0.01 C.C 0.0	0.0 0.01 0.01 0.01 0.01 0.01 0.0	0.0 C.C 0.11 0.03 0.C 0.0	0.0 C.C C.C 0.13 0.03 C.C 0.0	0.0 C.C C.C C.13 0.03 C.C 0.0	0.0 C.C C.14 C.C5 C.C C.C	8.0 0.0 C.C C.12 0.C4 0.C C.C	C.C C.13 C.C7 C.01 C.C0 C.C	C.C C.11 C.45 C.C7 C.C C.C C.C	G.C C.17 G.62 C.11 C.C G.C	C.C C.10 C.20 C.12 C.C G.C G.C
E DIRECT	ICA =	355.0	H												
Y=-664.4 # Y=-519.1 F Y=-224.1 # Y= 0.0 x Y= 224.1 # Y= 519.1 # Y= 684.4 *	0.0 C.0 C.0 C.0 C.0 C.C	0.0 c.c c.c c.c c.c c.c	0.0 0.0 0.0 0.0 0.0 0.0	8.8 6.0 6.0 6.0 0.0 0.0	0.0 0.05 0.31 0.05 0.0 0.0	0.0 0.17 0.86 C.17 0.0 0.0	0.C 0.27 1.16 C.17 0.C 0.C	C.C C.C 0.25 1.21 C.25 0.C 0.C	C.C C.C1 C.27 1.24 C.3C 0.CC 0.0	0.C C.13 2.54 1.33 C.31 C.C2 0.0	0.0 0.43 3.66 1.12 C.19 C.C3 0.C	C.0 C.73 J.53 G.72 C.C4 C.C2 C.C	C.C C.93 2.85 C.61 C.C C.C C.C	C.C 1.C2 3.Te C.ee 0.C C.C C.C	C.C 1.C5 7.14 C.et 0.C 0.C C.C
2 DIRECT	10H -	252.5	•												
Y==664.4 # Y==514.1 # Y==784.1 # Y= 64.1 # Y= 514.1 # Y= 319.1 M Y= 664.4 #	0.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 0.01 1.C8 C.01 C.C 0.0	0.0 0.1 3.90 0.18 C.C 0.0	C.Q C.Q Q.57 5.72 0.57 C.Q Q.Q	C.C C.Q E.1C 0.94 C.C C.C	C.C G.O 1.21 6.C3 1.21 0.0 C.C	0.C 0.G 1.36 5.52 1.16 0.0 C.C	0.C 0.C 27.64 5.84 1.35 0.0 C.C	C.C C.78 21.52 5.64 1.34 0.04 C.C	C.C C.SS 17.20 5.55 L.10 0.14 0.C	C.C 1.58 13.48 3.86 C.64 0.13 0.C	C.C 1.FR 1C.11 1.95 C.14 C.C6 G.C	C.Q 1.57 8.17 1.30 C.C C.C C.C	0.C 1.55 7.2C 1.2e C.C C.C 0.C	0.(C 1.59 7.11 1.23 C.C C.C 8.C
2 GIRECT	ICN •	156.0 1													
Y==664.4 P Y==519.1 P Y==284.1 P Y= C.C P Y= 284.1 P Y= 519.1 P Y= 519.1 P	C.0 C.C C.0 35.45 C.0 0.0 C.0	C.0 C.C C.31 32.69 C.31 0.0 C.0	0.0 0.0 1.16 25.20 1.16 0.0 0.0	0.0 1.82 18.21 1.82 C.0 0.0	6.0 C.0 7.10 13.57 2.10 C.C 0.0	0.0 2.2e 11.18 2.26 C.C 0.0	0.0 2.34 10.12 2.34 C.C 0.C	0.C 0.0 7.34 9.82 7.34 C.C C.C	C.C 0.01 2.27 9.44 2.30 C.CC C.C	C.C 0.14 4.29 7.58 2.07 C.C3 C.C3	C.C 0.44 4.44 1.13 0.C5 0.C	C.C C.74 4.17 1.47 C.28 C.C3 C.C	C.C C.S3 3.86 C.C C.C C.C C.C	C.C 1.C3 3.76 C.59 C.C C.C C.C	0.00 1.05 3.73 0.5 0.0 0.0 0.0
2 UERFCT	ICN -	45.3 M	•												
Y=-664.4 P Y=-51+.1 P Y=-264.1 P Y=-284.1 P Y= 284.1 P Y= 284.1 P Y= 519.1 P Y= 664.4 P	C.O C.O C.O C.O C.O C.O	C.C 9.0 C.C2 1.11 0.02 C.0 C.C	G.0 0.0 0.27 5.25 0.27 0.0 C.0	0.0 0.77 7.61 0.77 0.0 C.0	0.0 C.C 1.21 7.67 1.73 0.0 0.0	0.0 C.C 1.57 7.53 1.57 0.0 0.0	0.0 C.C 1.72 7.21 1.72 0.C 0.0	0.0 C.C 1.72 7.03 1.72 C.C Q.Q	0.0 C.C 1.45 6.68 1.67 C.CC C.C	0.0 0.0 5.39 1.47 C.C1 C.C	0.0 0.5 2.59 0.76 C.Cl C.Cl	C.C C.24 C.54 C.54 C.16 C.C1 C.C	C.C C.13 C.13 C.C C.C C.C C.C C.C	8. C 0.20 C.71 C.C7 C.C C.C C.C	C.20 C.12 C.73 C.C C.C C.C
2 DIRLET	ICN =	12.9 P	,							•		•			
V-664.4 # V-514.1 P V-26.1 V V- 0.0 P V- 284.1 P V- 514.1 P V- 514.1 P	0.0 c.c c.c c.c c.c c.c	0.0 0.0 0.0 0.0 C.0 C.0	6.0 6.0 6.0 6.0 6.0 6.0	0.0 0.17 1.60 C.17 C.0 C.0	C.G 0.0 0.45 3.74 0.45 0.0 0.0	C.C 0.0 1.12 4.90 1.12 0.0 0.0	C.C 0.C 1.22 4.78 1.;2 0.0 0.C	C.C L.2C 4.55 1.2C 0.0 0.0	C.C L.25 4.19 1.11 C.CC 0.0	0.0 C.C 1.13 2.95 C.E4 C.C1 Q.Q	0.0 0.43 1.04 0.13 0.21 0.01	0.0 0.02 0.12 0.12 0.04 0.00	C.C C.C3 C.11 C.C2 0.C C.C C.C	C.C 0.C6 C.14 C.C4 0.C C.C C.C	C.CC 0.C7 C.21 C.C2 0.C C.C C.C
8 DIALCTI	ick =	c.8 P													
Y664.4 Y519.1 Y284.1 Y- C.C P- 364.1 Y- 319.1 Y- 319.1 Y- 444.4 Y- 319.1 Y- 444.4 Y- 519.1 Y-	6.0 0.0 0.0 6.0 6.0 6.0	6.0 6.0 6.0 6.0 6.0	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.0 0.13 1.44 C.13 C.0 C.0	6.0 0.4 3.13 0.43 0.0 ¢.¢	0.0 1.10 4.78 1.10 0.0 6.6	0.C 0.0 1.30 4.64 1.30 0.G C.C	C.C 0.0 1.17 4.44 1.17 0.D C.C	C.C 0.C 4.52 1.00 0.00 C.C	C.C C.27 2.77 0.79 0.01 C.C	0.C C.C 0.35 0.44 C.2C 0.C1 0.C	0.0 C.C1 C.C3 C.C3 C.C0 C.C0 C.C0	0_0 C.C2 C.C2 C.C C.C C.C	0.0 c.cs c.it 0.04 c.c c.c c.c	0.C0 0.C4 C.IP 0.C2 0.C C.C G.C

\$\$\$C#/\$EC1+ 4444.76 5074.32 5134.58 5034.74 4868.48 4898.25 4024.15 7773.73 7645.72 7104.28 5335.32 3623.04 3024.45 2450.56 2920.02

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$(\Theta = 80 \text{ min})$

 $(\Omega = 90 \text{ min})$ # DIRECTION (HETERS)

6.0 337.7 1694.C 3006.9 6193.1 8306.0 9662.3 10000.0 10270.1 11355.8 13045.5 14554.5 16644.8 17725.9 18000.0

CCACENTRATICN (PG/CU.P)

E DIRLCT	11Ch = 1	\$65.0	•		•. • .										
Y==664.4 # Y==317.1 # Y==284.1 # Y==0.0 # Y= 284.1 # Y==319.1 # Y==664.4 #	C.O C.O C.O C.O C.O C.O	C.C C.C D.D C.C C.C D.D	0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0 0.0 0.0	C.C C.C C.C C.C C.C C.C	C.C 0.0 C.C 0.0 C.C C.C	C.C G.C G.CG C.C1 G.CG G.G C.C	C.C G.C G.CC C.CI C.CC G.G C.C	C.C 0.0 C.Cl C.CC 0.0 C.CC	C.C G.C G.C3 C.C1 G.C G.C	0.C C.C 0.C6 G.C1 0.C 0.C	C.C3 C.C3 C.C2 C.C3 C.C3 C.CC C.C2	C.C2 C.11 C.15 C.C5 C.CC C.C	0.00 C.10 C.19 C.C6 C.C1 C.C1	0.00 0.04 0.20 0.20 0.27 0.07 0.01
2 DIRECT	160N -	492.1	F												
Y=-664.4 # Y=-519.1 K Y=-286.1 # Y= C.0 # Y= 284.1 # Y= 519.1 # Y= 664.4 #	C.C C.C C.Q C.Q C.Q C.Q	C.1 C.C C.C C.C C.C 0.0 0.0	0.0 c.0 c.c c.c c.c c.c	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.C 0.0 4.0 0.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.C 0.0 C.(C 0.C1 0.C0 0.C	C.C 0.0 C.CC C.C1 0.CC C.C C.C	C.C 0.0 C.C C.C C.C C.C	C.C 0.C C.C C.C C.C C.C	C.C 0.0 0.0 C.C7 0.01 0.C C.C	C.C G.10 C.11 C.C3 C.C0 C.C	C.C C.C2 C.13 C.17 C.C5 C.C5 C.C5	0.CC C.C4 C.15 C.21 C.C7 C.C1 0.0	0.CC C.C4 C.21 0.22 C.C7 C.C1 0.C
2 GIRECT	ICN -	439.7	R												
V==264.4 # V==519.1 # V==284.1 # V==264.1 # V==264.1 # V==519.1 # V==864.4 #	C.0 C.0 C.C O.0 C.C C.0	C.C 0.0 0.0 C.C 0.0 C.0 C.0	C.C C.C C.C C.C C.C C.C	C.0 G.0 G.0 G.0 G.0 C.0	0.0 C.0 C.08 C.01 C.C C.C	0.0 6.C 0.07 0.07 6.02 C.C 0.0	0.0 C.C 0.C3 0.11 0.03 0.C 0.0	0.0 C.C C.C3 0.13 0.03 C.C 0.0	0.0 C.C C.I3 C.C C.C C.C C.C	0.0 C.C C.14 C.C5 C.C5 C.C	0.0 0.0 0.2 0.24 0.28 0.28 0.28	C.C3 C.24 C.16 C.12 C.C1 C.C1 C.C1	C.C C.12 C.51 C.15 C.C2 C.C	C.CC C.15 C.81 C.80 C.18 C.18 C.18 C.18 C.C	C.CC C.JC C.JE C.J7 C.17 C.C? C.C
2 DIRECT	ISCA +	355.0	*				•								
Y==664.4 # Y==519.1 # Y== 0.0 # Y== 0.0 # Y== 264.1 # Y==513.1 # Y==664.4 #	0.0 6.0 6.0 6.0 6.0 6.0	0.0 C.C C.C C.O C.O C.O	0.0 6.0 6.0 6.0 6.0 6.0	0.0 0.0 0.0 0.0 0.0 0.0	C.C 0.05 0.32 C.C5 C.0 C.0	0.0 0.17 0.86 C.17 0.0 0.0	0.C 0.26 1.12 C.je 0.0 0.0	C.C C.C 0.28 1.18 C.28 0.0 0.0	C.C C.C1 C.27 1.23 C.29 C.CC Q.0	G.C C.13 2.13 1.45 C.34 C.C; O.C	0.0 0.43 3.87 1.82 C.40 C.C5 0.0	G.C C.75 4.4C Z.17 C.46 C.C5 C.C5	C.C C.97 4.1C 2.19 C.52 C.11 C.C	C.C1 1.C8 4.47 2.45 G.56 C.12 C.C	C.C2 1.11 4.46 7.57 C.57 C.12 C.C
2 DIRECT	101 -	252.5	۲												
9=-664.4 P 9=-513.1 R 9=-284.1 4 9= 6.0 P 9= 284.1 M 9= 519.1 M 9= 664.4 P	C.0 C.0 C.0 C.0 C.0 C.0 C.0	0.0 6.0 1.07 6.01 1.07 6.01 6.0	G.C G.O G.18 3.91 G.18 C.C G.O	C.0 C.0 C.57 5.70 0.57 C.0 C.0	C.C G.95 4.14 0.95 C.O C.C	C.C 0.0 1.21 6.CC 1.21 C.O C.C	0.C 0.0 1.33 5.E1 1.33 0.C C.C	C.C 0.0 22.63 5.76 1.36 0.0 C.C	C.C C.26 21.52 5.65 1.23 0.04 C.C	C.C C.SS 17.S4 6.CS 1.25 0.14 C.C	C.C 1.EC 14.18 6.18 1.2C 0.20 0.0	C.C 1.53 11.68 6.10 1.37 C.23 C.23	C.C 3.C 5.85 5.73 1.42 C.Z 5 C.C	0.62 2.13 5.(3 5.79 1.45 6.24 0.6	C.C4 2.14 8.85 1.46 C.20 G.C
2 CIRECT	ICK = '	150.0 /	•												
Y=={{} Y==519.1 # Y==264.1 # Y== 0.0 # Y== 264.1 # Y== 519.1 # Y== 519.1 #	C.O C.C S.45 C.C D.O C.O	C.O C.C C.31 32.08 C.31 0.0 C.O	0.0 0.0 1.16 25.22 1.16 0.0 9.0	0.0 C.0 1.61 15.18 1.41 C.0 G.0	0.0 0.0 2.12 13.63 2.12 C.C 0.0	0.0 0.0 2.24 11.12 2.24 C.C 6.0	0.0 2.29 9.54 2.29 C.C 0.C	0.0 2.25 9.89 2.29 C.C C.C	C.C 0.01 2.28 5.5C 2.30 C.CC C.C	C.C 0.14 4.53 8.27 2.21 0.C3 C.C	C.C 0.47 5.75 2.16 2.29 0.09 C.C	C.C C.E3 6.22 7.59 2.28 C.17 C.C	C.C 1.CS 8.25 7.16 2.26 C.23 C.C	0.05 1.23 6.19 6.92 2.26 0.26 C.C	0.05 1.24 6.26 2.25 0.0
2 CIRECT	ICh +	65.3 P	•												
Y==644.4 • Y==519.1 P Y==j64.1 P Y==0.0 P Y==284.1 P Y==284.1 P Y==519.1 P Y==644.6 M	C.C 0.0 C.C 0.0 C.C C.C	C.C 0.C 1.02 0.02 C.O C.C	C.C 0.27 5.86 0.27 0.2 0.2 0.0 C.0	C.0 C.0 0.76 7.59 C.76 0.0 C.0	C.C 1.22 7.72 1.22 0.0 0.0	0.8 5.5 7.44 1.55 6.0 0.0	0.0 C.C 1.72 7.23 1.72 C.C 0.0	0.0 8.C 1.76 7.17 1.76 G.C 0.0	0.0 C.C 1.57 7.12 1.79 C.CC C.C	0.0 G.C 1.36 4.92 1.92 C.C1 C.C	0.0 0.02 1.74 6.43 2.08 C.C5 0.0	6.C 0.16 2.39 4.37 2.21 C.12 C.C	C.C C.32 2.25 £.17 2.29 0.19 C.C	0.65 0.42 3.67 e.63 2.32 6.23 6.6	C.CE 0.45 3.11 5.54 2.32 C.24 C.C
2 014651	IEN +	12.9 P	I												
V=-664.4 # V=-519.1 # V=-284.1 # V= 0.0 # V= 264.1 # V= 519.1 # V= 664.4 #	0.0 C.C C.C C.C C.C C.C	0.0 C.C 0.0 C.C C.C 0.0	0.0 0.0 0.0 0.0 0.0 0.0 6.0	6.0 0.17 1.42 0.17 6.0 6.0	C.C 0.0 3.45 3.74 0.65 0.0 C.O	C.C 0.0 1.10 4.84 1.10 0.0 0.0	C.E G.C 1.35 5.24 1.35 0.0 0.0	C.C G.G 1.41 5.31 1.41 0.0 0.0	C.C 1.EC 5.36 1.4E C.CC 0.0	0.0 C.C 1.13 5.53 1.63 C.C3 0.0	0-C 0.10 7-C0 5-63 1-E4 C.CE 0-0	0.C C.16 2.13 5.64 2.C8 C.15 C.15 C.C	C.C C.25 2.13 5.18 2.22 C.12 C.C	C.CS 0.31 2.43 5.41 2.23 C.25 C.C	C.C5 G.32 2.45 5.34 2.22 C.26 C.C
2 DIRECTI	ON -	(.C P									÷				
Y==664.6 F Y==513.1 M V==284.1 F Y==284.1 F Y= 384.1 M Y= 384.1 M Y= 384.1 M	C.C C.C C.C C.C C.C C.C C.C	C.C C.C C.C C.C C.C C.C C.C	0.0 0.0 0.0 0.0 0.0 0.0	0.0 C.2 0.15 1.40 C.15 C.15 C.0 6.0	0.0 0.43 3.54 0.63 C.C C.C	0.0 5.0 1.00 4.72 1.08 0.0 5.6	8.C 0.0 1.34 5.14 1.34 0.C C.C	8.C 0.0 1.40 5.22 1.4C 0.0 C.C	C.C 1.05 5.27 1.45 6.00 C.C	C.C L.7C J.43 L.43 0.03 C.C	0.0 C.1C 1.74 5.57 1.24 0.00 C.C	C.C C.14 2.C 5.5 7.C7 C.15 0.C	0.0 C.34 2.30 5.54 2.21 C.12 0.0	0.05 C.3C 2.34 5.37 2.73 C.25 0.0	Q_C5 C.32 2.41 5.3C 2.21 C.26 D.C

