

OPTIMUM PLASTIC DESIGN OF TAPERED  
GABLE FRAMES

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A Thesis  
Presented to  
the Faculty of the Department of Civil Engineering  
The University of Houston

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In Partial Fulfillment  
of the Requirements for the Degree  
Master of Science in Civil Engineering

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by  
Anwar M. Bydoun  
December, 1975

#### ACKNOWLEDGMENTS

The author wishes to express his deepest gratitude to Dr. G. Pincus for his indispensable guidance, encouragement and valuable suggestions during the preparation of this thesis.

## DEDICATION

To my family and brother, Dr. Talal, whose patience, understanding and financial support contributed immensely to this endeavor.

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## ABSTRACT

This study deals with the problem of plastic analysis and design of symmetrical tapered or uniform gable frames of the portal type subject to usual loads.

In the course of this study, a fully automatic plastic design procedure, based on an original conversational computer program written by H. B. Harrison, was developed and implemented on the UNIVAC 1108 computer. Input data was made free-format in order to eliminate the inconvenience present in using a fixed-format input. Another new feature of the program is the capability of dealing with tapered-column gable frames.

The procedure for analysis and design in the modified program accounts for the strength reduction in the plastic moment capacity caused by the presence of axial thrust in the frame members. The program also checks for the limiting width-to-thickness ratios of web and flange as given by the 1971 AISC specifications which serve to prevent local buckling failures.

## TABLE OF CONTENTS

|  |      |
|--|------|
| LIST OF FIGURES . . . . .  | vi   |
| NOMENCLATURE . . . . .   | viii |
| I. INTRODUCTION . . . . .  | 1    |
| 1.1 Background . . . . .   | 1    |
| 1.2 Stating The Problem . . . . .  | 2    |
| II. THE PLASTIC ANALYSIS OF THE STRUCTURE . . . . .  | 3    |
| 2.1 Assumptions . . . . .  | 3    |
| 2.2 Statical Method of Analysis . . . . .  | 4    |
| 2.3 Analysis Procedure . . . . .   | 5    |
| III. COMPRESSION MEMBERS . . . . .   | 11   |
| 3.1 Reduction of Plastic Moment Capacity of<br>Tapered Members Due to Axial Thrust . . . . . | 11   |
| 3.2 Discussion of the Interaction Formulas . . . . .   | 17   |
| IV. ADDITIONAL DESIGN CONSIDERATIONS . . . . .   | 25   |
| 4.1 Shear Force . . . . .  | 25   |
| 4.2 Local Buckling . . . . .   | 26   |
| 4.3 Connections . . . . .  | 26   |
| V. SUMMARY AND CONCLUSIONS . . . . .   | 30   |
| APPENDIX A SOLUTION OF LINEAR EQUATIONS . . . . .  | 32   |
| APPENDIX B USER'S MANUAL . . . . .   | 36   |
| APPENDIX C THE COMPUTER PROGRAM DFRAP . . . . .  | 47   |
| REFERENCES . . . . .   | 105  |

## LIST OF FIGURES

| Figure   |   | Page  |
|----------|---|-------|
| 2.1      | Stress strain curve of A7 structural steel,<br>idealized . . . . .  | 3     |
| 2.2      | Typical pressure distributions--wind and<br>vertical loading . . . . .  | 6     |
| 2.3      | Rigid frame twin cantilevers representations .  | 7     |
| 2.4      | Representation of determinate (a) bending<br>moment diagram; (b) shear force diagram;<br>(c) normal force diagram . . . . . | 9     |
| 2.5      | Graphical representation of bending moment<br>caused by reactants: M, R, S . . . . .  | 9     |
| 3.1      | Stress due to pure plastic moment . . . . .   | 11    |
| 3.2      | The first condition, natural axis in web . . .  | 12    |
| 3.3      | The second condition, natural axis in web . . .   | 14    |
| 3.4      | The third condition . . . . .   | 16    |
| 3.5-3.7  | Reduction of $Z_y$ , modulus of plasticity . . .  | 18-20 |
| 3.8-3.10 | Reduction of fully plastic moment by normal<br>forces . . . . .   | 22-24 |
| 4.1      | Type of connections in portal frames . . . .  | 27    |
| 4.2      | Diagonal stiffener . . . . .  | 28    |
| A.1      | Flow chart for the solution of system of<br>four linear equations . . . . .   | 34-35 |
| B.1      | Horizontal concentrated loads on left-hand<br>column . . . . .  | 40    |
| B.2      | Vertical concentrated loads on left-hand<br>rafter . . . . .  | 41    |
| B.3      | Gantry crane loads . . . . .  | 42    |
| B.4      | Representation of tapered bagle frame<br>sections . . . . .   | 44    |

| Figure                               | Page |
|--------------------------------------|------|
| B.5 Plastic design example . . . . . | 46   |

## NOMENCLATURE

|          |   |
|----------|---|
| $b_f$    | Flange width  |
| $d_1$    | Depth of member at the first end.   |
| $d_1$    | Depth of member at the second end.  |
| $d_n$    | Depth of member at distance $x$ from the first end.                             |
| $F_y$    | Specified minimum yield stress of type of steel being used (ksi).               |
| $A_s$    | Used in AISC Specification  |
| $F$      | A form factor for internal pressure.  |
| $h$      | The column height in a rigid industrial frame.                                  |
| $L$      | Length of a member; span of a portal frame.                                     |
| $M$      | Bending moment at apex of a portal frame.                                       |
| $M_p$    | The full plastic moment of a resistance of a section.                           |
| $M_{pc}$ | Reduced plastic moment capacity of a section.                                   |
| $P$      | An axial force, axial thrust, or a concentrated apex load.                      |
| $P_y$    | Plastic axial load; equal to profile area times specified minimum yield stress. |
| $R$      | The horizontal thrust at the apex of a frame.                                   |
| $r$      | Vertical rise from eaves to apex of an industrial rigid frame.                  |
| $S$      | The vertical shear force at the apex of a frame.                                |
| $t$      | Flange thickness.   |
| $w$      | Web thickness.  |

- $y_o$  Distance from neutral axis, when the section is under pure elastic moment, to the upper part of the assumed stress distribution, due to axial thrust.
- $y_1$  Distance from neutral axis, when the section is under pure plastic moment, to the neutral axis of the condition under consideration.
- $Z$  Plastic section modulus.
- $\beta_n$  The relative value of the distance from the neutral axis to the upper surface of a section and the section depth.
- $\tau_y$  Shear yield stress =  $y/3$ .
- $\sigma_y$  Is the Yield stress =  $F_y$  in any specified units.
- $\gamma$  The angle between the two flanges.
- $\theta$  Angle of inclination of rafter.
- cm Centimeter = 0.3937 inches.
- in Inches - 1/12 foot.
- kip 1000 pounds.
- kg Kilograms = 2.20462 pounds.
- lb Pounds
- ksi Stress in kips per square inch.
- tons metric = 2204.62 pounds.
- $\phi$  Angle of slope at diagonal stiffener.
- $\gamma_1$  Angle of taper at rafter.
- $\gamma_2$  Angle of taper of column.

## CHAPTER I

### INTRODUCTION

#### 1.1 Background

Plastic design, ultimate strength design, and limit design are terms which imply consideration of inelastic action. These techniques are now being employed in the design of continuous beams and building frames of one or two stories which sustain static loads. Furthermore, the principles of plastic theory justify many of the provisions in specifications formulated for elastic structures.

An engineering structure is satisfactorily designed if it can be built with the needed economy and if, throughout its useful life, it carries its intended loads and otherwise performs its intended function.

In the plastic analysis of an indeterminate structure three conditions must be considered:<sup>3</sup>

1. Mechanism condition;
2. Equilibrium condition;
3. Plastic-moment condition.

When all three of these conditions are satisfied, then the resulting analysis for ultimate load is correct because the two limit theorems basic to the plastic method are satisfied. These theorems can be stated as: theorem I (lower bound)

which is a statement of the ability of the material to adjust itself to carry the applied load if at all possible and gives the lower bound on, or safe value of, the ultimate loading. The maximum lower bound is the limit, or the ultimate load itself. Theorem II (upper bound), is a formal statement of the fact that if a path to failure exists, the structure will not stand up. Thus a load computed on the basis of an assumed mechanism will always be greater than, or at least equal to, the true ultimate load. The theorem so deals with the upper bound on, or unsafe value of, the plastic limit load. The minimum upper bound is the ultimate load itself.<sup>2</sup>

### 1.2 Stating the Problem

The problem consists of the plastic analysis and optimum plastic design of tapered gable frames under practical loading conditions using the modified computer program and the technique which has been previously used by H. B. Harrison.<sup>6</sup> The problem can be described as the design of a frame, with a given geometry and certain ultimate loads, such that its cost can be minimized.

## CHAPTER II

### THE PLASTIC ANALYSIS OF THE STRUCTURE

#### 2.1 Assumptions

The basic concepts and assumptions for the plastic analysis with regard to the plastic behavior of a structure are as follows:<sup>3</sup>

1. The material is ductile. It has the capacity of absorbing plastic deformation without the danger of fracture (Fig. 2.1).

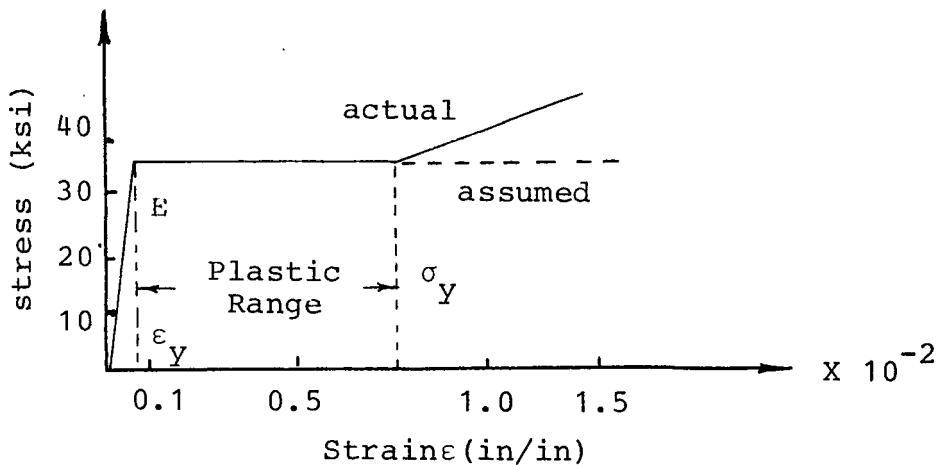


Fig. 2.1. Stress-strain curve of A7 structural steel, idealized.

2. Each section of a member has a maximum capacity, which causes plastification throughout the section.
3. Because of the steel ductility a plastic hinge

will be formed.

4. The deformations are so small that the equilibrium equations can be formulated for the undeformed structure.

5. The structure will reach its ultimate load when the mechanism of failure is created.

6. No instability will occur before the structure reaches its ultimate load.

7. Loads increase in proportion.

8. Sufficient number of plastic hinges have to be formed before the mechanism of failure is created.

## 2.2 Statical Method of Analysis

The objective of this method is to find an equilibrium moment diagram in which  $M < M_p$  such that a mechanism is formed. The procedure is as follows:<sup>2</sup>

1. Select redundants.

2. Find moment diagram for the determinate structure.

3. Find the moment diagram for the structure loaded by redundants.

4. Adding the two moment diagrams in such a way that a mechanism is formed.

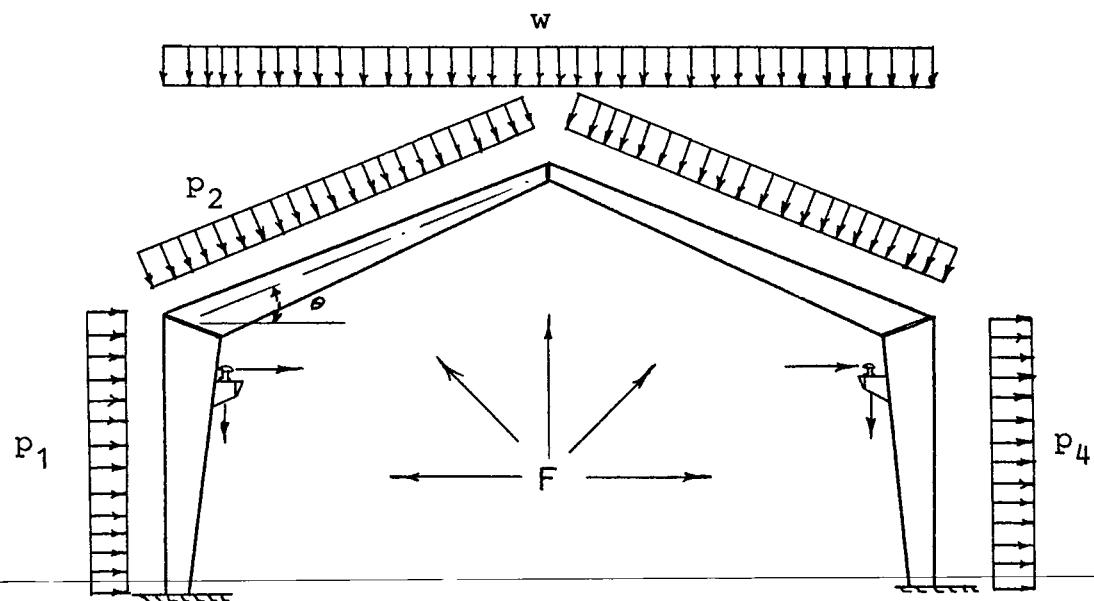
5. Check  $M < M_p$ .

After the mode of failure has been assumed, four linear simultaneous equations can be formulated to represent this mode of failure which can be solved to find  $M_p$ , (see

Appendix A). Then the moment at every section can be checked to see if it exceeds the reduced moment capacity of this section. If it does, the mode of failure has to be modified.<sup>6</sup>

### 2.3 Analysis Procedure

The most efficient method for performing plastic analysis in the case of tapered rigid gable frames is the twin-cantilever, statical method of analysis.<sup>6</sup> With reference to Fig. 2.3 the gable is cut at the apex, and the bending moment, shear force and axial thrust at any point in the frame are then evaluated statically in terms of the loads and the reactants M , S , and R. In a graphical representation, as in Fig. 2.4 and 2.5, it is convenient to separate the diagrams, the bending moment, the shear force and the normal force diagrams, into a determinate (Fig. 2.4 a, b, c) and indeterminate ones (Fig. 2.5 d, e, f). Then a plastic moment capacity distribution throughout the frame has to be assumed, or in other words, the section depth, the width and thickness and the web thickness has to be assumed. Next, a mechanism of failure is assumed, e.g. four locations are chosen at which the moment, of appropriate sign, are equal to the full plastic moment capacity at these sections. On the other hand, if the bases are pinned only two locations have to be nominated, and the four simultaneous equations expressing this mechanism can be formulated and solved for the three



$$p_1 = 0.8p$$

$$p_2 = (1.2 \sin \theta - 0.4)p$$

$$p_3 = 0.4p$$

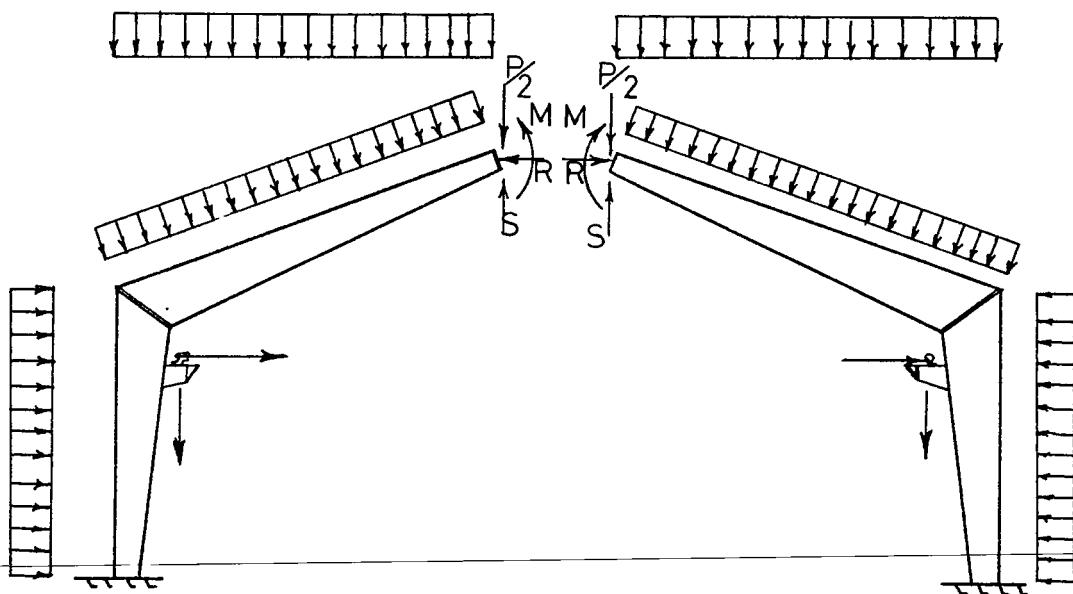
$$p_4 = 0.4p$$

where  $p = (\text{Wind Vel.})^2 / 400$

$F = (-0.2 \text{ or } 0.4)p$  for openings  $\geq 10\%$

$F = (-0.4 \text{ or } 0.8)p$  for openings  $< 30\%$

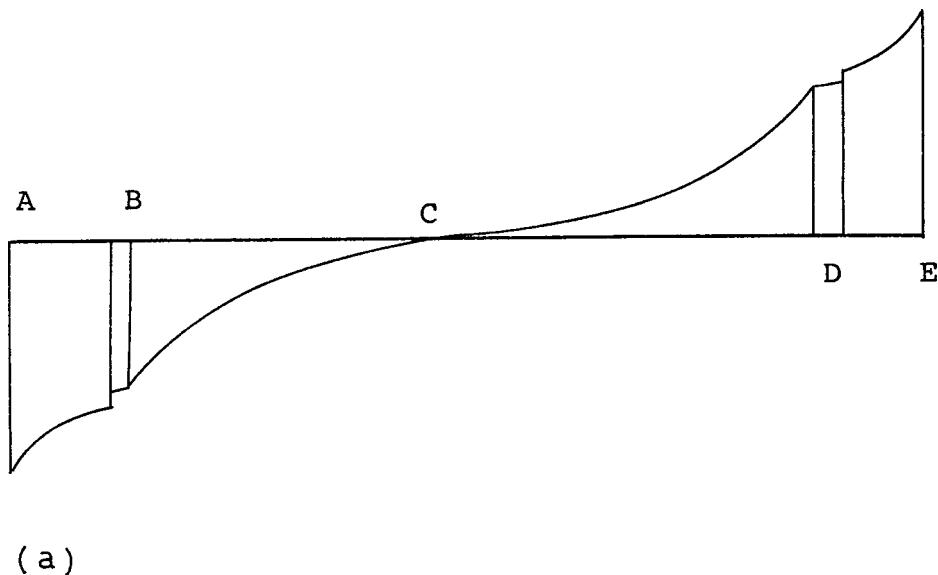
Fig. 2.2. Typical pressure distributions--wind and vertical loading.



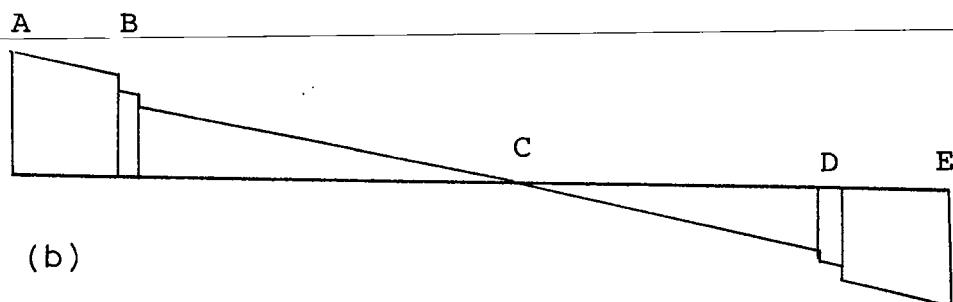
Apex - point load:  $P$

Reactants:  $M, R, S$

Fig. 2.3. Rigid frame twin cantilevers representations.



(a)



(b)

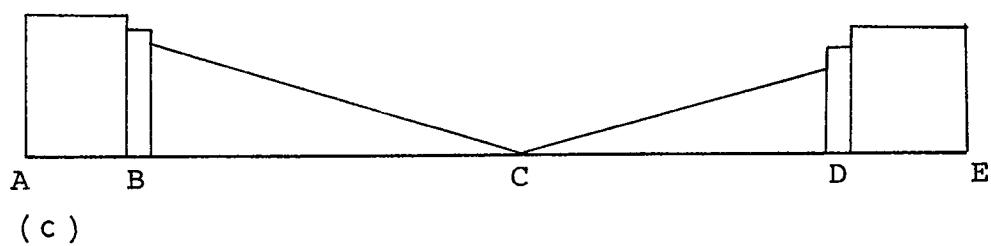


Fig. 2.4. Representation of determinate (a) bending moment diagram; (b) shear force diagram; (c) normal force diagram.

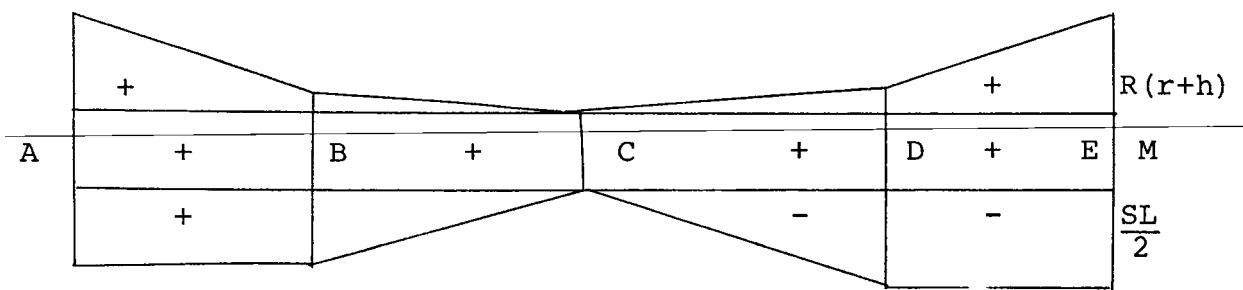
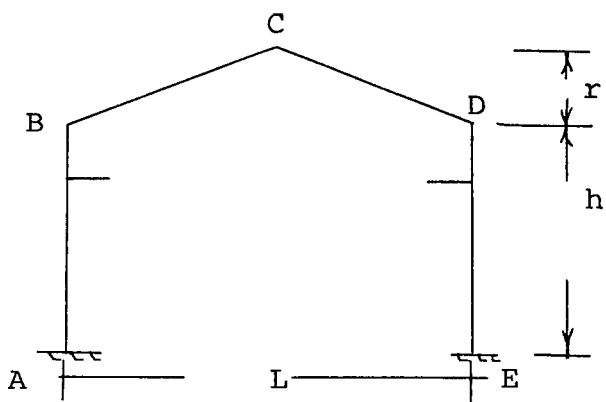


Fig. 2.5. Graphical representation of bending moment caused by reactants: M. R. S.

redundants M, S, and R. For the case of both bases pinned two of this four equations express the condition of zero moment at these locations. (In the case of one base is pinned and the other is fixed only three locations need to be nominated, and the fourth equation express the zero moment at this base, but this case is not available in the computer program represented in Appendix C. Although, this case can be programmed and a command for it can be added between lines 212 and 213.) Then the bending moment diagram, which corresponds to the assumed mechanism, can be found; using this diagram the normal and the shear force at any section can be determined, and the reduced plastic moment capacity can be evaluated. Then the evaluated plastic moment capacity can be used for the next iteration.

## CHAPTER III

### COMPRESSION MEMBERS

#### 3.1 Reduction of the Plastic Moment Capacity of Tapered Members Due To Axial Thrust

The presence of axial force tends to reduce the magnitude of the section plastic moment capacity in addition to causing column instability. Therefore, the moment capacity must be modified to account for this influence. Three different conditions can be discussed according to the location of the neutral axis.

#### The Plastic Modulus Of A Tapered Member

To find the member plastic modulus, the neutral axis of the section has to be located. Referring to Fig. 1 the factor  $\beta_n \leq 0.5$ , where  $\beta_n$  is the relative value of the neutral axis depth and the section depth, can be determined by taking the summation of the horizontal forces equal to zero.