\$#16#/3861+ 4999.76 5070.20 5144.44 5025.23 4899.54 4824.38 4777.76 7745.83 7774.45 7753.57 7755.69 7704.23 7642.32 7577.87 7556.78

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Sensivity Analysis

The merit of any model depends on its ability to represent the actual physics and chemistry.

It is evident from the diffusion equation that both the turbulent diffusivities and the velocity profiles have a direct influence on the values calculated by the present model. This is the reason why, as was previously discussed, general expressions are provided for the turbulent diffusivities and the velocity profiles by means of several parameters. In this way, different functional relationships can be easily tested with the present model.

All the parameters given in Table 6.10 must be supplied as input information by the user. Therefore, these parameters can be selected to obtain the best set which will represent the actual conditions of the problem to solve.

The influence of these parameters on the concentration distribution is discussed next. The basis of this analysis is the parametric study on hypothetical cases previously presented.

The influence of AM can be observed from a comparison between cases 2 and 3. As the wind velocity increases, i.e., AM decreases, the values for the mean concentration at a same position decrease. This effect is more evident for the concentration distribution at the ground level.

Table 6.10 : Parameters in Turbulent Diffusivity and Velocity Profiles

ISTB the stability class

ALPHA the coefficient in the equation for the y-component of the turbulent diffusivity vector

UST the geostrophic wind speed

- AM the exponent in the power law equation for the velocity profiles
- P the constant that determines the direction of the mean wind velocity
- TCH the parameter that specifies the time required for the y-component of the wind velocity to change from its value at t=0 to its maximum value, P•u

A comparison between cases 1 and 3 shows the effect of alpha on the concentration distribution. An increase in alpha (Case 3) means an increase in the y-component of the turbulent diffusivity vector, and thus more dispersion occurs in the y-direction. The net effect, as can be observed in Tables 6.4 and 6.5, is that the concentration at the centerline decreases, and at any y different than zero increases, when compared to the values at same x and z for case 1.

The effect of the chemical reaction on the concentration distribution can be observed from a comparison between cases 3 and 4. The concentration values at any location, x, y, z, in case 4, where a chemical reaction occurs, are smaller than the values in case 3, where the rate of reaction is zero.

A comparison between cases 4, 5, and 6 shows the effect of the atmospheric stability on the concentration distribution. As the atmospheric stability increases, the concentration values at the centerline increase, and the concentration distribution at ground level decreases. It can also be observed that as the instability increases, the pollutant dispersion increases, and the x-position of the maximum ground level concentration moves closer to the source.

Case 7 shows the effect of the stack height on the concentration distribution. It is evident that as the stack height increases, the concentration values at ground level decrease (compare with case 4).

Finally, the effect of the y-component of the wind velocity (case 8) is shown in Tables 6.6 and 6.7. In this case, V was directed in the positive y-direction (from -YMAX to YMAX), and the net effect is to move the plume more towards that direction.

From this analysis it can be concluded that the present model responds to variations in atmospheric conditions. Furthermore, the results indicate good agreement when compared to an actual response one would expect.

There are two additional parameters that will also influence the results of the present work. Although they do not affect the solution as much as the previous ones, some attention should be devoted to them.

One arises because of the use of RKGS subroutine for solving the system of first-order ordinary differential equations. This parameter is the upper error bound, which must be supplied as input information by the user, and is called PRMT(4) in the computer program.

Neither the truncation errors nor estimates of them are obtained in the calculational procedure performed by "RKGS". Therefore, control of accuracy and adjustment of the step size is done by comparison of the results due to double and single step size calculations.

The procedure is the following: a test value δ (see RKGS [14]), which is an approximate measure for the local truncation error, is compared to the given tolerance PRMT(4). If δ is greater the PRMT(4), the step size or increment of integration of the independent variable is

halved, and the procedure starts again. However, if δ is less than PRMT(4), the results are assumed to be correct.

It can be observed that the larger the value of PRMT(4), the faster the integration is performed. This indicates a decrease in the computing time required, but there is a possibility of obtaining less accurate results.

In order to avoid this inaccuracy on the results, the value of PRMT(4) is usually given small, i.e., in the range of 10^{-3} to 10^{-5} . However, its value and the value for PRMT(3), the initial step size, are actually dependent on the problem to be solved. Unfortunately, there is no general formula to evaluate these parameters. Therefore, for each particular problem, their values have to be studied in order to obtain a fast and accurate solution.

In the present work, this study was performed in the following way: initially, a small value for PRMT(4) was given, i.e., 10^{-5} . Then, this upper error bound was increased and the results for the concentration distribution compared to the previous ones. This procedure was stopped when a change in the second decimal on the concentration values was observed. The corresponding PRMT(4) was then used thereafter, for all the other cases solved.

A similar procedure was performed to obtain the best value for PRMT(3). In this case, several values were

tested and the corresponding computing times compared. It should be pointed out that this parameter affects only the number of bisections done on it, which is also related to the given tolerance PRMT(4).

For the present study, the best values obtained for these two parameters are given in Table 6.11.

Table 6.11 : PRMT(3) and PRMT(4) Values

	Prairie Grass Runs	Hypothetical Cases				
PRMT(3)	.05 min	2.5 min				
PRMT(4)	.01 min	1.0 min				

Used in the Simulations

It can be observed that the values of PRMT(3) and PRMT(4) for the Prairie Grass runs are much smaller than those for the hypothetical cases. The reason being the much smaller dimensions in the three coordinate directions, and thus a smaller step size of integration and smaller upper error bound were needed.

Orthogonal collocation introduces a second parameter that can affect the solution of the model. It is evident that the number of equations increases as the number of orthogonal points used to obtain the solution is increased. This situation will therefore increase the computer time requirements.

Having this in mind, the present method was developed for a variable number of orthogonal points in the x, y, and z directions. They must be supplied as input information by the user, and can be changed from one simulation to another. However, in the present program there is a restriction to use not more than 15 points in each direction and the product $(N_x+1)*N_y*N_z$ be less than 700. The reason being dimension and common statements presently used in the computer program. Actually, no limit exists except for computer capacities and time requirements to solve the problem.

A similar analysis to the one performed for PRMT(4) was done for the number of orthogonal points needed to obtain accurate results. It was concluded that 5 to 10 points in each direction were enough.

Facilities at U. of H. and Time Requirements

The computer facilities at the University of Houston consist of a UNIVAC 1108 digital computer at the University Computing Center and of an IBM 360 Model 44 digital computer in the Engineering Systems Simulation Laboratory of the Cullen College of Engineering.

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The computer time required to simulate atmospheric diffusion by the present method depends on the type of problem to be solved. For all the hypothetical cases simulated, with the exception of cases 5 and 10, the ratio of CPU time to real time was about 1/93 on the UNIVAC and 1/16 on the IBM. Case 5 was solved with a ratio of 1/53 on the UNIVAC and 1/10 on the IBM, and the two sources case had a ratio of 1/46 and 1/7, respectively.

The computer time required to solve each of the Prairie Grass runs was higher than the hypothetical cases. The reason being the small dimensions in the x, y, z directions, and thus a very small diffusion time. This required, as it was previously discussed, a very small step size in the integration of the differential equations and a very small upper error bound. In these cases, the ratio of CPU time to real time was about 1/2 on the UNIVAC and 3/1 on the IBM.

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Chapter VII

SUMMARY OF RESULTS AND RECOMMENDATIONS

Turbulent diffusion from single or multiple point sources in the atmosphere was successfully simulated using the K-theory and a new numerical technique, orthogonal collocation.

Excellent agreement was observed between simulated and experimental concentration profiles for ground level emission sources. The present model had also an excellent response to variations in atmospheric conditions. This was obtained by simulating hypothetical elevated source cases.

Empirical equations were used to describe the mean wind velocities and the turbulent diffusivities. Several parameters were included in these equations so that many atmospheric conditions can be simulated by the present technique.

The present method has several very significant advantages over other available methods, i.e.,

 A general 3-dimensional, unsteady state problem can be solved using a simpler numerical technique. Accurate results can be obtained in very reasonable amount of computer time.

- The method can handle multiple sources put at any position, depending only on the orthogonal points considered.
- Several meteorological effects are taken into consideration.
- 4.) Mean velocities and turbulent diffusivities can be functions of all three position coordinates, time, and meteorological conditions.
- 5.) Cases with or without an inversion layer, and with or without generation or deposition at the ground level can be solved.
- 6.) The concentration at the source or the emission rate can be given as input information. The method will also calculate the flux across y-z planes at x=constant.
- Chemical reactions in the atmosphere can be incorporated.
- 8.) Most of the assumptions involved in the present method are in the input information.

Although the present model gives an improved method for solving atmospheric diffusion problems, it should be extended such that any general case could be solved. These extensions should include:

- A better representation of true dispersion processes by means of improving the expressions for turbulent diffusivity and mean wind velocity profiles;
- The incorporation of more realistic chemical reactions and in general any type of removal processes;
- 3.) The incorporation of other atmospheric effects such as the Coriolis effect;
- Improvements in orthogonal collocation such as removing the restriction on the position of each point source; and
- 5.) Extension to area and line sources.

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APPENDIX

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APPENDIX A

COMPUTER PROGRAM LISTING

The computer program used in the present work is shown next. All statements are written in Fortran IV. This program can be executed in IBM 360 or UNIVAC 1108 digital computers.