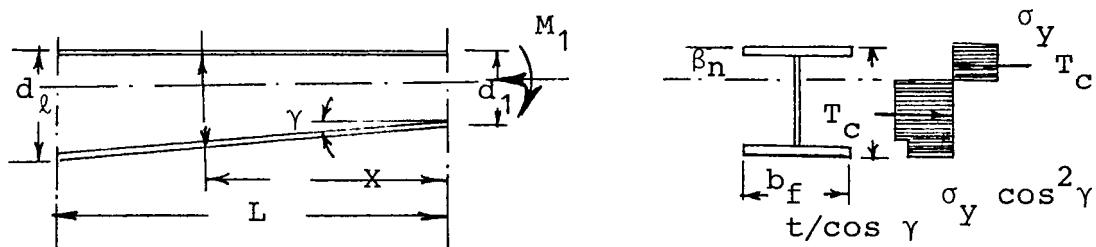


Fig. 3.1. Stress due to pure plastic moment.

$$\sum \overset{\rightarrow}{H} = 0$$

so  $T_c = -T_t$

$$\beta_n = 0.5 - t (1 - \cos\gamma) (b_f + w/\cos\gamma)/2 d_n w \quad (3.1)$$

where

$$d_n = \frac{d_\ell - d_1}{L} x + d_1 \quad (3.2)$$

$$\gamma = \tan^{-1} \frac{d_\ell - d_1}{L} \quad (3.3)$$

so, the plastic modulus of the section will be:

$$Z = \frac{w}{2} (\beta_n d_n - t)^2 + \frac{w}{2} [(1 - \beta_n) d_n - \frac{t}{\cos\gamma}]^2 + b_f t [\beta_n d_n + \cos\gamma d_n (1 - \beta_n) - t] \quad (3.4)$$

for the first condition, as shown in Fig. 2, the total stress-distribution,<sup>3</sup> sketch (a), may be divided into two parts: a part that is associated with the axial force, sketch (b), and a part that is associated with the bending moment, sketch (c).

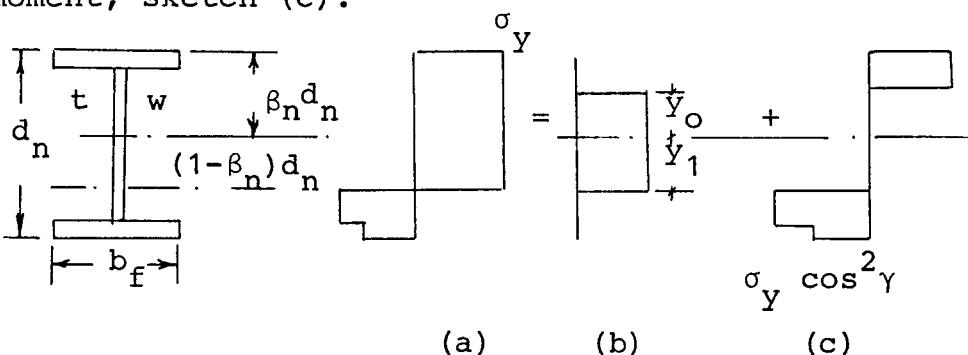


Fig. 3.2. The first condition, neutral axis in web. Representation of (b) stress due to axial force and (c) stress due to bending moment for a completely plastic cross section subjected to moment and axial force.

In this case the neutral axis is in the web ( $y_o \geq \beta_n d_n - t$ ). Because the section is in equilibrium, the axial force P is given by:

$$P = 2\sigma_y y_o w \quad (3.5)$$

where  $\sigma_y$  is the yield stress,  $y_o$  is the distance from the neutral axis of this condition to the neutral axis when the section undergoes pure plastic moment, and w is the web thickness. The corresponding bending moment is given by the following expression and represents plastic moment modified to include the effect of axial compression force.

$$M_{pc} = \sigma_y (z - w y_o^2) \quad (3.6)$$

where z is the plastic modulus of the section. Then, by substituting the value of  $y_o$  obtained from Eq. (3.5) into Eq. (3.6), the bending moment may be expressed as a function of the axial force P,

$$M_{pc} = M_p - P^2 / 4\sigma_y w, \quad P \leq 2 w \sigma_y (\beta_n d_n - t).$$

This relationship can be written in nondimensional form by dividing both sides by  $M_p = \sigma_y z$ . Thus:

$$\frac{M_{pc}}{M_p} = 1 - K_2^2 (P/P_y)^2 / 4 z w \quad (3.7)$$

where  $0 \leq P/P_y \leq 2 w (\beta_n d_n - t) / K_2$

$$K_2 = t (1 + \cos\gamma) (b_f - w/\cos\gamma) + w d_n$$

$$P_y = \sigma_y K_2 \quad (3.8)$$

The second condition, when the neutral axis is in the web but  $y_o > \beta_n d_n - t$  and  $y_1 < (1 - \beta_n) d_n - t/\cos\gamma$ , could be discussed using the same technique. An expression for  $M_{pc}$  as a function of  $P$  can be determined. Thus, the two forces  $P_1$  and  $P_2$ , which are in equilibrium with  $P$ , can be determined (Fig. 3).

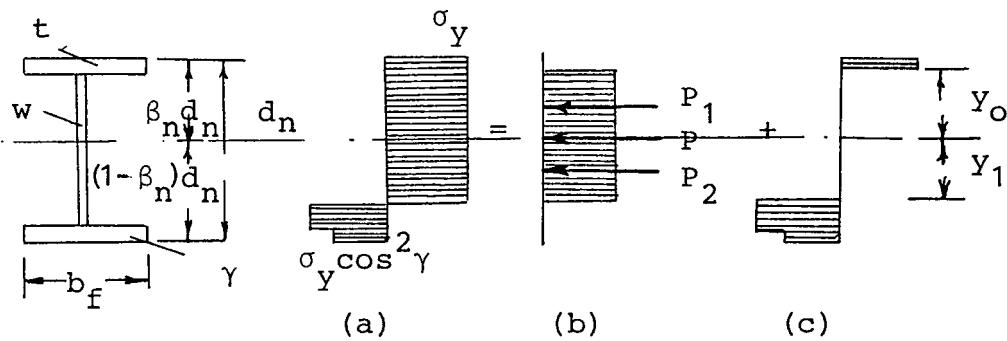


Fig. 3.3. The second condition, neutral axis in web and  $y_o \geq K_1$ .

Representation of (b) stress due to axial force and (c) stress due to bending moment for a completely plastic cross section subjected to bending moment and axial force.

$$\vec{P} = \vec{P}_1 + \vec{P}_2$$

where  $P$  is applied in the neutral axis of the case in which the section undergoes pure plastic bending moment. The relationship of  $y_o$  and  $y_1$  with  $P$  can be found by setting the moment of  $P_1$  and  $P_2$  at  $C$ , where  $C$  is the neutral axis of the case of pure moment, equal to zero.

$$M_{p1} = M_{p2}$$

hence  $y_1 = \sqrt{\frac{b_f}{w} y_o^2 + (\beta_n d_n - t)^2 (1 - \frac{b_f}{w})}$  (3.9)

or

$$y_1 = \sqrt{K_o y_o^2 + K_1^2 (1 - K_o)} \quad (3.10)$$

where

$$K_o = b_f/w \quad (3.11)$$

$$K_1 = (\beta_n d_n - t) \quad (3.12)$$

So  $P$ , the resultant of  $P_1$  and  $P_2$ , is equal to:

$$P = \frac{\sigma_y}{w} [y_1 + K_1 + K_o (y_o - K_1)] \quad (3.13)$$

From Eq. 3.10, Eq. 3.13 and 3.8  $y_o$  and  $y_1$  can be determined, and an expression for  $M_{pc}$ , the reduced plastic moment capacity, as a function of  $P/P_y$  can be determined. The resulting equations are:

$$\frac{M_{pc}}{M_p} = 1 - [w K_o y_o^2 + K_1^2 (1 - K_o)]/z \quad (3.14)$$

where

$$y_o = \frac{K_3 + K_1 (K_o - 1) \pm \sqrt{[K_3^2 + 2 K_3 K_1 (K_o - 1)]/K_o}}{(K_o - 1)} \quad (3.15)$$

$$K_3 = (K_2/w) P/P_y \quad (3.16)$$

$K_1$ ,  $K_2$ ,  $K_o$  and  $z$  are defined above.

By the same procedure, an expression for  $M_{pc}$  as a function of  $P/P_y$  can be determined when the neutral axis is in the flange instead of the web (Fig. 4).

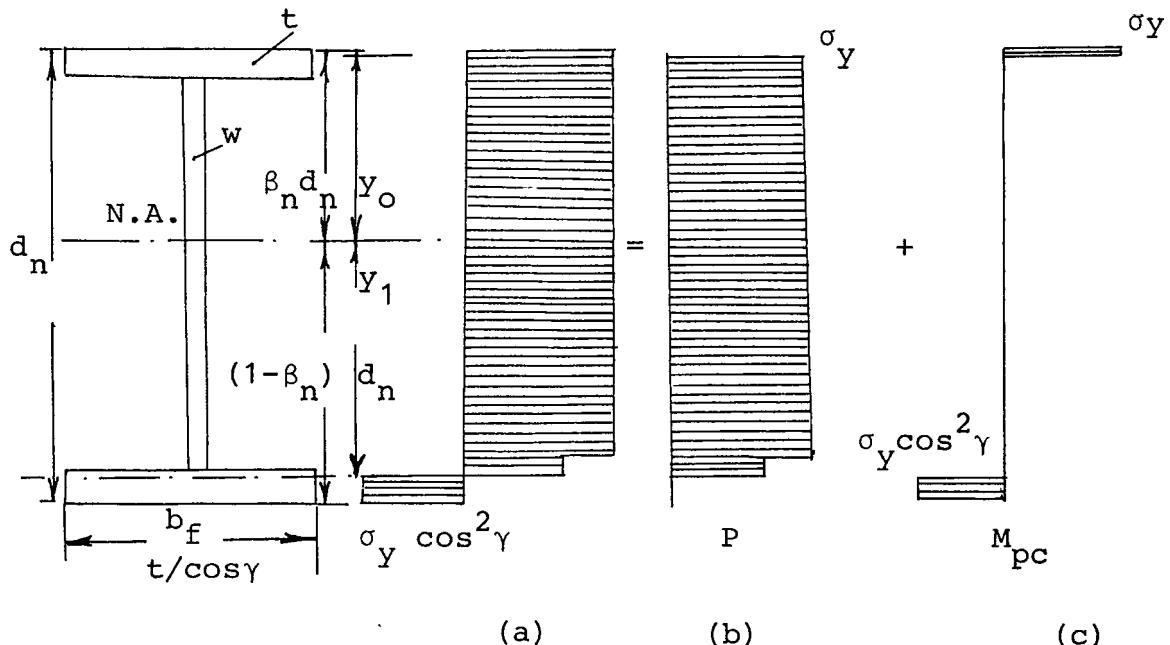


Fig. 3.4. The third condition, neutral axis is in flange. Representation of (b) stress due to axial force and (c) stress due to bending moment for a completely plastic cross section subjected to moment and axial force.

The relationship of  $y_1$  and  $y_o$  with  $P/P_y$  is given by the equations 3.17 and 3.18.

$$y_1 = \frac{1}{\cos \gamma} \sqrt{K_5^2 + y_o^2} \quad (3.17)$$

$$y_o = \frac{K_7 \pm \cos \gamma \sqrt{K_7^2 + K_o^2 K_5 (1 - \cos^2 \gamma)}}{K_o (1 - \cos^2 \gamma)} \quad (3.18)$$

where

$$\gamma \neq 0$$

$$K_4 = (1 - \beta_n) d_n - t / \cos \gamma$$

$$K_5 = K_1^2 \left( \frac{1}{K_o} - 1 \right) + K_4^2 (\cos^2 \gamma - \frac{1}{K_o})$$

$$K_6 = K_1 (K_o - 1) + K_4 (K_o \cos^2 \gamma - 1)$$

$$K_7 = K_3 + K_6$$

Thus, the non-dimensional expression for the reduced plastic moment of the section, when  $(1 - \beta_n) d_n \geq y_1 \geq K_4$  and  $\beta_n d_n \geq y_o \geq K_1$ ,  $M_{pc}/M_p$  is equal to:

$$\frac{M_{pc}}{M_p} = 1 - \frac{(K_1^2 + K_o y_o^2 - K_1^2 K_o) w}{z} \quad (3.19)$$

where  $K_o$ ,  $K_1$ , and  $y_o$  are given above.

### 3.2. Discussion of The Interaction Formulas:

After the relationships of  $z$ , and  $M$  with the angle  $\gamma$ , the angle between the two flanges has been determined (see Eqs. 3.4, 3.7, 3.14). Representations of this relationship are shown in Figs. 3.5 through 3.10. In this representation numerical data, for the section depth flange width and thickness and the web thickness, has been assumed, and by using the computer, subroutine three, the solution for this equation has been achieved.

In figures 3.5 through 3.7 the relationship of  $z_\gamma/z_o$ , where  $z_\gamma$  is the plastic section modulus when the angle between the two flanges is  $\gamma$  and  $z_o$  is the plastic

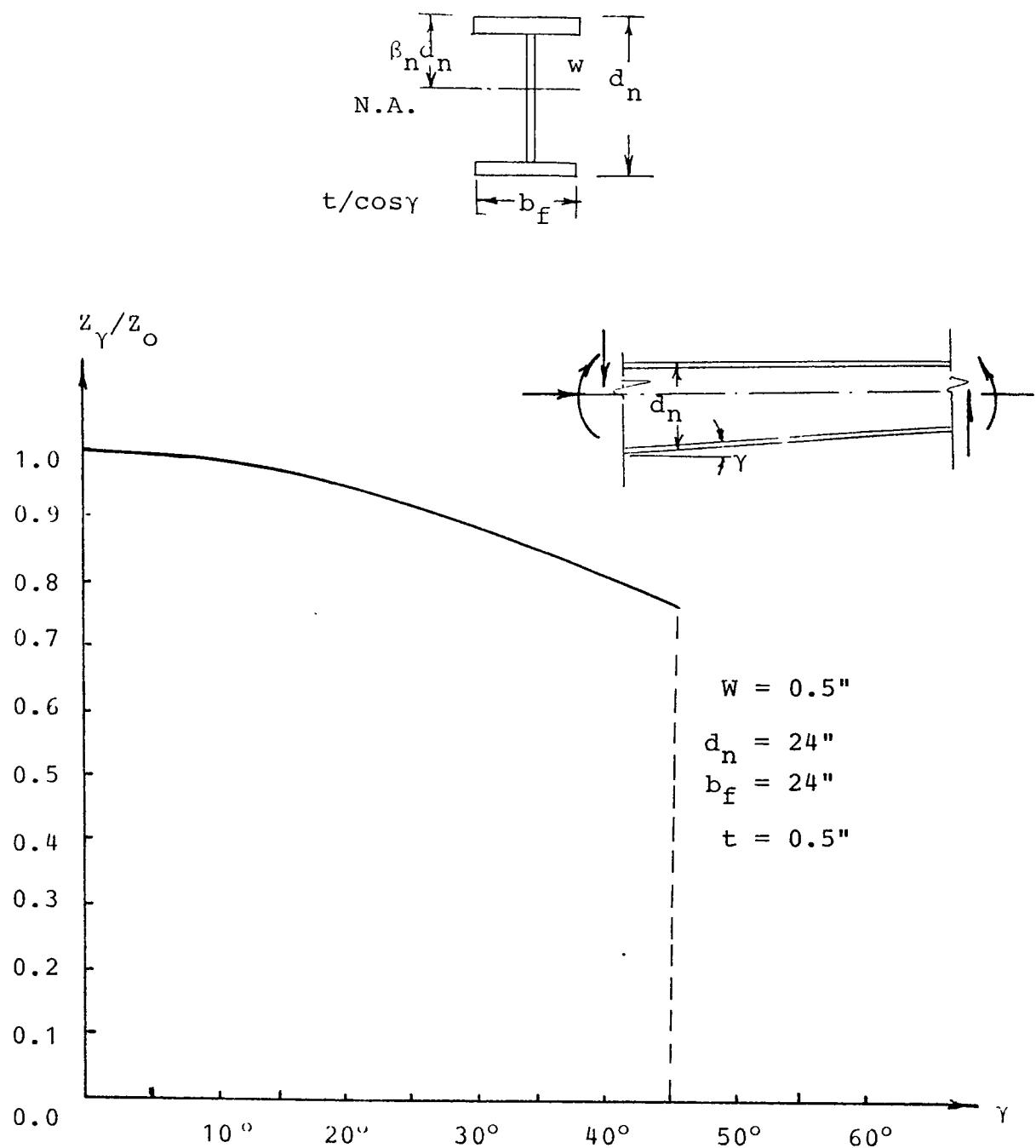


Fig. 3.5. Reduction of  $Z_\gamma$ , modulus of plasticity with the increase of  $\gamma$ , angle of taper, in column-beams.

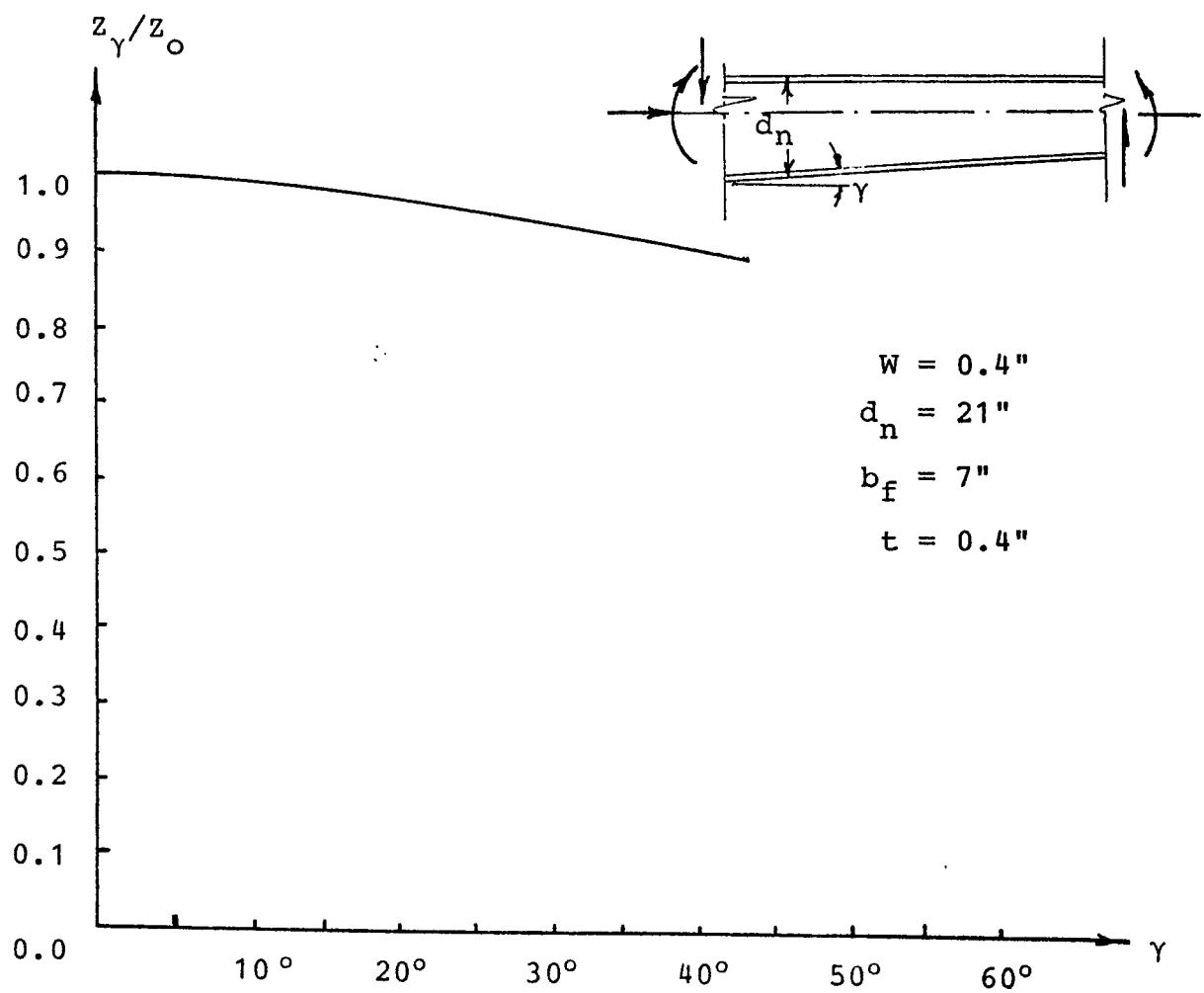
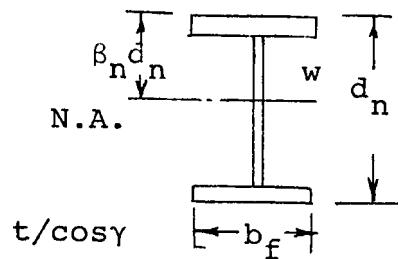


Fig. 3.6. Reduction of  $z_\gamma$ , modulus of plasticity with the increase of  $\gamma$ , angle of taper, in column-beams.

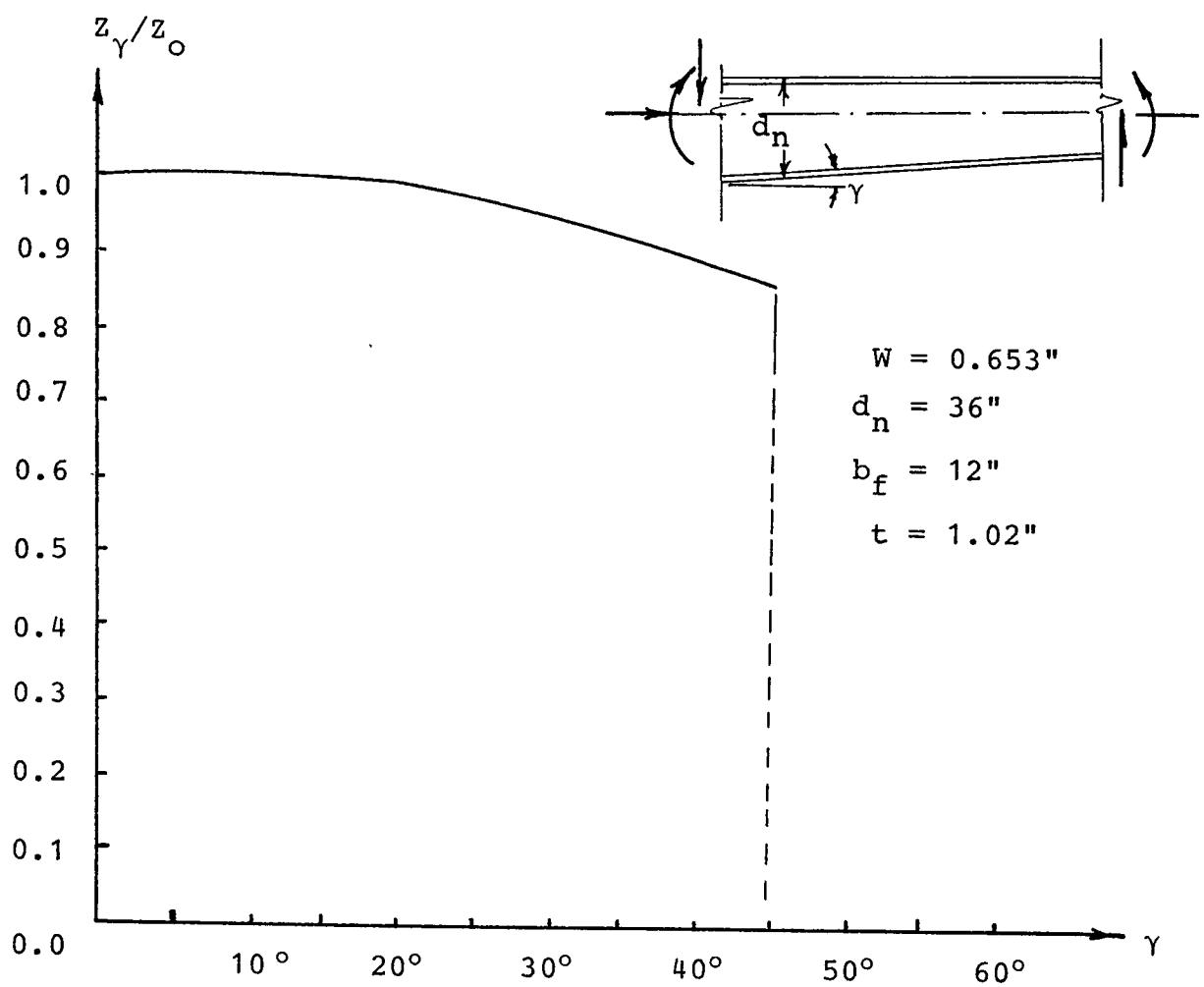
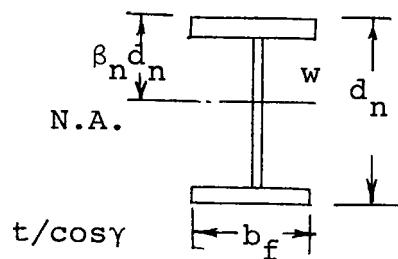


Fig. 3.7. Reduction of  $z_\gamma$ , modulus of plasticity with the increase of  $\gamma$ , angle of taper, in column-beams.

section modulus when  $\gamma$  is zero, with  $\gamma$  has been represented. These figures demonstrate how  $Z_\gamma$  decreases with the increase of  $\gamma$ , and it can be found that the decrease would not exceed 6 percent if  $\gamma$  is less than  $20^\circ$ , but the reduction in the modulus of plasticity will be 30 percent if the  $\gamma$  is equal to  $45^\circ$ . In this case an increase in the flange area, to retain the same modulus of plasticity, can be suggested.<sup>2</sup> It had been found that this needed increase in the flange area would not exceed 6 percent if  $\gamma$  is less than  $20^\circ$ . For practical purposes, both flanges should be made equal. Using the same procedure a representation of the relationship of  $M_{pc}/M_p$  with  $P/P_y$  can be seen in Figs. 3.8 through 3.10.