1:	C	******	******	**MAIN	10	•
2:	C			MAI'.	20	
3:	С	MODEL FUR ATMO	SPHERIC DIFFUSION BY ORTHOSONAL COLLOCATION	MAIN	40	
4:	C			MAIN	40	
5:	С	MIGUEL T. FLEI	SCHER	MAIN	50	
£ :	С			14 I N	60	
7:	6	******	* * * * * * * * * * * * * * * * * * * *	**MA I N	70	
9:	С			MAIN	но	
9:	С			MAIN	30	
10:	C	NOMENCLATUR		MAIN	100	
11:	С			MAIN	110	
12:	Ç			MAIN	120	
13:	C	IINIT, ENDS	- INITIAL AND FINAL SIMULATION TIMES	MAIN	130	
14:	С	Y	- CONCENTRATION (USED FOR PKCS)	MAIN	140	
15:	С	CC(I,J,K,L)	- CONCENTRATION AT POINT (J,K,L) DUE PRIMARILY TO	MAIN	150	
16:	С		ITH. SOURCE (USED FOR OUTPUT)	MAIN	160	
17:	С	l(J,K,L)	- CONCENTRATION AT POINT (J,K,L)(USED FOR ZERU AND AV	G)MAIN	170	
18:	С	NX	- NUMBER OF POINTS IN X DIRECTION	MAIN	1 * 0	
19:	С	\mathbf{M}	- NUMPER OF POINTS IN Y DIRECTION	MAIN	1 10	
20:	С	11Z	- NUMBER OF POINTS IN 7 DIPECTION	MAIN	200	
21:	С	XMAX(I)	- MAXIMUM DISTANCE FROM THE ITH. SOURCE IN THE K	MAIN	21)	
27:	С		DIRECTION	MAIN	220	
23:	C	YMAX	- MAXIMUM DISTANCE IN Y DIRECTION	MAIN	230	
24:	С	F1	- MAXIMUM HEIGHT ABOVE TERRALI	MAIN	240	
25:	С	<u> </u>	- TURBULENT EDBY DIFFUSIVITIES IN Y AND Z FIRECTIONS	MAIN	250	
26:	C	4 L P H A	- CONSTANT USED TO CALCULATE AKY	MAIN	2(1)	
27:	C	U, V, N	- COMPONENTS OF MEAN WIND VELOCITY VECTOR	MAIN	270	
28:	С	UST, M	- PARAMFTERS IN U EXPRESSION	ΜΔΙΝ	280	
29:	C.	TCH	- TIML WHEN V BECOMES 100 3	MAIN	2 10	
30:	Ç	Ч	- CONSTANT USED TO SPECIFY THE VALUE OF V	MAIN	300	
31:	С	UGR	- VELOCITY AT UPOUND LEVEL	MAIN	310	
32:	С	A K	- RATE OF REACTION	MATI.	320	
33:	C	ISTB	- STABILITY CLASS (1 VERY UNSTABLE, 6 VERY STABLE)	MAIN	3 .0	
34:	С	INVRS	- 1 OP O, IF 1 THERE IS INVERSION AT 7=H; IF O THERE	ISMAIN	740	
35:	C		NO INVERSION	MAIN	350	
PAUE	2					
------	---					
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36:	С	NSRCS - NUMPLE OF SOURCES	MAIN	300
37:	С	KS(I),LS(I)- PUSITION OF ITH. SOURCE IM THE Y AND Z DIRECTIONS	MAIN	370
38:	С	CO(I) - CONCENTRATION AT ITH. SOURCE	MAIN	350
37:	С		MAIM	3 10
40:	C.		MAIN	400
41:		FXTERNAL FCT, DUTP	MAIN	410
42:		DIMENSION PR'T(5),Y1(700),Y2(700),Y3(700),DEPY1(700),FERY2(700),	MAIN	420
43:		10ERY3(700),AUX(9,700),FA(15),FP(15),FC(15),RUNT(15),VEC(15),	MAIN	430
44:		2X7(15),AKY(15),AKZ(15),DAK7(15),COEFF(4),DKN(4),TDFKN(4)	MAIN	440
45:		DEFINE FILE 2(2000,225,0,IIPNT)	MAIN	450
46:		[FFI HE FILE 3(200,225,U,IPDI)	MAIN	460
47:		DEFINE FILE 4(200,225,U,IPOINT)	MAIN	470
48:		COMMON /PLK1/ 41(15,15),A2(15,15),A3(15,15),B1(15,15),B2(15,15),	MAIN	480
49:		1:3(15,15),R1(15),R3(15),R5(15),P6(15),R361(15),R362(15),APA1(15),	MAIM	490
50:		2?36I(15,15),0(700),U(15),V(1)),XXX(3,15),XYY(15),XZ7(15),XMAX(3),	MΔ Ι.1	500
51:		3YMAX,H,AK,TCH,P,KS(3),LS(_),CO(3),NX,NY,M7,N1,M2,N3,KX,KY,K7,	ΜΔΙΝ	510
52:		4V1(15),W2(15),W3(15),WSRCS,IFLU,MFLJ,CC(3,15,15,15),PFDEL,PRMT3,	MAIN	520
53:		5 I N VR S	MAIN	530
54:	C		MAIN	540
55:	ί		MAIN	550
56:	С	READ AND WRITE IMPUT DATA	ΜΔΙΝ	561)
57:	С		MAIN	570
58:	L		MAIN	ና አብ
59:		DATA CREFK/2940.,540.,258.,90./	MAIN	540
60:		DATA DKN/125.,125.,100.,75./	MVI 1	600
61:		wRITE(6,205)	MAIN	610
62:		WRITH(6,206)	MAIM	620
03:		READ(5,100) TINIT, ENDS, PRMT(2), PMMT(4)	MAIN	€ 30
64:		NPITH(6,200) TINIT,ENDS,PPMT(3),PPMT(4)	MAIN	641)
65:		READ(5,109) NSRCS	NVIV	559
66:		MRITE(6,209) NSPCS	MA I 1	660
67:		PEAU(5,106) PRDEL	MAIN	670
68:		WRITH(6,208) PRDEL	MAIN	6 71)
69:		READ(5,101) NX,NY,NZ	MAIN	670
70:		NRITE(6,201) NX, NY, NZ	MAIN	700
71:		PEAD(5,102) YMAX,H	MAIN	710

		PAUE
72:	WRITE(4,202) YMAX,H	MAIN 720
73:	NU 89 IEL,NSRUS	MAIN 730
74:	PFAD(5,1C4) KS(1),LS(1),CU(1),XMAX(1)	MAIN 740
/5: 7/ -	85 WRITE(6,204) KS(I), LS(I), UN(I), XMAX(I)	MAIN 7.0
76:	KEAD(5,105) ISTB, 19VRS, 4LPHA, AK	
11:	WFIF(E,207) = ISFS, INVRS, ALPHA, AK	MAIN 770
79:	PEAD(5, 193) UST, UGR, AM, TCH, P	MAIN 780
79:	W^{2} IF (6,203) USE, USE, M, ICH, P	MA1*1 (70
90:		MAIN 800
81:		MAIN PIO
82:	C INITIAL COMPTTIONS	MAIN P20
83:		MAIN 830
84:		MAIN 840
85:	(101M = (NX + 1) * (Y * V)	MAIN RJO
86:	PRMT3 = PRTT(3)	MAIN 800
87:	00.51 I = 1, NDIM	MAIN P70
98:	$(\mathbf{U}(\mathbf{I}) = \mathbf{O})$	MAIN 830
89:	Y1(I) = 0.	MAIN 830
90:	$Y_{2}(I) = C$.	MAIN 900
91:	51 Y3(I)=0.	MAIN 910
92 :	$S \cup N = 0$.	MAIN 920
93:	K K = I I I M - I	MAIN 910
94:	.)∩ 52 I=1,KK	MAIN 940
95:	DERY1(I)=1./FLOAT(NDIM)	MAIN 950
96:	52 SUM=SUM+DERY1(I)	MAIN 900
· 97:	DEFY1(NDIM)=1SUM	MAIN 970
98:	PO = 86 I = 1, ND IM	MAIL 980
99:	UFPY2(I)=UERY1(I)	MAIN 330
100:	$86 \cup FRY_{3}(I) = DERY1(I)$	MAINIOCO
101:	1 = N X + 2	MAIN1010
102:	N2 = NY + 2	MAIN1020
103:	N3=NZ+2	MAIN1030
104:	$K \times = N \times + 1$	MAIN1040
105:	k X = M X + 1	MAIN1050
106:	k7=17+1	MAINIOCO
167:	I(1 - 5) = I = 1, MSPCS	MAIN1070

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108:		1.03 ± 1.01	MA IN1020
109:		111 53 K=2,KY	MAIN1090
110:		UN 53 L=1,N3	MAINIIOU
111:		53 (C(I,J,K,L)=().	MAIN1110
112:	С		MAIN1120
113:	С		MAIN1130
114:	С	CALCULATION OF ORTHOGONAL POINTS, CUADRATURE VEIGHTS, AND	MAIN1140
115:	С	"ATRICES A AND B	MAIN1150
116:	C		MAIN1150
117:	Ċ		MAIN1170
118:	Ĉ	X DIRECTION	MAIN1140
119:	Ċ		MAIN1130
120:	Ċ		MAT.11200
121:		(AII) JCPI (NX $(1 \cdot 1 \cdot 1 \cdot 0 \cdot 0 \cdot EA \cdot EB \cdot EU \cdot ROOT)$	MAI11210
122:		10^{2} 1^{-1} 1^{-1}	MAIN1220
123:		$X \times (1 \cdot I) = P \cap \Omega T (I) * X^{\prime} \Delta X (I)$	MAINIZIO
124:		IE(NSPCS, EQ.1) OD TO 20	MAIN1240
125:		$X \times X (2 \cdot I) = R ((0 T (I) * X') \times X (2) + X \wedge \Delta X (1))$	ΔIN1250
126:		IE(NSRCS-E0.2) GO TO 20	MAIN1260
127:		$X \times (3, 1) = P \cap (1) \neq X \setminus (3) + X \cap (2)$	MAIN1270
129:		20 CONTINUE	MAIN 1280
129:		$\Theta = 60 = 1 + 1$	MAIN1220
130:			MAINIZIO
131:		$(C - 70) K = 1 \cdot N^{-1}$	MATH1310
132:		$7(1) \wedge 1(1 + K) = JE((K))$	M& IN1 520
1 3 4 :		$CALL UEOPR(NX \cdot 1 \cdot 1 \cdot 1 \cdot 1 \cdot 2 \cdot EA \cdot EP \cdot EC \cdot POOT \cdot VEC)$	MATVIZA
134:		00.80 K=1.11	MAIN 1340
135:		80 (J1(I-K)=VEC(K))	MATNERSO
1 16:		60 CONTINUE	MAIN1360
137:		(AII - DEOPR(NX, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1,	MA 1111 470
138:	ſ		MAIN13/0
1.9:	ŕ		MAINISON
140:	ř	Y DIPECTION	NATN1400
141:	ř		MATN1410
147:	ſ		MATE 1420
143:	0	LALL JCJI(NY,1,1,0,,0,,FA,FB,FC,ROOT)	MAIN1430

PACE

144 •		100.21 I=1.N2	MATNIAAO
145:		$21 \times Y \times (1) = R \cup C \times (1) \times 2 \times Y = Y = X \times M \wedge Y = Y = M \wedge Y$	MAINIASO
146:		10.61 1=1.N2	MAIN1460
147:			MAIN1470
148•		$\frac{1}{10} 71 \mathbf{K} = 1 \mathbf{N} = 2$	
140+		$\frac{1}{1} \frac{1}{1} \frac{1}$	
150+			MA IN1500
151.		UNEL DEURNY (11919191929EAGER 9EU9EU) EUEED 1 () 91 V-1 NO	
121.		(1) OI = 1 + NZ	MAINICOO
1.724		O[L(2(1)N) = VL(1N)	MAINISZU
153:			MAINIS 30
154:	~	CALL DENPR("IY, I, I, I, S, FA, FL', FU, R IUI, WZ)	MAINI540
155:	Ċ		MA1*11550
156:	C		MAIN1560
157:	С	7 DIRECTION	MA 1111570
158:	С		MAIN1530
159:	С		MAIN1590
150:		CALL JCPI(NZ,1,1,0.,0.,FA,FB,FC,ROOT)	MAIN1600
161:		LO 92 I=1,N3	MAIN1610
162:		$\frac{1}{2}$ $\frac{1}{1}$ = ROOT(I)	MAIN 1620
163:		DI = 22 I = 1, H3	MAIN1630
164:		22 ×Z7(I)=R()OT(I)*H	MAIN1640
165:		10 62 I=1,N3	MAIN 1650
165:		UALL DEOPR(N7,1,1,1,1,EA,EP,EC,ROCT,VEC)	MA IN 1650
167:		UD 72 K=1,N3	MAIN1670
169:		72 $13(I,K) = VEC(K)$	MAIN1630
169:		CALL DEOPR(NZ+1+1+1+2+EA+EB+EL+POPT+VEC)	MA IN 1690
170:		NO 82 K=1.N3	MAI11700
171:		82 + 3(1 - K) = VFC(K)	MAL 1710
172:			MA IN 1720
173:			MAIN1730
174 •	r	CHEE DIGULARY TYTYTYJY AFTIYGYRCHINI TY	
175.	r C		
174.	r	CALCHEATTUN OF EXDRESSIONS USED IN FOUR	
177.	Č	CALCOLATION OF TARKESSIONS USED IN THIEL	
170+	С С		
1725	C		MA191780
1/7:		1 + (1 + 1) + (1 + 5) = 50 + 10 = 10	MA 101770

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180:	U(1) = UGP	MAIN1800
181:	TDEKN(ISTH)=DKN(ISTB)/H	MA I'11810
182:	TI'SKN=1100./H	MAIN1820
103:	HO 2 L=2,13	NAIN1830
184:	IF(X2(L)-TDFK')(ISTR)) 11,12,12	MAIN1P40
135:	11 AKZ(F)=COEFK(ISTR)*X7(L)/TDFKN(ISTR)+(O.	MAINIRFO
186:	HAKZ(L)=COSEK(ISTE)/HKM(ISTE)	MAINIREO
187:	U(L)=UST*(XZ(L)*H/DKN(ISTB))** 4M	MAIN1970
188:	JO TO 16	MAINIBRO
189:	12 IF(X7(L)-TUSKN) 13,13,14	MAI 118 90
190:	13 AK/(L)=CCEFK(ISTB)+60.	MAIN1900
191:	$I \Lambda K Z (L) = 0$.	MAIN1910
192:	JA TA 15	MAI11920
193:	14 AKZ(L)=CHEFK(IST*)*H*(1XZ(L))/109.+60.	MA IN 1930
194:	DAKZ(L)=-COFFK(ISTB)/100.	MA IN1940
195:	15 U(L)=UST	MAIN19FU
196:	16 $AKY(L) = ALPHA*(COEFF(ISTP)+60.)$	MAIN1950
197:	2 LUNTINUE	MAIN1970
198:	UN TO 50	MAIN1980
197:	10 IF(ISTR.FQ.6) GD TH 40	0691141930
200:	10 3 L=2,113	MA 112000
201:	$4KZ(L)=\epsilon()$.	MAIN2010
202:	U(L) = UST	MAIN2020
203:	DAKZ(L) = 0.	MAIN20JO
204:	3 AKY(L)=ALPHA*AKZ(L)	MAIN2040
2 05 :	JO TO 50	MAIN2050
206:	40 + P + 4 = 2,13	MAIN2060
207:	AK7(L)=30.	MAIN2070
208:	U(L)=UST	MAIN2080
209:	$D \wedge KZ(L) = 0$.	MAIM2000
210:	4 "KY(L)=ALPHA*AK7(L)	MA IN2100
211:	30 CONTINUE	MAIN2110
212:	LO 41 L=2,KZ	MAIN2120
213:	R3(L)=DAKZ(L)/H**2	0c1211AM
· 214:	<pre><pre><pre><pre><pre><pre><pre><pre></pre></pre></pre></pre></pre></pre></pre></pre>	MAIN2140
215:	41 ?/(L)=AKZ(L)/H**2	MAI 12150