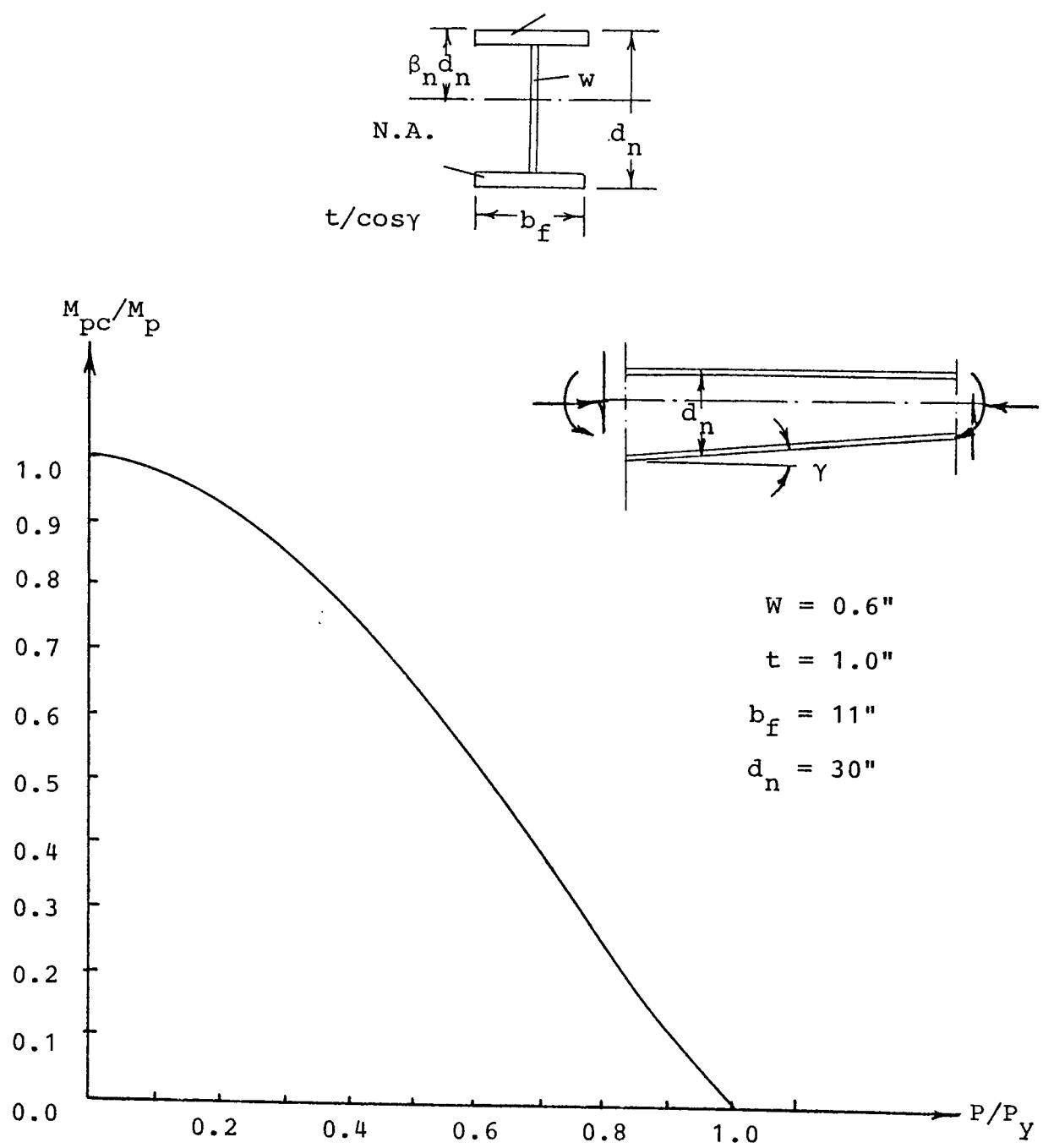


Fig. 3.9. Reduction of fully plastic moment by normal forces.

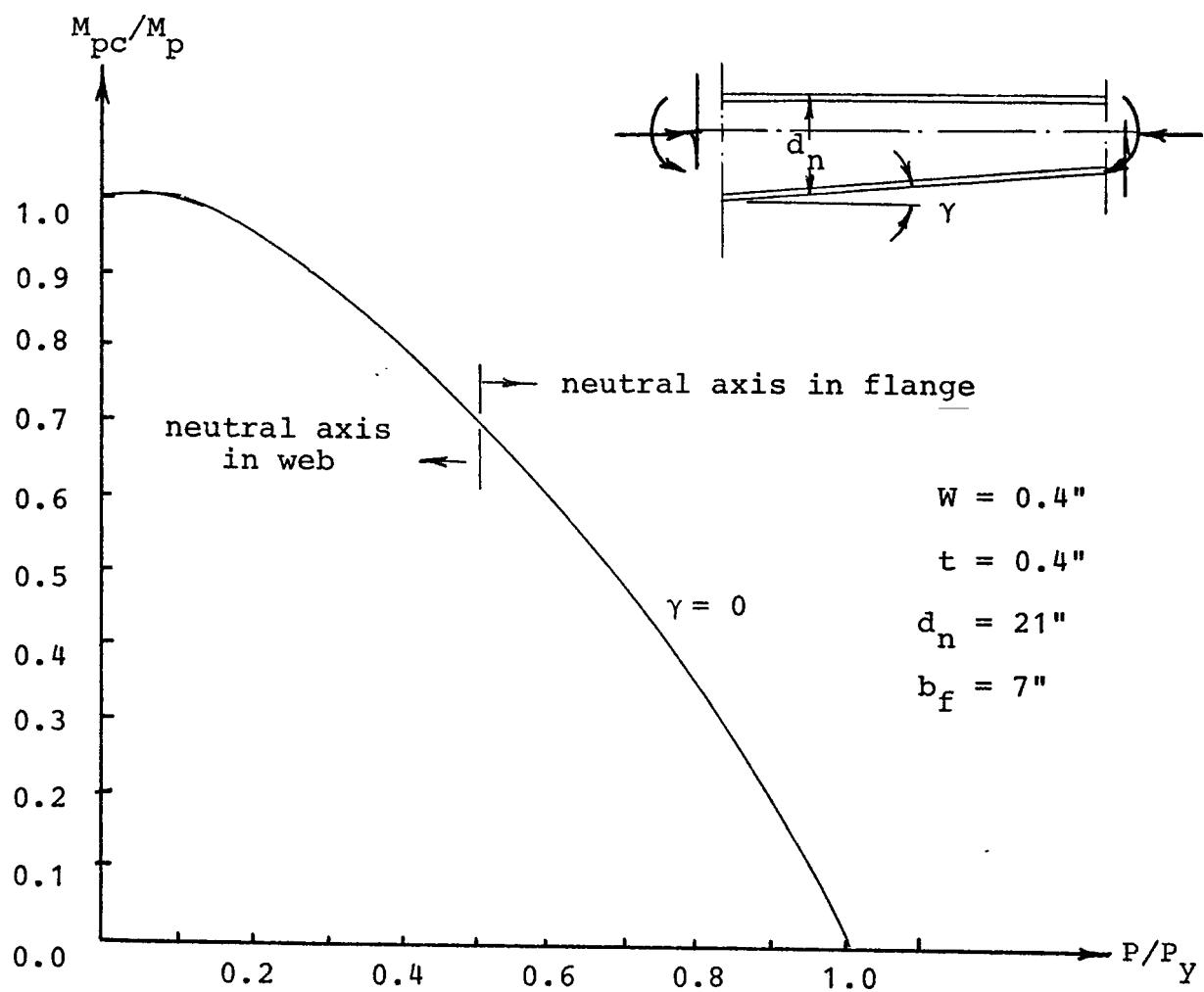
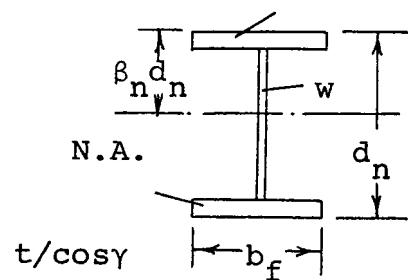


Fig. 3.8. Reduction of fully plastic moment by normal force.

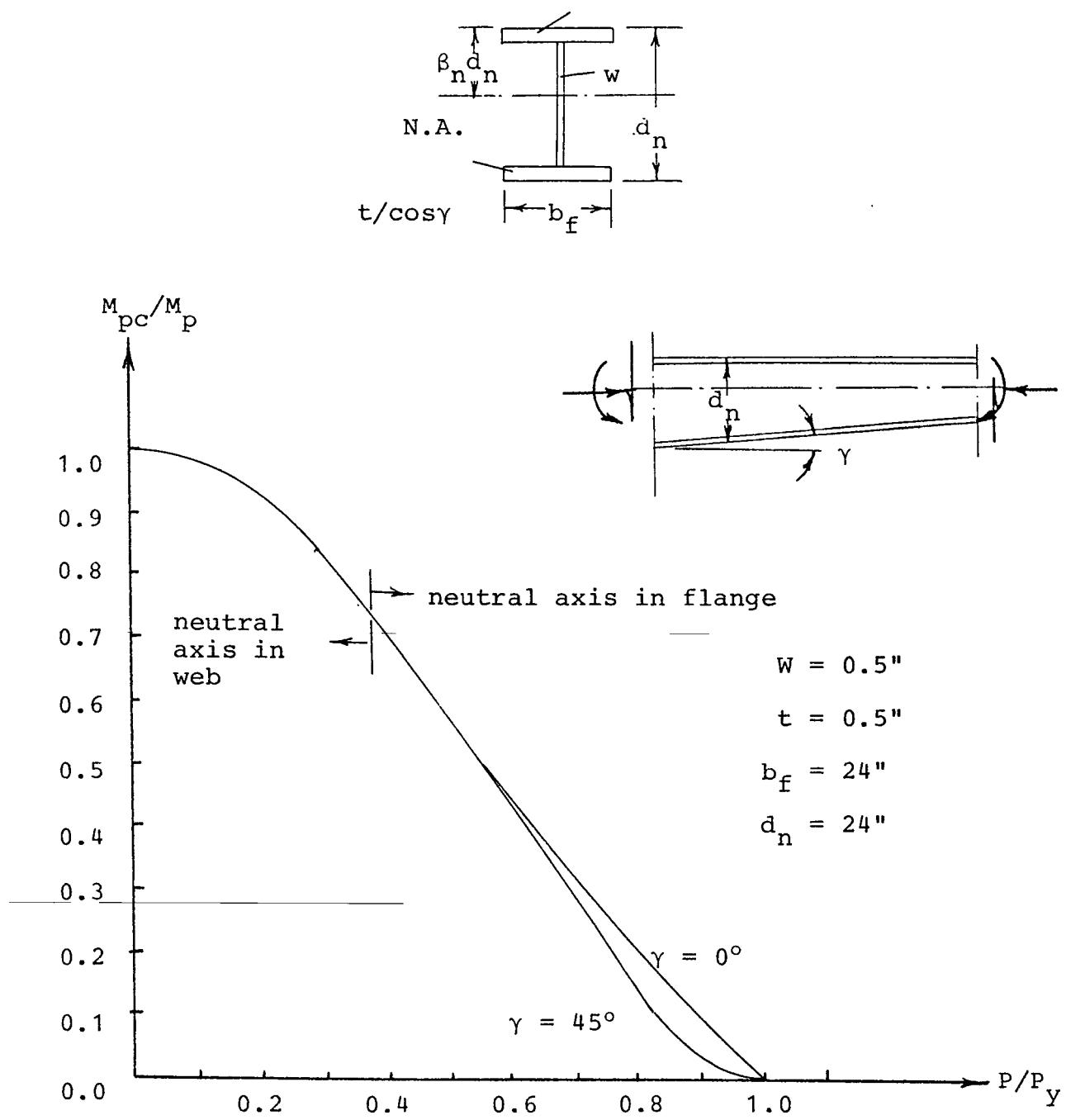


Fig. 3.10. Reduction of fully plastic moment by normal force.

## CHAPTER IV

### ADDITIONAL DESIGN CONSIDERATIONS

This chapter considers important factors which must be evaluated when the plastic theory is applied in the design of a structure, just as in the case of allowable-stress design. Besides the influence of the applied normal forces on the plastic moment capacity of the section, which have been discussed in the previous chapter, the influence of shear force, local buckling of flanges and webs, lateral instability, and repeated loading have to be evaluated.

#### 4.1. Shear Force

No reduction of plastic moment due to shear is required if the magnitude of the shear force  $V$  at the ultimate load does not exceed

$$V \leq 0.55 F_y w d \quad (4.1)$$

---

where  $w$  is the web thickness, and  $d$  is the section depth. On the other hand if the member is reinforced by diagonal stiffeners or a doubler plate, the web does not need to correspond to the previous limitation.<sup>4</sup>

#### 4.2. Local Buckling

The problem of elastic and inelastic buckling below

the yield stress in flanges has been discussed and analyzed theoretically by G. Haaijer and B. Thurlimann.<sup>2</sup> The width-thickness ratio for flanges of built-up sections which have one web and undergo axial thrust and hinge rotation under ultimate loading shall not exceed the following values, which have been recommended by the American Institute of Steel Construction in 1973:<sup>3</sup>

| $\frac{F_y}{Y}$ | $\frac{b_f}{2} \frac{t_f}{t_f}$ |
|-----------------|---------------------------------|
| 36              | 8.5                             |
| 42              | 8.0                             |
| 45              | 7.4                             |
| 50              | 7.0                             |
| 55              | 6.6                             |
| —               | 6.3                             |
| 65              | 6.0                             |

Also the depth-thickness ratio of members subjected to plastic bending shall not exceed the value given by the formula 4.1 or 4.2.<sup>3</sup>

---


$$\frac{d}{w} = \frac{412}{F_y} \left(1 - 1.4 \frac{P}{P_y}\right) \text{ when } \frac{P}{P_y} \leq 0.27 \quad (4.2)$$

$$\frac{d}{w} = \frac{257}{F_y} \quad \text{when } \frac{P}{P_y} \geq 0.27 \quad (4.3)$$

#### 4.3. Connections

In the design of connections for portal frames, both strength and rigidity are prime requirements, but these qualities

have to be achieved without either undue fabrication cost or difficulty during erection.<sup>7</sup>

The various types of connections which are typical of those that might be encountered in portal frames are as designated in Fig. 4.1.

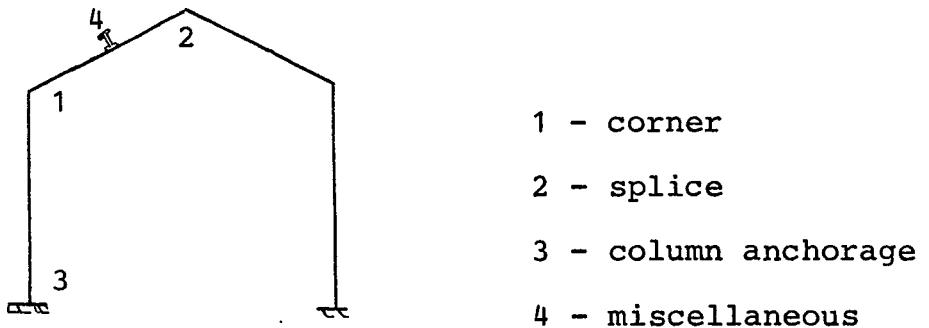


Fig. 4.1. Types of connections in portal frames.

Primary attention is given herein to corner connections (the other types can be found in Ref. 4). Method of analysis are based on the assumption of stress distribution at ultimate load which satisfies the equilibrium but does not violate the plasticity conditions. Thus, the solution constitutes lower bound to connection capacity.

The maximum possible forces in the two flanges and the diagonal stiffener can be determined from an equilibrium of horizontal components of these forces as follows:

$$\begin{aligned} \sigma_y A_s \cos \theta - \sigma_y b_f t \cos(\gamma_1 + \theta) + \sigma_y A_f t \sin \gamma_2 &= 0. \\ A_s = b_f t \left[ \frac{\cos(\gamma_1 + \theta) - \sin \gamma_2}{\cos \phi} \right] & \end{aligned} \quad (4.4)$$

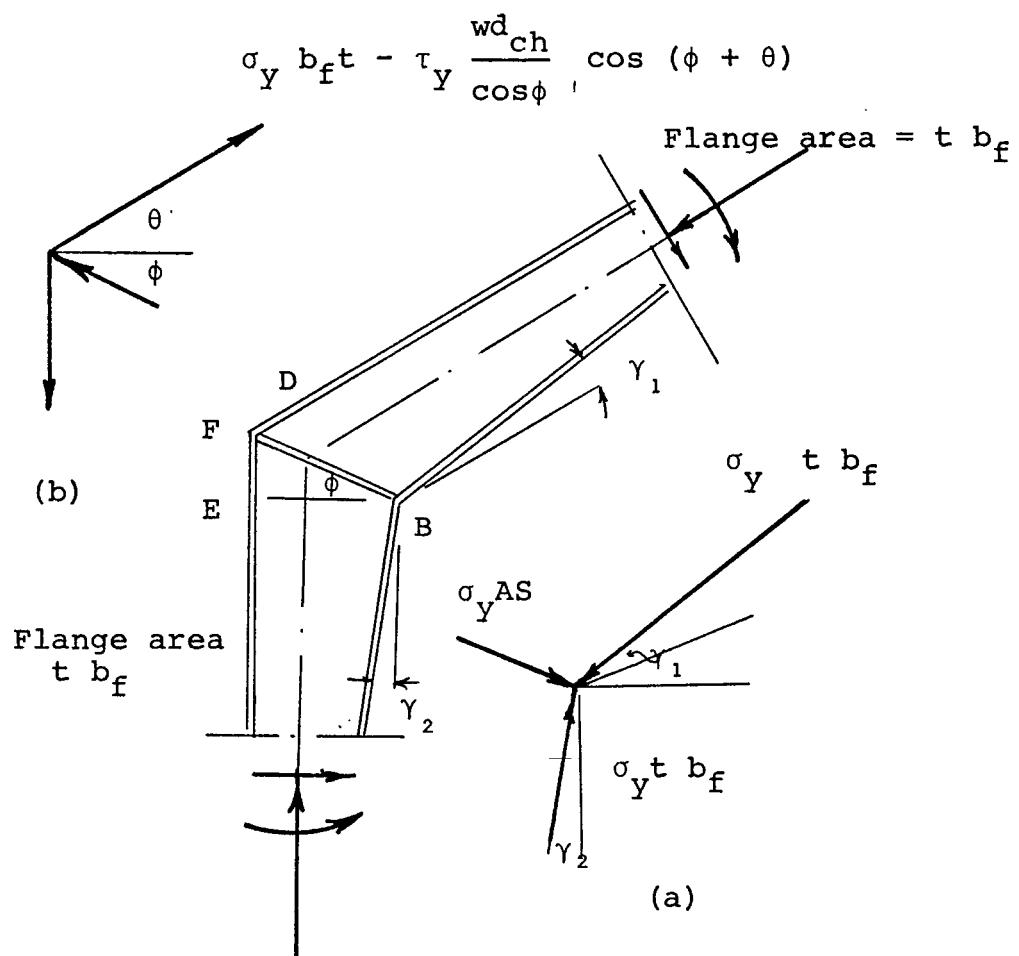


Fig. 4.2. Diagonal stiffener forces in a corner connection of tapered gable frame.

in which

$A_s$  = area of symmetrical pair of diagonal stiffeners

$b_f$  = flange width

$t$  = flange thickness

$\theta$  = angle of inclination of rafter

$\phi$  = angle of slope of diagonal stiffener

$\gamma_1$  = angle of taper of rafter

$\gamma_2$  = angle of taper of column

In Fig. 4.2 (b) the maximum possible forces at outer corner F are shown. Equilibrium of horizontal components gives:

$$\sigma_y b_f \cos \theta - \tau_y \frac{w d_{ch}}{\cos} \cos(\phi + \theta) \cos \theta - \sigma_y A_s \cos \phi = 0$$

Substituting  $\tau_y = \sigma_y / \sqrt{3}$  and solving for the required stiffener area gives

$$A_s = \left[ \frac{b_f t}{\cos \phi} - \frac{w d_{ch} \cos (\phi + \theta)}{\sqrt{3} \cos^2 \phi} \right] \cos \theta \quad (4.5)$$

in which  $w$  is web thickness  $d_{ch}$  depth of column at section EB, (see Fig. 4.2). If Eq. 4.5 should result in zero or negative value for  $A_s$ , the implication is that the shear capacity of the web is adequate to transmit the outer flange force and the diagonal stiffeners are needed only to transmit the unbalanced force of the inner flange. Since in any case the web will carry some force, Eq. 4.4, which is based on equilibrium at the inner corner, will control.<sup>2</sup>

## CHAPTER V

### SUMMARY AND CONCLUSIONS

#### Summary

The computerized plastic analysis and design of tapered, or uniform, gable frames, when both bases are fixed or pinned, under practical loading conditions has been developed and presented in Chapter II.

The plastic section modulus of taper members (variable depth web) has been derived in Chapter III. Also, three interaction equations to evaluate the reduced plastic moment capacity, the reduction caused by the axial thrust only, have been derived and checked by using Subroutine SUB3, see Appendix C. Graphs representing these three equations have been provided.

A list of the developed computer program has been given in Appendix C. This program has the capability of performing the analysis of tapered gable frames in addition to uniform ones and can also check for the validity of the AISC recommended design criteria for premature failure due to local buckling.

#### Conclusions

It is found in studying the results of the numerical example given in Appendix C that the developed

computer program has the capability of handling the analysis of tapered gable frames in addition to uniform ones. In using the free-field format to input the numerical data in addition to using both subroutine SUB1 and subroutine UNITS to input the alphanumerical data resulted in giving the program the desired flexibility to minimize some of the common mistakes usually made while inputting data in a rigid, specified format. Also, subroutine units made it possible to have the output in any specified units (see Appendix C). By inspecting the user's manual, provided in Appendix B, it can be seen that this program does not require in its use any great understanding of the computer.

The computer program can be modified to account for more than the studied base conditions or other loading configurations.

---

On the other hand, the same technique can be used to solve gable frames with nonuniform cross sections other than or in addition to these already included. All that is needed, to study another type, is a subroutine that will calculate  $M_{pc}$  as a function of  $X$ , where  $X$  is the relative location of the section with respect to one end of the member. A command for this type of modification may be easily added.

## APPENDIX A

## Solution of Linear Equations

A system of linear algebraic equations in N unknowns can be written as the following single matrix equation:

$$\begin{bmatrix} a_{11} & a_{12} & a_{13} & \dots & a_{1n} \\ a_{21} & a_{11} & a_{12} & \dots & a_{2n} \\ a_{31} & a_{32} & a_{33} & \dots & a_{3n} \\ \dots & \dots & \dots & \dots & \dots \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \\ \vdots \\ x_n \end{Bmatrix} = \begin{Bmatrix} c_1 \\ c_2 \\ c_3 \\ \vdots \\ c_n \end{Bmatrix} \quad (1)$$

or simply

$$A X = C \quad (2)$$

where  $A$  is the coefficient matrix,  $X$  is the unknown vector and  $C$  is the constant vector. Using this notation the solution will be as follows:

$$X = A^{-1} C \quad (3)$$

This solution can be achieved by using Gauss reduction technique<sup>9</sup> and the resulting equations can be written as follows:

$$a_{kj}^k = a_{kk}^{k-1}/a_{kj}^{k-1}, \quad j \neq k; \quad a_{kk}^k = 1/a_{kk}^{k-1} \quad (4)$$

$$a_{ij}^k = a_{ij}^{k-1} - a_{ik}^{k-1} a_{kj}^k, \quad j \neq k;$$

$$a_{ik}^{k-1} = -a_{ik}^{k-1} a_{kk}^k, \quad i \neq k, \quad j \neq k \quad (5)$$

The flow chart for this solution is presented in Fig. A.1.

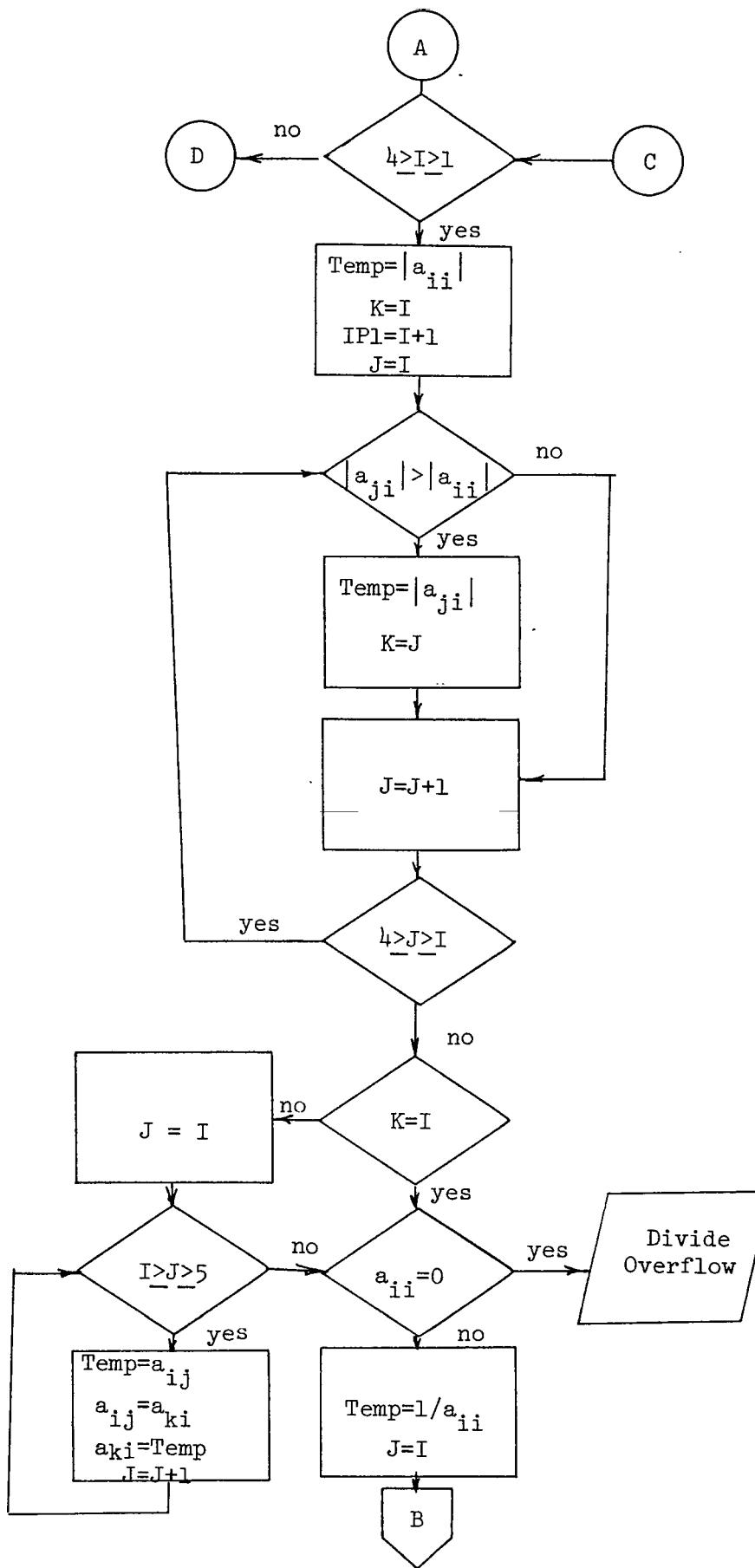


Fig. A.1. Flow chart of the solution of the system of four simultaneous linear equations.

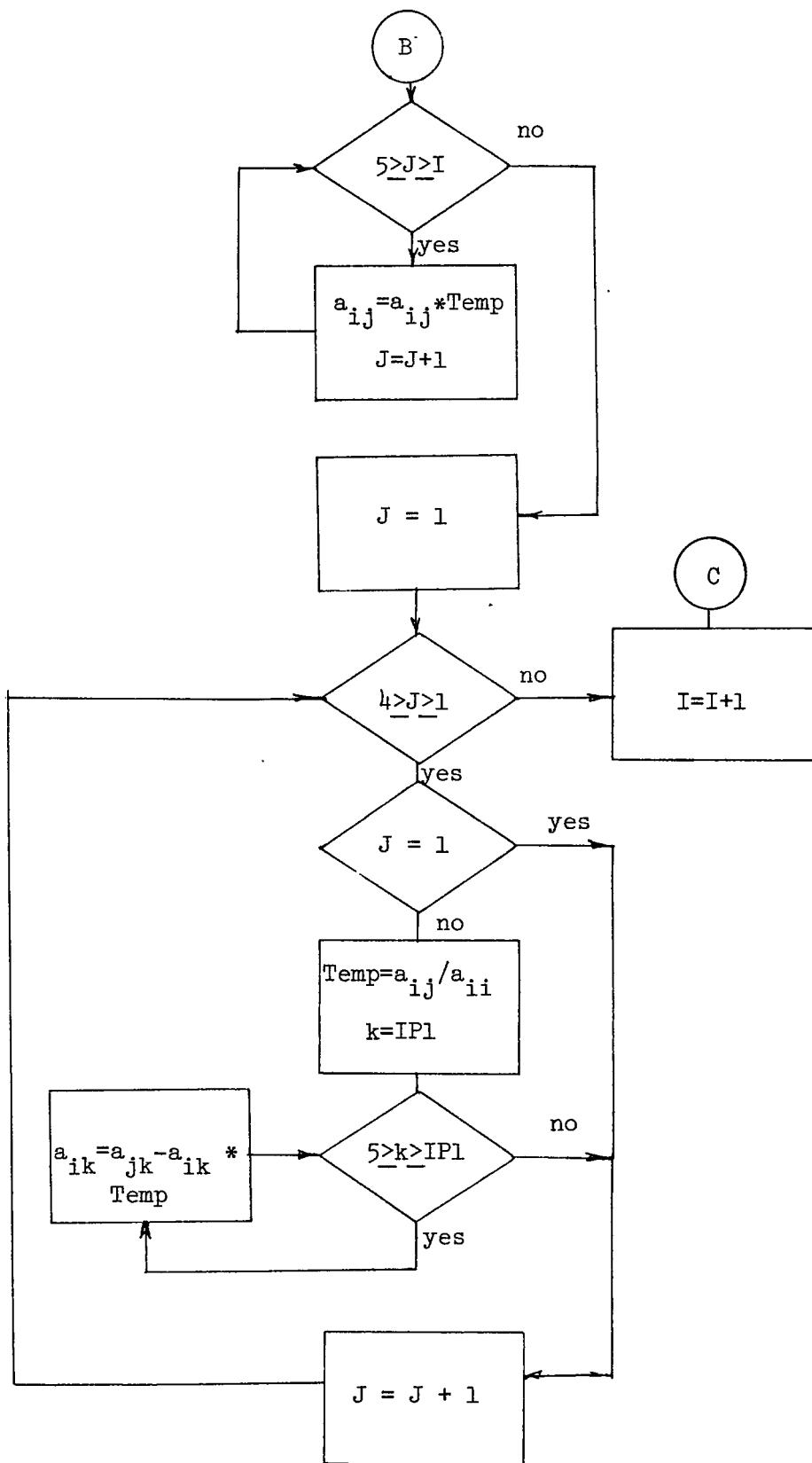


Fig. A.1. (continued)

## APPENDIX B

### User's Manual

#### DFRAP

##### Primary Purpose

The program DFRAP, as listed in Appendix C, has been restricted to deal with symmetrical rigid frames of the portal type. However, the analysis capabilities fall into two categories. The first one is the analysis of symmetrical uniform rigid frames of the portal type. The second capability is the analysis of symmetrical tapered frames of the portal type, when both bases are fixed or pinned.

##### Conventions

All 80 columns of a card can be used for the input of the alphanumerical data. A free-field format has been used to allow free-field input, which employ a standard G type editing code with no fixed fields; in other words, free-field input is a set of external fields separated by commas. The end of the input image is a field end.

##### 1. Problem Initiation

This card must be the first one in the data deck. It can be any alphanumeric name, which has been chosen by the user to identify his problem, and has a maximum length of

70 characters.

Example

JOB # 1. DESIGN OF TAPERED GABLE FRAME.

2. Problem Type

This command is used to classify the frame members, as tapered or uniform sections.

Example

TAPERED

UNIFORM

or alternatively in abbreviation form.

TAP

UNIF

3. Units

The following are the only available units for the system:

Length unit:           INCHES, FT., CM, or METERS

Force unit:           LB., KIPS, KG., or metric TONS

Angular unit:          DEGREES. (This unit does not  
                          need to be specified.)

Example

LB   FT

The user must specify the units otherwise the system will run

terminated.

#### 4. Frame Dimensions Input

To input the dimensions of the frame, four variables must be specified at this stage. These variables are the span, the eave height, the degrees of pitch, and the frame spacing. The input record must be in the following format and order:

SPAN, EAVE HEIGHT, DEGREE OF PITCH, AND FRAME SPACING.

##### Example

72. , 33. , 15. , 24.

This record describes a frame with 72.0 length units span, 33.0 length units eave height, 15.0 degrees of pitch and 24.0 length units spacing of frames.

#### 5. Working Load Data Input

The working load is the uniform vertical loading intensity measured in the chosen units.

##### Example

10.0

Which describes the intensity of 10.0 force units per square length units.

#### 6. Wind Forces Input

These forces are normal to the covering and optional. To input these forces the user will be asked if there are

any pressures normal to covering; the answer must be YES or NO. According to the answer the control will be transferred to the next step e.g.: YES--so the next record has to specify four normal surface pressures for the left-hand column, left-hand rafter, right-hand rafter and right-hand column respectively.

Example

24. , 6.289 , 0.0 , 0.0

#### 7. Vertical Concentrated Apex Load and Load Factor Input

Example

0.0 , 1.4

This record will specify a zero concentrated apex load and 1.4 load factor.

#### 8. Concentrated Loads Other Than the Apex Load Input

If there are any concentrated loads other than the apex load the user must input YES. Otherwise he has to input NO. Then the user will be asked if there are any horizontal concentrated loads on the left-hand column. The answer must be YES or NO. If the answer is YES, the user has to answer the next question: How many? And on the next cards the magnitude and the point of application of these loads must be given.

**Example**

|        |        |
|--------|--------|
| 3      |        |
| 500.0  | , 2.0  |
| 1000.0 | , 6.0  |
| 1200.0 | , 15.0 |

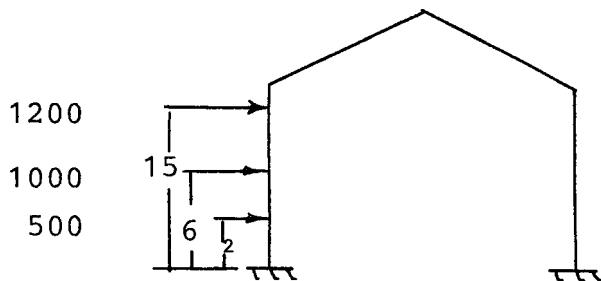


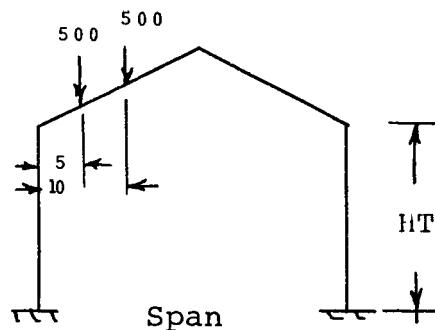
Fig. B.1. Horizontal concentrated loads on left-hand column.

The same procedure is applicable for any horizontal load applied on the right-hand column with a positive sign if acting inwards.

**9. Concentrated Vertical Loads Acting On the Rafter Input**

The user will be asked if there is any vertical loading on the left-hand rafter, if the answer is YES the user will be asked: How many? On the next cards the magnitude and the point of application of these loads must be given.

Example, Fig. B.2



YES

2

500.0 , 5.0

500.0 , 10.0

Fig. B.2. Vertical concentrated loads on left-hand rafter.

The same procedure is applicable if there is any concentrated vertical loads on the right-hand rafter.

#### 10. Gantry Crane Loads Input

The user will be asked if there are any gantry crane loads. If the answer is YES, the user has to input the height of the crane rails and the offset on the first record, the magnitude of vertical reactants of both left and right-hand columns respectively on the following record. On the third record he has to input the horizontal reactants of the gantry crane rails on left and right-hand column respectively.

Example, Fig. B.3

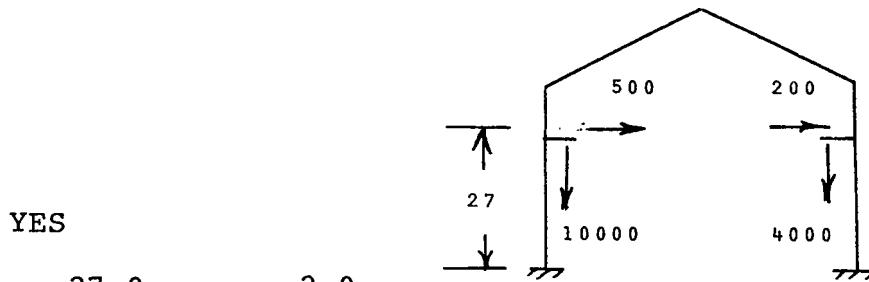


Fig. B.3. Gantry crane loads.

11. At this stage, the user will be asked if he wants to see the twin cantilever bending moment distribution. The answer must be YES or NO.

#### 12. Input the Base Supports

The user has to specify the bases condition as FIXED or PINNED. The left-hand base first, each base on a separate record.

Example

FIXED , or PINNED

or alternatively in abbreviation form:

FIX , or PIN.

13. At this stage the computer, if the frame type is tapered, will read one record. This record has to specify the section depth at four locations, at the column base and

top, the rafter near the column and the rafter near the apex respectively. This record also has to specify the thickness of the web and the flange, the width of the flange, and the specified minimum yield stress of the type of steel used ( $F_y$  in ksi), as used in the AISC specification.

Example, Fig. B.4.

D1 , D2 , D3 , D4 , w , t ,  $b_f$  ,  $F_y$

This record can be written on more than one record if needed. This can be done by skipping the rest of the card. The units must be the same as specified above at stage three except that  $F_y$  must be in ksi.

#### 14. Nomination of Plastic Hinges Positions

The nomination of two plastic hinges is required if both bases are PINNED, or four positions have to be nominated if both bases are FIXED. These positions have to be assigned an integer constant with sign which serves to indicate the sign at the assumed hinge.

#### 15. The Bases Reactions

If the user wants to see the bases reactions he has to input YES; otherwise he has to answer NO.

16. At this stage the user will be asked if he wants to try another hinge pattern. The answer must be YES or NO; if it is YES, he has to repeat step 14; on the other hand,

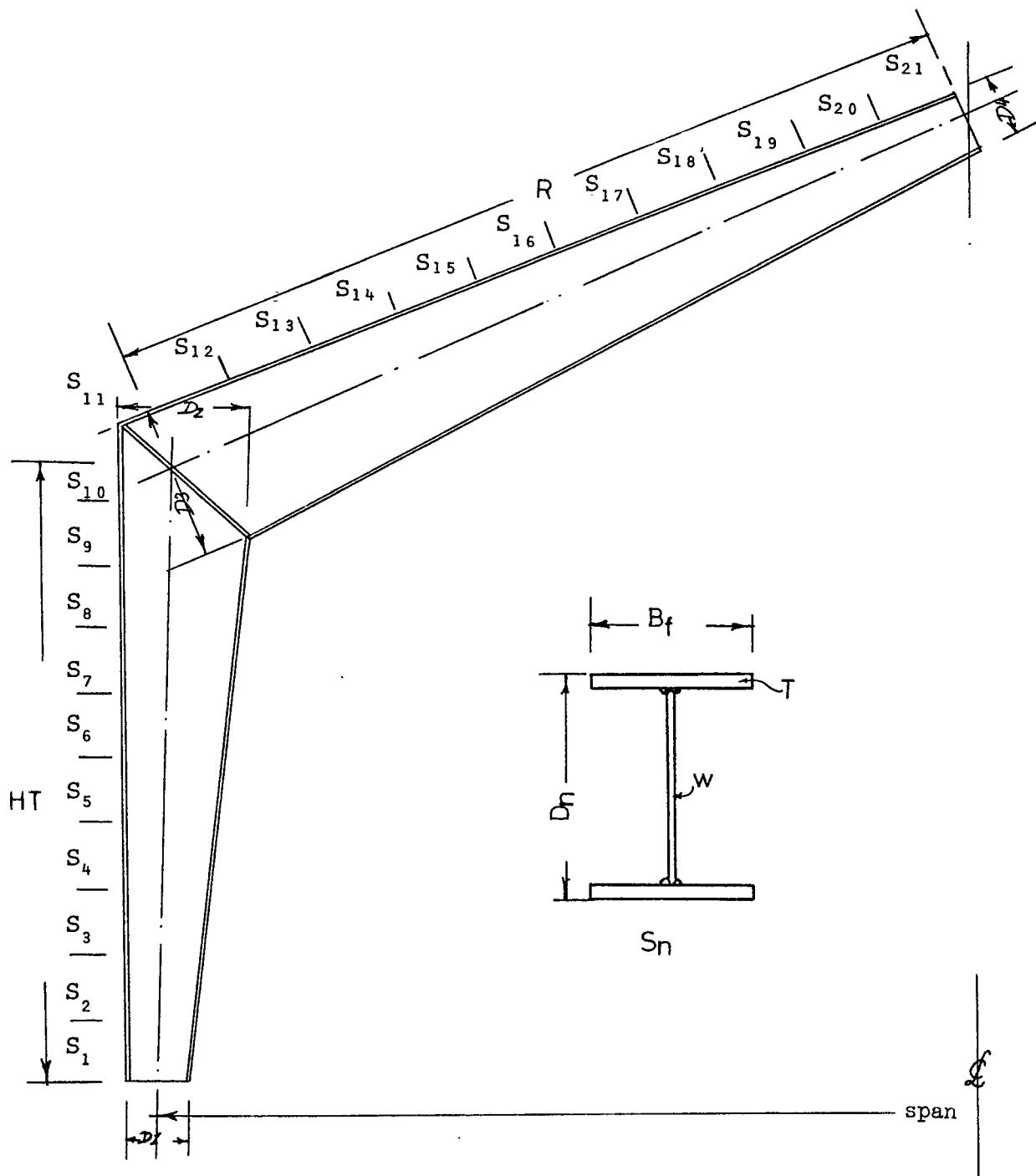


Fig. B.4. Representation of tapered gable frame sections.

if the answer is NO, the operator will transfer the control to step 17.

17. The user at this stage will be asked if he has another frame to be considered. If YES it will transfer the control to step one above--otherwise the computer run will stop.

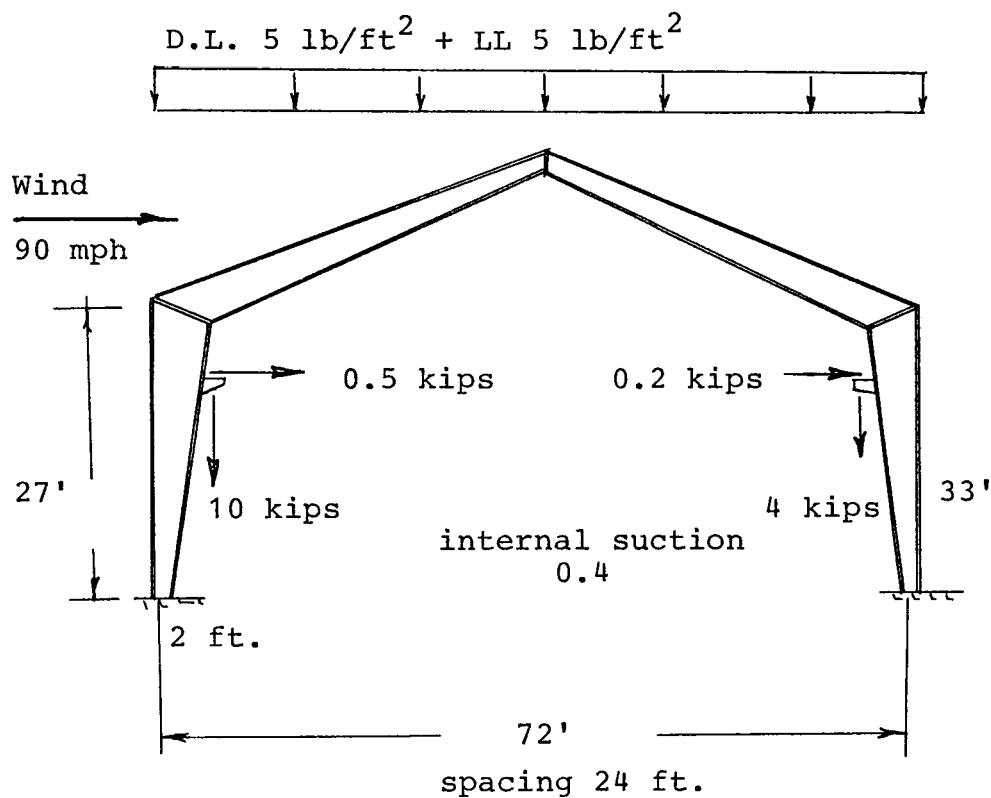


Fig. B.5. Plastic design example.