216:		UN 30 L=2,KZ	MAIN2160
217:		<pre><361(L)=K3(L)*A3(L,1)+R6(L)*F3(L,1)</pre>	MAIN7170
218:		30 × 362(L)=23(L)*43(L,N3)+R6(L)*83(L,N3)	MATH2130
219:)FN=43(N3,1)*A3(1,N3)-A3(1,1)*A3(N3,N3)	MAIN2190
220:		10.31 I = 2, KZ	MAIN2200
221:		31 APA1(I)=(A3(1,1)*A3(N3,I)-A3(N3,1)*A3(1,I))/DEN	MAIN2210
222:		1032 L=2, K7	MAIN2220
223:		$100 \ 32 \ I=2, KZ$	MA I 12230
224:		32 R 36I(L,I)=R3(L)*A3(L,I)+R6(L)*P3(L,I)	MA 1N2240
225:	Ç		MAIN22JO
226:	С		MAIN22GO
227:	С	WRITE INITIAL CONDITIONS	MAIN2270
229:	С		MAIN2220
229:	С		MAIN2230
230:		MFLS=1	MAIN2300
231:		IFLS=0	MA 1N 2310
232:		PRMT(1)=().	MA IN2320
233:		PPMT(2) = 0.	MA I N 2 3 3 0
234:		CALL PUTP(G.,Y1,DERY1,O,MUIM,PRMT)	MA IN 2340
235:	С		MA IN2350
236:	С		MAIN2360
237:	С	INTEGRATION USING RKCS	MA 112370
238:	С		MAIN2380
239:	C		MAIN2390
240:		MFLG=2	MA I M24 00
241:		PRMT(1) = TINIT	MA IN 2410
242:		PRMT(2)=[IIDS	MA I N2420
243:		IFLG=1	MA I 12430
244:		LALL RKGS(PRMT,Y1,DERY1,NDIM,IHLF,FCT,OUTP,AUX)	MA 1N2440
245:		$IF(NSRCS \cdot EQ \cdot 1) GO TO 59$	MA 1112450
246:		IFLG=2	MAIN2450
247:		CALL RKUS(PRMT,Y2,DERY2,MDIM,INLF,FUT,OUTP,AUX)	MAIN2470
248:		IF(NSRCS.E(.2) G0 TO 59	MAIN2480
249:		IFL G= 3	MAIN24 70
250:		CALL PKSJ(PRMT,Y3, DERY3, NDIM, IHLF, FCT, OUTP, AUX)	MAI 12500
251:		39 CONTINUE	MA 112510

252:	С	MAIN 2520
253:	100 FORMAT(4F10.5)	MAI 12530
254:	101 FORMAT(315)	MAIN2540
255:	102 FORMAT(2F15.7)	MAIN2530
256:	103 FARMAT(5F10.3)	MAIN2560
257:	104 FORMAT(21),2F15.4)	MAI.12570
258:	105 FORMAT(212,F10.2,L12.3)	MAI 125 10
259:	105 FORMAT(F10.4)	MAIN25 10
260:	109 FOPMAT(I3)	MAIN2600
261:	200 FORMAT(5x, TINIT = ", F5.1, 5x, "EMDS = ", F6.1, " MIN", 5x, "PRMT(3)	= ", MAIN261)
262:	$1F_{0}, 2, !$ MIN', 5X, ! PPMT(4) = !, F8.5, /)	MA 1112620
263:	201 FORMAT(5X, MX = ", I3, 5X, MY = ", I3, 5X, M7 = ", I3, /)	MA1112630
264:	202 EDRMAT(5X, 'YMAX =', E7.1, ' M', 5X, 'H =', E8.2, ' M', /)	MAI 112640
265:	203 FORMAT(5X, 'UST =', F8.2, ' M/MIN', 5X, 'USK ='. F9.2, ' M/MIN', 5X, '	A" = MAIN265)
256:	1, F7. 3, 5X, 'TCH = ', F3. 1, ' MIN', 3X, 'P = ', F6. 2)	MA [12660
267:	204 FORMAT(5K, 1KS = 1, 13, 3X, 1 LS = 1, 13, 2X, 1 CO = 1, F9.2, 1 MJ/CU.M1	,3X, MAI 12670
268:	$1^{*} \times MAX = *, F \rightarrow 1, * M^{*}, /)$	"AIN2630
269:	205 EARMAT(1)(/), 55X, "INPUT HATA")	MAI12640
270:	206 FNPMIT(35X, ***** *****, //)	MAIN2700
271:	207 FORMAT(5*, STADILITY CLASS = ', 14, 5*, ' I IVPS = ', 13, 5*, ' ALPHA	= ', F7MA 112710
272:	1.7.5K, 4K = ",E17.2, 1/MIN",/)	MA1112720
273:	208 FORMAT(5X, PRUEL = ', F6.2, ' MI (',/)	MAIN2730
274:	207 FORMAT(5x, MUMBEP OF SOURCES = M, 14, /)	MA 112740
275:	STOP	MAIN2750
276:	FND	MAIN2760

1:		SUBRJUTIIE FCT(X,Y,DERY)	FCT	10
2:	С		FCT	20
3:	C		FCT	517
4:	し	THIS SUBROUTINE COMPUTES THE DERIVATIVES(RICHT HAND SIDES) OF THE	FCT	40
5:	С	SYSTEM TO GIVEN VALUES OF X(TIME) AND Y(CONCENTRATION)	FCT	ちい
6:	С		FCT	4 C
7:	6		FCT	70
8:		DIMENSION Y(700), DERY(700), PAC(700), C70(700), CZ1(700), CZ2(700),	FCT	30
1:		1CZI(700),CYK(700),R2(15),K52I(15,15,15),CUU(15,15)	FCT	÷0
10:		LOMMON /BLK1/ A1(15,15),42(15,15),43(15,15),F1(15,15),B2(15,15),	FCT	100/
11:		113(15,15),R1(15),R3(15),R5(15),R6(15),R361(17),R362(15),APA1(15),	FCT	110
12:		2R36I(15,15), 0(700), U(15), V(15), X*X(3,15), XYY(15), XZ7(15), XMAX(3),	FCT	120
13:		3YMAX,H,AK,TCH,P,KS(3),LS(3),CO(3),NX, JY,NZ,N1,N2,N3,KX,KY,KZ,	FCT	130
14:		4v1(1)),W2(15),W3(15),NSRCS,IFLC,"FLJ,CC(3,15,15,15),PRDEL,PRMT3,	FCT	140
15:		SINVRS	FCT	150
16:		INTESER VAR1, VAR2, VAR3, VAR4, VAR5, VAR6, VAR7	FCT	130
17:		, 11) I M= (NX+1) * 1 Y*1 7	FCT	170
18:		10.4 L = 2.4 K T	FCT	150
19:		4 R1(L)=U(L)/YMAX(IFLG)	FCT	1 70
20:	С		FCT	200
21:	С	VARIATION IN THE V VELOCITY	FCT	210
22:	С		FCT	220
23:		IF(X-T(H) 1,7,2	FCT	230
24:		1 CONTINUL	FCT	240
25:		110 5 L=2, KZ	FCT	250
26:		V(L)=P*U(L)*(X/TCH)	FCT	ر) ں 2
27:		5 R2(L)=V(L)/(2.*YMNX)	FCT	270
28:		UNITO 3	FCT	230
27:		2 UENTINUE	FCT	2 70
30:		10 6 L = 2, KZ	FCT	300
31:		V(L)=P*U(L)	FCT	310
32:		6 K2(L)=V(L)/(2.*YMAX)	FCT	320
33:		3 CUNTINUE	FCT	350
34:	C		FCT	340
35:	С	COMPUTATION OF THE DERIVATIVES	FCT	350

36:	С			FCT	360
37:	-		110 - 33 L = 2, KZ	FCT	370
32:			10 33 K = 2, KY	FCT	380)
39:			11(1) 33 I = 2.KY	FCT	310
4():		33	R52I(L,K,I)=R5(L)*P2(K,I)-R2(L)*A2(K,I)	FCT	400
41:			00.39 LL = 1, NDIM	FCT	410
47:			(70(LL)=0.	FCT	42)
43:			LZI(LL) = J.	FCT	430
44:			$L_{2}(LL) = 0$.	FCT	441)
45:		37	CYK(LL)=0.	FCT	450
46:			J = 1	FC T	460
47:			J J = K X	FCT	47()
48:			IF(IFLG.EQ.1) JD TO 35	FCT	480
47:			IX=IFIX(X/(PRMT301))	FLT	4 70
50:			II「K=(IFLG-2)*97+(IX+1)	FCT	500
51:			<pre>PEAD(4'IIBK) ((CCU(K,L),K=2,KY),L=1,13)</pre>	FCT	510
52:		35	CONTINUE	FCT	520
53:			0 51 LK=2,KY	FCT	537
54:			00 51 LZ=2,K7	FCT	540
55:			un 50 LJ=J,JJ	FCT	5.0
56:			$\sqrt{4}$ R1=KS(IFL G -1)	FCT	560
57:			VAR2=LS(IFLG-1)	FCT	570
58:			√ AP 3=L J - (J - 1) + 1	FCT	500
59:			IF(IFL3.EQ.1) 30 TO 40	FCT	F 7()
ა0:			IF(LK.EQ.KS(IFLG).AND.LZ.EU.LS(IFLG)) GO TU 10	FCT	600
61:			PAC(LJ)=0.	FCT	619
62:			IF(CCC(V\R1,VAR2).5T.O.) PAC(LJ)=CCC(LK,L7)*A1(VAP3,1)	FCT	620
63:			SO TO 20	FCT	6.20
64:		10	PAC(LJ)=CO(IFLG)*A1(VAR3,1)	FCT	640
05:			IF(CCC(VAR1,VAR2).GT.O.) PAC(LJ)=PAC(LJ)+CUC(LK,L7)*A1(VAR3,1)	FCT	650
66:			SO TO 20	FCT	600
67:		40	IF(LK.EQ.KS(1).AND.L7.FQ.LS(1)) GO TO 11	FCT	670
:83			PAC(LJ)=0.	FCT	(S.I)
69:			UN TA 20	FCT	06.0
70:		11	PAC(LJ)=CO(1)*A1(VAP3,1)	FCT	700
71:		20	CUNTINUE	FCT	710

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72:	1.0 60 I=1,KX	T 720
73:	۷۸۹4=I+1 FC	T 730
74:	VAR5=I+J-1 FC	T 740
75:	60 PAC(LJ)=PAC(LJ)+A1(VAR3,VAP4)*Y(VAR5) FC	(הכ T
75:	UC 70 I=2,KZ FC	T 760
77:	VAR6=LJ+KX*(I-LZ) FL	T 770
78:	LZ2(LJ)=LZ2(LJ)+APA1(I)*Y(VAP6)*FLOAT(INVPS) FC	T 750
79:	L7()(LJ)=CZO(LJ)+A3(1,I)*Y(VAP6) FC	T 790
80:	70 $UZI(LJ)=UZI(LJ)+R30I(LZ,I)*Y(VAPG)$ FU	T 800
81:	CZ1(LJ)=(-A3(1,N3)*C72(LJ)-C70(LJ))/A3(1,1) FC	T 210
82:	UD 80 I=2,KY FC	T 820
83:	VAR7=LJ+(I-LK)*(K/-1)*KX FC	T 830
84:	PO + YK(LJ) =CYK(LJ) + <>2I(LZ,LK,I) *Y(√AR7) FC	T 840
85:	50 DERY(LJ)=-R1(L7)*PAC(LJ)+<361(L7)*CZ1(LJ)+R362(LZ)*CZ2(LJ)+CZI(LJ)FC	r 850
86:	1+CYK(lj)+C(lj)-AK*Y(lj) FC	T 860
87:	J=JJ+1 FC	T 870
88:	JJ = J + KX - 1FC	r 870
89:	51 CUNTINUE FC	0 F R
90:	RETURN FC	T 900
? 1:	FND FC	F 910

1:		SUBROUTI'F JCBI(N,NO,M1,AL,PE,FA,FP,FC,KOOT)	JCII	10
2:	C		JCPI	20
3:	С		JCFI	30
4:	С	THIS SUBROUTINE COMPUTES THE ROOTS OF AN NTH-DEGREE JACOBI POLYNOMIAL	JCBI	4()
5:	ſ,	AND THE DERIVATIVES OF THE POLYNOMIAL AT THESE POINTS	JCLI	\cdot
5:	С		JC ^ I	13
7:	С		JCHI	70
8:	ί	I - THE NUMBER OF INTERIOR POINTS	JCHI	60
₹:	С	MO - 1 OR O IF X=O IS INCLUDED OR MOT	JCPI	- 30
10:	С	- SAME AS MO, FOR THE POINT X=1	JCPI	100
11:	С	AL,BC - THE QUANTITIES ALEA AND BETA FOR THE WEIGHTING	JCHI	110
12:	C	FUNCTION	JCFI	120
13:	С	FA,FC,FC - THE FIRST,SECOND AND THIRD DERIVATIVES OF THE	JCPI	130
14:	C .	POLY JUMIAL AT THE ROOTS	JCBI	140
15:	С	2001 - OUTPUT ARRAY CONTAINING THE NENOENI ROOTS OF THE	JULI	150
16:	С	POLY IONIAL	JCET	150
17:	С		JCBI	170
19:	С		JCHI	190
19:		DIMENSION FA(1), FU(1), FC(1), ROOT(1)	JCBI	1 +0
20:		4P=AL+PE	JCHI	200
21:		1D=8F-AL	JCPI	210
22:		10=B=*AL	JCFI	220
23:		FA(1) = (AD/(AB+2.)+1.)/2.	JCBI	230
24:		FP(1)=0.	JCII	240
25:		10 I = 2, 4	JCCI	2.50
26:		/1=FLOAT(I)-1.	JCPT	250
27:		<i>I</i> = AB + 2 • *71	JCBI	270
28:		F^(I)=(AH/Z*AU/(7+2.)+1.)/2.	JCLI	280
29:		IF(I-2) 11,12,11	JCCI	217
30:	12	FF(I)=(Ar+AP+21)/2/2/(7+1.)	JCBI	300
31:		,0TO 10	JCPI	310
32:	11	L=Z*L	JCPI	320
33:		$Y = Z 1 * (\Lambda^{n} + Z 1)$	JCPI	330
34:		A=A*(7b+A)	JCPI	340
35:		+P(I) = Y/7/(Z-1)	JCHI	350