The solution for this problem is presented in Appendix C.

```

1*      DIMENSION DTMS(41), REAS(41), REAK(41), FQ(4,5), SH(41), TH(41)
2*      COMMON MPN(41), RESM(41), ANS(41), H(80), THRUST(41)
3*      2, SHEAR(41)
4*      REAL MPN
5*      INTEGER RBAS, AHS, YLS, PIN, FIX, ERR, FORCE, CM, TON, FT
6*      1, TAP, UNIF, ASTR
7*      DATA YES/'YES'/, NO/'NO'/, ERR/'ERR'/, CM/'CM'/, TON/'TON'/
8*      1, PIN/'PINNED'/, FIX/'FIXED'/, FT/'FT'/, KIPS/'KIPS'/, LB/'LB'/
9*      2, TAP/'TAP'/, UNIF/'UNIF'/
10*     3, ASTR/'***'/, BLANK/' '/
11*    1 FORMAT(   )
12*    2 FORMAT( 8UA1)
13*    3 FORMAT( T16, 4G(***))
14*    5 WRITE(6,3)
15*    6 WRITE(6,10)
16*    7 WRITE(6,3)
17*    7 FORMAT(//1X,72(''),/1X,'*',70A1,'*',/1X, 72('')//)
18*    8 FORMAT(3X,A4//)
19*    10 FORMAT( T16,'* PLASTIC DESIGN OF INDUSTRIAL RIGID FRAMES **')
20* C*****C
21* C PROBLEM INITIATION C
22* C*****C
23*    9 WRITE(6,16)
24*    16 FORMAT(* PROBLEM INITIATION*)
25*    READ (5,2) M
26*    DO 9 K1=1,80
27*    9 IF(M(K1).NE.BLANK) GO TO 13
28*    13 WRITE(6,7) (H(K2),K2=K1,70)
29*    17 WRITE(6,17)
30*    17 FORMAT(* TYPE OF MEMBER SAY TAPERED,OR UNIFORM*)
31*    CALL SUB1 (LTYF)
32*    11 WRITE(6,11) LTYF
33*    11 FORMAT(// * TYPE *, A5//)
34*    18 WRITE(6,18)
35*    18 FORMAT(* INPUT THE SPECIFIED UNITS ,LENGTH AND FORCE*)
36*    CALL UNITS(LENGTH,FORCE,CF,CL)
37* C*****C
38* C SPAN ,LEAVES HEIGHT,(FIGURE OF PITCH AND FRAME SPACING C
39* C INPUT THE FRAME DIMENSIONS C
40* C*****C
41*    19 WRITE(6,19)

```

## THE COMPUTER PROGRAM DFRAP

## APPENDIX C

```

42*      19  FORMAT(1* INPUT THE FRAME DIMENSIONS SPAN,EAVES HEIGHT,DEGREE OF PITCH AND
43*          2 FRAMES SPACING//)
44*          READ(5,1)      SPAN,HT,THETA,BAY
45*          WRITE(6,15)   SPAN,LENGTH, HT,LENGTH,THETA ,BAY,LENGTH
46*      15  FORMAT(1*     SPAN           =*,G15.5,A4 , ///
47*                  EAVS HEIGHT      =*,G15.5, A4 , // /
48*                  DEGREE OF PITCH    =*,G15.5,*DEGREE* , ///
49*                  FRAME SPACING     =*,G15.5, A4 //)
50*          WRITE(6,21)
51*      21  FORMAT(1* INPUT THE WORKING LOAD DATA VERTICAL UNIFORM LOADING INTENSITY*)
52*          C*****C*****C*****C*****C*****C*****C*****C*****C*****C*****C
53*          C   INPUT THE WORKING LOAD DATA - VERTICAL UNIFORM LOADING INTENSITY, C
54*          C*****C*****C*****C*****C*****C*****C*****C*****C*****C*****C
55*          READ(5,1)  WSFET
56*          WRITE(6,26) WSFET , FORCE,LENGTH
57*      26  FORMAT(1*  WORKING LOAD//,/,* VERTICAL UNIFORM LOADING INTENSIT
58*          IY = *,G15.5,A4,* /SQUARE *, A4//)
59*          C*****C*****C*****C*****C*****C*****C*****C*****C*****C
60*          C  ARE THERE PRESSURES NORMAL TO COVERING ?      C
61*          C*****C*****C*****C*****C*****C*****C*****C*****C*****C
62*          WRITE(6,22)
63*      22  FORMAT(1*  ARE THERE PRESSURES NORMAL TO COVERING ? //)
64*          CALL SUR1(NUM)
65*          IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
66*          WRITE(6,8) NUM
67*          IF (NUM=YES) 41,30,41
68*          C*****C*****C*****C*****C*****C*****C*****C*****C*****C
69*          C  INPUT 4 NORMAL SURFACE PRESSURES FOR THE LEFT-HAND COLUMN,LEFT-HAND
70*          C RAFTER,RIGHT-HAND RAFTER AND RIGHT-HAND COLUMN WITH PRESSURE POSITIVE
71*          C IF ACTING INWARDS ON THE BUILDING.          C
72*          C*****C*****C*****C*****C*****C*****C*****C*****C*****C
73*          30  WRITE(6,31)
74*      31  FORMAT(1* INPUT FOUR NORMAL SURFACE PRESSURES FOR THE /*,
75*          2 LEFT-HAND COLUMN,LEFT-HAND RAFTER,RIGHT-HAND RAFTER//,
76*          3* AND RIGHT-HAND COLUMN WITH PRESSURE POSITIVE IF*/,
77*          4 * ACTING INWARDS ON THE BUILDING*/)
78*          READ(5,1) AAA,BBB,CCC,DDD
79*          WRITE(6,40) AAA,FORCE,LENGTH,BBB,FORCE,LENGTH,CCC,FORCE,LENGTH,
80*          DDD,FORCE,LENGTH
81*      40  FORMAT(1* PRESSURE ON THE LEFT-HAND COLUMN =*,G15.5,A4,* /SQUARE *
82*          1*A4,/*/* PRESSURE ON THE LEFT-HAND RAFTER =*,G15.5, A4,* /SQUARE *
83*          2 ,A4,/*/* PRESSURE ON THE RIGHT-HAND RAFTER=*,G15.5,A4,* /SQUARE *
84*          3*A4,/*/* PRESSURE ON THE RIGHT-HAND COLUMN=*,G15.5,A4,* /SQUARE *
85*          4 ,A4//)

```

```

88*      C*****C*****C*****C*****C*****C*****C
89*      C VERTICAL CONCENTRATED APLX LOAD AND LOAD FACTOR   C
90*      C*****C*****C*****C*****C*****C*****C
91*      WRITE(6,39)
92*      39 FORMAT(' INPUT VERTICAL CONCENTRATED APEX LOAD AND LOAD FACTOR')/
93*      41 RLAT(5,1) PP,Z
94*      WRITE(6,44) PR,FORCE,Z
95*      44 FORMAT(' CONCENTRATED APEX LOAD =',G15.5,A4,'//'
96*      ' LOAD FACTOR=',G15.5 '///')
97*      W=RAY*Z*WS*FT
98*      PR=PP*Z
99*      VL=U*E
100*      VR=O*U
101*      HL= .0
102*      HR=O*U
103*      DO 45 I=1,41
104*      S(I)=U*U
105*      T(I)=U*U
106*      45 DTMS(I)=O*U
107*      A=RAY*Z*RAY
108*      ANGLE=PI*FT*3.14159265/18.0*L
109*      C=CCC*Z*RAY
110*      D=IND*Z*RAY
111*      E=BFB*Z*RAY
112*      R=(SPAN*SIN(ANGLE)/COS(ANGLE))*0.5
113*      DRH=C*R*HT
114*      NZ=1*SPAN*0.5
115*      NQ=Z*COS(ANGLE)
116*      NB=Z*SIN(ANGLE)
117*      L=V*SPAN**2/8*L
118*      PR1=U.5*PP
119*      PRZ=PP1*COS(ANGLE)
120*      PR3=PP1*SIN(ANGLE)
121*      CRH=C*R*HT
122*      SLC=1.0/COS(ANGLE)
123*      CSC=SIN(ANGLE)
124*      Q=(SPAN*SLC)**2/8*
125*      NZ=HT**2/2.
126*      C*****C*****C*****C*****C*****C
127*      C ARE THERE ANY OTHER CONCENTRATED LOADS?   C
128*      C*****C*****C*****C*****C*****C
129*      WRITE(6,47)
130*      47 FORMAT(' ARE THERE ANY OTHER CONCENTRATED LOADS? ')///
131*      CALL SUB1(NUM)
132*      IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
133*      WRITE(6,B) NUM
134*      IF (NUM.YES) 49,850,49

```

```

135*      49  DU 5U I=1,10
136*      AK=FLOAT(1)/10.
137*      J1=11-I
138*      DTMS(J1)=DTMS(J1)-R8-b*u-PRH*AK-HZ*AK**2-PP*   SPAN/4.0
139*      TH(J1)=TH(J1)-'2 -3*D.5*SPAN -PP1
140*      SH(J1)=SH(J1)+4*D.5*SEC*COSC*SPAN+A*AK*HT
141*      KEAS(J1)=SPAN/2.
142*      KLEAR(J1)=K+AK*HT
143*      J2=21-I
144*      DTMS(J2)=DTMS(J2)-(1.8+4*B)*AK**2*u-PP*AK*SPAN/4.0
145*      KEAS(J2)=AK*SPAN/2*u
146*      KEAS(J2)=K*AK
147*      SH(J2)=SH(J2)+4*AK +PP2 +B*AK*D.5*SPAN *SEC
148*      TH(J2)=TH(J2)-A*AK-PP3.
149*      J3=21+I
150*      TH(J3)=TH(J3)-.6*AK -PP3
151*      SH(J3)=SH(J3)-4*AK -PP2 -C*AK -SPAN*SEC*u.5
152*      DTMS(J3)=DTMS(J3)-(1.8+6*C)*AK**2-PP*AK*SPAN/4.0
153*      KEAS(J3)=-AK*SPAN/2.
154*      KLEAR(J3)=K*AK
155*      J4=31+I
156*      TH(J4)=TH(J4)-2*C*SPAN*u.5 -PP1
157*      SH(J4)=SH(J4) -C*SPAN*D.5*SEC*COSC -AK*HT *D
158*      DTMS(J4)=DTMS(J4)-W8-W*C-CRH*AK-HZ*D*AK**2-PP*SPAN/4.
159*      KEAS(J4)=-SPAN/2*
160*      50  KLEAR(J4)=K+AK*HT
161*      SH(21)=SH(21)+ PP2
162*      TH(21)=TH(21)-PP3
163*      C*****C
164*      C  GANTRY CRANE LOADS  C
165*      C*****C
166*      WRITE(6,51)
167*      51  FORMAT(' ARE THERE ANY GANTRY CRANE LOADS? '///)
168*      CALL SUB1(NUM)
169*      IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
170*      WRITE(6,8) NUM
171*      IF (NUM =YES) 52 ,710 ,52
172*      C*****C
173*      C  ASK IF YOU WANT TO SEE THE TWIN CANTILEVER MOMENT C
174*      C*****C
175*      52  WRITE(6,53)
176*      53  FORMAT (' DO YOU WANT TO SEE THE TWIN CANTILEVER MOMENT? '///)
177*      CALL SUB1(NUM)
178*      IF (NUM.NE. N1.AND.NUM.NE.YES) GO TO 475
179*      WRITE(6,8) NUM
180*      IF (NUM = YES ) 62,55,62

```

```

181*      C*****C*****C*****C*****C*****C
182*      C PRINT THE TWIN CANTILEVER MOMENTS   C
183*      C*****C*****C*****C*****C*****C
184*      55  WRITE(6,57)
185*      57  FORMAT (' THE TWIN CANTILEVER MOMENT SHEAR AND THRUST : //',1X,
186*           155(' -'),/ ' SECTION NO.',T2G,'MOMENT',T33,'SHEAR',T47,'THRUST',
187*           2 /1X,155(' -'))
188*      DO 56 I=1,4
189*      58  WRITE (6,61) 1,DTMS(I),SH(I),TH(I)
190*      61  FORMAT (17,T18,G10.5,T30,G10.5,T45,G10.5 )
191*      62  CONTINUE
192*      C*****C*****C*****C*****C*****C
193*      C WHAT KIND OF BASES ?   C
194*      C*****C*****C*****C*****C
195*      65  WRITE(6,65)
196*      65  FORMAT (55(' -'),//,
197*           ' WHAT KIND OF BASES? //, * INPUT FIXED OR PINNED ,LEFT-HAND
198*           1 BASE FIRST'//)
199*      CALL SUB1(RBAS)
200*      IF( RBAS.NE.FIX.AND.RBAS.NE. PIN) GO TO 475
201*      CALL SUB1(LBAS)
202*      IF( LBAS.NE.FIX.AND.LBAS.NE. PIN) GO TO 475
203*      WRITE(6,70) LBAS,RBAS
204*      70  FORMAT(
205*           1 ' LEFT-HAND BASE IS ',A6, // * RIGHT-HAND BASE IS ',A6//)
206*      IF( LTYP.EQ.UNIF) GO TO 80
207*      WRITE(6,73)
208*      73  FORMAT(* INPUT THE SECTION DIMENSIONS,D1,D2,D3,D4,W,T,BF,FY*)
209*      CALL SURZISEC,HT,SPAN,LENGTH,CF,CL,D1,D2,D3,D4,W,T,SEGMA,      BF,
210*           1 ,RBAS,LBAS,COS1,COS2,FY)
211*      80  IF(RBAS .EQ. PIN .AND. LBAS .EQ. PIN) GU TO 100
212*      IF(RBAS .EQ. FIX .AND. LBAS .EQ. FIX) GU TO 150
213*      C*****C*****C*****C*****C
214*      C ANALYSIS FOR PIN BASES   C
215*      C*****C*****C*****C*****C
216*      100  DO 110 I=1,4
217*      110  EW(I,1)=1.0
218*             EW( 1,2)=REAR(1)
219*             EW( 1,3)=REAS(1)
220*             EW( 1,4)=0.0
221*             EW( 1,5)=-DTMS(1)
222*             EW(2,2)=REAR(41)
223*             EW(2,4)= 0.0
224*             EW(2,3)=REAS(41)
225*             EW(2,5)=-UTMS(41)
226*      C*****C*****C*****C*****C*****C
227*      C NOMINATE POSITIONS OF TWO PLASTIC HINGES WITH SIGN   C

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```

229*
230*
231*
232*   115 FORMAT(' THE TWO NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN
233*   1ANG ',Z15//)
234*   JH1=IA55(1#1)
235*   JH2=IA55(1#2)
236*   Lw(3,2)=REAK(JH1)
237*   Ew(3,3)=REAS(JH1)
238*   Ew(3,5)=-UTMS(JH1)
239*   Ew(4,2)=REAK(JH2)
240*   Ew(4,3)=REAS(JH2)
241*   Ew(4,5)=-UTMS(JH2)
242*   IF(LTYP .EQ. UNIF) GO TO 135
243*   120 Lw(3,4)=MPN(JH1)
244*   Ew(4,4)=-IPN(JH2)
245*   GU TO 260
246*   130 Ew(3,4)=-MPN(JH1)
247*   Ew(4,4)=MPN(JH2)
248*   135 IF (IH1) 140,146,145
249*   140 Ew(3,4)=1
250*   Ew(3,4)=-1
251*   Ew(4,4)=-1
252*   145 Ew(3,4)=-1
253*   Ew(4,4)=1
254*   GU TO 260
255*
256* C ANALYSTS FOR FIXED BASES.          C
257*
258* C*****C
259* C NOMINATE POSITION OF FOUR PLASTIC HINGES      C
260* C*****C
261*   150 WRITE(6,174)
262*   174 FORMAT(' NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO IND
263*   ICATE THE MOMENT SIGN//')
264*   READ (5,1) IH1,IH2,IH3,IH4
265*   WRITE (6,155) IH1,IH2,IH3,IH4
266*   155 FORMAT(' THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN
267*   1'//,', TO INDICATE THE MOMENT SIGN ARE:'//,,4I5//')
268*   JH1=IABS(IH1)
269*   JH2=IABS(IH2)
270*   JH3=IABS(IH3)
271*   JH4=IABS(IH4)
272*   DU 160 I=1,4
273*   160 Lw(1,1)=1.0

```

|      |                                   |
|------|-----------------------------------|
| 274* | $E_u(1,2) = REAR(JH1)$            |
| 275* | $E_u(1,3) = REAS(JH1)$            |
| 276* | $E_u(1,5) = -UTS(JH1)$            |
| 277* | $E_u(2,2) = REAR(JH2)$            |
| 278* | $E_u(2,3) = REAS(JH2)$            |
| 279* | $E_u(2,5) = -UTS(JH2)$            |
| 280* | $E_u(3,2) = -TR(JH3)$             |
| 281* | $E_u(3,4) = REAS(JH3)$            |
| 282* | $E_u(3,5) = -UTS(JH3)$            |
| 283* | $E_u(4,2) = REAR(JH4)$            |
| 284* | $E_u(4,3) = -CAS(JH4)$            |
| 285* | $E_u(4,5) = -UTS(JH4)$            |
| 286* | IF (11YP.F .EQ. 0) THEN GO TO 271 |
| 287* | 11(IH1) 17 , 17 , 16.             |
| 288* | 170 $E_u(1,4) = MPH(JH1)$         |
| 289* | GU TO 190                         |
| 290* | 180 $E_u(1,4) = -MPH(JH1)$        |
| 291* | 190 1F(IH2) 200,200,195           |
| 292* | 195 $E_u(2,4) = -MPN(JH2)$        |
| 293* | GU TO 220                         |
| 294* | 200 $E_u(2,4) = MPH(JH2)$         |
| 295* | 220 1F(IH3) 230,230,240           |
| 296* | 230 $E_u(3,4) = MPH(JH3)$         |
| 297* | GU TO 250                         |
| 298* | 240 $E_u(3,4) = -MPN(JH3)$        |
| 299* | 250 1F(IH4) 260,260,270           |
| 300* | 260 $E_u(4,4) = MPH(JH4)$         |
| 301* | GU TO 240                         |
| 302* | 270 $E_u(4,4) = -MPN(JH4)$        |
| 303* | GU TO 280                         |
| 304* | 271 1F(IH1) 272,272,273           |
| 305* | 272 $E_u(1,4) = 1$                |
| 306* | GU TO 274                         |
| 307* | 273 $E_u(1,4) = -1$               |
| 308* | 274 1F(IH2) 275,275,276           |
| 309* | 275 $E_u(2,4) = 1$                |
| 310* | GU TO 277                         |
| 311* | 276 $E_u(2,4) = -1$               |
| 312* | 277 1F(IH3) 276,276,279           |
| 313* | 278 $E_u(3,4) = 1.0$              |
| 314* | GU TO 281                         |
| 315* | 279 $E_u(3,4) = -1.0$             |
| 316* | 281 1F(IH4) 282,282,283           |
| 317* | 282 $E_u(4,4) = 1.0$              |
| 318* | GU TO 286                         |
| 319* | 283 $E_u(4,4) = -1.0$             |

```

320*      280  DO 390 I=1,4
321*          IF I=I+1
322*              TEMP = ABS(EQ(I,I))
323*          K=1
324*          DO 300 J=1,4
325*              IF (AHS(EU(J,I))-TEMP) .GT. 300,300,290
326*      290  K=J
327*          TEMP=AHS(EQ(J,J))
328*      300  CONTINUE
329*          IF (K -1) 310, 330,310
330*      310  DO 320 J=1,5
331*              TEMP =EV(I,J)
332*              EU(I,J) =EU(K,J)
333*      320  EU(K,J)=TEMP
334*      330  IF (EW(I,I)) 340,600, 350
335*      350  TEMP=1.0 /EW(I,I)
336*          DO 360 J=1,5
337*      360  EU(I,J)=EV(I,J)*TEMP
338*      370  DO 390 J=1,4
339*          IF (I-J) 370,390,370
340*      370  TEMP =EV(J,I)
341*          DO 380 K=1P1,5
342*      380  EJ(J,K)=EV(J,K)-TEMP*EV(I,K)
343*      390  CONTINUE
344*          DO 395 I= 1,10
345*              JZ=21-I
346*              JS=32-I
347*              J4= 42-I
348*              SH(I)= SH(I)-EV(Z,5)
349*              TH(I)= TH(I)+EV(3,5)
350*              SHEAR(J2)= SH(J2)-EV(3,5)/SEC-EV(Z,5)*COSC
351*              THRUST(J2)= TH(J2)+EV(3,5)*COSC-EV(Z,5)/SEC
352*              SHEAR(J3)= SH(J3)+EV(3,5)/SEC+EV(Z,5)*COSC
353*              THRUST(J3)= TH(J3)-EV(3,5)*COSC-EV(Z,5)/SEC
354*              SHEAR(J4)= SH(J4)+EV(3,5)
355*              THRUST(J4)= TH(J4)-EV(3,5)
356*      395  CONTINUE
357*      WRITE (6,396)
358*      396  FORMAT(1X,55(''),/* SECTION NO. *,7A, * SHEAR *, BX,* THRUST*
359*      1 ,/, 55(''))
360*              SHEAR(21)=SH(21)+EV(3,5)/SEC+EV(Z,5)*COSC
361*              THRUST(21)=TH(Z1)+EV(3,5)*COSC-EV(Z,5)/SEC
362*              DO 397 I= 1,4
363*      397  WRITE (6,398) I, SFAM(I), THRUST(I)
364*      398  FORMAT( 5X,15,5X,20I5*4)
365*      WRITE(6,399)

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360*      399  FORMAT(1X,55(''')/
367*          DU 420  I=1,41
368*          420  RL5M(I)=DTMS(I)+EM(1,5)+REAR(I)*LU(2,5)+REAS(I)*EN(3,5)
369*  C*****+
370*  C      LOCATION HENDING MOMENT  MT>NP   L
371*  C*****+
372*      WRITE (6,222)
373*      222  FORMAT(///' LOCATION OF BEARING MOMENT MT >MPC:  //',1X,60('''),/
374*      ' SECTION NO. ', 133,' PLASTIC MOMENT CAPACITY ',/,+1A,60('''))
375*      IF (LTYP .EQ. 0) IF ( ) GO TO 419
376*      CALL      SUB3(CUS1,CUS2,D1,D2,D3,D4,,BF,T,SEGMA,RBAS,LBAS,FY)
377*      GU TO 412
378*      419  ANPNE= EN(4,5)
379*      430  FORMAT (//2X,T53,U5.5,A3,A5,10,A5//)
380*      WRITE (6,430)  AMPM    ,FORCE,LENGTH
381*      DU 409  I=1,41
382*      IF (ABS(LRESM(I))-0.79*ANPN) 406,405,405
383*      405  IF (ABS(LRESM(I))-1.01*ANPN) 407,408,408
384*      406  ANS(I)= NO
385*      GU TO 411
386*      407  ANS(I)=ASTF
387*      GU TO 411
388*      408  ANS(I)=YES
389*      IF (LBAS.EQ. PIN) ANS(I)=NO
390*      IF (RBAS.EQ. PIN) ANS(41)=NO
391*      411  WRITE(6,421),  RESM(I),ANS(I)
392*      419  CONTINUE
393*      GU TO 425
394*      412  DU 410  I=1,41
395*      410  WRITE(6,421) I,RFSM(I),ANS(I)  ,MPH(I)
396*      421  FORMAT(15,SX,G12.3,4X,A3,1GX ,G15.3)
397*  C*****+
398*  C      BASE REACTIONS
399*  C*****+
400*      425  WRITE(6,423)
401*      423  FORMAT (55('''),///
402*      1 * DO YOU WANT TO SEE THE BASE REACTIONS?  *///
403*      CALL SUR1(NUM)
404*      IF (NUM.NE. 1.0+ANU+NUM+1.0+YES) GO TO 475
405*      WRITE(6,8) NUM
406*      IF (NUM-YES)440,440,440
407*      426  VA=PP/2.0+(A+B)*SPAN/2.0-FW(3,5)+VL+SVL
408*      VE=PP/2.0+(B+C)*SPAN/2.0 +VR+SVR +EG(3,5)
409*      HE=EL+HT+C+ER(2,5)+HR+SIR
410*      HA=PA+HT +B+ER(2,5)+HL+SIL
411*      VLE (6,427)  VA,,,VE,HE
412*      427  FORMAT(14H V1=,,4.4,2X,5H H1=,,6.4,2X,
413*              15H V41=,,4.4,2X,5H H41=,,6.4//)

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414*      440 WRITE(6,450)
415*      450 FORMAT(* DO YOU WANT TO TRY ANOTHER HINGE PATTERN? /**)
416*          CALL SU(1)(UM)
417*          IF (NUM.EQ.0) GO TO 470
418*          WRITE(6,*) 'NO'
419*          IF (NUM.EQ.1) GO TO 460
420*          WRITE(6,460)
421*          C ASK IF THERE IS ANOTHER FRAME
422*          C
423*          460 WRITE(6,470)
424*          CALL SU(1)(UM)
425*          470 FORMAT(* DO YOU WANT TO ANALYSE ANOTHER FRAME? /**)
426*          WRITE(6,8) NUM
427*          IF (NUM.NE.0) AND (NUM.NE.1) GO TO 475
428*          IF (NUM.EQ.1) GO TO 475
429*          475 READ(5,475)(T(I),I=1,40)
430*          480 FORMAT(RA1)
431*          WRITE(6,490)(M(I),I=1,80)
432*          490 FORMAT(* THIS CAN'T/IX,8RA1,/ * DOES NOT CORRESPOND TO INPUT ROW
433*          LINE*,/5(T15,B10*,ARNING*))
434*          500 STOP
435*          600 WRITE(6,650)
436*          650 FORMAT(5X,'ZERO DIVISION IN EQUATION')
437*          GO TO 425
438*          710 WRITE(6,660)
439*          660 FORMAT(* INPUT THE HEIGHT OF THE GANTRY CRANE RAILS AND*
440*          1 * THE OFFSET OF THE CRANE RAILS*)
441*          READ(5, 1 ) HC,EC
442*          WRITE(6,720) 1 HC,LENGTH,EC,LENGTH
443*          720 FORMAT(* HEIGHT OF CRANE RAIL =*,G15.5,A4 , // * OFFSET OF CRANE
444*          1 RAIL = *,G15.5 ,A4 //)
445*          C
446*          C INPUT WORKING VALUES OF VERTICAL CRANE FORCES LEFT-HAND COLUMN FIRST
447*          C
448*          WRITE(6,660)
449*          660 FORMAT(* INPUT WORKING VALUES OF VERTICAL GANTRY CRANE FORCES LEF
450*          T-HAND COLUMN FIRST /**)
451*          READ(5, 1 ) VL,VN
452*          WRITE(6, 780) VL,FORCE,VR,FORCE
453*          780 FORMAT(* WORKING VALUES OF VERTICAL CRANE FORCES *//,* WORKING VE
454*          IRtical CRANE FORCE OF LEFT-HAND COLUMN= *,G15.5,A4,/// * WORKING
455*          2 VERTICAL CRANE FORCE OF RIGHT-HAND COLUMN= *,G15.5 , A4 //)
456*          C
457*          C INPUT WORKING VALUES OF HORIZONTAL CRANE FORCES LEFT-HAND COLUMN FIRST
458*          C

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459*      KLAPE (S,I)  HL,HI
460*      WRITE (6,85)  HL,FORCL,FR,FORCE
461*      805 FORMAT(' WORKING VALUES OF HORIZONTAL CRANE FORCES //,')
462*      1HORIZONTAL CRANE FORCE ON LEFT-HAND COLUMN =',G15.5,A4 ',//'
463*      2HORIZONTAL CRANE FORCE ON RIGHT-HAND COLUMN=',G15.5,A4 //')
464*      VL=Z*VL
465*      VX=Z*VR
466*      HL=Z*HL
467*      HI=Z*HR
468*      DU F20.1= 1,11
469*      AK= FLOAT(I)/10. -0.1
470*      HKM=HC-AK*HT
471*      IF (HARM) 820,41, ,810
472*      810  UTMS(I)=DI*S(I)-VL*EC-HL*HARM
473*      SH(I)=SH(I)+PL
474*      TH(I)=TH(I)-VL
475*      820  CONTINUE
476*      DU R40. I=1,11
477*      AK= FLOAT(I)/10. -0.1
478*      HARM=HC-AK*HT
479*      IF (HARM) 840,830,030
480*      830  J=42-I
481*      UTMS(J)= UTMS(I)-VR*EC +HR*HARM
482*      SH(J)=SH(J)+HR
483*      TH(J)=TH(J)-VR
484*      840  CONTINUE
485*      GO TO 52
486*      C*****C*****C*****C*****C*****C*****C*****C*****C
487*      C HORIZONTAL CONCENTRATED LOADS ON COLUMN AND POSITIVE IF ACTING C
488*      C INWARDS. VERTICAL CONCENTRATED LOADS ON RAFTERS ARE POSITIVE IF C
489*      C ACTING DOWNWARDS . C
490*      C*****C*****C*****C*****C*****C*****C*****C*****C
491*      C*****C*****C*****C*****C*****C*****C*****C*****C
492*      C HORIZONTAL LOADS ON LEFT-HAND COLUMN? C
493*      C*****C*****C*****C*****C*****C*****C*****C*****C
494*      845  FORMAT(' ARE THERE ANY HORIZONTAL LOADS ON LEFT-HAND COLUMN? //')
495*      850  WRITE(6,845)
496*      CALL SUB1(NUM)
497*      IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
498*      WRITE(6,8) NUM
499*      IF( NUM=YES)  910,860,910
500*      860  WRITE(6,865)
501*      865  FORMAT(' HOW MANY? //')
502*      READ(5,1) NPL
503*      WRITE(6,1)NPL

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504*      SHL=0.0
505*      WRITE(6,924)
506*      DU 905 I=1,NPL
507*      READ(S, 1) SP,HP
508*      WRITE(6,871)SP,FORCE,HP , LENGTH
509*      870 FORMAT(' THE MAGNITUDE OF THE FORCE=',G10.5,A6, //'* THE POINT OF*
510*      *APPLICATION=',G15.5,A6//')
511*      SHZ=SP
512*      SHL=SHL+SP
513*      DU 906 J=1,11
514*      AK=FLOAT(J)/10.0-U+1
515*      HARM=HP-AK*HT
516*      IF(HARM)906,890,840
517*      890 DTMS(J)=DTMS(J)-SP*HARM ,
518*      SH(J)= SH(J)+SP
519*      900 CONTNU
520*      ****
521*      C   HORIZONTAL LOADS ON RIGHT-HAND COLUMN   C
522*      ****
523*      910 WRITE(6,915)
524*      915 FORMAT(' ARE THERE ANY HORIZONTAL LOADS ON RIGHT-HAND COLUMN? //')
525*      CALL SUB1(NUM)
526*      IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
527*      WRITE(6,8) NUM
528*      IF (NUM=YES)950,920,950
529*      920 WRITE(6,925)
530*      925 FORMAT(' HOW MANY?//')
531*      READ(5,1)NPR
532*      WRITE(6,1)NPR
533*      SHR=U.0
534*      924 FORMAT (' INPUT THE FORCE AND THE POINT OF APPLICATION')
535*      WRITE(6,924)
536*      DU 940 I=1,NPR
537*      READ(5,1) SP,HP

538*      WRITE(6,870)SP,FORCE,HP,LENGTH
539*      SP=2*SP
540*      SHR=SHR+SP
541*      DU 940 J=1,11
542*      AK=FLOAT(J)/10.0-U+1
543*      HARM =HP-AK*HT
544*      IF(HARM)940,930,930
545*      930 K=42-J
546*      DTMS(K)=DTMS(K)+SP*HARM
547*      SH(J)= SH(J)-SP
548*      940 CONTNU
549*      ****
550*      C   VERTICAL LOADS ON LEFT-HAND RAFTER   C
551*      ****

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552*      950  WRITE(6,955)
553*      955  FORMAT(' ARE THERE ANY HORIZONTAL LOADS ON LEFT-HAND RAFTER ? //')
554*      CALL SUB1(NUM)
555*      IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
556*      WRITE(6,B) NUM
557*      IF (NUM=YES) 1000,963,1000
558*      960  WRITE(6,925)
559*      READ(S,1)NPL
560*      WRITE(6,1)NPL
561*      SVL=U.J
562*      WRITE(6,924)
563*      DO 995 J=1,NPL
564*      READ (S, 1 )SP,HP
565*      WRITE(6,87)SP,FORCE,HP,LENGTH
566*      SP=2*SP
567*      SVL=SVL+S.
568*      HP=ABS(HP)
569*      DO 995 J=1,11
570*      AK=FLOAT(J)/10.0+0.1
571*      K=10+J
572*      HARM=SPAN/2.0-HP-AK*SPAN/2.0
573*      IF (HARM)997,YPL,YPL
574*      980  DTMS(K)= DTMS(K)-SP*(SPAN*0.5-HP)
575*      TH(K)=TH(K)-SP*COSC
576*      SH(K)=SH(K)+SP/SEC
577*      990  CUNTINUE
578*      DO 995 J=1,13
579*      995  DTMS(J)=DTMS(J)-SP*(SPAN/2.0-HP)
580*      DO 996 J=1,10
581*      996  TH(J)= TH(J)-SVL
582*      1000  WRITE(6,997)
583*      997  FORMAT(' ARE THERE ANY VERTICAL LOADS ON RIGHT-HAND RAFTER ? //')
584*      CALL SUB1(NUM)
585*      IF (NUM.NE. NO.AND.NUM.NE.YES) GO TO 475
586*      WRITE(6,B) NUM
587*      ***** C
588*      C   VERTICAL LOADS On RIGHT-HAND RAFTER   C
589*      ***** C
590*      IF (NU4=YES) 49,1010,49
591*      1010  WRITE(6,925)
592*      READ(S, 1 ) NPL
593*      WRITE(6,1)NPP
594*      SVL=U.U
595*      WRITE(6,924)
596*      DO 1040 I=1,NPL
597*      WRITE(6,87)SP,FORCE,HP,LENGTH
598*      READ (S, 1 )SP,HP
599*      SP=2*SP

```

```

600*      SVR=SYR+5,J
601*      DL 1030 J=1,11
602*      AK=FLUAT(J)/FLU,L=J+1
603*      K=L+J
604*      HAN'=AK*SPAN/2+C-MP
605*      IF(HANR)=1,2,1020,1020
606*      1020 DTM5(K)=DT5(K)-SP*HANR
607*      SH(K)=SH(K)-SP/SEC
608*      TH(K)=TH(K)-SP*COSC
609*      1030 CONTINUE
610*      DU 1040 J=32,41
611*      1040 DTM5(J)=DT5(J)-SP*(SPAN/2+j-MP)
612*      DU 1050 J=32,41
613*      1050 TH(J)=TH(J)-SP
614*      DU TO 49
615*      END

```

END OF COMPIRATION: NO DIAGNOSTICS.

```

1*      SUBROUTINE SUB1(NU1)
2*      COMMON MPN(41),RESM(41),ANS(41),M(80)
3*      INTEGER YES,Y,F,S,U,NO,BLANK
4*      I,F,P,X,FIX,PIN,ERR
5*      2,A,TAP,U,T,UNIF
6*      DATA BLANK/' /,YES/"YES"/,NO/"NO"/, FIX/"FIXED"/,PIN/"PINNED"/
7*      I,P/"P"/,F/"F"/,I/"I"/,X/"X"/,N/"N"/,U/"0"/, Y/"Y"/, E/"E"/
8*      I,S/"S"/,ERR/"ERR"/
9*      2,U/"U"/,T/"T"/,A/"A"/,TAP/"TAP"/, UNIF/"UNIF"/
10*     READ(5,100) M
11*    100 FORMAT(20A1)
12*    DU 30 J=1,79
13*    IF (M(J)=NE.BLANK)GO TO 39
14*    30 CONTINUE
15*    39 IF(J.GE.79) GO TO 40
16*    IF(M(J).EQ.Y .AND. M(J+1).EQ. E .AND. M(J+2).EQ. S )GO TO 42
17*    IF(M(J).EQ. F .AND. M(J+1).EQ. I .AND. M(J+2).EQ. X )GO TO 43
18*    IF(M(J).EQ. P .AND. M(J+1).EQ. I .AND. M(J+2).EQ. N )GO TO 44
19*    IF(M(J).EQ.T*ANH*M(J+1).EQ.A*AND*M(J+2).EQ.P) GO TO 50
20*    IF(M(J).EQ.U*ANH*M(J+1).EQ.N*AND*M(J+2).EQ.I*AND*M(J+3).EQ.F)
21*    I GO TO 60
22*    40 IF(M(J).EQ. N .AND. M(J+1).EQ. U ) GO TO 45
23*    NUM=ERR
24*    WRITE (6,100) M
25*    GO TO 46
26*    42 NUM=YES
27*    GO TO 46
28*    43 NUM=FIX
29*    GO TO 46

30*    44 NUM=PIN
31*    GO TO 46
32*    45 NUM=NO
33*    GO TO 46
34*    50 NUM=TAP
35*    GO TO 46
36*    60 NUM=UNIF
37*    46 RETURN
38*    END

```

END OF COMPIRATION: NO DIAGNOSTICS.

```

1* SUBROUTINE SUB2(SEC,HT,SPAN,LENGTH,CF,CL,D1,D2,D3,D4,W,T,SEGMA,BF,
2*   RIAS,LBAS,COST,LCOS2,FY)
3* COMMON MPN(41),RESM(41),ANS(41),H(80)
4* INTEGER PIN,RBAS
5* REAL MPN
6* DATA PIN/'PINNED'
7* FBN(DN,W,T,BF,CGM)=0.5-(T/(2.0*DN*W))*(1.0-CGM)*(BF+W/CGM)
8* FMPN(DN,T,W,BF,COSG,BN)=(RN+DN-T)*2.0/2.0+U+5*T*BF*(2.0*BN*
9*   DN-T)+U+5*W*((1.0-RN)*DN-T/COSG)*2.0+U+0.5*BF*T*COSG*(2.0*(1.0-RN)
10*   -DT-T/COSG)
11*   R=SFC*SPAN+0.5
12*   10 FORMAT( )
13*   READ(5,11) D1,D2,D3,D4,W,T,BF,FY
14*   WRITE(6,12) U1,LENGTH,D2,LENGTH,U3,LENGTH,D4,LENGTH,R,LENGTH,T,
15*   LENGTH,BF,LENGTH,FY
16*   12 FORMAT(' SECTION DEPTH A1 : //',
17*   ' COLUMN BASE ',T20,'=',T27,G15.5,A5,///
18*   ' COLUMN TOP ', T20,'=',T27,G15.5,A5,///
19*   ' COLUMN TOP WHICH //, CORRESPONDS TO THE // RAFTER BOTTOM',/T20,
20*   ' = ', T27,G15.5,A5,///
21*   ' AFEA*T20,'=',T27,G15.5,A5,///
22*   ' WEB THICKNESS',T20,'=',T27,G15.5,A5,///
23*   ' FLANGE THICKNESS',T20,'=',T27,G15.5,A5,///
24*   ' FLANGE WIDTH',T20,'=',T27,G15.5,A5,///
25*   ' STEEL',T20,'=',T27,G15.5,'KS1'///)
26*   SEGMA=FY*CF/(CL*CL)
27*   BFT=U.5*PF/T
28*   IF(FY>E4*30.0*AND*U.5*LF/T*LE.0.5) GU TO 30
29*   IF(FY>E4*42.0*AND*U.5*BF/T*LE.0.5) GU TO 30
30*   IF(FY>E4*45.0*AND*U.5*BF/T*LE.7.4) GU TO 30

```

```

31*      IF(FY.EQ.50.0.AND.U.5*BF/T .LE.7.0) GO TO 30
32*      IF(FY.EQ.55.0.AND.U.5*BF/T .LE.6.0) GO TO 30
33*      IF(FY.EQ.60.0.AND.U.5*BF/T .LE.6.3) GO TO 30
34*      IF(FY.EQ.65.0.AND.U.5*BF/T .LE.6.0) GO TO 30
35*      GO TO 50
36*      30  COS1 =HT/S,PT((T**2+(D2-D1)**2)
37*      COS2=R/SNHT((K**2+(D3-D4)**2)
38*      DO 40 I=1,11
39*      J1=42-I
40*      J2=22-I
41*      J3=20+I
42*      X=FLOAT(I)/10.0 -1.0
43*      D1=1.02-D1*X+D1
44*      D2=(D3-D4)*X+D4
45*      BN2=FBN(DN2,W,T,BF,COS2)
46*      BN1=FBN(D1,0,T,BR,COS1)
47*      IF(I.EQ.11) GO TO 35
48*      MPN(I)=(FMPN(DN1,T,A,BF,COS1 ,BN1 ))*SEGMA
49*      MPN(J1)= MPN(I)
50*      35 MPN(J2)=(FMPN(DN2,T,W,BF,COS2 ,BN2 ))*SEGMA
51*      40 MPN(J3)= MPN(J2)
52*      GO TO 100
53*      50  IF(   FY.EQ.36.0.OR.
54*           1   FY.EQ.42.0.OR.
55*           1   FY.EQ.45.0.OR.
56*           1   FY.EQ.50.0.OR.
57*           1   FY.EQ.55.0.OR.
58*           1   FY.EQ.60.0.OR.
59*           1   FY.EQ.65.0 ) GO TO 70
60*      WRITE(6,65) FY
61*      65 FORMAT(' THIS STEEL DOES NOT CORRESPONDS TO THE AISC SPECIFI
62*      ICATIONS OF 1973 FY=',G15.3,' KSI ')
63*      STOP
64*      70  WRITE(6,75) BFT
65*      75 FORMAT(' THIS SECTION UNDERGOES FAILURE DUE TO LOCAL BUCKLING'
66*           1 , ' IN FLANGE'
67*           1 // ' CHANGE BF/2T' , ' =',G15.5)
68*      STOP
69*      100 If(RBAS .EQ.PIN) MPN(41)=0.0
70*      If(LBAS .EQ.PIN) MPN( 1)=0.0
71*      RETURN
72*      END

```

END OF COMPIRATION:

NO DIAGNOSTICS.

```

1*      SUBROUTINE SUB3(COS1,COS2,D1,D2,D3,D4,",BF,T,SEGMA,RBAS,LBAS,FY)
2*      COMMON MPN(41),RESM(41),ANS(41),N(80),THRUST(41),SHEAR(41)
3*      INTEGER YES,NO, ANS,ASTR ,PIN ,RBAS
4*      REAL MPI.
5*      DATA ASTR/*****/, NO/*NO*/,YES/*YES*/
6*      DATA PIN/*PIN-NF0*/
7*      Z(FAC1,FAC4,CGM,T)=
8*      1   W*((FAC1+FAC4+FAC4)*D+5+FAC0*T*(FAC1+FAC4*CGM+T))
9*      FDN(DN,A,T,BF,CGM)= D+5-(T/(2.0*DN*W))*(1.0-CGM)*(BF+W/CGM)
10*     FAC0=BF/W
11*     DU 410 J=1,41
12*     IF(J=10) 110,110,120
13*     110 X=FLOAT(J)/10.-0.1
14*     Di=(D2-D1)*X+D1
15*     CGM=COS1
16*     GU T,J 170
17*     120 IF(J=21) 130,130,140
18*     130 ..,J=10
19*     X=FLOAT(J1)/10.0 -0.1
20*     Di=D3-(D3-D4)*X
21*     CGM=COS2
22*     GU T,J 170
23*     140 IF(J=31) 150,150,160
24*     150 J1=J-1
25*     X=FLOAT(J1)/10.0
26*     Di=D4+(D3-D4)*X

```

```

27*      CGM=COS2
28*      160  J1=J-31
29*      X=FLOAT(J1)/10.0
30*      DIV=D2-(D2-D1)*X
31*      CGM=COS1
32*      170  BN=FBN(BN,V,T,F,LGM)
33*      FAC1=BN*DN-T
34*      WUT= DN/W
35*      FAC2= T*(1.0+CGM)+(BF-K/CGM)+W*DN
36*      PPG=AES(THMUST(J))/FAC2
37*      PHY=PPG/SEGMA
38*      FAC4= (1.0-HN)*DN-T/CGM
39*      FAC5=-(FAC1**2*(FAC0-1.0)-FAC4**2*(FAC0*CGM**2-1.0))/FAC0
40*      FAC6=FAC1*(FAC0-1.0)+FAC4*(FAC0*CGM**2-1.0)
41*      FAC3=(FAC2/W)*(PHY)
42*      Z1=Z(FAC1,FAC4,CGM,T)
43*      FAC10= SQRT((2.0*FAC1*FAC3*(FAC0-1.0)+FAC3**2)/FAC0)
44*      Y0=0.5*FAC3
45*      Y1=Y0
46*      IF(Y0.LE.FAC1) GO TO 40
47*      IF(CGM.EQ.1.0) GO TO 75
48*      YU=(FAC3+FAC1*(FAC0-1.0)-FAC10)/(FAC0-1.0)
49*      IF(Y0.LT.FAC1.OR.Y0.GT.HN+DN) GO TO 55
50*      Y1=SQRT(FAC0*Y0**2+FAC1**2*(1.0-FAC0))
51*      IF(Y0.LE.BN*DN.AND.Y1.LE.FAC4) GO TO 60
52*      55  YU=(FAC3+FAC1*(FAC0-1.0)+FAC10)/(FAC0-1.0)
53*      IF(YU.LT.FAC1.OR.YU.GT.DN) GO TO 44
54*      Y1=SQRT(FAC0*Y0**2+FAC1**2*(1.0-FAC0))
55*      IF(Y0.LE.HN*DN.AND.Y1.LE.FAC4) GO TO 60
56*      44  FAC7=FAC3+FAC6
57*      FAC9=FAC7**2+FAC0**2*FAC5*(1.0-CGM**2)
58*      FAC8=SQRT(FAC9)
59*      YU=(FAC7+CGM*FAC8)/(FAC0*(1.0-CGM**2))
60*      Y1=SQRT(FAC5+YU**2)/CGM
61*      IF(Y1.GE.FAC4.AND.Y1.LE.(1.0-HN)*DN.AND.Y0.GE.FAC1.AND.Y0.LE.
62*          1.0*DN) GO TO 60
63*      YU=(FAC7-CGM*FAC8)/(FAC0*(1.0-CGM**2))
64*      Y1=SQRT(FAC5+YU**2)/CGM
65*      IF(Y1.GE.FAC4.AND.Y1.LE.(1.0-BN)*DN.AND.Y0.GE.FAC1.AND.Y0.LE.
66*          1.0*DN) GO TO 60
67*      40  MPN(J)= SEGMA*(Z1-(FAC2**2*PHY**2)/(4.0*K))
68*      GO TO 80
69*      60  MPN(J)= SEGMA*(Z1-W*(FAC0*Y0**2+FAC1**2*(1.0-FAC0)))
70*      GO TO 80
71*      75  YU=(0.5*FAC3+FAC1*(FAC0-1.0))/FAC0

```

```

72*          Y1=Y0
73*          GO TO 60
74*          60  IF(RBAS .EQ.PIN)  MPN(4)=0.0
75*          IF(LBAS .EQ.PIN)  NPN(1)=0.0
76*          I=J
77*          IF(ABS(SHEAR(J)) .LE. 0.55*SEGMA*W*DN)  GO TO 90
78*          WRITE(6,220)   I
79*          220  FORMAT(' SECTION NO.',IS,' WEP NEED TO BE STIFFENED',//,
80*                   ' BECAUSE THE SHEAR FORCE IS > THE ALLOWABLE',//,
81*                   ' 0.55*FY*L*DN=',G15.5)
82*          90  IF (PPY.GT.C*27) GO TO 18C
83*          DTH=412.0*(1.0-1.4*PPY)/SQRT(FY)
84*          180 DTH=257.0/SQRT(FY)
85*          GO TO 190
86*          190 IF(DN/W.GT.DTH) GO TO 20D
87*          GO TO 401
88*          200 WRITE(6,210)   I ,AUT ,DTH
89*          210 FORMAT(1X,12('*')//,*' WARNING * //,1X,12('*'),/
90*                   ' SECTION NO.',IS,' UNDERGOES FAILURE DUE TO LOCAL BUCKL
91*                   ' INING IN WEB',//,' DN/W=',G15.5,' THE ALLOWABLE IS =',G15.5//)
92*          401 IF (ABS(RESM(1))-0.99*MPN(1)) 406,405,405
93*          405 IF (ABS(RESM(1))-1.01*MPN(1)) 407,408,408
94*          406 ANS(1)=1.0
95*          407 ANS(1)=ASTR
96*          408 ANS(1)=YES
97*          410 CONTINUE
98*          RETURN
99*          END
100*

```

END OF COMPILATION: NO DIAGNOSTICS.

```

1*      SUBROUTINE UNITS(LENGTH,FORCE,CF,CL)
2*      DIMENSION MI(60)
3*      DATA KG/*KG*/,TON/*TON*/,LB/*LB*/,T/*T*/,G/*G*/,U/*O*/,B/*B*/,
4*           FT/*FT*/,IN/*IN*/,CH/*CF*/,C/*C*/,BLANK/* /,I/*I*/,KIPS
5*           2/*KIPS*/,F/*F*/,P/*P*/,S/*S*/,R/*R*/,L/*L*/,N/*N*/,M/*M*/
6*           REAR(5,1) NI
7*           1 FORMAT(40A1)
8*           JZ=1
9*           KL=1
10*          5 DO 10 J1=JZ,FU
11*             IF(MI(J1)=BLANK) 20,1C,20
12*             10 CONTINUE
13*             20 IF( (MI(J1) .EQ. K .AND. MI(J1+1) .EQ. G) GO TO 40
14*             IF( (MI(J1) .EQ. F .AND. MI(J1+1) .EQ. P .AND. MI(J1+2) .EQ. P .AND. MI(J1+3)
15*             .EQ. S ) GO TO 50
16*             IF( (MI(J1) .EQ. L .AND. MI(J1+1) .EQ. H) GO TO 60
17*             IF( (MI(J1) .EQ. T .AND. MI(J1+1) .EQ. U .AND. MI(J1+2) .EQ. N ) GO TO 70
18*             IF( (MI(J1) .EQ. F .AND. MI(J1+1) .EQ. T) GO TO 60
19*             IF( (MI(J1) .EQ. C .AND. MI(J1+1) .EQ. R) GO TO 80
20*             IF( (MI(J1) .EQ. I .AND. MI(J1+1) .EQ. N) GO TO 100
21*             IF( (MI(J1) .EQ. M) GO TO 110
22*             WRITE(6,33)
23*             33 FORMAT(1X,SS(*,*),/* USER DOES NOT SPECIFIED RICHT UNITS*/,/
24*                   1A1,(*)*)
25*             STOP
26*             40 FORCE=NG

```

```

28*      CF=453.592
29*      GO TO 120
30*      50      FORCE=KIPS
31*      LF= 1.0
32*      GO TO 120
33*      60      FORCE= LB
34*      CF= 166.667
35*      GO TO 120
36*      70      FORCE =TON
37*      CF=0.453592
38*      GO TO 120
39*      80      LENGTH=FT
40*      CL=1.0/12.0
41*      GO TO 120
42*      90      LENGTH= CM
43*      CL= 2.54001
44*      GO TO 120
45*      100     LENGTH= IN
46*      CL= 1.0
47*      GO TO 120
48*      110     LENGTH= M
49*      CL= 0.0254001
50*      120     KL =KL+1
51*      IF (KL.GT.2) GO TO 200
52*      DO 130 J2=J1,P0
53*      IF (PI(J2)=BLANK) 130, 5,130
54*      130     CONTINUE
55*      200     WRITE(6,44L) FORCE,LENGTH
56*      440     FORMAT (//IX, ' FORCE UNIT ',A4,//' LENGTH UNIT ', A4)
57*      RETURN
58*      END

```

END OF COMPILEATION: NO DIAGNOSTICS.