36:	10	LONTINUE	JCHI 360
37:		$\chi = ()$	JC * I 370
38:		LO 20 I=1,N	JCPI 380
39:		I = I - I	JCFE IHJC
4():	25	x+)=().	JCFI 400
41:		X:1=1.	JCBI 410
42:		X())]=().	JUPI 420
43:		$\times M1 = 0$.	JCPI 410
44:		10 30 J=1,N	JCRI 440
45:		<p=(fa(j)-x)*xn-fn(j)*xn< td=""><td>JCBI 450</td></p=(fa(j)-x)*xn-fn(j)*xn<>	JCBI 450
46:		XP1=(FA(J)-X)*XN1-FB(J)*XD1-XN	JC3I 460
47:		$\mathbf{x}(\mathbf{t}) = \mathbf{X} \mathbf{A}$	JCP1 470
49:		XU1=XN1	JCHI 440
49:		X, 1 = X δ	JCEI 490
50:	30	× M1 = X P1	JCHI 500
51:		<u>ZC = 1.</u>	JCBI 510
52:		$\underline{7} = X \cdot J \times N \cdot 1$	JCHI 520
53:		IF(I1) 21,21,22	JCじI 530
54:	22	()(123 J=1,I1	JCBI 540
55:	23	ZC = Z(-Z/(X-R)OT(J))	JCHI 550
56:	21	Z=Z/7C	JCPI 560
57:		$\lambda = X - Z$	JCBI 570
58:		IF(AdS(7)-1.E-7) 26,26,25	JCHI 530
59:	26	RCOT(I)=K	JCHI 590
60:		X = X + Λ • () () U հ	JCFI ECO
51:	20	CONTINUE	JUBI E10
62:		r T = N+NO+11	JC,I 050
53:		IF(NO-1) 35,36,35	JCE1 630
64:	36	1 () 42 I=1,N	JCBI 640
65:		$\mathbf{J} = \mathbf{N} + 1 - \mathbf{I}$	JCDI 600
66:	42	PO(J + 1) = ROOT(J)	JCHI 660
67:		< O O T (1) = 0.	JCBI 670
68:	35	IF(N1-1) 38,37,38	JCEI 680
67:	37	$R(D) \top (NT) = 1 $	JCF3 IAD
70:	38	0(° 40 I=1,NT	JCPI 7CO
71:		X=PONT(I)	JCPI 710

PACE 2

			PACE	3
12:		FA(I)=1.	JCBI	72)
73:		F - (I) = O.	JCPI	730
74:		+C(I)=0.	JCF I	74)
75:		$I \cap 4(J=1, NT)$	JCHI	750
76:		IF(J-I) 41,40,41	JCHI	700
77:	41	Y = X - ROOT(J)	JCPI	770
78:		FC(I)=Y*FC(I)+3.*FE(I)	JCFI	720
79:		FP(I)=Y*F8(I)+2.*FA(I)	JCFI	790
80:		⊢A(I)=Y*FA(I)	JC1 I	800
81:	4()	CONTINUE	JC t I	810
82:		RETUPN .	JCFI	820
83:		END	JC ? I	P 30

1:		SUFROUTINE DEOPR(J,NO,M1,I,I),F4,E8,EC,RODI,VECT)	DEOP	10
2:	С		DEOP	20
3:	С		UFCP	50
4:	ί	THIS SUFROUTINE CALCULATES THE A AND P NATPICES AND THE QUAJRATURE	DEDP	4)
5 :	С	WEIGHTS FROM THE QUANTITIES DERIVED IN SUPPOUTINE JCHI	DECP	50
6:	C,		DELIP	40
7:	С		DECP	70
8:	С	N,NO,NI – SAME AS JCPI	PEOP	40
7:	С	I - DIFFERENTIAL OPERATOR AT X=ROOT(I)	NENP	7 0
10:	С	ID - INDICATOR, I.F. 1 FOR MATRIX A, 2 FOR MATRIX R,	UFOP	100
11:	С	3 FOR QUADRATURE WEIGHTS W	ĽEPP	110
12:	С	FA,FP,FC - COMPUTED IN JCBI	DEUP	127
13:	С	POGT - CUMPUTED IN JUBI	DEUD	130
14:	С	VECT - CONTAINS THE COMPUTED VECTOR	DEDP	140
15:	С		DECP	1.0
16:	С		DEOP	150
17:		<pre>UIMENSIO FA(1), FP(1), FC(1), ROOT(1), VCCT(1)</pre>	DENP	170
18:		4T = N + NO + NI	DEUP	189
17:		1F(1) - 3 1,10,10	['F()P	190
20:	1	$100 \ 200 \ J=1, 11T$	DEOP	200
21:		IF(J-I) 21,2,21	() F () P	210
22:	2	IF(IU-1) 5,4,5)EU.b	220
23:	4	$VECT(I) = EE(I) / EA(I) / 2 \cdot$	UFIJP	230
24:		UDTO 20	DFOP	241)
25:	5	VFCT(I)=FC(I)/FA(I)/3.	DEUP	2.J
26:		GPT0 20	DEOP	2:0
27:	21	Y=R0)T(I)-R())T(J)	DECIP	270
23:		VECT(J) = FA(I)/FA(J)/Y	DECP	200
29:		IF(IU-2) 20,22,20)F(P	2 70
30:	22	VFCT(J)=VEUT(J)*(FB(I)/FA(I)-2./Y)	DECP	300
31:	20	CONTINUE	DECP	310
32:		JOTO 50	DEOP	320
33:	10	$Y = O_{\bullet}$	DENP	330
34:		IF(ILI-4) 31,30,31	DELIP	340
35:	31	uD 25 J=1,NT	1) F Li b	350

36:		X=RJUT(J)	11 FCP 360
37:		1 X = X * (] • -	DEDP 370
38:		IF(NO) 24,27,26	DEOP 330
39:	27	$\Delta X = \Delta X / X / X$	UE()P 300
40:	26	IF(N1) 28,27,23	DF()P 460
41:	2.3	n X = A X / (1 - X) / (1 - X)	0F0P 410
42:	28	<pre>/+CT(J)=^X/F^(J)/F^(J)</pre>	UFOP 420
43:	25	Y = Y + V ECT(J)	NECP 44)
44:		COTO 50	UE0P 440
45:	30)n 35 J=1,NT	DENP 400
46:		X=ROAT(J)	UECP 460
47:		IF(NU) 36,37,36	UF0P 470
48:	37	$\Delta X = 1 \cdot / X$	UFCP 420
49:	36	IF(N1) 38,33,38	DECP 430
50:	37	<pre>/ X=1./(1X)</pre>	UEMP 560
51:	312	$V \perp C T(J) = \Delta X / F \Delta (J) / F \Delta (J)$	DEUP 517
52:	35	Y = Y + V ECT(J)	CFCP 520
53:	60	10.61 J=1, NT	DE0P 530
54:	61	$V \cap C T(J) = V \subseteq C T(J) / Y$	UEOP 540
55:	50	PETUR 1	0F0P 5.0
56:		E 10	DECP 560

1:	_	SUPROUTIJE OUTP(X,Y, HERY, IHLE, NDIN, PRMT)	ΠΗΤΡ	10
2:	C		NUTP	20
3:	C		NUTP	30
4:	C	THIS SUBROUTINE GIVES THE DUTPUT OF THE MODEL	CUTP	40
5:	(,		UUTP	50
- 6 :	C		OUTP	60
(:	C	PROEL - THE INTERVAL AT WHICH THE RESULTS WILL DE PRINTED	CUTP	70
8:	C	I OR THEFA - TIME PETWEEN THE POLLUTANT RELEASE(TIMIT) AND THE	PUTP	0 %
9:	C	INITIATION OF THE AVERAGING TIME	Ουτρ	+0
10:	С	DMEGA - END OF THE AVERAGING TIME	NUTP	100
11:	C	CX(I,J) - FLUX ACPOSS Y-Z PLANE AT Y' JTH. POINT DUE TO ITH.	OUTP	110
12:	С	AND PRECEDING SOURCES	NUTP	120
13:	С		OUTP	1+0
14:	Ç		CUTP	140
15:		DIMENSION PRMT(5),Y(700),DERY(700),CZC(700),CZ1(700),CZ2(700),	(UTP	160
15:		1(13, 15, 15), 0X(3, 15), IHL(3, 50)	NUTP	150
17:		COMMUN /PLK1/ A1(15,15), A2(15,15), A3(15,15), 21(15,15), 22(15,15),	CUTP	170
18:		1 + 3(15, 15), R1(15), R3(15), R5(15), R(15), R3(1(15), R3(2(15), APA1(15)), R3(2(15)), R3(2(15	NUTP	180
19:		2 ₹ 36I (15,15), J (700), U (15), V (15), X X (3,15), X Y Y (15), X Z Z (15), X M A X (3),	OUTP	140
20:		3YMAX,H,AK,TCH,P,KS(3),LS(3),CO(3),NX,YY,N7,N1,N2,N3,KX,KY,KZ,	CUTP	200
21:		4%1(15),W2(15),W3(15),NSPCS,IFLG,MFLG,CC(3,15,15,15),PRDEL,PRMT3,	OUTP	210
??:		SINVRS	OUTP	220
23:		INTEGER VARIO, VARII, VARI2, VARI3, VARI4, VARI5	DUTP	230
24:		IFLAG=IFLG	NUTP	240
25:		IF(MELS.EQ.1) 30 TO 15	UUTP	0ر 2
26:		IF(NSRCS.NE.1) GO TO 70	NUTP	260
27:		$IF(X \cdot FQ \cdot 0 \cdot 0) IIII = 0$	CUTP	270
28:		IF(X - EQ - 0 + 0) = GQ = TQ - 71	OUTP	280
29:		AI=FLOAT(IIII)	DUTP	200
3():		AAII = AAI - X/PPUEL	CUTP	300
31:		$I \land I = I \vdash I \land (\land \land I I \neq 1 \cup) .)$	NUTP	310
32:		IF(IA1.EV.0) 30 TO 71	ΠΗΤΡ	320
33:		GO TO 800	CIUTP	330
34:		70 IF(X.E0.0.0) II=0	NUTP	340
35:		IF(X,FQ,Q,Q) 30 T(1 71	OUTP	250

26.			
10.			
57. i			101P 379
38:		$11 = 1 + 1 \times (A + 1 + 1) (O_{\bullet})$	CUTP 380
39:		$IF(I1 \cdot EQ \cdot 0) = 50 + 71$	ULL 3 30
40:		GO TO 800	OUTP 400
41:	71	CONTINUE	CUTP 410
42:		J = 1	NUTP 420
43:		J]= k X	OUTP 430
44:		10 2 ID=1,11Y	CUTP 440
45:		10 1 LJ=J,JJ	CUT9 450
46:		LZO(LJ) = 0.	NUTP 460
47:		CZ2(LJ)=0.	OUTP 470
48:		LO 5 I=2,KZ	CUTP 480
47:		V \P1)=LJ+YX*(I-2)	OUTP 490
50:		UZ2(LJ)=UZ2(LJ)+APA1(I)*Y(VAR10)*FL(AT(I)VRS)	OUTP 500
51:	5	(70(1,1)=(70)(1,1)+33(1,1)*Y(VA(10))	CUT2 510
52:	-	(71(1J) = (-A3(1.N3) * C72(1J) - C70(1J))/(3(1.1))	CUTP 520
53:	1	LENTINUE	OUTP 530
54:	-	$J = T [J \neq N] T \neq K \neq T$	CHTP 540
55:		1 = 1 + KX - 1	OUTP 550
56:	2		
57:	2	IhD = 2	OUTP 570
58.		1 + 1 = 1	CHTP 580
5.2.			
. 0.		$\frac{1}{2} = \frac{1}{2} + \frac{1}$	
61.			
47.			
2.		$[11] \mathcal{L} = [1 - 1] \mathcal{L} = [1 - 1] \mathcal{L}$	
() •	1	$(J_{j} \land j \land j) = (L_{1} \land j)$	
041	21		
051			OUTP AND
56:		f'4 = M2	OUTP 650
67:		101 20 L=2, KZ	CUTP 670
68:		J=2	CUTP 680
67:		$11(1 \ 31 \ I = M3, M4$	NUTP 690
70:		C(J,K,L)=Y(I)	0UTP 700
71:	31	J = J + 1	CUTP 710

72:	113=14+1	OUTP 720
73:	► 4 = I JI)*KX	PUTP 730
74:	[N(i) = I N(i) + 1	OUTP 740
75:	20 CONTINUE	OUTP 7-0
7E:	J=2	CUTP 760
77:	L 1 22 I=M1, M2	OUTP 770
79:	C(J,K,N3) = CZ2(I)	OUTP 780
79:	22. $J = J + 1$	CUTP 770
80:	M 1 = M 3	NUTP ROU
81:	112=M4	OUTP 810
92:	10 CONTINUE	CUTP 820
83:	LALL ZERU(C)	CUTP 630
34:	IF(NSRCS.EQ.1) ON TO 65	UUTP 840
85:	IX = IFIX(X/(PRMT301))	CUTP R O
86:	IIUK = (IFLC-1)*97+(IX+1)	CUTP RCO
87:	WRITE(4'IIEK) ((C(N1,K,L),K=2,KY),L=1,N3)	CUTP P70
83:	I I = I I + 1	OUTP 800
99 :	$IF(X \cdot EQ \cdot () \cdot ()) III = 0$	CUTP 840
90:	IF(X.EQ.0.0) GO TU 15	0UTP 900
91 :	AII=CLOAT(III)	CUTP 910
J2:	AIII=AII-X/PRDFL	NUTP 920
93:	$III = I F I \times (A I I I \times 100)$.)	UNTH 830
) 4:	IF(II1.E(.0) 30 TO 65	NUTP 940
75:	GD TU 800	NUTP 950
96 :	65 LONTINUE	CUTP 960
∃7:	I I I = I I I + 1	OUTP 970
98:	IF(IFLG.FQ.1.AND. ISKCS.ED.1) IIII=IIIII+1	CUTP 920
44:	IIX=IFIX(X/(PROEL01))	CUTP 970
100:	IF(IFLG.EW.NSRCS) GO TO RO	NUTP1C63
101:	$(1 \ 9) \ J=2, N $	OUTP 1010
102:	[BLK=J+(IFLG-1)*4/*1F+II**15	CUTP1020
103:	<pre>>∩ WRITF(2'IILK) ((C(J,K,L),K=2,KY),L=1,V3)</pre>	OUTP1030
104:	IHL(IFLS,IIX+1)=IHLF	OUTP1040
105:	-0 TO 800	OUTP1050
106:	BO CALL AVG(C,IIX,IFLAG)	OUTP1060
107:	IF(IIX+F5+1+AND+N3RCS+E0+1) 30 TH 500	0UTP1070