BXQT PROG.ANWA

\*\*\*\*\*  
\* PLASTIC DESIGN OF INDUSTRIAL RIGID FRAMES \*  
\*\*\*\*\*

PROBLEM INITIATION

\*\*\*\*\*  
\* ANALYSIS NO 1  
\*\*\*\*\*

TYPE OF MEMBER SAY TAPERED,OR UNIFORM

TYPE TAP

INPUT THE SPECIFIED UNITS ,LENGTH AND FORCE

FORCE UNIT LB

LENGTH UNIT FT

INPUT THE FRAME DIMENSIONS SPAN,EAVES HEIGHT,DEGREE OF PITCH AND FRAMES SPACING

SPAN = 72.000 FT

EAVS HEIGHT = 33.000 FT

DEGREE OF PITCH = 15.000 DEGREE

FRAME SPACING = 24.000 FT

INPUT THE WORKING LOAD DATA VERTICAL UNIFORM LOADING INTENSITY

WORKING LOAD

VERTICAL UNIFORM LOADING INTENSITY = 10.000 LB /SQUARE FT

ARE THERE PRESSURES NORMAL TO COVERING ?

YES

INPUT FOUR NORMAL SURFACE PRESSURES FOR THE  
LEFT-HAND COLUMN, LEFT-HAND RAFTER, RIGHT-HAND RAFTER  
AND RIGHT-HAND COLUMN WITH PRESSURE POSITIVE IF  
ACTING INWARDS ON THE BUILDING.

PRESSURE ON THE LEFT-HAND COLUMN = 24.300 LB /SQUARE FT

PRESSURE ON THE LEFT-HAND RAFTER = 6.2890 LB /SQUARE FT

PRESSURE ON THE RIGHT-HAND RAFTER= .00000 LB /SQUARE FT

PRESSURE ON THE RIGHT-HAND COLUMN= .00000 LB /SQUARE FT

INPUT VERTICAL CONCENTRATED APEX LOAD AND LOAD FACTOR

CONCENTRATED APEX LOAD = .00000 LB

LOAD FACTOR= 1.4000

ARE THERE ANY OTHER CONCENTRATED LOADS?

NO

ARE THERE ANY GANTRY CRANE LOADS?

YES

INPUT THE HEIGHT OF THE GANTRY CRANE RAILS AND  
THE OFFSET OF THE CRANE RAILS

HEIGHT OF CRANE RAIL = 27.000 FT

OFFSET OF CRANE RAIL = 2.0000 FT

INPUT WORKING VALUES OF VERTICAL GANTRY CRANE FORCES LEFT-HAND COLUMN FIRST

WORKING VALUES OF VERTICAL CRANE FORCES

WORKING VERTICAL CRANE FORCE ON LEFT-HAND COLUMN= 10000. LB

WORKING VERTICAL CRANE FORCE ON RIGHT-HAND COLUMN= 4000.0 LB

WORKING VALUES OF HORIZONTAL CRANE FORCES

HORIZONTAL CRANE FORCE ON LEFT-HAND COLUMN = 500.00 LB

HORIZONTAL CRANE FORCE ON RIGHT-HAND COLUMN= 200.00 LB

DO YOU WANT TO SEE THE TWIN CANTILEVER MOMENT?

YES

THE TWIN CANTILEVER MOMENT SHEAR AND THRUST :

| SECTION NO. | MOMENT    | SHEAR  | THRUST  |
|-------------|-----------|--------|---------|
| 1           | -92323+06 | 29682. | -33703. |
| 2           | -82972+06 | 26986. | -33703. |
| 3           | -74511+06 | 24293. | -33703. |
| 4           | -66938+06 | 21549. | -33703. |
| 5           | -60255+06 | 18905. | -33703. |
| 6           | -54461+06 | 16210. | -33703. |
| 7           | -49557+06 | 13516. | -33703. |
| 8           | -45541+06 | 10521. | -33703. |
| 9           | -42414+06 | 8127.1 | -33703. |
| 10          | -37566+06 | 4732.7 | -19703. |
| 11          | -36449+06 | 19559. | -3130.7 |
| 12          | -29524+06 | 17603. | -2617.6 |
| 13          | -23327+06 | 15647. | -25.4.5 |
| 14          | -17860+06 | 13692. | -2191.5 |
| 15          | -13122+06 | 11736. | -1878.4 |
| 16          | -91122.   | 9779.7 | -1545.3 |
| 17          | -58318.   | 7823.7 | -1252.3 |
| 18          | -32804.   | 5367.8 | -934.20 |
| 19          | -14580.   | 3411.9 | -626.14 |
| 20          | -3644.9   | 1955.9 | -313.07 |
| 21          | .00000    | .00000 | .00000  |

|    |           |         |         |
|----|-----------|---------|---------|
| 22 | -2177.3   | -1160.4 | -313.07 |
| 23 | -8709.1   | -2336.8 | -626.14 |
| 24 | -19596.   | -3505.2 | -939.20 |
| 25 | -34836.   | -4673.5 | -1252.3 |
| 26 | -54432.   | -5841.9 | -1565.3 |
| 27 | -78382.   | -7010.3 | -1878.4 |
| 28 | -10669+.6 | -8178.7 | -2191.5 |
| 29 | -13935+.6 | -9347.1 | -254.5  |
| 30 | -17636+.6 | -10515. | -2617.6 |
| 31 | -21773+.6 | -11684. | -3130.7 |
| 32 | -21773+.6 | .00000  | -12096. |
| 33 | -22074+.6 | 280.00  | -17696. |
| 34 | -22784+.6 | 280.00  | -17696. |
| 35 | -22691+.6 | 280.00  | -17696. |
| 36 | -22599+.6 | 280.00  | -17696. |
| 37 | -22506+.6 | 280.00  | -17696. |
| 38 | -22414+.6 | 280.00  | -17696. |
| 39 | -22322+.6 | 280.00  | -17696. |
| 40 | -22229+.6 | 280.00  | -17696. |
| 41 | -22137+.6 | 280.00  | -17696. |

---

WHAT KIND OF BASES?

INPUT FIXED OR PINNED ,LEFT-HAND BASE FIRST

LEFT-HAND BASE IS FIXED

RIGHT-HAND BASE IS FIXED

INPUT THE SECTION DIMENSIONS,D1,D2,D3,D4,W,T,BF,FY  
SECTION DEPTH AT :

COLUMN BASE = .86000 FT

COLUMN TOP = 1.3500 FT

COLUMN TOP WHICH  
RESPONDS TO THE  
RAFTER BOTTOM

= 1.2000 FT

APEX = .85000 FT

WEB THICKNESS = .45000-01FT

FLANGE THICKNESS = .52090-01FT

FLANGE WIDTH = .50000 FT

] STEEL = 36,000 KSI

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1 17 -31 41

| SECTION NO. | SHEAR           | THRUST           |
|-------------|-----------------|------------------|
| 1           | •1857+05        | -•267 <u>+05</u> |
| 2           | •1568+05        | -•267 <u>+05</u> |
| 3           | •1319+05        | -•267 <u>+05</u> |
| 4           | •1049+05        | -•267 <u>+05</u> |
| 5           | 7797•           | -•267 <u>+05</u> |
| 6           | 5103•           | -•267 <u>+05</u> |
| 7           | 2408•           | -•267 <u>+05</u> |
| 8           | -286•0          | -•267 <u>+05</u> |
| 9           | -298 <u>0</u> • | -•267 <u>+05</u> |
| 10          | -6375•          | -•127 <u>+05</u> |
| 11          | 9918•           | -•120 <u>+05</u> |
| 12          | 7962•           | -•117 <u>+05</u> |
| 13          | 6000•           | -•114 <u>+05</u> |
| 14          | 405 <u>0</u> •  | -•111 <u>+05</u> |
| 15          | 2094•           | -•107 <u>+05</u> |
| 16          | 138•0           | -•104 <u>+05</u> |
| 17          | -1118•          | -•101 <u>+05</u> |
| 18          | -3774•          | -•965•           |
| 19          | -573 <u>0</u> • | -•9542•          |
| 20          | -768 <u>0</u> • | -•9229•          |
| 21          | 9642•           | -•8916•          |
| 22          | 8473•           | -•128 <u>+05</u> |
| 23          | 7345•           | -•1317+05        |
| 24          | 6136•           | -•134 <u>+05</u> |
| 25          | 4968•           | -•137 <u>+05</u> |
| 26          | 3800•           | -•1411+05        |
| 27          | 2631•           | -•1442+05        |
| 28          | 1463•           | -•147 <u>+05</u> |
| 29          | 294•6           | -•150 <u>+05</u> |
| 30          | -673•6          | -•153 <u>+05</u> |
| 31          | -2042•          | -•1567+05        |
| 32          | •1111+05        | -•141 <u>+05</u> |
| 33          | •1139+05        | -•247 <u>+05</u> |
| 34          | •1139+05        | -•247 <u>+05</u> |
| 35          | •1139+05        | -•247 <u>+05</u> |
| 36          | •1139+05        | -•247 <u>+05</u> |
| 37          | •1139+05        | -•247 <u>+05</u> |
| 38          | •1139+05        | -•247 <u>+05</u> |
| 39          | •1139+05        | -•247 <u>+05</u> |
| 40          | •1139+05        | -•247 <u>+05</u> |
| 41          | •1139+05        | -•247 <u>+05</u> |

## LOCATION OF PENDING MOMENT MT &gt;MPC:

| SECTION NO. | PLASTIC MOMENT CAPACITY |     |
|-------------|-------------------------|-----|
| 1           | - .987+05               | NO  |
| 2           | - .419+C5               | NO  |
| 3           | .608+04                 | NO  |
| 4           | .451+05                 | NO  |
| 5           | .753+05                 | NO  |
| 6           | .966+05                 | NO  |
| 7           | .1u9+06                 | NO  |
| 8           | .113+06                 | NO  |
| 9           | .107+06                 | NO  |
| 10          | .119+06                 | NO  |
| 11          | .935+05                 | NO  |
| 12          | .127+06                 | NO  |
| 13          | .153+06                 | NO  |
| 14          | .172+06                 | NO  |
| 15          | .183+06                 | NO  |
| 16          | .187+06                 | YES |
| 17          | .184+06                 | YES |
| 18          | .174+06                 | YES |
| 19          | .156+06                 | *** |
| 20          | .131+06                 | NO  |
| 21          | .986+C5                 | NO  |
| 22          | .819+05                 | NO  |
| 23          | .609+05                 | NO  |
| 24          | .355+C5                 | NO  |
| 25          | .575+04                 | NO  |
| 26          | - .264+05               | NO  |
| 27          | - .668+05               | NO  |
| 28          | - .11u+06               | NO  |
| 29          | - .157+06               | NO  |
| 30          | - .2u8+06               | NO  |
| 31          | - .264+06               | *** |
| 32          | - .228+06               | NO  |
| 33          | - .202+06               | NO  |
| 34          | - .164+06               | NO  |
| 35          | - .127+06               | NO  |
| 36          | - .892+05               | NO  |
| 37          | - .516+05               | NO  |
| 38          | - .14u+05               | NO  |
| 39          | .236+05                 | NO  |
| 40          | .o12+05                 | NO  |
| 41          | .947+05                 | NO  |

; DO YOU WANT TO SEE THE BASE REACTIONS?

YES

] V1= +1460+05 H1= +1857+05 V41= +1261+05 H41= +1139+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

YES

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1 19 -33 41

| SECTION NO. | SHEAR    | THRUST    |
|-------------|----------|-----------|
| 1           | +1802+05 | -+2722+05 |
| 2           | +1532+05 | -+2722+05 |
| 3           | +1263+05 | -+2722+05 |
| 4           | 9933+    | -+2722+05 |
| 5           | 7238+    | -+2722+05 |
| 6           | 4544+    | -+2722+05 |
| 7           | 1850+    | -+2722+05 |
| 8           | -8444.7  | -+2722+05 |
| 9           | -3139+   | -+2722+05 |
| 10          | -6433+   | -+1322+05 |
| 11          | +1028+05 | -+1272+05 |
| 12          | 8324+    | -+1241+05 |
| 13          | 6368+    | -+1210+05 |
| 14          | 4412+    | -+1170+05 |
| 15          | 4560+    | -+1147+05 |
| 16          | 5000.1   | -+1116+05 |
| 17          | -1456+   | -+1084+05 |
| 18          | -3412+   | -+1053+05 |
| 19          | -5368+   | -+1022+05 |
| 20          | -7324+   | -+990.4   |
| 21          | 9260+    | -+95.91   |
| 22          | 8111+    | -+1326+05 |
| 23          | 6943+    | -+1357+05 |
| 24          | 5774+    | -+1384+05 |
| 25          | 4600+    | -+1420+05 |
| 26          | 3458+    | -+1451+05 |
| 27          | 2209+    | -+1482+05 |
| 28          | 1101+    | -+1514+05 |
| 29          | -67044   | -+1545+05 |
| 30          | -1236+   | -+1576+05 |
| 31          | -2464+   | -+1606+05 |
| 32          | +1167+05 | -+1852+05 |
| 33          | +1195+05 | -+2410+05 |
| 34          | +1195+05 | -+2410+05 |
| 35          | +1195+05 | -+2410+05 |
| 36          | +1195+05 | -+2410+05 |
| 37          | +1195+05 | -+2410+05 |
| 38          | +1195+05 | -+2410+05 |
| 39          | +1195+05 | -+2410+05 |
| 40          | +1195+05 | -+2410+05 |
| 41          | +1195+05 | -+2410+05 |

LOCATION OF BENDING MOMENT MT >MPC:

| SECTION NO. |         | PLASTIC MOMENT CAPACITY |
|-------------|---------|-------------------------|
| 1           | -118+06 | .142+06                 |
| 2           | -626+05 | .153+06                 |
| 3           | -164+05 | .164+06                 |
| 4           | .208+05 | .176+06                 |
| 5           | .491+05 | .188+06                 |
| 6           | .686+05 | .200+06                 |
| 7           | .791+05 | .212+06                 |
| 8           | .808+05 | .225+06                 |
| 9           | .735+05 | .238+06                 |
| 10          | .835+05 | .252+06                 |
| 11          | .562+05 | .225+06                 |
| 12          | .909+05 | .216+06                 |
| 13          | .118+06 | .207+06                 |
| 14          | .138+06 | .198+06                 |
| 15          | .151+06 | .189+06                 |
| 16          | .157+06 | .181+06                 |
| 17          | .155+06 | .172+06                 |
| 18          | .146+06 | .164+06                 |
| 19          | .129+06 | .156+06                 |
| 20          | .106+06 | .148+06                 |
| 21          | .748+05 | .141+06                 |
| 22          | .616+05 | .401+06                 |
| 23          | .420+05 | .384+06                 |
| 24          | .190+05 | .369+06                 |
| 25          | -832+04 | .353+06                 |
| 26          | -400+05 | .338+06                 |
| 27          | -760+05 | .323+06                 |
| 28          | -116+06 | .308+06                 |
| 29          | -161+06 | .294+06                 |
| 30          | -211+06 | .279+06                 |
| 31          | -264+06 | * * *                   |
| 32          | -225+06 | .265+06                 |
| 33          | -198+06 | .252+06                 |
| 34          | -158+06 | .238+06                 |
| 35          | -119+06 | .225+06                 |
| 36          | -795+05 | .212+06                 |
| 37          | -400+05 | .200+06                 |
| 38          | -608.   | .186+06                 |
| 39          | .388+05 | .164+06                 |
| 40          | .782+05 | .153+06                 |
| 41          | .116+06 | .142+06                 |

DO YOU WANT TO SEE THE BASE REACTIONS?

YES

V1= +1513+05 H1= +1802+05 V41= +1208+05 H41= +1195+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

YES

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1 14 -33 41

| SECTION NO. | SHEAR    | THRUST    |
|-------------|----------|-----------|
| 1           | +1647+05 | -+2710+05 |
| 2           | +1578+05 | -+2710+05 |
| 3           | +1319+05 | -+2710+05 |
| 4           | +1039+05 | -+2710+05 |
| 5           | 7497+    | -+2710+05 |
| 6           | 5002+    | -+2710+05 |
| 7           | 2308+    | -+2710+05 |
| 8           | -386.3   | -+2710+05 |

|    |                  |                   |
|----|------------------|-------------------|
| 9  | -3081.           | -+271 <u>+</u> .5 |
| 10 | -647 <u>+</u> 05 | -+131 <u>+</u> 05 |
| 11 | +1026+05         | -+122 <u>+</u> 05 |
| 12 | 8321.            | -+1193+05         |
| 13 | 6365.            | -+1162+05         |
| 14 | 4404.            | -+1131+15         |
| 15 | 2453.            | -+1092+05         |
| 16 | 497.2            | -+1060+15         |
| 17 | -1459.           | -+1037+05         |
| 18 | -3415.           | -+1006+05         |
| 19 | -5371.           | -+9742.           |
| 20 | -7327.           | -+9429.           |
| 21 | 9282.            | -+9116.           |
| 22 | 8114.            | -+1283+05         |
| 23 | 6946.            | -+1310+05         |
| 24 | 5777.            | -+1344+05         |
| 25 | 4609.            | -+1374+05         |
| 26 | 3441.            | -+1410+05         |
| 27 | 2272.            | -+1441+05         |
| 28 | 1101.            | -+1473+05         |
| 29 | -64.57           | -+1504+05         |
| 30 | -1233.           | -+1535+05         |
| 31 | -2401.           | -+1567+05         |
| 32 | +1121+05         | -+1870+05         |
| 33 | +1149+05         | -+2430+05         |
| 34 | +1149+05         | -+243 <u>+</u> .5 |
| 35 | +1149+05         | -+2430+05         |
| 36 | +1149+05         | -+2430+05         |
| 37 | +1149+05         | -+2430+05         |
| 38 | +1149+05         | -+2430+05         |
| 39 | +1149+05         | -+243 <u>+</u> .5 |
| 40 | +1149+05         | -+2430+05         |
| 41 | +1149+05         | -+2430+05         |

LOCATION OF BENDING MOMENT MT >MPC:

| SECTION NO. |           | PLASTIC MOMENT CAPACITY |
|-------------|-----------|-------------------------|
| 1           | - .113+06 | NO                      |
| 2           | - .565+05 | NO                      |
| 3           | - .687+04 | NO                      |
| 4           | .299+05   | NO                      |
| 5           | .597+05   | NO                      |
| 6           | .807+05   | NO                      |
| 7           | .927+05   | NO                      |
| 8           | .959+05   | NO                      |
| 9           | .902+05   | NO                      |
| 10          | .102+06   | NO                      |
| 11          | .759+05   | NO                      |
| 12          | .111+06   | NO                      |
| 13          | .138+06   | NO                      |
| 14          | .158+06   | NO                      |
| 15          | .171+06   | NO                      |
| 16          | .176+06   | NO                      |
| 17          | .174+06   | YES                     |
| 18          | .165+06   | ***                     |
| 19          | .149+06   | NO                      |
| 20          | .125+06   | NO                      |
| 21          | .944+05   | NO                      |
| 22          | .792+05   | NO                      |
| 23          | .597+05   | NO                      |
| 24          | .359+05   | NO                      |
| 25          | .766+04   | NO                      |
| 26          | - .249+05 | NO                      |
| 27          | - .618+05 | NO                      |
| 28          | - .103+06 | NO                      |
| 29          | - .149+06 | NO                      |
| 30          | - .199+06 | NO                      |
| 31          | - .253+06 | NO                      |
| 32          | - .216+06 | NO                      |
| 33          | - .190+06 | NO                      |
| 34          | - .152+06 | NO                      |
| 35          | - .114+06 | NO                      |
| 36          | - .764+05 | NO                      |
| 37          | - .305+05 | NO                      |
| 38          | - .579-   | NO                      |
| 39          | .373+05   | NO                      |
| 40          | .752+05   | NO                      |

41            .113+06      NO

.142+06

[ ]

DO YOU WANT TO SEE THE CASE REACTIONS?