109:	IF(IIX.E0.1) CU TU 74	0UTP1050
109:	10 67 J=2,N1	OUTP10-0
110:	10 67 K=2,KY	CUTP1100
111:	10 e7 L=1, N3	OUTP1110
112:	67 (C(IFLG,J,K,L)=C(J,K,L)	OUTP1120
113:	IF(X.FQ.0.0) 60 TO 800	CUTP1135
114:	$T = X - 2 \cdot * PP DEL$	CUTP1140
115:	IHL(IFLC,IIX+1)=IHLF	GUTP11 0
116:	IF(NSPCS+FQ+1) 60 TO 18	OUTP1100
117:	74 IIFLJ=IFLS	OUTP1170
118:	78 IIFLG-1	OUTP1180
119:	IF(IIFLG.E0.0) GO TO 18	OUTP11→0
120:	10.75 J=2.01	CUTP1200
121:	1°LI=J+(IIFL3-1)*4?*15+IIX*13	DUTP1210
122:	75 REAU(?'IPLI) ((C(J,K,L),K=2,KY),L=1,N3)	NUTP1220
123:	U^LL AVG(C,IIX,IIFLG)	CUTP1230
124:	IF(IIX.EQ.1.AND.IIFLG.EQ.1) GO TO 800	OUT21240
125:	(-7), J=2, N, 1	NUTP1250
126:	Hft = 7c K = 2, KY	UUTF1500
127:	10176 L=1, N3	CUTP1270
128:	76 CC(IIFLS,J,K,L)=C(J,K,L)	00TP1280
129:	G() T() 78	∩UTP12→0
130:	15 T=0.	CUTP1300
131:	I I X = ()	OUTP1410
132:	100.91 J=1,NSRCS	OUTP1320
133:	91 IHL(J ,1)=()	NUTP1330
134:	IF(MFLG.EC.1) GU TO 16	CUTP1340
135:	CALL AVG(C,IIX,IFLAG)	OUTP1350
136:		GUTP1360
137:	UN TO 800	CUTP1370
138:	18 CONTINUE	UTP1340
139:	VAR11 = KS(1)	DUTP1390
140:	VAP 12 = LS(1)	CUTP1400
141:	$C(1, 1, V \land R 11, V \land R 12) = CO(1)$	CUTP1410
142:	IF (NSRCS.EQ.1) OF TO 16	OUTP1420
143:	00.17 I=2, NSRCS	OUTP1430

144:		$\sqrt{AR} 1 = I - I$	QUTP1440
145:		V AR 14=K S(I)	PUTP1450
146:		VAR1F=LS(I)	NUTP1460
147:	17	CC(VAR13,N1,VAR14,VAR15)=CC(VAR13,J1,VAR14,VAR15)+CO(I)	OUTP1470
149:	16	wRITF(6,131) T,X,(IHL(J,IIX+1),J=1,NSRCS)	CUTP1480
149:		WRITL(6,204)	CUTP14 70
150:		WRITE(6,210) XXX(1,1),((XXX(J,I),I=2,M1),J=1,MSRCS)	OUTP1500
151:		ARITE(6,215)	CUTP1510
152:		WRITE(6,208)	CUTP1520
153:		L=[]3+]	CUTP1530
154:	53	L=L-1	OUTP1540
155:		WRITE(6,205) X72(L)	CUTP1550
156:		10 41 K=2,KY	NUTP1560
157:		wPITF(6,134) XYY(K),CC(1,1,K,L),((CC(I,J,K,L),J=2,N1),I=1,NSKCS)	OUTP1570
158:	41	CUNTINUE	QUTP1580
159:		IF(L.E0.1) GO TO 40	OUTP1590
160:		SO TO 53	OUTP1600
161:	40	CONTINUE	CUTP1010
162:		.10 50 I=1, NSRCS	CUTP1620
163:		10 50 J=1,11	OUTP1630
154:		$() \times \{ I, J \} = 0.$	OUTP1640
165:		DO 52 K=2,KY	CUTP16>0
166:		いつ ック L=1,N3	OUTP1660
167:	52	UX(I,J)=OX(I,J)+CU(I,J,K,L)*U(L)*w2(K)*w3(L)	OUTP1670
168:		QX(I,J)=(X(I,J)/60000.*H*2.*Y*AX	NUTP1630
169:	50	CONTINUE	AUTP1690
170:		WRITE(6,220) $OX(1,1),((OX(J,I),I=2,M1),J=1,NSE(S))$	NUTP17C0
171:	200 V	LONTINUE	OUTP1710
172:	C		CUTP1720
173:	1 + 1	FURMAT(8(/),30X, THETA = ', F6.2, MIN', /, 30X, ONEGA = ', F6.2, MIN'	, NUTP17JO
174:		110X, 'IHLF =', 3I3)	UUTP1740
175:	134	FORMAT(1X, *Y=*, F6.1, * M *, 15F8.2)	CUTP1750
176:	204	FORMAT(//,39X,"X DIRECTION (METELS)",/)	OUTP17r0
177:	205	FORMAT(//,5X,'Z DIRECTION =',F10.1,' M',//)	NUTP 1770
178:	203	+{}}^\/ (5\X, ************************************	CUTP17+0
177:	210	FORM (T(11x+F9+1+1x+14F9+1)	OUTP1790

180:	215 FORMAT(//,38X,'CONCENTRATION(MG/CU.M)')	OUTP1800
181:	220 FOPMAT(4(/),1X,"QX(GR/SEC)=",15EP.2)	OUTP 1810
182:	4FTUP1	OUTP1820
183:	END	GUTP1830

1:		SUBROUTINE ZERO(C)	ZERM	10
2:	ι		7 F F C	20
3:	Û		ZFPN	30
4:	С	THIS SUPROUTINE CLEANS THE OUTPUT OF THE MODEL	ZERN	4()
5:	C		ZEIN	эO
6:	C		7 E 12 ()	(()
7:		DIMENSION C(15,15,15)	ZERU	70
8:		COMMON / EK1/ A1(15,15), A2(15,15), A3(15,15), P1(10,15), P2(10,15),	ZERN	ч0
9:		163(15,15),K1(15),P3(15),P5(15),R>(15),R3(1(15),R362(15),APA1(15),	ZFRC	40
10:		2P36I(15,15),Q(700),U(15),V(15),XXX(3,15),XYY(15),XZZ(15),XMAX(3),	ZERN	100
11:		3YMAX,H,AK,TCH,P,KS(3),LS(3),CO(3),MX,MY,MZ,M1,M2,M3,KX,KY,KZ,	ZERO	110
12:		4wl(15),w2(15),W3(15),NSRCS,IFLG, 'FLG,CC(3,15,15,15),PDUEL,PP-4T3,	7 E ?12	120
13:		5 I VVRS	ZENU	130
14:		J=2	7⊧ぺ∩	140
15:		55 K=KS(IFL?)	ZEVD	150
16:		L=LS(IFLG)	ZヒK()	170
17:		IF(C(J,K,L).LF.O.) GO TO 10	7E?+1	170
18:		12 L=L+1	ZERCI	1 '' 0
19:		IF(L.GT.NJ) GO TO 14	ZFRC	120
20:		IF(C(J,K,L).LE.U.) OF TO 20	7FR()	200
21:		JO TO 12	ZERN	210
22:		14 L=LS(IFLC)	ZELU	220
23:		15 L=L-1	ZERO	230
24:		IF(L.LT.1) GO TO 30	ZEPH	240
25:		IF(C(J,K,L).LE.O.) SP TD 25	ZERO	250
26:		50 TO 15	ZFP()	260
27:		20 DO 21 I=L,N3	ZERN	270
28:		21 $C(J, K, I) = 0$.	ZERD	250
27:		GO TO 14	ZERO	210
30:		25 (J,K,L)=0.	ZERO	300
31:		L = L - 1	ZERN	310
32:		IF(L.EQ.0) GO TO 30	ZERO	3211
33:		CO TO 25	ZFRU	330
34:		30 L=LS(IFLU)	ZERN	340
35:		K=KS(IFL^)	ZEPN	3,0

•			PAG	E 2
36:	33	L=LS(IFL)	7 E 4	J 24()
57:		κ=κ+1	758	0 370
38:		IY D = 1	7EP	0 380
37:		IF(K.FQ.N2) CO TH 32	ZER	0 370
40:	52	IF(C(J,K,L).LE.0.) 30 TO 35	7[2	() 4()()
41:		UN TH 37	ZER	0 410
42:	35	L 7 35 I=1,13	ZER	<u>n 420</u>
4 5 :	35	U(J,K,I)=0.	ZER	0 430
44:		IF(IYD.FO.1) 60 TO 85	ZER	በ 44ባ
45:		CO TO 86	ZER	A 450
46:	85	UO 80 I=K,KY	ZFR	0 460
47:	60	L(J, I, L) = 0.	ZER	0 47J
48:		G TO 87	ZEK	በ 4ዖን
47:	86	k I ,11) = K	ZEK	0 470
50:	EB	<pre>((J,KIND,L)=0.</pre>	ZER	1 500
51:		IF(KIND.FQ.1) 30 TO 87	ZER	0 510
52:		$k I \downarrow I () = k I \downarrow I () - I$	ZER	0 520
53:		60 TH 88	ZEK	0 530
54:	87	CONTINUL	7ER	() 540
55:		IF(IYD.EC.1) CO TU 33	ZEF	N 550
56:		CO TO 51	ZES	n 560
57:	37	L=L+1	7 E R	0 570
58:		IF(L.GT.13) CO TO 38	ZER	0 590
59:		IF(C(J,K,L).LE.O.) 30 TO 39	ZEP	(+ 5)0
60:		J TO 37	ZEP	0 600
51:	39	UN 40 I=L,N3	ZER	0 610
62:	4()	$(\mathbf{J},\mathbf{K},\mathbf{I})=\mathbf{O}.$	ZER	n 620
63:	8 ز	L=LS(IFLG)	ZEH	n 630
64:	68	L=L-1	7 F R	በ 640
65:		IF(L.GE.1) GO TO 63	ZER	N 650
66:		IF(IYD.EQ.1) GO TO 33	ZFI	0 660
67:		IF(IYU.E(.2) GO TO 51	7 E R	n 670
68:	63	IF(C(J,K,L).LE.O.) OF TO 42	Z H K	FI 6 10
69:		30 TO 68	7 म स	0 690
70:	42	<pre>U(J,K,L)=().</pre>	7E ?	0 700
71:		L=L-1	7 F F	0 710

72: $1F(L, 0F, 1)$ 00 T0 42 $2FP0$ 72073: $1F(1YD, Ew, 1)$ 00 T0 33 $2FR0$ 74074: $1F(1YD, Ew, 2)$ 00 T0 51 $2FP0$ 74075: $32 L \pm LS(1FLw)$ $2FR0$ 75076: $k \pm KS(1FLw)$ $2FP0$ 74077: $51 L \pm LS(1FLw)$ $2FP0$ 74078: $k \pm K - 1$ $2FP0$ 74079: $1YD=2$ $2FP0$ 74080: $1F(K + EQ + 1)$ 60 T0 50 $2FP0$ 74081: GD T0 52 $2ER0$ 70082:50 J=J+1 $2EP0$ 74083: $1F(J, ST + N1)$ 50 T0 60 $2EP0$ 74084: GO T0 55 $2EP0$ 74085:10 (0 58 T = J + N1 $2EP0$ 74086:10 58 L = J + N1 $2EP0$ 74087: LO 59 L = 1 + N3 $2EP0$ 74089:59 U (1, K, L) = 0. $2EP0$ 87089:60 CONTINUL $2FP0$ 87090: $R ETURN$ $2FP0$ 87091: FND $2EP0$ 910				
73: $IF(IYD,ES,I)$ CO TO 33 $ZFRO 750$ 74: $IF(IYD,EO,2)$ GO TD 51 $ZFRO 750$ 75: 32 L=LS(IFLG) $ZERO 750$ 76: $K=KS(IFLG)$ $ZERO 750$ 77: 51 L=LS(IFLG) $ZERO 750$ 77: 51 L=LS(IFLG) $ZERO 750$ 79: $IYD=2$ $ZERO 750$ 80: $IF(K,EQ,I)$ GO TO 50 $ZERO 750$ 81: GO TO 52 $ZERO 750$ 82: 50 J=J+1 $ZERO 750$ 83: $IF(J,ST,MI)$ GO TO 60 $ZERO 750$ 84: GO TO 55 $ZERO 750$ 85: 10 (C 58 L=J,NI $ZERO 750$ 86: 10 58 K=2,KY $ZERO 750$ 87: $LO 58$ L=J,NI $ZERO 750$ 89: 59 $U(I,K,L)=0$ $ZERO 750$ 89: 60 CONTINUL $ZERO 750$ 90: $RETURN$ $ZERO 790$ 20: $ZERO 790$ $ZERO 790$ 89: 60 CONTINUL $ZERO 790$ 90: $RETURN$ $ZERO 790$ 91: END $ZERO 910$	72:		IF(L.CF.1) GO TO 42	ZER(1 720
74: IF(IYD.E0.2) GO TO 51 ZFPO 740 75: 32 L=LS(IFLG) ZERO 750 76: k=KS(IFLG) ZEPO 750 77: 51 L=LS(IFLG) ZEPO 750 77: 51 L=LS(IFLG) ZEPO 750 78: k=K-1 ZEPO 750 79: IYD=2 ZEPO 750 80: IF(K.EQ.1) GO TO 50 ZEPO 750 81: GO TO 52 ZEPO 720 82: 50 J=J+1 ZEPO 720 83: IF(J.GT.*1) GO TO 60 ZEPO 720 84: GO TO 55 ZEPO 720 85: 10 +O 58 I=J,N1 ZEPO 720 86: HO 58 K=2,KY ZEPO 740 86: HO 58 K=2,KY ZEPO 740 87: LO 58 L=1,N3 ZEPO 740 87: LO 58 L=1,N3 ZEPO 740 89: 60 CONTINUL ZEPO 870 89: 60 CONTINUL ZEPO 870 89: 60 CONTINUL ZEPO 90 90: RETURN ZERO 90 91: FND ZERO 910	73:		IF(IYD.E1) CO TO 33	ZER0 730
75: 32 L=LS(IFLG) ZER0 750 76: K=KS(IFLG) ZER0 750 77: 51 L=LS(IFLG) ZF10 770 78: K=K-1 760 760 79: IYD=2 ZER0 700 80: IF(K,EQ.1) 60 T0 50 ZER0 700 81: G0 T0 52 ZER0 700 82: 50 J=J+1 ZER0 700 83: IF(J.ST.M1) 60 T0 60 ZER0 700 84: G0 T0 55 ZER0 700 85: 10 (0 58 I=J,M1 ZER0 700 86: 00 58 K=2,KY ZER0 700 87: LO 58 L=1,N3 ZEF0 700 89: 60 CONTINUL ZEF0 700 90: RETURN ZEF0 900 91: FND ZER0 910	74:		IF(IYD.E0.2) GO TO 51	ZFRO 740
76: K=KS(IFLJ) ZEPO 7.00 77: 51 L=LS(IFLC) ZF0 770 78: K=K-1 ZEPO 700 79: IYD=2 ZEPO 700 80: IF(K.EQ.1) GD TD 50 ZERD 700 81: GD TD 52 ZERD 710 82: 50 J=J+1 ZERD 720 83: IF(J.ST.M1) GD TD 60 ZERD 720 84: GD TD 55 ZERD 740 85: 10 (D 58 I=J,N1 ZERD 740 86: HD 58 K=2,KY ZERD 740 87: LD 58 L=1,N3 ZERD 720 87: LD 58 L=1,N3 ZERD 720 87: D 58 L=1,N3 ZERD 720 87: LD 58 L=1,N3 ZERD 720 87: LD 58 L=1,N3 ZERD 720 89: 60 CONTINUL ZERD 720 89: 60 CONTINUL ZERD 720 90: RETURN ZERD 920 91: END ZERD 910	75:	32	L=LS(IFLJ)	ZERO 750
77: 51 $L=LS(IFLC)$ $ZF:0$ 770 $79:$ $K=K-1$ $7FR()$ $7FR()$ $7FR()$ $79:$ $IYD=2$ $ZER()$ 790 $80:$ $IF(K+EQ+1)$ GO $TI()$ 50 $91:$ GO TO 52 $ZER()$ $82:$ 50 $J=J+1$ $ZERO$ $P20$ $83:$ $IF(J+GT+M1)$ GO TO $ZERO$ $84:$ GO TO 56 $ZERO$ $84:$ GO TO 56 $ZERO$ $85:$ 10 CSR $I=J+M1$ $ZERO$ $86:$ 10 58 $I=J+M1$ $ZERO$ $86:$ 10 58 $K=2+KY$ $ZERO$ $87:$ LO 58 $L=1,N3$ $ZEYO$ $87:$ LO 58 $L=1,N3$ $ZEYO$ $89:$ 60 $CONTINUL$ $ZERO$ 900 $90:$ $RETURM$ $ZERO$ 910	76:		K=KS(IFL)	ZERO 700
79: K=K-1 7FR0 780 79: IYD=2 ZER0 700 80: IF(K.EQ.1) 60 T0 50 ZFR0 800 81: GD TO 52 ZER0 910 82: 50 J=J+1 ZER0 920 83: IF(J.ST.*1) GO TO 60 ZER0 930 94: GD TO 55 ZER0 940 85: 10 (C 58 I=J,*1 ZER0 940 86: HO 58 K=2,*Y ZER0 920 87: LO 59 L=1,*3 ZER0 920 87: LO 59 L=1,*3 ZER0 920 89: 60 CONTINUL ZER0 920 90: RETUR*1 ZER0 920 91: FND ZER0 910	77:	51	L=LS(IFL)	ZF 20 770
79: IYD=2 ZER0 790 80: IF(K.EQ.1) GO TO 50 ZFEG 800 81: GO TO 52 ZERO 910 82: 50 J=J+1 ZERO 920 83: IF(J.GT.M1) GO TO 60 ZERO 930 94: GO TO 55 ZERO 940 85: 10 (O 58 I=J,N1 ZERO 920 86: D0 58 K=2,KY ZERO 920 87: LO 59 L=1,N3 ZERO 920 89: 60 CONTINUL ZERO 920 90: RETURN ZERO 900 91: FND ZERO 910	79:		K = K - 1	7 FR() 7 HO
80: IF (K.EQ.1) GO TO 50 ZFRO F00 81: GO TO 52 ZERO P10 82: 50 J=J+1 ZERO P20 83: IF (J.GT.M1) GO TO 60 ZERO P30 84: GO TO 55 ZERO P40 85: 10 + O 58 I=J,M1 ZERO P20 86: DO 58 K=2,KY ZERO P20 87: LO 58 L=1,N3 ZERO P20 89: 60 CONTINUL ZERO P30 90: RETURM ZERO 940 91: FND ZERO 910	79:		IYD=2	ZERO 790
81: GP TP 52 ZERU P10 82: 50 J=J+1 ZERO P20 83: IF(J.GT.M1) GO TD 60 ZERO P30 94: GO TP 55 ZERO P40 85: 10 (P 58 I=J,M1 ZERO P20 86: HO 58 K=2,KY ZERO P20 87: LO 58 L=1,N3 ZERO P20 89: 60 CONTINUL ZERO P20 90: RETURM ZERO 900 91: FND ZERO 910	30:		IF (K.EQ.1) GO TO 50	ZEND 800
82: 50 J=J+1 ZER0 20 83: IF(J.ST.M1) S0 T0 60 ZER0 20 84: GD T0 55 ZER0 240 85: 10 (0 58 I=J,M1 ZER0 20 86: H0 58 K=2,KY ZER0 20 87: L0 58 L=1,M3 ZE20 20 88: 59 L(I,K,L)=0. ZE20 20 89: 60 CONTINUL ZER0 20 90: RETURM ZER0 90 91: END ZER0 910	91:		GP TP 52	ZERO P10
83: IF(J.GT.M1) GO TO 60 ZERO 230 94: GO TO 55 ZERO 240 85: 10 (O 58 I=J,W1 ZERO 250 86: 00 58 K=2,KY ZERO 250 87: LO 58 L=1,N3 ZERO 270 88: 59 U(I,K,L)=0. ZERO 820 89: 60 CONTINUL ZERO 820 90: RETURM ZERO 900 91: FND ZERO 910	82:	50	J = J + J	ZERO 820
94: GD_TD_55 ZEPD_P40 85: 10 (D_58_I=J,M1) ZER0_P50 86: 00 58_K=2,KY ZERD_P50 87: ED_58_L=1,N3 ZEPD_P70 88: 59_U(I,K,L)=0. ZERD_890 89: 60_CONTINUL ZERD_890 90: RETURN ZERD_900 91: END ZERD_910	83:		IF(J.ST.*1) SO TO 60	ZERN 830
85: 10 (0.58 I=J,N1 ZERU P20 86: 00.58 K=2,KY ZERO P20 87: 00.58 L=1,N3 ZEP0 P20 88: 59 U(I,K,L)=0. ZEP0 890 89: 60 CONTINUL ZERO 920 90: RETURN ZERO 920 91: END ZERO 910	94:		50 TC 55	ZERO 240
86: 10 58 K=2,KY 7ERO P(0) 87: LO 58 L=1,N3 ZEP0 P(0) 88: 59 L(I,K,L)=0. ZEP0 BP0 89: 60 CONTINUE ZERO P(0) 90: RETURN ZERO P(0) 91: END ZERO P(0)	85:	10	(P 58 I=J,11	ZERO POO
87: LO 58 L=1,N3 ZEPO 870 88: 58 U(I,K,L)=0. ZEPO 880 89: 60 CONTINUL ZERO 900 90: RETURN ZERO 900 91: END ZERO 910	86:		U() 58 K=2, KY	ZERN PLO
88: 58: 58: 2ERD 880 89: 60 CONTINUE ZERD 90 90: RETURN ZERD 900 91: END ZERD 910	87:		L(1, 58, L=1, N)	ZE20 870
89: 60 CONTINUL ZERD 90 90: RETURN ZERD 900 91: END ZERD 910	88:	58	U(I,K,L)=0.	ZERO 880
90: RETURN ZERD 900 91: END ZERD 910	89:	60	CONTINUE	2FK1 930
91: FND ZERD 910	90:		RETURN	ZERD 900
	91:			ZERD 910

1:		SUBROUTINE AVG(C,II,IFLAG)	۵۷ ت	10
2:	С		AVG	20
3:	С		AVG	30
4:	С	THIS SUBROUTINE COMPUTES THE AVERAGE CONCENTRATIONS (AVERAGING	AVG	40
5:	С	$TIM_{c} = 2*PROFL)$	AVG	4,5
6:	ι		AVS	εC
7:	С		AVG	7(
8:		DIMENSION C(15,13,15),0(15,15,15)	AVG	50
ን:		LEMMON /PLK1/ A1(15,15),A2(15,15),A3(15,15),P1(15,15),32(15,15),	AV,	÷
10:		113(15,15),P1(15),R3(15),R5(15),R6(15),P3,1(15),R362(15),APA1(15),	AVG	100
11:		23341(15,15),0(700),U(15),V(15),XXX(3,15),XYY(15),XZ7(15),XMAX(3),	AV C	110
12:		3YMAX, H, AK, TCH, P, KS(3), IS(3), CO(3), 'IK, 'IY, N7, N1, N2, N3, KX, KY, KZ,	AVJ	120
13:		441(15),W2(15),W3(15),MSRCS,IFLC,MFLC,CC(7,15,15,15),PRDEL,PRMT3,	AVS	130
14:		51NVR S	4V 3	14(
15:		IF(II.JT.2) SO TO 90	AVU	1.50
16:		M = I I + I	AVC	14.0
17:		(0) 10 J=2, N1	AVG	17(
18:		IPLK=J+(IFLAG-1)*1++(M-1)*45	AVG	180
17:		10 WPITE(3'IPLK) ((C(J,K,L),K=2,KY),L=1,M3)	AVC	1.20
20:		IF(II.EQ.2) SU TO 80	AVG	201
21:		RETURN	AVC	210
22:)0 M=2	AVG	220
23:		10 15 J=2,N1	AVS	230
24:		1hLK=J+(1FL4G-1)*15+(M-1)*45	AVG	240
25:		15 ₹FAU(3'IPLK) ((D(J,K,L),K=2,KY),L=1,43)	AVC	2 > (
26:		۸ [.] = 1	AVC	2.00
27:		10 25 J=2.N1	AV()	270
28:		IPLK=J+(IFLAG-1)*15+(M-1)*45	AVU	230
29:		25 WRITE(3'1)LK) (()(J.K.L).K=2.KY).L=1.()3)	AVC	2 1
30:		M=3	AVG	300
31:		10 35 J=2,11	AVJ	310
32:		IPIK = J + (IFLAC - 1) * 15 + (M - 1) * 45	AV ;	323
33:		35 KEAP(3!I+LK) ((D(J,K,L),K=2,KY),L=1,3)	AVG	3.20
34:		M=2	AV G	340
35:		LO 45 J=2,N1	AV G	350

1

.

36:		18LK=J+(IFLAG−1)*15+(M−1)*45	Δ٧,	160
37:	45	WRITI(3*IELK) ((()(J,K,L),K=2,KY),L=1,13)		370
38:		1 = 3	∧v G	300
39:		10 5% J=2,N1	۸V ن	390
4():		I }LK=J+(IFLAG-1)*15+(M-1)*45	AVC	400
41:	55	WRITE(3'IBLK) ((C(J,K,L),K=2,KY),L=1,N3)	ΔVG	410
42:	80	CONTINUE	AVG	420
43:		M = 1	AVG	4 3 U
44:		(1, 7), J=2, N=1	4VG	440
45:		IPLK=J+(IFLAG-1)*15+(M-1)*45	AV 6	450
46:	70	READ(3'IBLK) ((C(J,K,L),K=2,KY),L=1,43)	AV.	460
47:		™=2	AVG	47)
48:		UN 71 J=2,N1	AV C	4 10
49:		IPLK=J+(IFLAG-1)*15+(M-1)*45	AV ;	4 70
50:	71	READ(3'IFLK) ((D(J,K,L),K=2,KY),L=1,M3)	Δ∨β	500
51:		10 7? J=?,N1	ΔVG	510
52:		U() 72 K=2,KY	AVU	520
53:		10 72 L=1,N3	AVG	530
54:	72	C(J,K,L) = C(J,K,L) + D(J,K,L)	Δ 🗸 ι,	540
55:		M= 3	AVC	ちらり
56:		10073 J=2, N1	AVJ	550
57:		IPLK=J+(IFLAG-1)*15+(M-1)*45	Δνς	۶ 70
58:	73	k[/D(3'I3LK) ((D(J,K,L),K=2,KY),L=1,13)	AV J	530
59:		1074 J=2, N1	AVU	571)
60:		111 74 K=2,KY	AV?	600
61:		LN 74 L=1,N3	AVG	610
62:	74	C(J,K,L)=(C(J,K,L)+D(J,K,L))/3.	AV 5	620
63:		RETURN	CVA	630
64:		E*H)	AV;	540

1:		SUPRODITINE	RKGS(PRMT,Y,DFRY,NDIM,IHLF,FCT,OUTP,AUX)	PKGS	10
2:	C			PKCS	20
3:	C			RKUS	4)
4:	C,	THIS SU ROUTI	NE SOLVES A SYSTEM OF FIRST ORIER ORDINARY LIFFERFATIAL	RKGS	40
: ر	C	EQUATIONS WIT	H GIVEN INITIAL CONDITIONS	RKCS	()ر
6:	С			RKJS	ίŪ
7:	C			RKCS	70
٩:	6	PK₩T	- AT INPUT OUTPUT VECTOR WITH DIMENSION GREATER OR	RKCS	20
9:	С		EQUAL TO 5	BKCS	30
10:	С	PPMT(1)	- LOWER BOUND OF THE INTERVAL	RKSS	100
11:	C	PPMT(2)	- UPPER BOUND OF THE INTERVAL	RKGS	110
12:	С	PRMT(3)	- INITIAL INCREMENT OF THE I DEPENDENT VARIABLE	RKCS	120
13:	C	PRNT(4)	- UPPER ERROR FOUND	RKOS	130
14:	C.	PRMT(5)	- NO INPUT PARAMETER. IT IS DUNLESS THE USER WANTS TO	RKGS	140
15:	С		TERMINATE RESS AT ANY CUTPUT POINT	RKUS	140
16:	C	UF PY	- INPUT VECTOR OF LARAR WEIGHTS. LATERON IS THE VECTO?	RKGS	160
17:	C,		DF DERIVATIVES	RKGS	170
18:	C	NDIM	- THE NUMBER OF EQUATIONS IN THE SYSTEM	RK GS	160
19:	С	I HILF	- THE NUMPER OF DISECTIONS OF THE INITIAL INCREMENT	RKGS	1.40
20:	C	чUх	- AN AUXILIARY STURAGE ARRAY (R ROVS AND WIM COLUMNS)	RKSS	200
21:	C			PKGS	210
22:	С			PKUS	220
23:		DIMENSIOM	Y(1), DERY(1), AUX(8, 1), A(4), F(4), C(4), PRMT(1)	RKOS	23()
24:		$100 \ 1 \ I=1, N$		PKGS	247
25:		1 AUX(8, I) = .	()666667*()FRY(I)	RKUS	2.0
26:		X = ORMT(1)		PKCS	260
27:		XF*ID=PRMT(2)	RKGS	270
29:		H=PRt T(3)		RK ÚS	280
29:		PPMT(5)=).		546S	210
30:		LALL FCT(X	,Y,DERY)	EKUS	300
31:	C			BK CS	310
32:	С	ERRUP TEST		RKGS	320
33:	С			KK35	() د 3
34:		[F(H*(XE√D	-X))38,37,2	RKGS	340
35:	C			rk CS	350