YES

V1= .1500+05    H1= .1847+05    V41= .1221+05    H41= .1149+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

YES

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1    12    -33    41

| SECTION NO. | SHEAR    | THRUST    |
|-------------|----------|-----------|
| 1           | .1954+05 | -.2680+05 |
| 2           | .16e5+05 | -.2680+05 |
| 3           | .1415+05 | -.2680+05 |
| 4           | .1146+05 | -.2680+05 |
| 5           | 8764-    | -.2680+05 |
| 6           | 607u-    | -.2680+05 |
| 7           | 3375-    | -.2680+05 |
| 8           | 68u.7    | -.2680+05 |

|    |          |            |
|----|----------|------------|
| 9  | -2614.   | -+2684+05  |
| 10 | -5408.   | -+1284+05  |
| 11 | +1027+05 | -+1114+05  |
| 12 | 8315.    | -+1083+05  |
| 13 | 6359.    | -+1051+05  |
| 14 | 4404.    | -+1024+05  |
| 15 | 2440.    | -+94+05    |
| 16 | 4916.    | -+975.     |
| 17 | -1464.   | -+9262.    |
| 18 | -3420.   | -+8949.    |
| 19 | -5370.   | -+836.     |
| 20 | -7332.   | -+8323.    |
| 21 | 9286.    | -+10.      |
| 22 | 8124.    | -+112,+05  |
| 23 | 6901.    | -+1021+05  |
| 24 | 5783.    | -+1252+05  |
| 25 | 4615.    | -+1203+05  |
| 26 | 3446.    | -+1315+05  |
| 27 | 2276.    | -+1346+05  |
| 28 | 1109.    | -+1377+05  |
| 29 | -59+02   | -+1437+05  |
| 30 | -1247.   | -+1446,+05 |
| 31 | -2396.   | -+1471+05  |
| 32 | +1014+05 | -+1494+05  |
| 33 | +1042+05 | -+2454+05  |
| 34 | +1042+05 | -+2454+05  |
| 35 | +1042+05 | -+2454+05  |
| 36 | +1042+05 | -+2454+05  |
| 37 | +1042+05 | -+2454+05  |
| 38 | +1042+05 | -+2454+05  |
| 39 | +1042+05 | -+2454+05  |
| 40 | +1042+05 | -+2454+05  |
| 41 | +1042+05 | -+2454+05  |

LOCATION OF PENDING MOMENT MT >MPC:

| SECTION NO. |         | PLASTIC MOMENT CAPACITY |
|-------------|---------|-------------------------|
| 1           | -103+06 | NO                      |
| 2           | -425+05 | NO                      |
| 3           | .866+04 | NO                      |
| 4           | .509+05 | NO                      |
| 5           | .843+05 | NO                      |
| 6           | .109+06 | NO                      |
| 7           | .124+06 | NO                      |
| 8           | .131+06 | NO                      |
| 9           | .129+06 | NO                      |
| 10          | .144+06 | NO                      |
| 11          | .122+06 | NO                      |
| 12          | .156+06 | NO                      |
| 13          | .164+06 | NO                      |
| 14          | .204+06 | YES                     |
| 15          | .216+06 | YES                     |
| 16          | .222+06 | YES                     |
| 17          | .226+06 | YES                     |
| 18          | .211+06 | YES                     |
| 19          | .195+06 | YES                     |
| 20          | .171+06 | YES                     |
| 21          | .140+06 | ***                     |
| 22          | .123+06 | NO                      |
| 23          | .101+06 | NO                      |
| 24          | .751+05 | NO                      |
| 25          | .448+05 | NO                      |
| 26          | .102+05 | NO                      |
| 27          | -288+05 | NO                      |
| 28          | -722+05 | NO                      |
| 29          | -120+06 | NO                      |
| 30          | -172+06 | NO                      |
| 31          | -228+06 | NO                      |
| 32          | -195+06 | NO                      |
| 33          | -172+06 | NO                      |
| 34          | -138+06 | NO                      |
| 35          | -104+06 | NO                      |

|    |           |    |         |
|----|-----------|----|---------|
| 36 | - .693+05 | NO | .256+06 |
| 37 | - .349+05 | NO | .168+06 |
| 38 | -526.     | NO | .176+06 |
| 39 | .339+05   | NO | .164+06 |
| 40 | .683+05   | NO | .153+06 |
| 41 | .153+06   | NO | .142+06 |

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DO YOU WANT TO SEE THE BASE REACTIONS?

YES

V1= .1471+05 H1= .1954+05 V41= .1250+05 H41= .1042+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

YES

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1 12 -25 41

| SECTION NO. | SHEAR    | THRUST    |
|-------------|----------|-----------|
| 1           | +1576+05 | -+2533+05 |
| 2           | +1366+05 | -+2533+05 |
| 3           | +1637+05 | -+2533+05 |
| 4           | 767+0    | -+2533+05 |
| 5           | 4970+0   | -+2533+05 |
| 6           | 2283+0   | -+2533+05 |
| 7           | -411+2   | -+2533+05 |
| 8           | -3106+0  | -+2533+05 |
| 9           | -5800+0  | -+2533+05 |
| 10          | -9194+0  | -+1133+05 |
| 11          | 7867+0   | -+1442+05 |
| 12          | 5911+0   | -+1410+05 |
| 13          | 3955+0   | -+1374+05 |
| 14          | 1994+0   | -+134+05  |
| 15          | 42+04    | -+1310+05 |
| 16          | -1913+0  | -+1283+05 |
| 17          | -3604+0  | -+1254+05 |
| 18          | -6845+0  | -+1221+05 |
| 19          | -7761+0  | -+1191+05 |
| 20          | -9737+0  | -+1160+05 |
| 21          | +1169+05 | -+1121+05 |
| 22          | +1052+05 | -+1593+05 |
| 23          | 9356+0   | -+1625+05 |
| 24          | 8180+0   | -+1650+05 |
| 25          | 7014+0   | -+1687+05 |
| 26          | 5851+0   | -+1714+05 |
| 27          | 4682+0   | -+1750+05 |
| 28          | 3514+0   | -+1781+05 |
| 29          | 2346+0   | -+1812+05 |
| 30          | 1177+0   | -+1844+05 |
| 31          | 8+739    | -+1875+05 |
| 32          | +1343+05 | -+2047+05 |
| 33          | +1421+05 | -+2607+05 |
| 34          | +1421+05 | -+2607+05 |
| 35          | +1421+05 | -+2607+05 |
| 36          | +1421+05 | -+2607+05 |
| 37          | +1421+05 | -+2607+05 |
| 38          | +1421+05 | -+2607+05 |
| 39          | +1421+05 | -+2607+05 |
| 40          | +1421+05 | -+2607+05 |
| 41          | +1421+05 | -+2607+05 |

LOCATION OF BENDING MOMENT MT >MPC:

| SECTION NO. |           | PLASTIC MOMENT CAPACITY |
|-------------|-----------|-------------------------|
| 1           | - .495+05 | .142+06                 |
| 2           | - .192+04 | .153+06                 |
| 3           | .367+05   | .164+06                 |
| 4           | .665+05   | .176+06                 |
| 5           | .874+05   | .188+06                 |
| 6           | .994+05   | .200+06                 |
| 7           | .102+06   | .212+06                 |
| 8           | .966+05   | .225+06                 |
| 9           | .819+05   | .238+06                 |
| 10          | .845+05   | .252+06                 |
| 11          | .497+05   | .225+06                 |
| 12          | .754+05   | .216+06                 |
| 13          | .937+05   | .207+06                 |
| 14          | .105+06   | .198+06                 |
| 15          | .109+06   | .149+06                 |
| 16          | .105+06   | .151+06                 |
| 17          | .944+05   | .172+06                 |
| 18          | .763+05   | .164+06                 |
| 19          | .510+05   | .156+06                 |
| 20          | .183+05   | .144+06                 |
| 21          | - .216+05 | .142+06                 |
| 22          | - .405+05 | .400+06                 |
| 23          | - .637+05 | .364+06                 |
| 24          | - .913+05 | .360+06                 |
| 25          | - .123+06 | .353+06                 |
| 26          | - .160+06 | .338+06                 |
| 27          | - .200+06 | .323+06                 |
| 28          | - .245+06 | .318+06                 |
| 29          | - .295+06 | .293+06                 |
| 30          | - .348+06 | YFS                     |

|    |           |     |         |
|----|-----------|-----|---------|
| 31 | - .406+06 | YES | .265+06 |
| 32 | - .360+06 | YES | .252+06 |
| 33 | - .326+06 | YES | .238+06 |
| 34 | - .279+06 | YES | .225+06 |
| 35 | - .232+06 | YES | .212+06 |
| 36 | - .195+06 | NO  | .210+06 |
| 37 | - .138+06 | NO  | .186+06 |
| 38 | - .911+05 | NO  | .176+06 |
| 39 | - .443+05 | NO  | .164+06 |
| 40 | .262+04   | NO  | .153+06 |
| 41 | .495+05   | NO  | .142+06 |

---

DO YOU WANT TO SEE THE BASE REACTIONS?

YES

V1= .1324+05 H1= .1576+05 V41= .1398+05 H41= .1421+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

NO

DO YOU WANT TO ANALYSE ANOTHER FRAME?

YES

\*\*\*\*\*  
\* PLASTIC DESIGN OF INDUSTRIAL RIGID FRAMES \*  
\*\*\*\*\*

PROBLEM INITIATION

\*\*\*\*\*  
\* ANALYSIS NO 2  
\*\*\*\*\*

TYPE OF MEMBER SAY TAPERED,OR UNIFORM

TYPE UNIF

INPUT THE SPECIFIED UNITS ,LENGTH AND FORCE

FORCE UNIT LB

LENGTH UNIT FT  
INPUT THE FRAME DIMENSIONS SPAN,EAVES HEIGHT,DEGREE OF PITCH AND FRAMES SPACING

SPAN = 72.000 FT

EAVES HEIGHT = 33.000 FT

DEGREE OF PITCH = 15.000 DEGREE

FRAME SPACING = 24.000 FT

INPUT THE WORKING LOAD DATA VERTICAL UNIFORM LOADING INTENSITY

WORKING LOAD

VERTICAL UNIFORM LOADING INTENSITY = 10.000 LB /SQUARE FT

ARE THERE PRESSURES NORMAL TO COVERING ?

YES

INPUT FOUR NORMAL SURFACE PRESSURES FOR THE  
LEFT-HAND COLUMN,LEFT-HAND RAFTER,RIGHT-HAND RAFTER  
AND RIGHT-HAND COLUMN WITH PRESSURE POSITIVE IF  
ACTING INWARDS ON THE BUILDING.

PRESSURE ON THE LEFT-HAND COLUMN = 24.300 LB /SQUARE FT

PRESSURE ON THE LEFT-HAND RAFTER = 0.2890 LB /SQUARE FT

PRESSURE ON THE RIGHT-HAND RAFTER = .00000 LB /SQUARE FT

PRESSURE ON THE RIGHT-HAND COLUMN= .00000 LB /SQUARE FT

) INPUT VERTICAL CONCENTRATED APEX LOAD AND LOAD FACTOR

) CONCENTRATED APEX LOAD = .00000 LB

) LOAD FACTOR= 1.4000

ARE THERE ANY OTHER CONCENTRATED LOADS?

NO

ARE THERE ANY GANTRY CRANE LOADS?

YES

INPUT THE HEIGHT OF THE GANTRY CRANE RAILS AND  
THE OFFSET OF THE CRANE RAILS

HEIGHT OF CRANE RAIL = 27.000 FT

OFFSET OF CRANE RAIL = 2.0000 FT

INPUT WORKING VALUES OF VERTICAL GANTRY CRANE FORCES LEFT-HAND COLUMN FIRST

WORKING VALUES OF VERTICAL CRANE FORCES

WORKING VERTICAL CRANE FORCE ON LEFT-HAND COLUMN= 10000. LB

WORKING VERTICAL CRANE FORCE ON RIGHT-HAND COLUMN= 4000.0 LB

WORKING VALUES OF HORIZONTAL CRANE FORCES

HORIZONTAL CRANE FORCE ON LEFT-HAND COLUMN = 500.00 LB

HORIZONTAL CRANE FORCE ON RIGHT-HAND COLUMN= 200.00 LB

DO YOU WANT TO SEE THE TWIN CANTILEVER MOMENT?

YES

THE TWIN CANTILEVER MOMENT SHEAR AND THRUST :

-----

-----

| SECTION NO. | MOMENT     | SHEAR   | THRUST  |
|-------------|------------|---------|---------|
| 1           | -+92323+u6 | 29682.  | -337u3. |
| 2           | -+82972+u6 | 26958.  | -33/u3. |
| 3           | -+74511+u6 | 24243.  | -337u3. |
| 4           | -+66938+u6 | 21599.  | -33/u3. |
| 5           | -+60255+u6 | 189u5.  | -337u3. |
| 6           | -+54461+u6 | 16210.  | -33703. |
| 7           | -+49557+u6 | 13516.  | -337u3. |
| 8           | -+45541+u0 | 11921.  | -33/u3. |
| 9           | -+42414+u6 | 8127.1  | -337u3. |
| 10          | -+37566+u6 | 4732.7  | -197u3. |
| 11          | -+36449+u6 | 19559.  | -313u.7 |
| 12          | -+29524+u6 | 176u3.  | -2817.6 |
| 13          | -+23327+u6 | 15647.  | -25u4.5 |
| 14          | -+17860+u6 | 13692.  | -2191.5 |
| 15          | -+13122+u6 | 11736.  | -1878.4 |
| 16          | -91122.    | 9779.7  | -1565.3 |
| 17          | -58318.    | 7823.7  | -1252.3 |
| 18          | -32804.    | 5867.8  | -939.20 |
| 19          | -14580.    | 3911.9  | -626.14 |
| 20          | -36444.9   | 1955.9  | -313.07 |
| 21          | .00000     | .00000  | .00000  |
| 22          | -2177.3    | -1168.4 | -313.07 |
| 23          | -8709.1    | -2336.8 | -626.14 |
| 24          | -19596.    | -35L5.2 | -939.20 |
| 25          | -34836.    | -4673.5 | -1252.3 |
| 26          | -54432.    | -5841.9 | -1565.3 |
| 27          | -78382.    | -751u3  | -1878.4 |
| 28          | -+10669+u6 | -8178.7 | -2191.5 |
| 29          | -+13935+u6 | -9347.1 | -2564.5 |
| 30          | -+17636+u6 | -1L515. | -2817.6 |
| 31          | -+21773+u6 | -11684. | -313u.7 |
| 32          | -+21773+u6 | .00000  | -12096. |
| 33          | -+22876+u6 | 280.u0  | -17696. |
| 34          | -+22784+u6 | 280.00  | -17696. |
| 35          | -+22691+u6 | 280.u0  | -17696. |
| 36          | -+22599+u6 | 280.00  | -17696. |
| 37          | -+22506+u6 | 280.00  | -17696. |
| 38          | -+22414+u6 | 280.00  | -17696. |
| 39          | -+22327+u6 | 280.u0  | -17696. |
| 40          | -+22229+u6 | 280.00  | -17696. |
| 41          | -+22137+u6 | 280.00  | -17696. |

WHAT KIND OF BASES?

INPUT FIXED OR PINNED ,LEFT-HAND BASE FIRST

LEFT-HAND BASE IS FIXED

RIGHT-HAND BASE IS FIXED

NOMINATE POSITIONS OF FOUR PLASTIC HINGES .THE SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1    17    -31    41

| SECTION NO. | SHEAR    | THRUST   |
|-------------|----------|----------|
| 1           | .1467+05 | -2877+05 |
| 2           | .1637+05 | -2877+05 |
| 3           | .1368+05 | -2877+05 |
| 4           | .1498+05 | -2877+05 |
| 5           | 8;40.    | -2877+05 |
| 6           | 5596.    | -2877+05 |
| 7           | 2902.    | -2877+05 |
| 8           | 207.1    | -2877+05 |
| 9           | -2487.   | -2877+05 |
| 10          | -5882.   | -1477+05 |
| 11          | .1265+05 | -1211+05 |
| 12          | .1009+05 | -1174+05 |
| 13          | 8135.    | -1146+05 |
| 14          | 6179.    | -1117+05 |
| 15          | 4223.    | -1085+05 |
| 16          | 2267.    | -1054+05 |
| 17          | 311.0    | -1023+05 |
| 18          | -1645.   | -9915.   |
| 19          | -3601.   | -9602.   |
| 20          | -5557.   | -9289.   |
| 21          | 7515.    | -8976.   |
| 22          | 6344.    | -1184+05 |
| 23          | 5170.    | -1216+05 |
| 24          | 4008.    | -1247+05 |
| 25          | 2839.    | -1276+05 |
| 26          | 1671.    | -1304+05 |
| 27          | 502.5    | -1341+05 |
| 28          | -665.9   | -1372+05 |
| 29          | -1634.   | -1463+05 |
| 30          | -3043.   | -1435+05 |
| 31          | -4171.   | -1466+05 |
| 32          | .1461+05 | -1703+05 |
| 33          | .1089+05 | -2263+05 |
| 34          | .1409+05 | -2263+05 |
| 35          | .1689+05 | -2263+05 |
| 36          | .1669+05 | -2263+05 |
| 37          | .1089+05 | -2263+05 |
| 38          | .1009+05 | -2263+05 |
| 39          | .1689+05 | -2263+05 |
| 40          | .1689+05 | -2263+05 |
| 41          | .1089+05 | -2263+05 |

LOCATION OF PENDING MOMENT AT >FC:

| SECTION NO. | PLASTIC MUNFT CAPACITY |
|-------------|------------------------|
|-------------|------------------------|

.17332+06 LB . FT

|    |          |     |
|----|----------|-----|
| 1  | -.173+06 | *** |
| 2  | -.115+06 | NO  |
| 3  | -.653+05 | NO  |
| 4  | -.246+05 | NO  |
| 5  | .725+C4  | NO  |
| 6  | .302+05  | NO  |
| 7  | .442+05  | NO  |
| 8  | .443+05  | NO  |
| 9  | .455+05  | NO  |
| 10 | .590+05  | NO  |
| 11 | .351+05  | NO  |
| 12 | .764+05  | NO  |
| 13 | .110+06  | NO  |
| 14 | .137+06  | NO  |
| 15 | .156+06  | NO  |
| 16 | .169+06  | NO  |
| 17 | .173+06  | *** |
| 18 | .171+06  | NO  |
| 19 | .161+06  | NO  |
| 20 | .144+06  | NO  |
| 21 | .120+06  | NO  |
| 22 | .110+06  | NO  |
| 23 | .959+05  | NO  |
| 24 | .775+05  | NO  |
| 25 | .547+05  | NO  |
| 26 | .276+05  | NO  |
| 27 | -.388+04 | NO  |
| 28 | -.397+05 | NO  |
| 29 | -.799+05 | NO  |
| 30 | -.124+06 | NO  |
| 31 | -.173+06 | *** |
| 32 | -.138+06 | NO  |
| 33 | -.114+06 | NO  |
| 34 | -.783+05 | NO  |
| 35 | -.424+05 | NO  |
| 36 | -.644+04 | NO  |
| 37 | .295+05  | NO  |
| 38 | .655+05  | NO  |
| 39 | .101+06  | NO  |
| 40 | .137+06  | .   |
| 41 | .173+06  | *** |

DO YOU WANT TO SEE THE BASE REACTIONS?

YES

V1= .2877+05 H1= .1907+05 VH1= .2263+05 H41= .1069+05

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DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

YES

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1 19 -31 41

40              •1669+05      -•2272+05  
 41              •1669+05      -•2272+05

LOCATION OF BENDING MOMENT MT > MPC:

| SECTION NO. | PLASTIC MOMENT CAPACITY |
|-------------|-------------------------|
|             | •16447+06 LB FT         |
| 1           | •176+06 ***             |
| 2           | •111+06 NO              |
| 3           | •606+05 NO              |
| 4           | •192+05 NO              |
| 5           | •133+05 NO              |
| 6           | •368+05 NO              |
| 7           | •515+05 NO              |
| 8           | •573+05 NO              |
| 9           | •542+05 NO              |
| 10          | •684+05 NO              |
| 11          | •452+05 NO              |
| 12          | •863+05 NO              |
| 13          | •126+06 NO              |
| 14          | •147+06 NO              |
| 15          | •166+06 NO              |
| 16          | •178+06 YES             |
| 17          | •183+06 YES             |
| 18          | •180+06 YES             |
| 19          | •170+06 ***             |
| 20          | •153+06 NO              |
| 21          | •128+06 NO              |
| 22          | •118+06 NO              |
| 23          | •103+06 NO              |
| 24          | •845+05 NO              |
| 25          | •612+05 NO              |
| 26          | •336+05 NO              |
| 27          | •158+04 NO              |

|    |           |     |
|----|-----------|-----|
| 28 | - .348+05 | NO  |
| 29 | - .755+05 | NO  |
| 30 | - .121+06 | NO  |
| 31 | - .170+06 | *** |
| 32 | - .136+06 | NO  |
| 33 | - .112+06 | NO  |
| 34 | - .770+05 | NO  |
| 35 | - .417+05 | NO  |
| 36 | - .644+04 | NO  |
| 37 | .286+05   | NO  |
| 38 | .641+05   | NO  |
| 39 | .994+05   | NO  |
| 40 | .135+06   | NO  |
| 41 | .170+06   | *** |

DO YOU WANT TO SEE THE BASE REACTIONS?

YES

V1= .2868+05 H1= .1927+05 V41= .2272+05 H41= .1069+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

YES

NOMINATE POSITIONS OF FOUR PLASTIC HINGES WITH SIGN TO INDICATE THE MOMENT SIGN

THE FOUR NOMINATED POSITIONS OF PLASTIC HINGES WITH SIGN

TO INDICATE THE MOMENT SIGN ARE:

-1 20 -31 41

| SECTION NO. | SHEAR    | THRUST    |
|-------------|----------|-----------|
| 1           | +1956+05 | -+2854+05 |
| 2           | +1606+05 | -+2854+05 |
| 3           | +1417+05 | -+2854+05 |
| 4           | +1147+05 | -+2854+05 |
| 5           | 8760.    | -+2854+05 |
| 6           | 6186.    | -+2854+05 |
| 7           | 3392.    | -+2854+05 |
| 8           | 697.3    | -+2854+05 |
| 9           | -1971.   | -+2854+05 |
| 10          | -5391.   | -+1457+05 |
| 11          | +1196+05 | -+1157+05 |
| 12          | +1000+05 | -+1120+05 |
| 13          | 8045.    | -+1193+05 |
| 14          | 6189.    | -+1164+05 |
| 15          | 4133.    | -+1132+05 |
| 16          | 2177.    | -+1081+05 |
| 17          | 2200.    | -9696.    |
| 18          | -1735.   | -9383.    |
| 19          | -3691.   | -9170.    |
| 20          | -5647.   | -8757.    |
| 21          | 7603.    | -8444.    |
| 22          | 6435.    | -+1143+05 |
| 23          | 5200.    | -+1174+05 |
| 24          | 4098.    | -+1205+05 |
| 25          | 2924.    | -+1237+05 |
| 26          | 1701.    | -+1261+05 |
| 27          | 542.6    | -+1244+05 |
| 28          | -575.8   | -+1331+05 |
| 29          | -1744.   | -+1364+05 |
| 30          | -2913.   | -+1393+05 |
| 31          | -4061.   | -+1424+05 |
| 32          | +1012+05 | -+1725+05 |
| 33          | +1040+05 | -+2245+05 |
| 34          | +1040+05 | -+2280+05 |
| 35          | +1040+05 | -+2285+05 |
| 36          | +1040+05 | -+2285+05 |
| 37          | +1040+05 | -+2285+05 |
| 38          | +1040+05 | -+2285+05 |
| 39          | +1040+05 | -+2285+05 |
| 40          | +1040+05 | -+2285+05 |
| 41          | +1040+05 | -+2285+05 |

## LOCATION OF BENDING MOMENT MT &gt;MPC:

| SECTION NO. | PLASTIC BENDING MOMENT CAPACITY | • 16623+06 |      |
|-------------|---------------------------------|------------|------|
|             |                                 | LB         | • FT |
| 1           | -•165+06                        | ***        |      |
| 2           | -•105+06                        | NO         |      |
| 3           | -•539+05                        | NO         |      |
| 4           | -•116+05                        | NO         |      |
| 5           | •218+05                         | NO         |      |
| 6           | •463+05                         | NO         |      |
| 7           | •626+05                         | NO         |      |
| 8           | •697+05                         | NO         |      |
| 9           | •666+05                         | NO         |      |
| 10          | •816+05                         | NO         |      |
| 11          | •594+05                         | NO         |      |
| 12          | •100+06                         | NO         |      |
| 13          | •134+06                         | NO         |      |
| 14          | •160+06                         | NO         |      |
| 15          | •179+06                         | YES        |      |
| 16          | •191+06                         | YES        |      |
| 17          | •196+06                         | YES        |      |
| 18          | •193+06                         | YES        |      |
| 19          | •183+06                         | YES        |      |
| 20          | •165+06                         | ***        |      |
| 21          | •141+06                         | NO         |      |
| 22          | •130+06                         | NO         |      |
| 23          | •114+06                         | NO         |      |
| 24          | •945+05                         | NO         |      |
| 25          | •705+05                         | NO         |      |
| 26          | •421+05                         | NO         |      |
| 27          | •933+04                         | NO         |      |
| 28          | -•278+05                        | NO         |      |
| 29          | -•692+05                        | NO         |      |
| 30          | -•115+06                        | NO         |      |
| 31          | -•165+06                        | ***        |      |
| 32          | -•132+06                        | NO         |      |
| 33          | -•109+06                        | NO         |      |
| 34          | -•751+05                        | NO         |      |

|    |           |     |
|----|-----------|-----|
| 35 | = .404+05 | NO  |
| 36 | = .644+04 | NO  |
| 37 | .279+05   | NO  |
| 38 | .622+05   | NO  |
| 39 | .966+05   | NO  |
| 40 | .131+06   | NO  |
| 41 | .165+06   | *** |

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DO YOU WANT TO SEE THE BASE REACTIONS?

YES

V1= .2854+05 H1= .1956+05 V41= .2285+05 H41= .1046+05

DO YOU WANT TO TRY ANOTHER HINGE PATTERN?

NO

DO YOU WANT TO ANALYSE ANOTHER FRAME?

NO

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