36 :	r	PREPARATIONS FOR RUNCE-KUTTA NETHOD	
37:	č		
38:		2 (1) = 5	PKCS 340
37:		4(2) = .2928932	BK (25 - 340)
40:		A(3) = 1,707107	2K . 5 400
41:		(4) = .1665667	BKGS 410
42:		(1) = 2.	RK 15 420
43:		1'(2) = 1.	RK - S - 4 10
44:		+(3)=1.	RK 15 440
45:		F(4) = 2	2KGS 450
46:		C(1) = .5	8KC5 400
47:		L(2) = .2928932	RKUS 470
48:		(.(3) = 1.707107	RK65 480
47:		C(4) = .5	8K 4 10
50:	С		BKUS 500
51:	ί	PREPARATIONS OF FIRST RUNGE-KUTTA STEP	RK 25 510
52:	С		8KUS 520
53:		DO 3 I=1,NDIM	RKUS 530
54:		$\wedge \cup \gamma (1, 1) = \gamma (1)$	RKOS 540
55:		UX(2,I) = DEPY(I)	RK45 540
56:		$(1) \times (3, 1) = 0$.	RK 15 560
57:		$3 + (J \times (\ell, I) = 0.$	RKOS 570
58:		I P □ C = O	RKGS FRO
59:		t1=t1+t1	RK65 570
60:		IHLF=-1	RK15 600
úl:		I STEP=0	PKGS 610
62:		$I \in MD = 0$	RK 65 620
63:	C		RKUS 630
64:	С	START OF A RUNGE-KUTTA STEP	RKUS (40
65:	C		RK65 650
60:		4 IF((X+H-XEND)*H)7,6,5	RKGS (+)
67:		5 H=XE 10-X	RK 35 670
68:		$6 1 \in \mathcal{M}(i) = 1$	RKGS ENO
69:	С		RKUS 600
70:	С	RECURDING OF INITIAL VALUES OF THIS STEP	KK35 700
71:	C.		RK65 710

72:		7 CALL DUTP(X,Y,DEPY,IRFC,MUIM,PPMT)	2KUS 720
75:		1F(PRM1(5))40,8,40	PKG 730
74:			RKUS 740
75:		9 15/EP=15/EP+1	PKG5 750
77.	C C	CTART OF INUTRACT RUNC, WHITE LOOD	KK 72 100
70.	с С	START OF INDERMOST RUNGE-KOTTA LUNP	PKUS 770
70.	L	1 - 1	KKGS TEU
00.0			
00.			
5L+ 07+		0J=0(J) C J=0(J)	
0.2.		7'1 L1 1-L914()10 D1-14*()50((1)	
04÷			
501		<pre></pre>	3893 COU
00• 07•			
01+			
0.0.4		11 ///////////////////////////////////	
n J •			
904		$\begin{array}{c} 12 \\ 12 \\ 12 \\ 12 \\ 12 \\ 12 \\ 12 \\ 12 $	PK-2 010
·1• 02•		17 (J=7)17914913 17 V=V+ 5*4	
97.		$13 A = AT \bullet \Im \Phi \Pi$	PK 15 920
·····			
94 • 05 •	1		
	r.		PK 35 9.0
90. 97.	c c		
28.	U	15 TEATSSTN1/.1/.20	5KC2 080
90. 00.	ſ	15 11 (11 51/10)10)20	
100.	c c	IN LASE ITEST-O THERE IS NO DOSSIDILITY FOR TESTING OF "COMPACY	
101.	ĉ	IN CAPE TREST=0 TREST I HO POSSIATETALANA TESTINS DI ACCONACT	UK 2 5 1 0 1 0
101.		14 DO 17 T=1.NUTM	PK 051020
103.		13 10 11 1-11 010 17 1012 0.1 - 2010	RK (\$1040
104.		IT ()/(+)I/-((I)) ITEST=1	BKC21040
105:		T \$TEP=T\$TEP+T\$TEP=2	PK6510+0
106:			PK (S10/0
107:			PKC \$1070
T . / / •			1. 1. 7. 9 T. 7 L. 7

108:		H=•9+H	RK351040
109:		$U \cap (1) = 1 $, NPI 1	RKG\$1040
110:		Y(I) = AUX(I, I)	RK CS1100
111:		ERY(I) = AUX(2, I)	RKUS1110
112:		$13 + 10 \times (5, I) = AU \times (3, I)$	PKGS1120
113:		GNTO 9	RKGS1132
114:	С		RK55114)
115:	C,	IN CASE ITEST=1 TESTING OF ACCURACY IS POSSIFLE	RK 351150
116:	С		RK US11_0
117:		20 IMOD=ISTEP/2	RK051170
118:		IF(ISTEP-IMOD-IMOD)21,23,21	RKC51180
119:		21 CALL FCT(X,Y,DERY)	PK 651190
120:		DO 22 I=1, NDIM	PK051200
121:		$(U \times (5, I) = Y (I))$	RKOS1210
122:		2? $\Delta UX(7, I) = UERY(I)$	PK651220
123:		GOTO 9	RK 551230
124:	С		PKGS1240
125:	C	COMPUTATION OF TEST VALUE DELT	RKGS1250
126:	C		RK 351260
127:		23 DELT=0.	RK((S127)
128:		$1^{\circ} (24 I = 1, N) I H$	KK 3 S 1 2 H 0
129:		24)FLT=DELT+AUX(8,I)*APS(AUX(4,I)-Y(I))	PKGS12→0
130:		IF(DELT-PRMT(4))28,28,28	RK651300
131:	С		RK051310
132:	(.	ERROP IS TOD GREAT	RKGS1320
133:	С		RKGS1330
134:		25 IF(IHLF-10)26,36,36	KK3S1340
135:		25 DO 27 I=1,NDIM	RK 551350
136:		27 $AUX(4,I) = AUX(5,I)$	RK 051360
137:		ISTEP=ISTEP+ISTEP-4	RK351370
138:		$x = x - \frac{1}{2}$	RKGS13HO
139:		$I \in N D = 0$	PK 351370
140:		UNTU 18	RK3°14JU
141:	С		RK351410
142:	С	RESULT VALUES ARE GOOD	FKUS1420
143:	С		PKGS1430

144:		28 CALL FCT(X,Y,DERY)	RK~S1440
145:		UO 29 I=1,NUIM	RKGS1450
146:		$\Delta \cup X (1, I) = Y (I)$	RK 651460
147:		AUX(2,I)=DERY(I)	RK JS1470
148:		$(U \times (3, I) = (U \times (6, I))$	RKG\$1430
149:		Y(I)=AUX(5,I)	RK (5514 70
150:		29 DERY(I)=^UX(7,I)	RK JS15 DO
151:		UALL OUTP(X-H,Y,OERY,IHLF,NDI4,PPNT)	RK 351510
152:		[F(PRMT(5))40,30,40	RK051520
153:		30 + 0 + 31 = 1, 101 M	RK 551530
154:		Y(I) = A(JX(I, I))	PK 351540
155:		31 LERY(I) = AUX(2,I)	RKGS15.)
156:		IPFC=IHLF	RK651560
157:		IF(IEND)32,32,39	RKJ\$1570
158:	С		RK351540
159:	ί	INCREMENT GETS DOUBLED	RKGS1590
160:	С		RK551600
151:		32 IHLF=IHLF-1	RK7S1610
162:		ISTEP=ISTEP/2	PK651620
163:		H=H+H	RK351630
164:		IF(IHLF)4,33,33	RK351640
165:		33 IMOU=ISTOP/2	RK 651650
166:		IF(ISTEP-IMOD-IMOD)4,34,4	RK 3S16E0
107:		34 IF(UHLT02*?RMT(4))35,35,4	RK351670
168:		35 IHLF=IHLF-1	RKSS1690
159:		ISTEP=ISTEP/2	RK 351670
170:		F1=F1+F1	RKUS1700
171:		CTO 4	PK951710
172:	C.		3K GS 1720
173:	С	RETURNS TO CALLING PROGRAM	RK351730
174:	С		RK981740
175:		35 IHLF=11	RK (5517)0
176:		CALL FCT(X,Y, JFRY)	RKCS17.0
177:		UNTO 39	RK ; S1770
178:		37 IIILF=12	PK ~ S1740
179:		()T() 39	RK US 17 70

		PASE S
180:	38 1HLF=13	RK 651400
181:	3) LALL OUTP(X,Y,DERY,IHLE,NHIM,PRMT)	RKJS1810
182:	40 RETURM	RKCS1920
183:	F 11)	2K 351830

APPENDIX B

INPUT DATA REQUIRED

As examples, the input information for the hypothetical cases 8 and 10 are given next.

INPLT DATA (Case #8)

 YINIT = 0.0
 ENCS = 90.0 MIN
 PRMT(3) = 2.5C MIN
 PRMT(4) = 1.CCCCO

 NUMBER OF SCURCES = 1
 PREEL = 5.CO NIN

 NX = 6
 NY = 7
 N2 = 7

 YMAX = 7C0.0 H
 H = 505.CO H

 KS = 5
 LS = 4
 CC = 35.45 MG/CU.M

 XMAX = 19000.0 H
 STABILITY CLASS = 4
 INVRS = 1

 ALPHA = 2.CC
 AK = C.58E-C4 1/MIN

 UST = 3CC.CC M/MIN
 LGR = 30.CO M/MIN
 AM = 0.250

INPUT DATA (Case #10)

 TINIT = 0.C
 EKCS = 90.0 PIN
 PRHT(3) = 2.50 PIN
 PRHT(4) = 1.CCCCO

 NUPBER OF SOURCES = 2
 2

 PROEL = 5.CO MIN
 NX = 6
 NY = 7
 NZ = 7

 NX = 6
 NY = 7
 NZ = 7
 7

 YMAX = 7CC.0 F
 H = 505.CO F
 KS = 5
 LS = 4
 CO = 35.45 MG/CU.M
 XMAX = 1CCCO.0 F

 KS = 5
 LS = 4
 CO = 35.45 MG/CU.M
 XMAX = 1CCCO.0 F
 KS = 4
 LS = 5
 CO = 21.27 FG/CU.F
 XPAX = 2COO.C F

 STABILITY CLASS = 4
 INVRS = 1
 ALPHA = 2.CO
 AK = C.52EE-C6 1/PIK
 USI = 3CO.CC M/PIK
 LGR = 30.CO P/PIK
 AF = 0.250
 TCH = 15.C MIN
 P = 0.0

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APPENDIX C

NOMENCLATURE

a	Constant in equation (4.8)
a i	Parameters defined by equation (4.2)
a _{ij}	Parameters defined by equation (4.41)
= A	Matrix defined by equation (4.19)
AM	Represents the exponent m
b	Constant in equation (4.29)
Ē	Matrix defined by equation (4.20)
с	Constant in equation (4.29)
°j	Coefficients in equation (4.3)
c _i	Instantaneous concentration of species i, mg/m^3
C _B	Background concentration
C _o (i)	Concentration at i th source
d _i	Coefficients defined by equation (4.11)
d _{ij}	Coefficients defined by equation (4.42)
D _i	Molecular diffusivity of species i, m^2/min
D '	Eddy diffusivity tensor used in equation (3.1)
e _i	Constants defined by equation (4.43)
fj	Constants defined by equation (4.50)
G _i	Constant given by equation (4.9)

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G^{*} Constant in equation (4.33)

H Maximum height above terrain (in some cases refers to the elevation of the inversion base)

INVRS Constant used in equation (4.105)

ISTB Stability class (1 very unstable, 6 very stable)

(JS,KS,LS) (x,y,z) coordinate position of a point source

k Reaction rate constant, min⁻¹

k Von Karman's constant

K Turbulent diffusivity, m²/min

Height defined by equation (3.11)

L Effective emission height, m

m Exponent in equation (4.65)

N Number of interior collocation points

N_j jth independent Gaussian white noise

NSRCS Number of sources

l

p Exponent in equation (3.10)

P Constant in equation (4.67)

P First orthogonal polynomial

P; ith orthogonal polynomial

P Power spectral density of the jth white noise

PRMT(3) Initial step size of integration

PRMT(4) Upper error bound in "RKGS"

Q Source emission rate, mg/m³min

Q(i) Strength of ith continuous point source, kg/s $\overline{\overline{Q}}$ Matrix used in orthogonal collocation theory

 Q_x Mass rate through y-z plane at x = constant, kg/s

Transition probability density function $Q(x,t/x^*,t^*)$ $\overline{\overline{R}}$ Matrix used in orthogonal collocation theory Rate of generation of species i R; Number of species s Spatial-temporal distribution of particle sources S(x,t)t Time, min ㅠ Matrix used in orthogonal collocation theory Parameter in equation (4.67) TCH Velocity in x-direction, m/min u Variable in equation (4.31) u jth component of fluid velocity, m/min ^ui Friction velocity u* UGR Mean wind velocity in the x-direction at ground level UST Geostrophic wind speed, m/min Velocity in y-direction, m/min v ν* Pseudo-velocity defined by equation (3.1) Velocity in z-direction, m/min W w(x)Weighting function Wj Quadrature weights Cartesian coordinate in mean wind direction х Refers to independent variable in orthogonal х collocation theory Collocation points ×i \mathbf{x}_{\max} Maximum distance in the x-direction

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У	Cartesian coordinate in horizontal direction
У	Refers to dependent variable in orthogonal colloca- tion theory
y _{max}	Maximum distance in the y-direction
z	Cartesian coordinate in vertical direction
^z o	Characteristic surface roughness length
z1	Reference height

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Greek Symbols

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α	Constant in equation (4.63)
α	Exponent in equation (4.5)
∇	Gradient vector
⊽ ²	Laplacian
β	Exponent in equation (4.5)
Г	Gamma function
δ	Test value used in "RKGS"
^δ ij	Kronecker delta function
Δ	Knee height, m
ζ	Function that represents K_z
η	Function that represents v
θ	Time between the pollutant release and the initiation of the averaging time, min
ξ	Function that represents K
σ	Standard deviation
φ(x,y)	Function of the independent variables x and y, given by equation (4.41)

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 ψ Function that represents u

 Ω End of the averaging time, min

Superscripts

- Denotes the fluctuating component when used in a concentration or velocity variable
- Denotes the mean value when used on a velocity variable

* Refers to shifted polynomials

= Refers to a matrix

Subscripts

i	i th species
i	Index in collocation equations
i	Index used in orthogonal collocation theory
j	j th direction in Cartesian coordinates
j	Index used in orthogonal collocation theory
j	Represents the x-direction in collocation equations
k	Represents the y-direction in collocation equations
٤	Represents the z-direction in collocation equations
n	Index used in orthogonal collocation theory
S	Denotes ground level
x	Refers to x coordinate direction
У	Refers to y coordinate direction
Z	Refers to z coordinate direction
~	Denotes a vector quantity

Brackets

< > Denotes ensemble averages

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